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Jupiter Through 1990*

Andrey B. Sergeyevsky

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CALIFORNIA INSTITUTE OF TECHNOLOGY
PASADENA, CALIFORNIA

September 1, 1975



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PREFACE

This document is divided into three volumes. Volume I comprises the mission design data for Venus, Volume II the data for Mars, and Volume III the data for Jupiter.

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MISSION DESIGN DATA FOR VENUS, MARS AND JUPITER THROUGH 1990

Andrey B. Sergeyevsky

A. INTRODUCTION

This document presents mission design data for direct transfer trajectories from Earth to three planets – Venus, Mars and Jupiter, extending previously published information (see Refs. 1, 2, 3, 4 and 5) through the 1990 departure opportunity.

The primary purpose of this effort is to provide the mission analyst with graphical information, sufficient for preliminary mission design and evaluation. The data follows closely the format of Reference 4 and reflects methods of Reference 2. A specially modified version of the Space Research Conic Program (SPARC) (see Ref. 6) was used to generate the trajectory information presented. The data were automatically contour-plotted on the SC4020 plotter using the General Plot Program (GPP) (see Ref. 7), then hand retouched and labeled. A special program (VIEWPE) was constructed to provide planetary positional data in graphical form, plotted on the SC4020, and presented in original format.

The data are arranged in three sections by arrival planet, in natural sequence. Each section consists of two parts – the trajectory characteristics for all available opportunities to the particular planet, in chronological order, followed by that planet's positional data for every calendar year, from 1975 to 1995.

The persevering and encouraging insistence of management, especially that of Mr. Willard E. Bollman to carry this effort through to completion, as well as the graphic and editorial support of Mr. Richard W. Rackus are gratefully acknowledged.

B. DESCRIPTION OF TRAJECTORY CHARACTERISTICS DATA

I. General

The data represent trajectory performance information plotted in the departure date/arrival date space, thus

defining all possible transfer trajectories between the two bodies, within the time-span considered. Fourteen individual parameters are contour-plotted on the departure energy (C_3) background contour chart, for each opportunity. The following opportunities are presented:

To Venus: 1975, 1976/7, 1978, 1980, 1981, 1983, 1984/5, 1986, 1988, 1989/90.

To Mars: 1979, 1981/2, 1983/4, 1985/6, 1988, 1990.

To Jupiter: 1977, 1978, 1979, 1980/81, 1981/82, 1983, 1984, 1985, 1986, 1987, 1988, 1989, 1990.

2. Definition of Terms

The following parameters are displayed on the contour plots:

C_3 = Earth departure energy (km^2/sec^2); same as the square of departure hyperbolic excess velocity $V_\infty^2 = C_3 = V_I^2 - 2GM/R_I$, where

V_I = conic injection velocity (km/sec)

GM = gravitational constant times mass of the attracting body, from Reference 8:

GM_{VENUS} = 0.32486010E6
(km^3/sec^2)

GM_{EARTH} = 0.39860115E6

GM_{MARS} = 0.42828444E5

$GM_{JUPITER}$ = 0.12670772E9

$R_I = R_S + h_I$, Injection radius (km), sum of surface radius R_S PLANET and injection altitude h_I , where (see Ref. 8):	DLA	Geocentric declination (vs. mean Earth equator of launch date) of the departure V_∞ - vector. May impose launch constraints. (deg)
$R_{S_{VENUS}} = 6052$ (km)	ZAL	Angle between departure V_∞ vector and Sun-Earth vector. Equivalent to Earth-probe-Sun angle, several days out. (deg)
$R_{S_{EARTH}} = 6378.16$	INC	Heliocentric inclination of transfer trajectory with mean ecliptic (Earth orbital) plane of launch date. (deg)
$R_{S_{MARS}} = 3393.4$	ZAP	Angle between arrival V_∞ vector and the arrival planet-to-Sun vector. Equivalent to planet-probe-Sun angle at far encounter; for subsolar impact would be equal to 180° .(deg).
$R_{S_{JUPITER}} = 71372$	ETS	Angle in arrival B-plane, measured from T-axis, clockwise, to projection of Sun-to-planet vector. Equivalent to solar occultation region center-line. (deg)
TF Time of flight (Days)	LVI	Planetocentric latitude of vertical impact vs arrival planet equator. Note that Venusian north is below ecliptic, while Mars' and Jupiter's is above. Equivalent to declination of the incoming asymptote (i.e., the negative of incoming V_∞ vector) in planetary equator system.
CD Earth to planet communication distance at arrival (km)	ZAE	Angle between arrival V_∞ vector and the planet-to-Earth vector. Equivalent to planet-probe-Earth angle at far encounter. (Deg.)
VHP Arrival hyperbolic excess velocity	ETE	Angle in arrival B-plane, measured from T-axis, clockwise, to projection of Earth-to-planet vector. Equivalent to Earth occultation region centerline. (deg)
$V_\infty = \sqrt{V^2 - \frac{2GM}{R}}$, (km/sec), where V = Heliocentric conic arrival velocity at heliocentric radius R (km).	THA	Angle in arrival B-plane, from T-axis, clockwise, to major axis of error dispersion ellipse (0 - 180 deg).
Arrival Planet Orbit insertion velocity increment ΔV , at perihelion, may be computed from V_∞ :	SG1	Semi-major axis magnitude of B-plane dispersion ellipse, resulting from a spherically distributed V_∞ velocity vector error of 0.1 m/sec on departure asymptote (km).
$\Delta V = \sqrt{V_\infty^2 + \frac{GM}{R_p}} - \sqrt{\frac{2GM R_A}{R_p(R_A + R_p)}}$	SG2	Semi-minor axis of above dispersion ellipse (km).
where R_p and R_A are planetocentric perihelion and apohelion radii (km), respectively. Similarly, if specific capture orbit period P (sec) and perihelion radius R_p are desired:		
$\Delta V = \sqrt{V^2 + \frac{2GM}{R_p}} - \sqrt{\frac{2GM}{R_p} - 3\sqrt{\left(\frac{2GM\pi}{P}\right)^2}}$		
B-PLANE A plane normal to the incoming V_∞ - vector and passing through the center of planet.		
T-AXIS Axis in B-plane, parallel to ecliptic (Earth mean orbital) plane (see Figure 1).		

SG3 Arrival time dispersion, normal to B-plane,
for above error model (sec).

YR/M/D Year, Month, Date.

C. DESCRIPTION OF PLANETARY POSITIONAL DATA

1. General

The data represent planetary geometry-related information plotted versus calendar arrival date at the target planet. Each set of seven plots represents the annual time history of 19 parameters, and may be used for flyby and orbiter missions.

2. Description of Curve Labels

P	Target planet, equivalent to probe approaching or in orbit about target planet.
E	Earth
S	Sun
CA	Cone Angle, i.e., Sun-probe-object (Earth or Canopus, etc.) angle. (See Figure 2.)
KA	Clock Angle, i.e., angle between projections of the Probe-Canopus and probe-object vectors into the plane normal to the sun-line (for which CA = 90°). (See Figure 2.)
RISEXX	Rise time (GMT) of planet through 6° horizon mask at DSN Station No. XX. (e.g., XX = 14 = GOLDSTONE, 43 = CANBERRA, 63 = MADRID.)
SETXX	Set time (GMT) of planet through 6° horizon mask at DSN Station No. XX.

3. Description of Plots

Plot	Y-axis label	
a)	DECLIN	Geocentric Earth equatorial declination of planet (P), planetocentric planetary equatorial declination of Earth (E) and Sun (S). Note that Venusian north is below ecliptic.
b)	EC.LON	Heliocentric ecliptic longitude of planet.
c)	CA,KA	Cone (ECA) and Clock (EKA) angle of Earth and cone angle of Canopus (CCA) as seen from a Sun-Canopus oriented spacecraft near target planet, P (see Figure 2).
d)	DISTANCE	Sun-Planet distance (SP) and Earth-Planet communication distance (EP) in mill. km.
e)	SUN-EARTH-PLANET	Sun-Earth-Planet angle (SEP), indicating times of superior ($SEP \approx 0$) and inferior ($SEP \approx 180^\circ$) conjunction; $SEP > 5^\circ$ is a communications constraint.
f)	STATION RISE/SET	Rise and Set times (GMT) of planet at 3 DSN Stations on Earth, 6° mask.

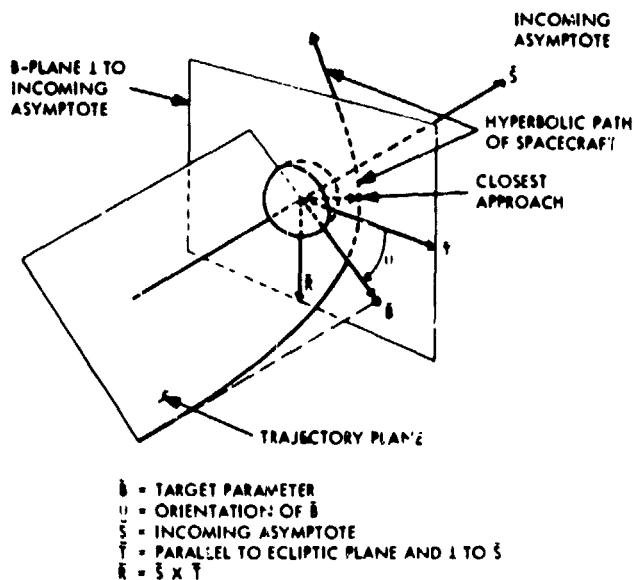


Figure 1. Definition of B-Plane

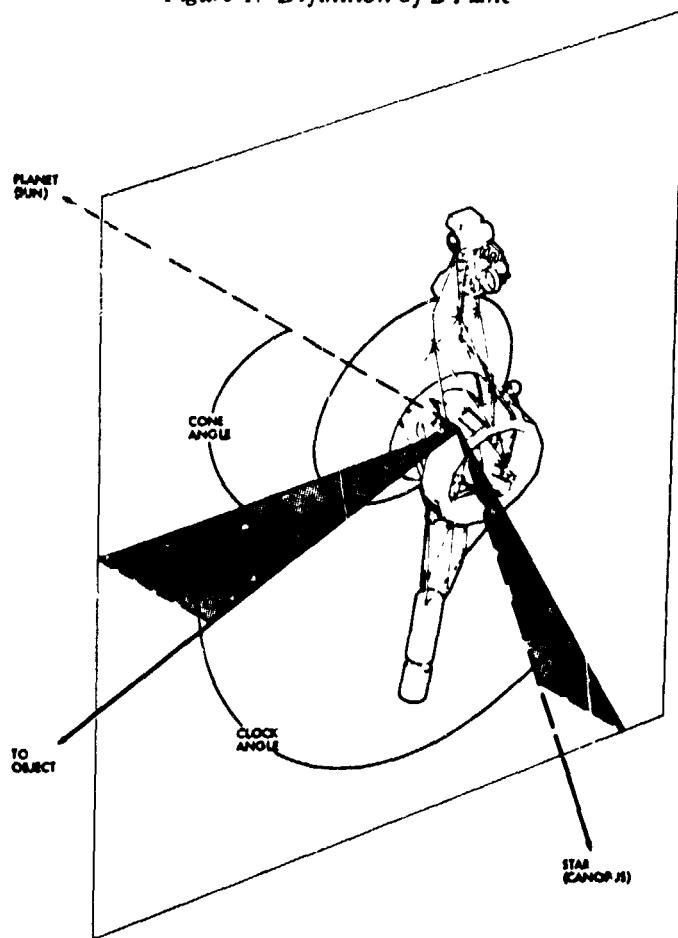


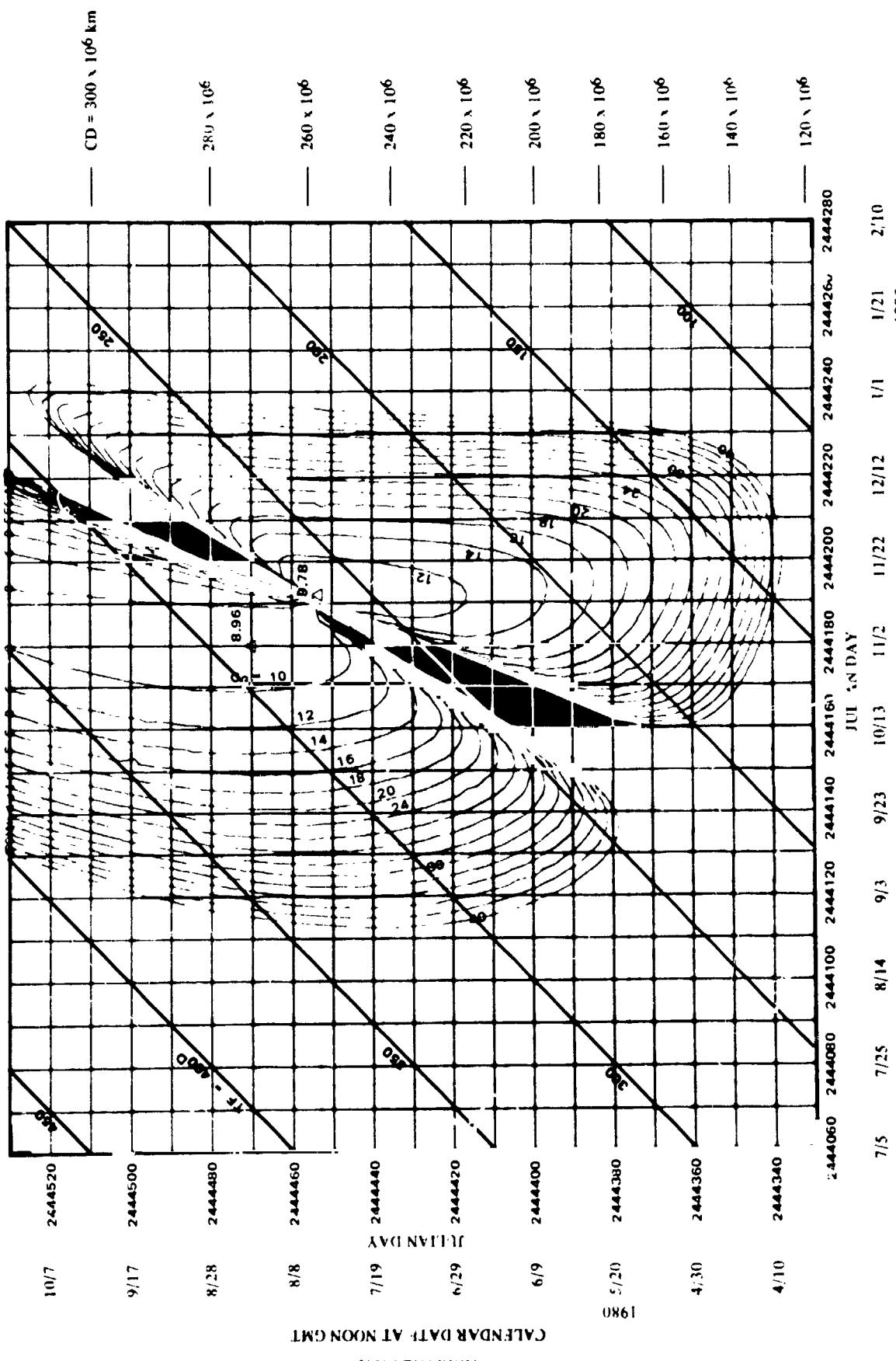
Figure 2. Definition of Cone and Clock Angle

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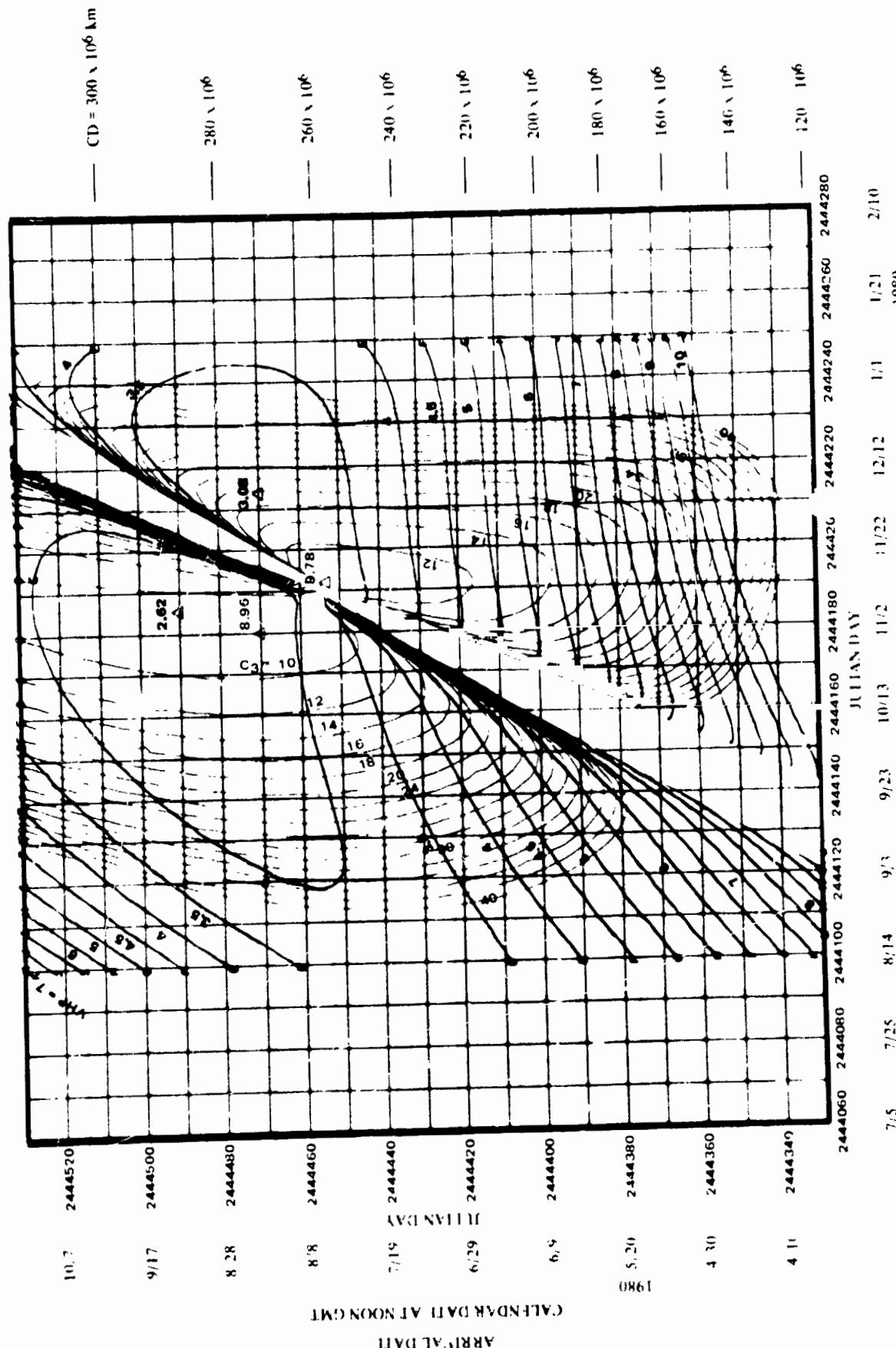
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C₃
♂
1979

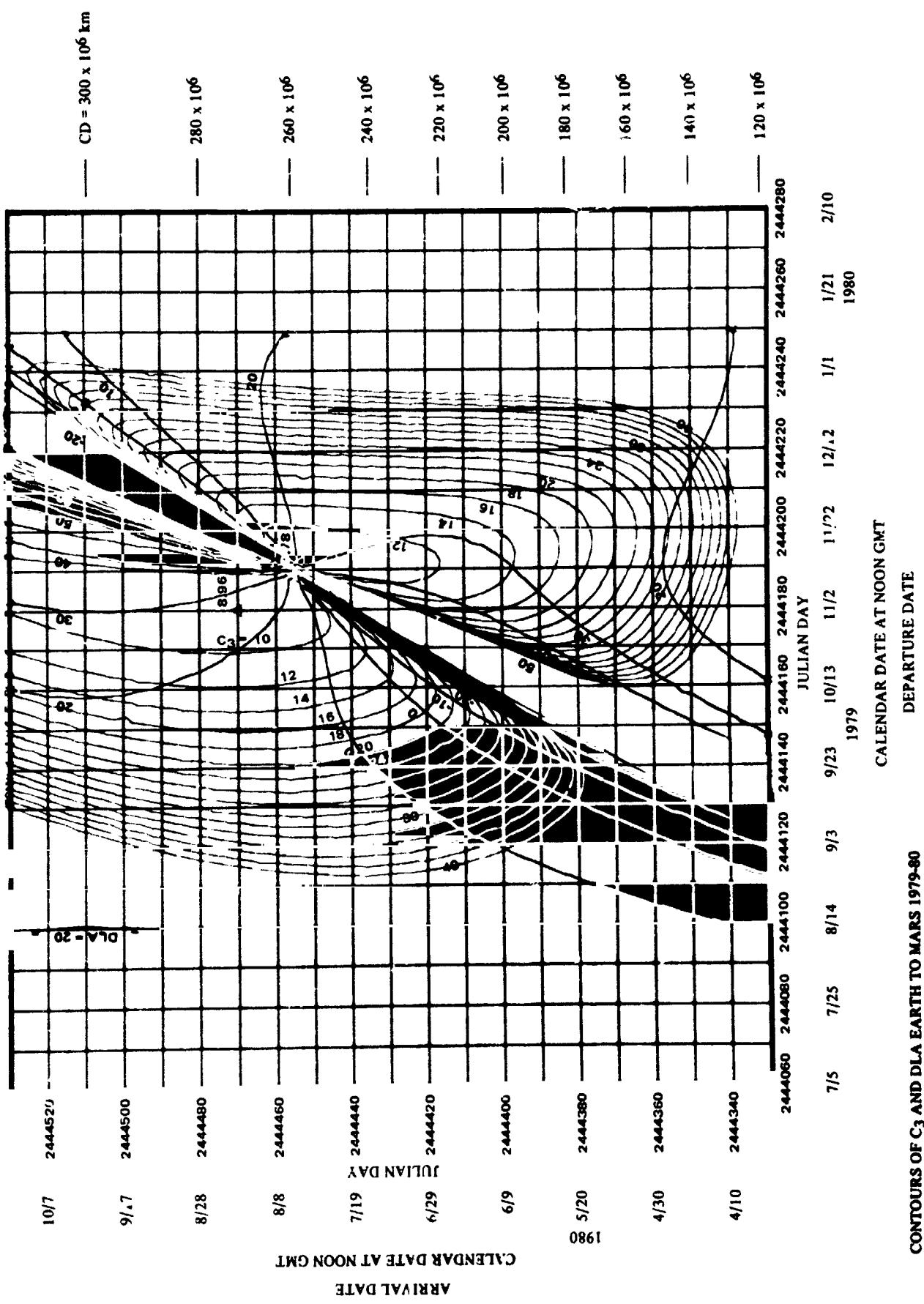


CONTOURS OF C₃ AND FLIGHT TIMES EARTH TO MARS 1979-80

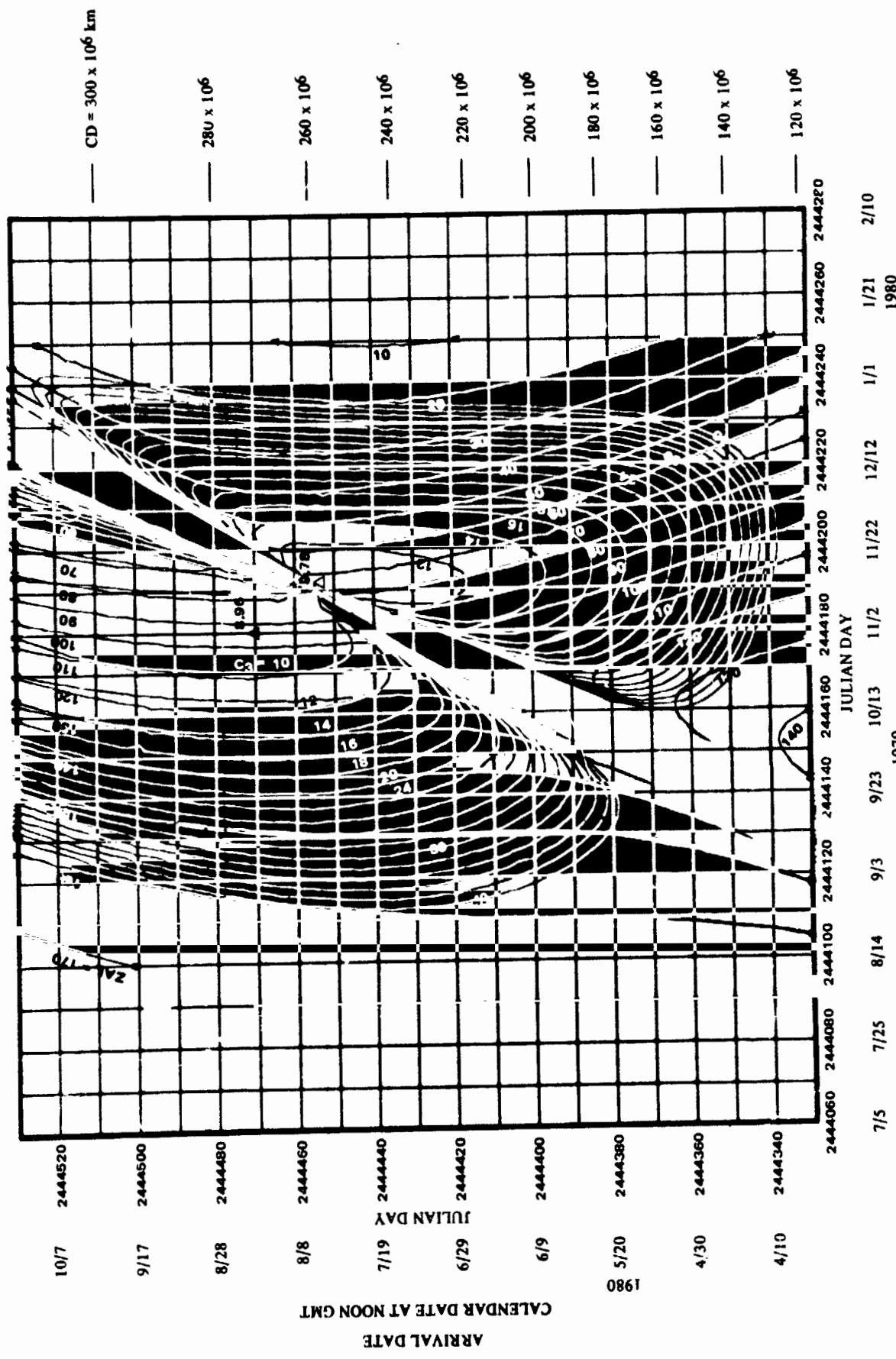
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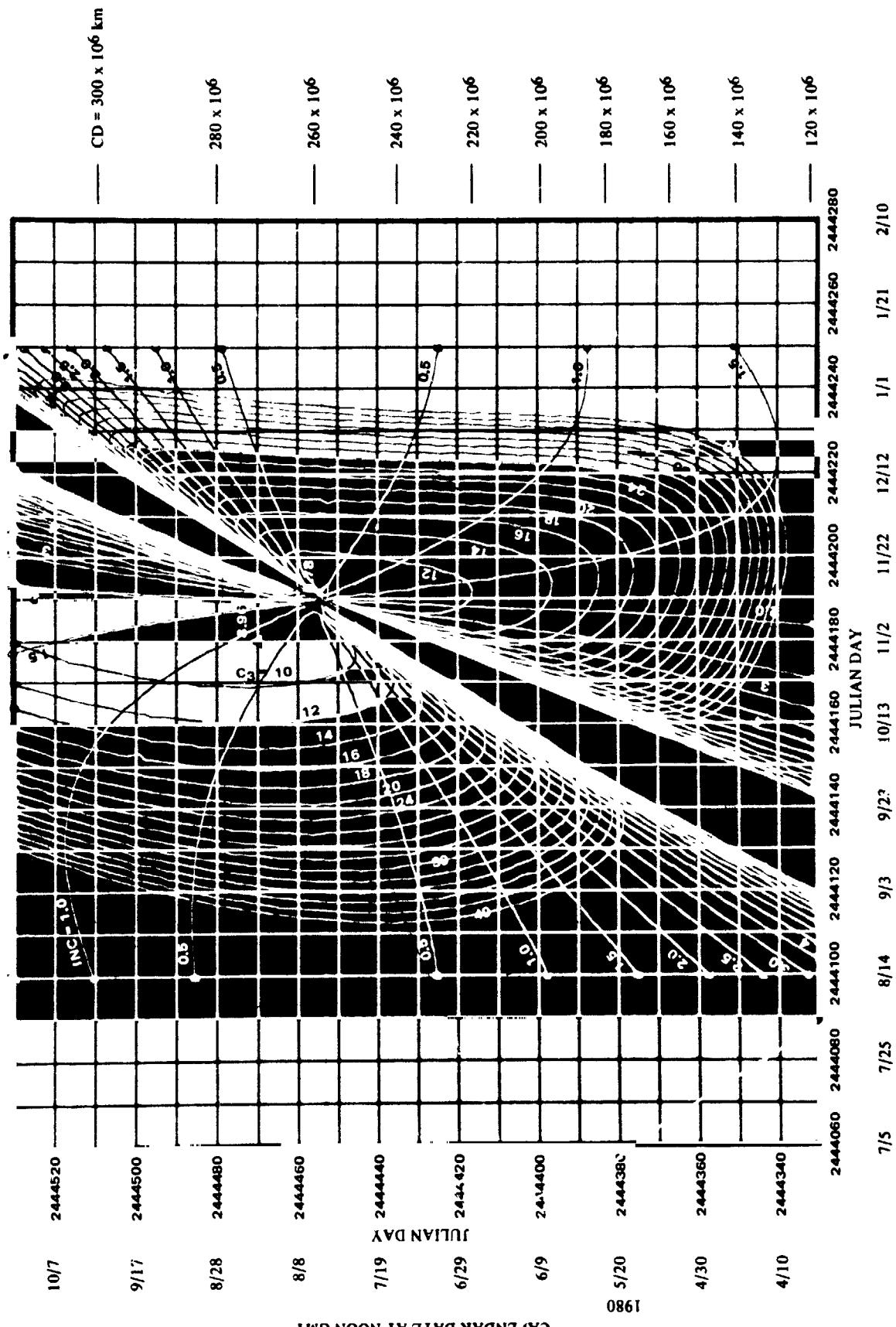
DLA
Q
1979



ZAL
Q
1979

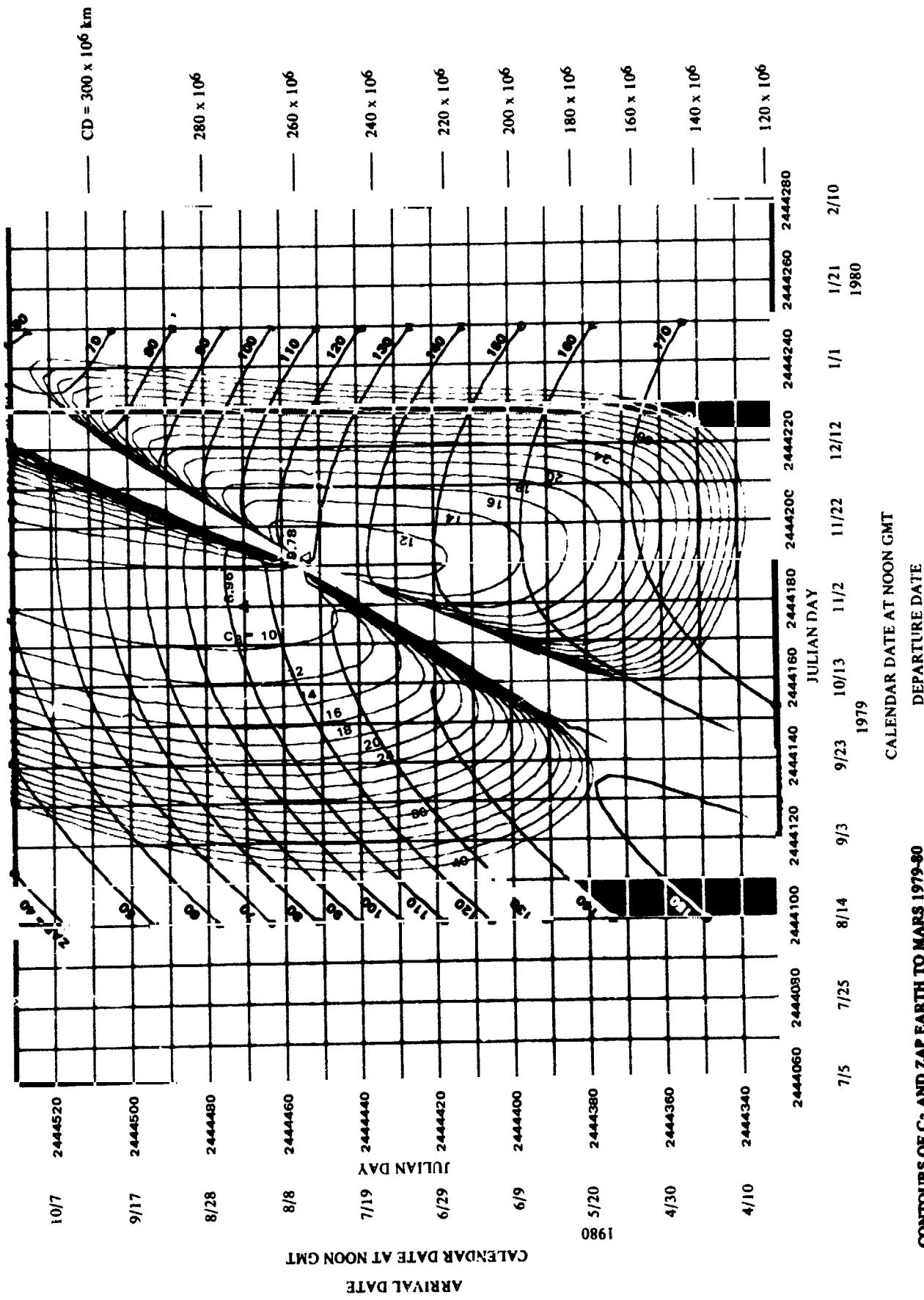


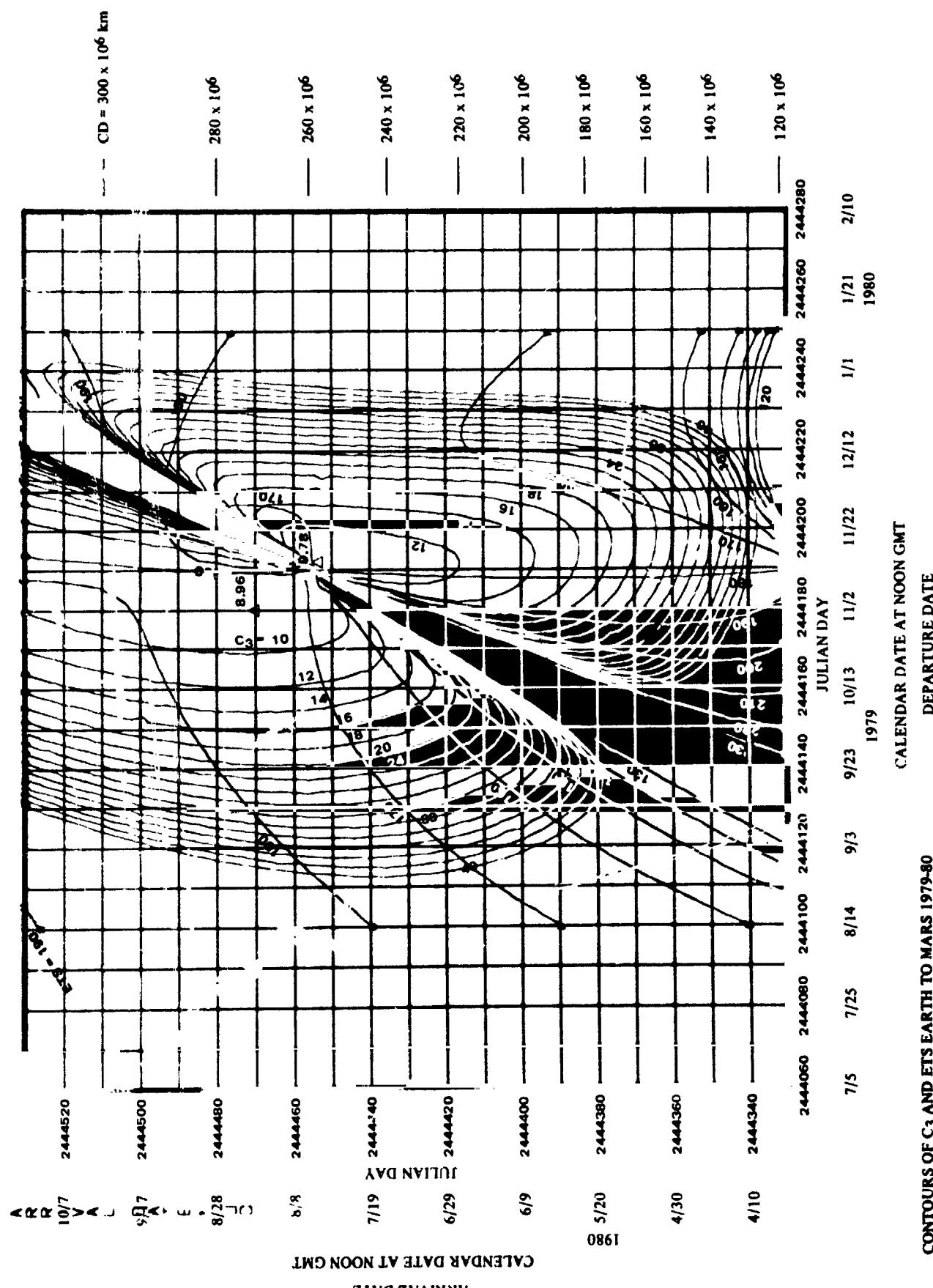
INC
Q
1979



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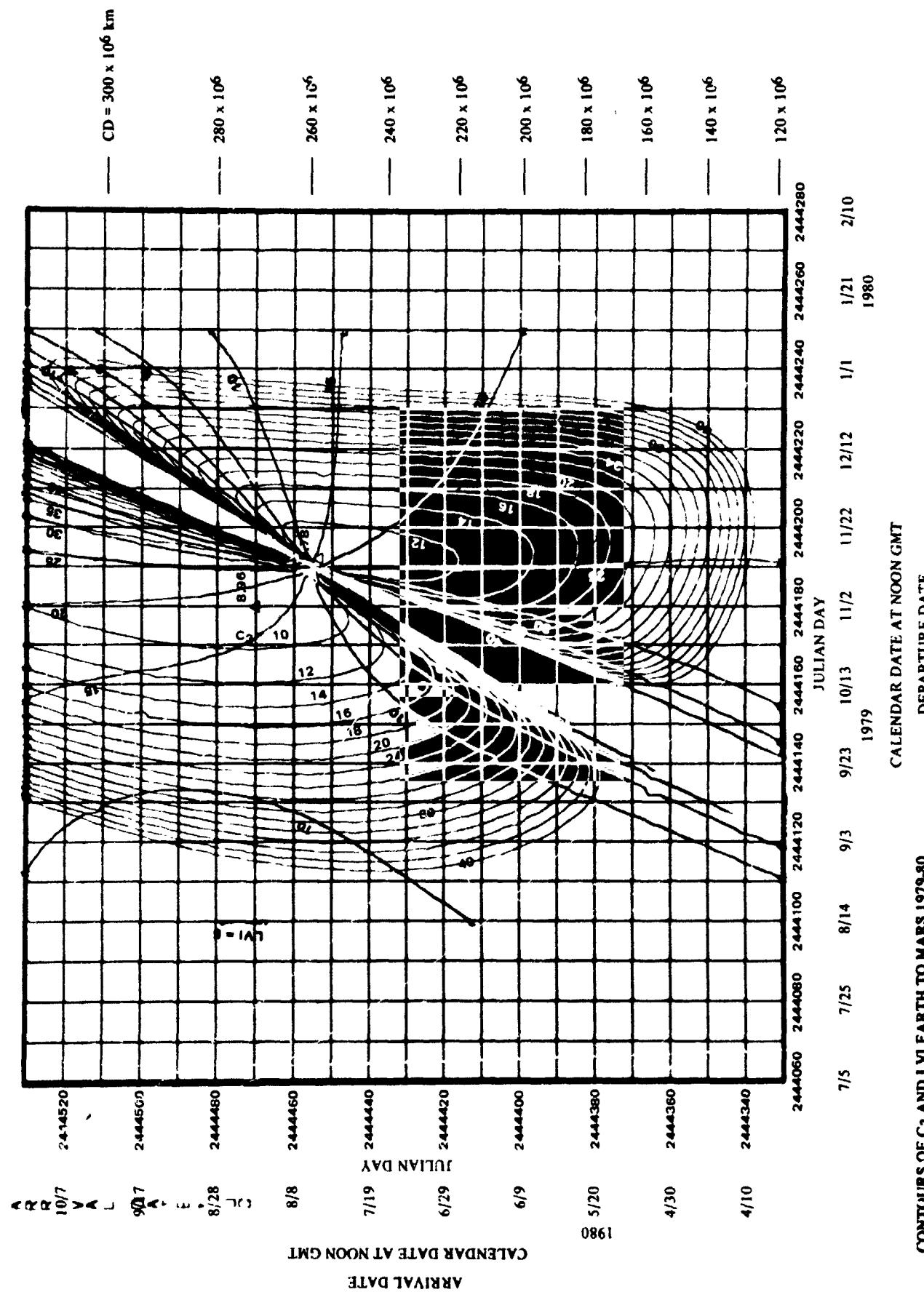
ZAP
♂
1979

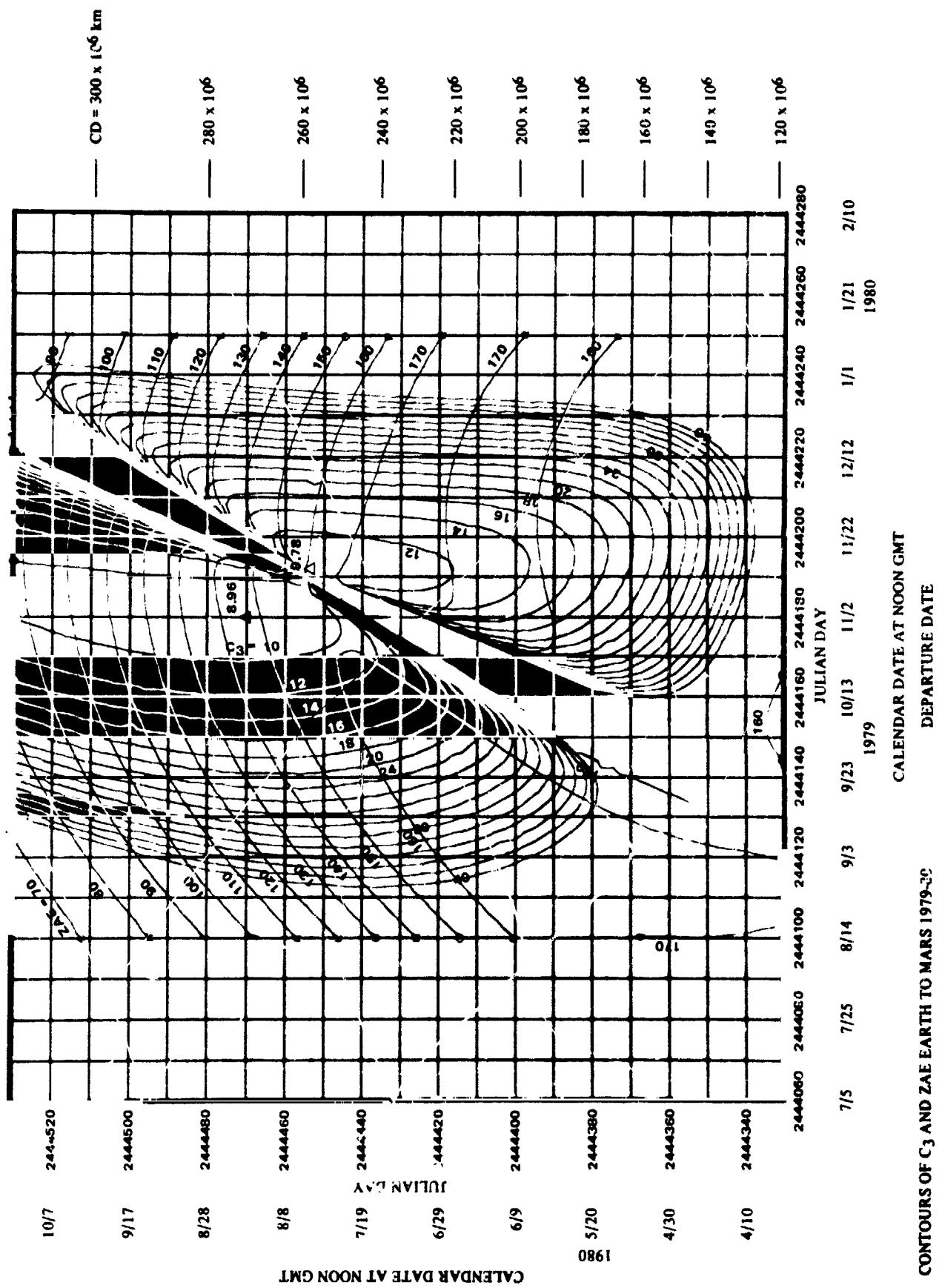




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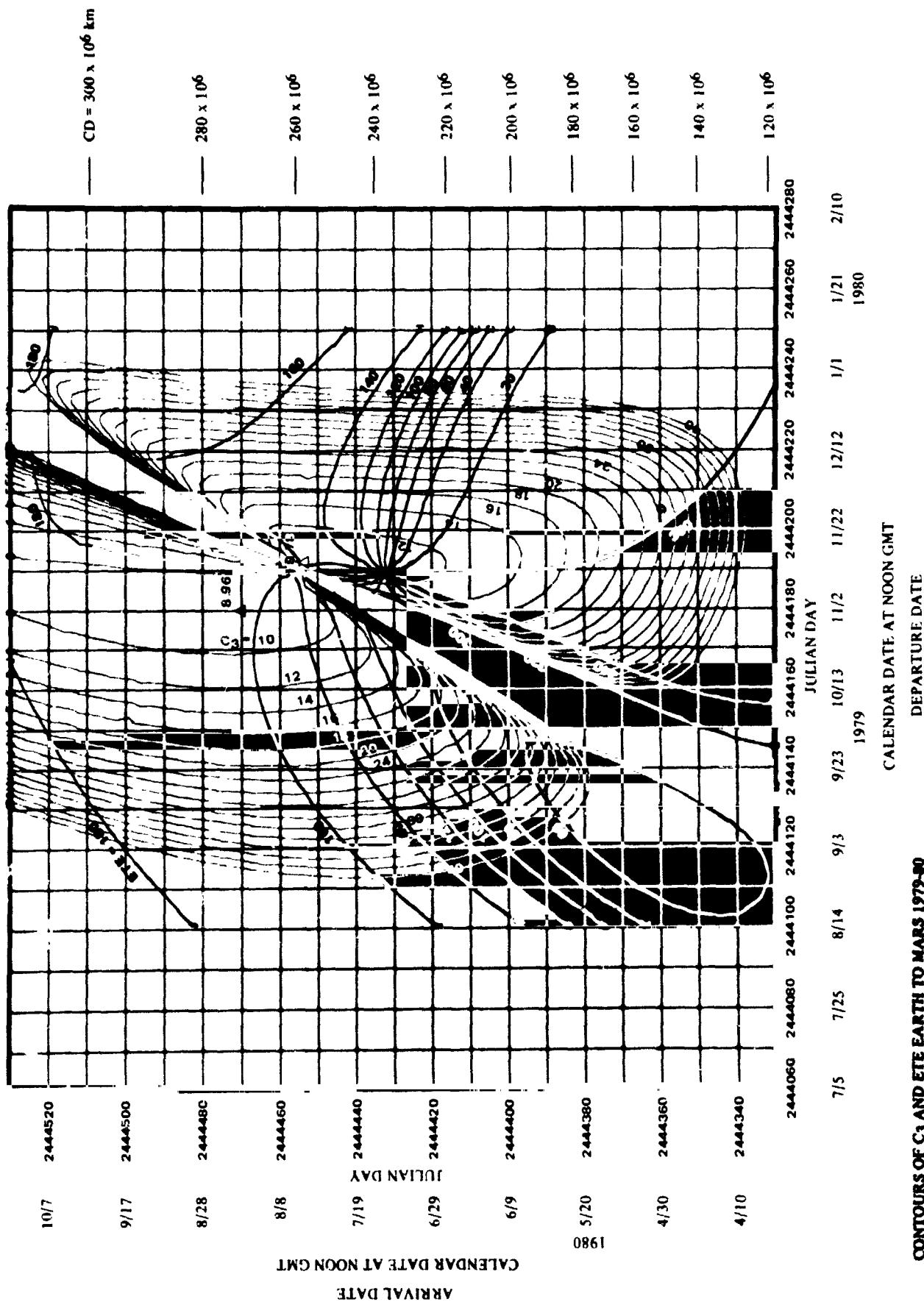
LVI
♂
1979

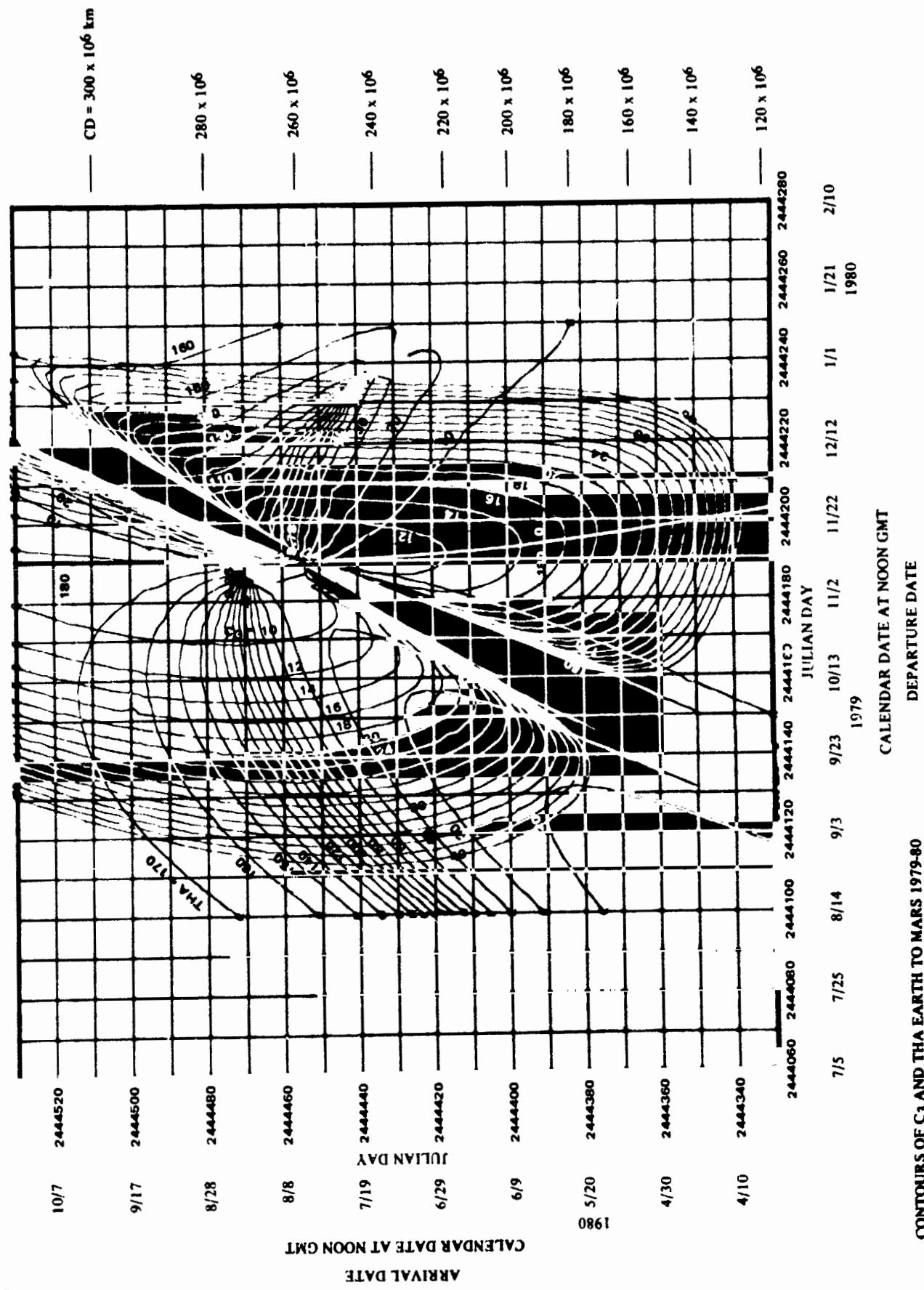


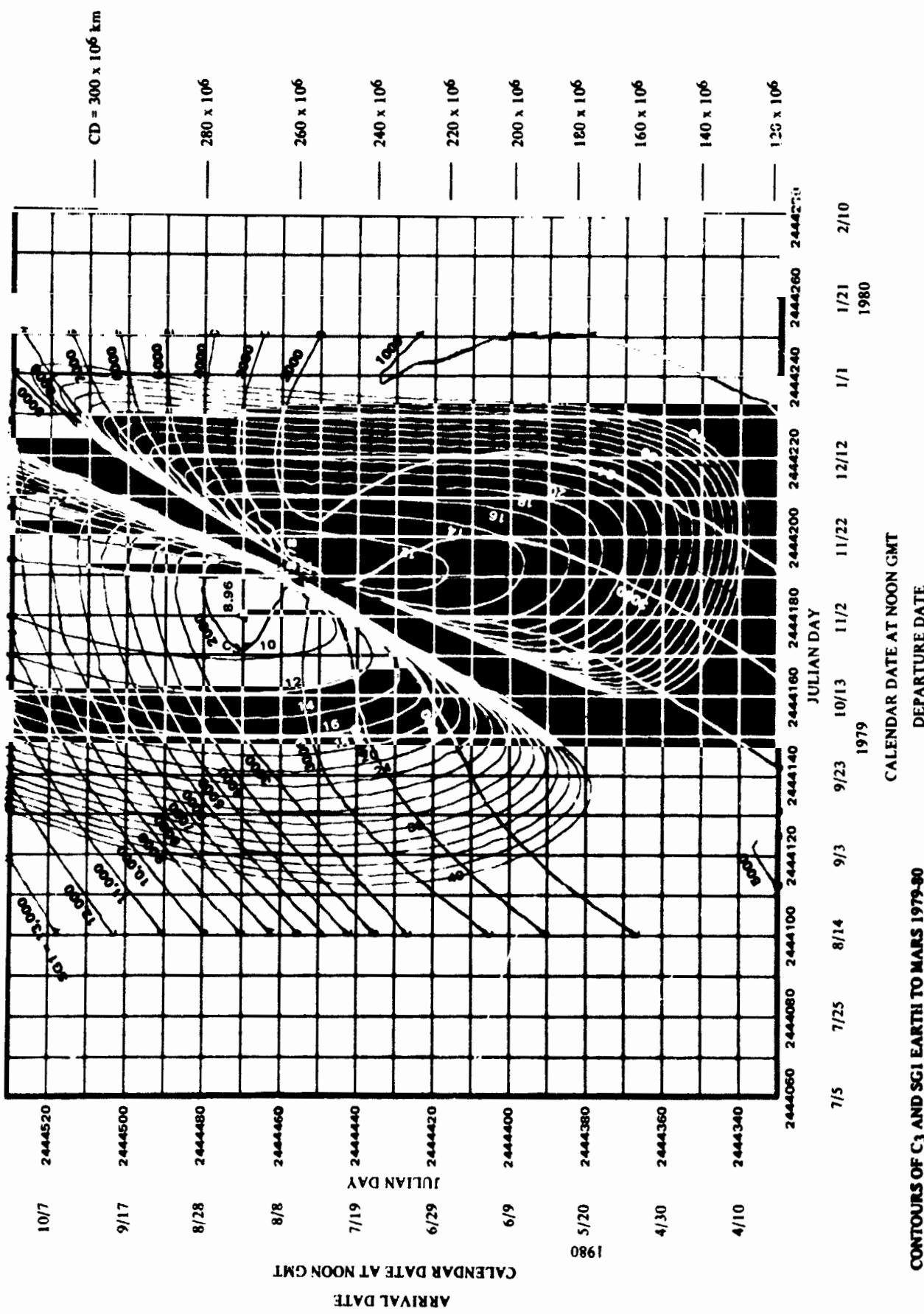


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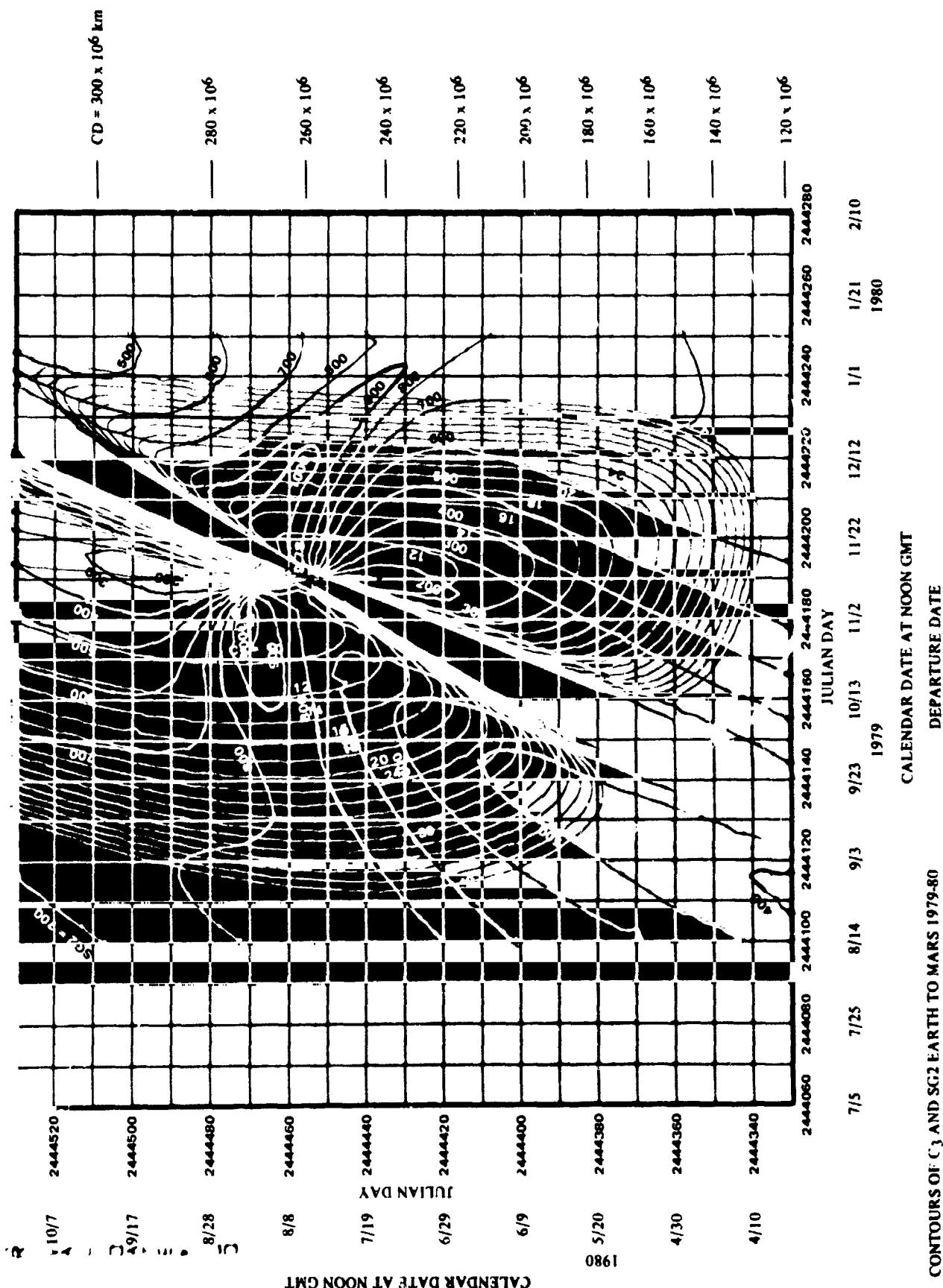
ETE
1979



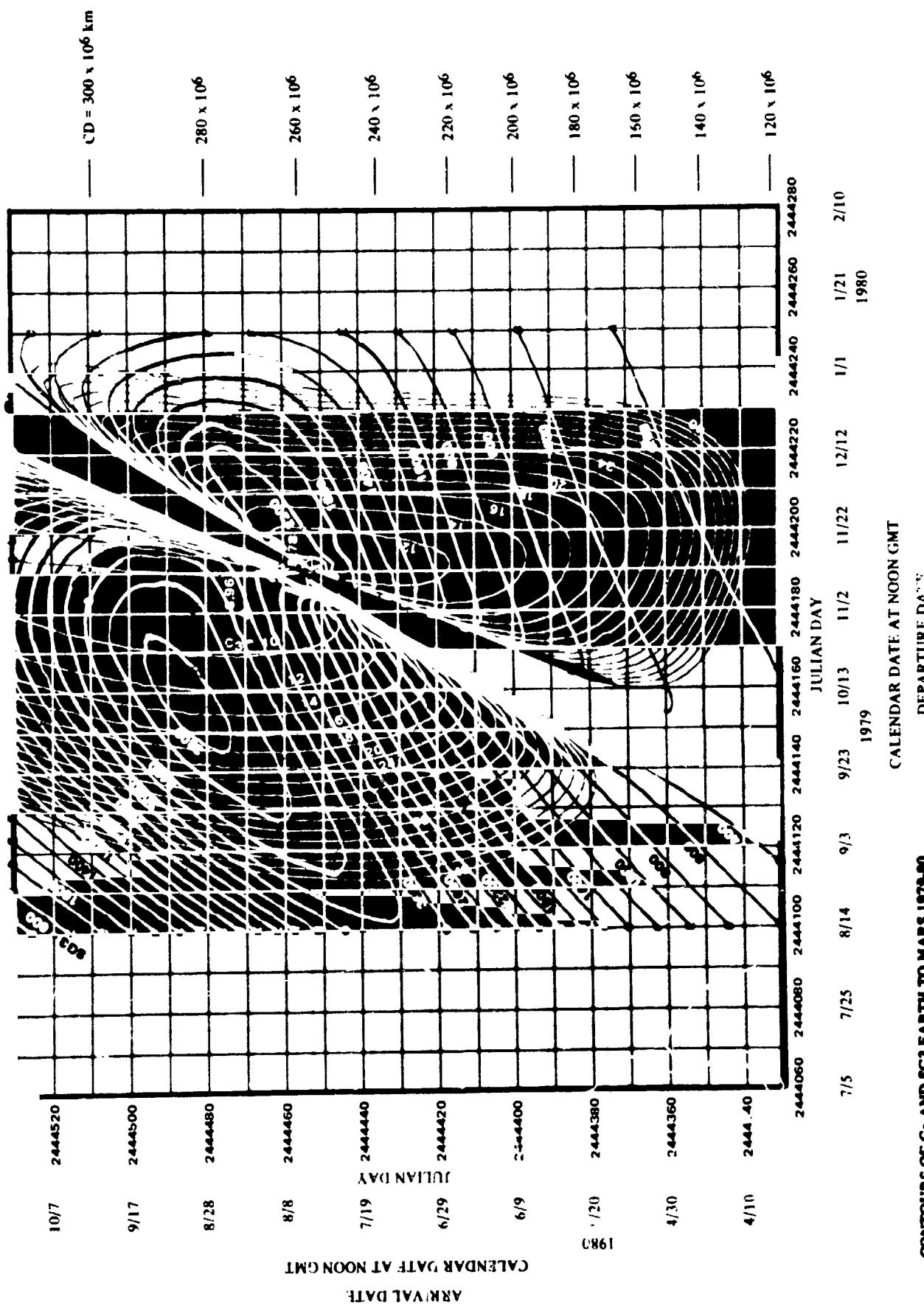




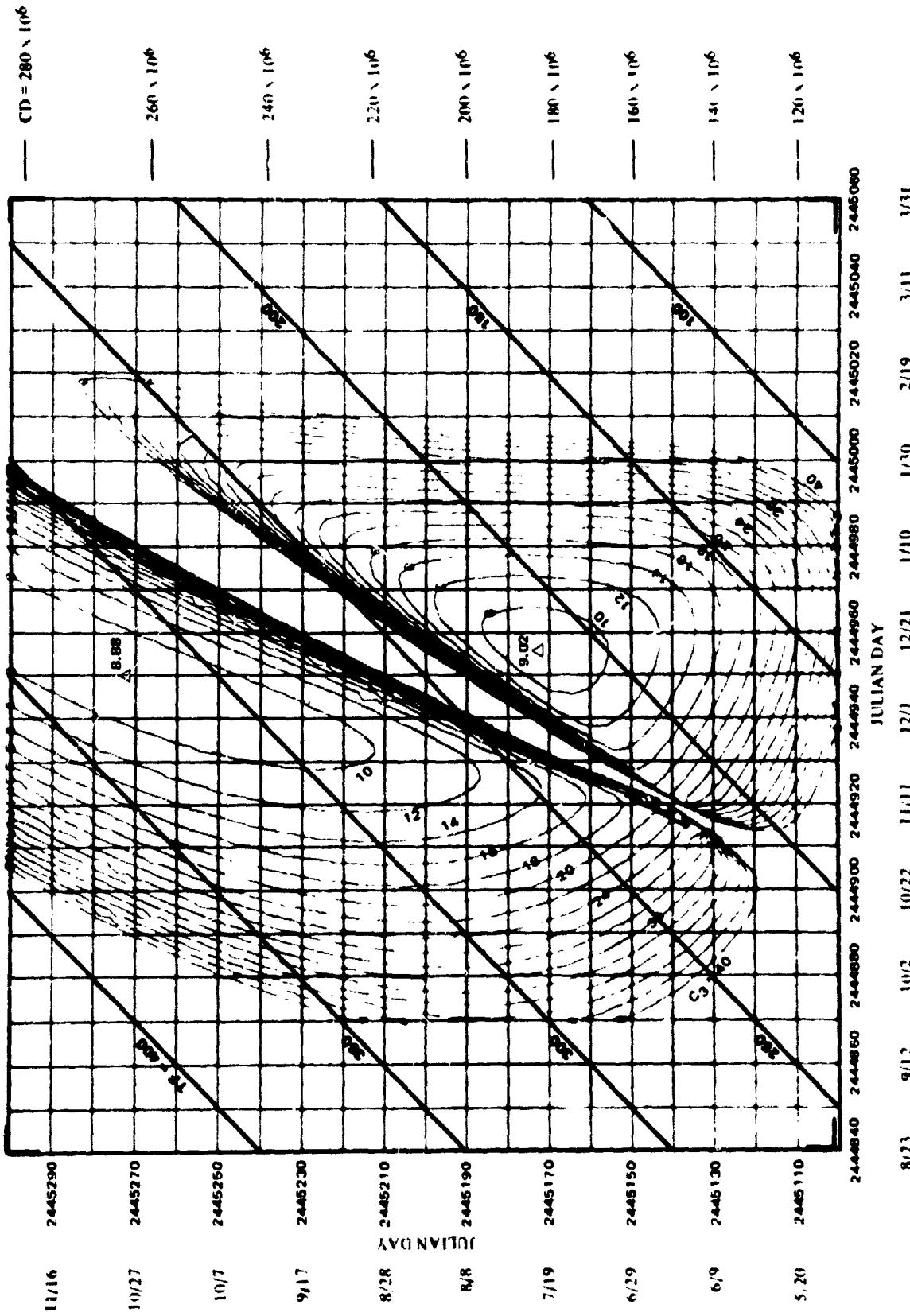
SG2
Q
1979



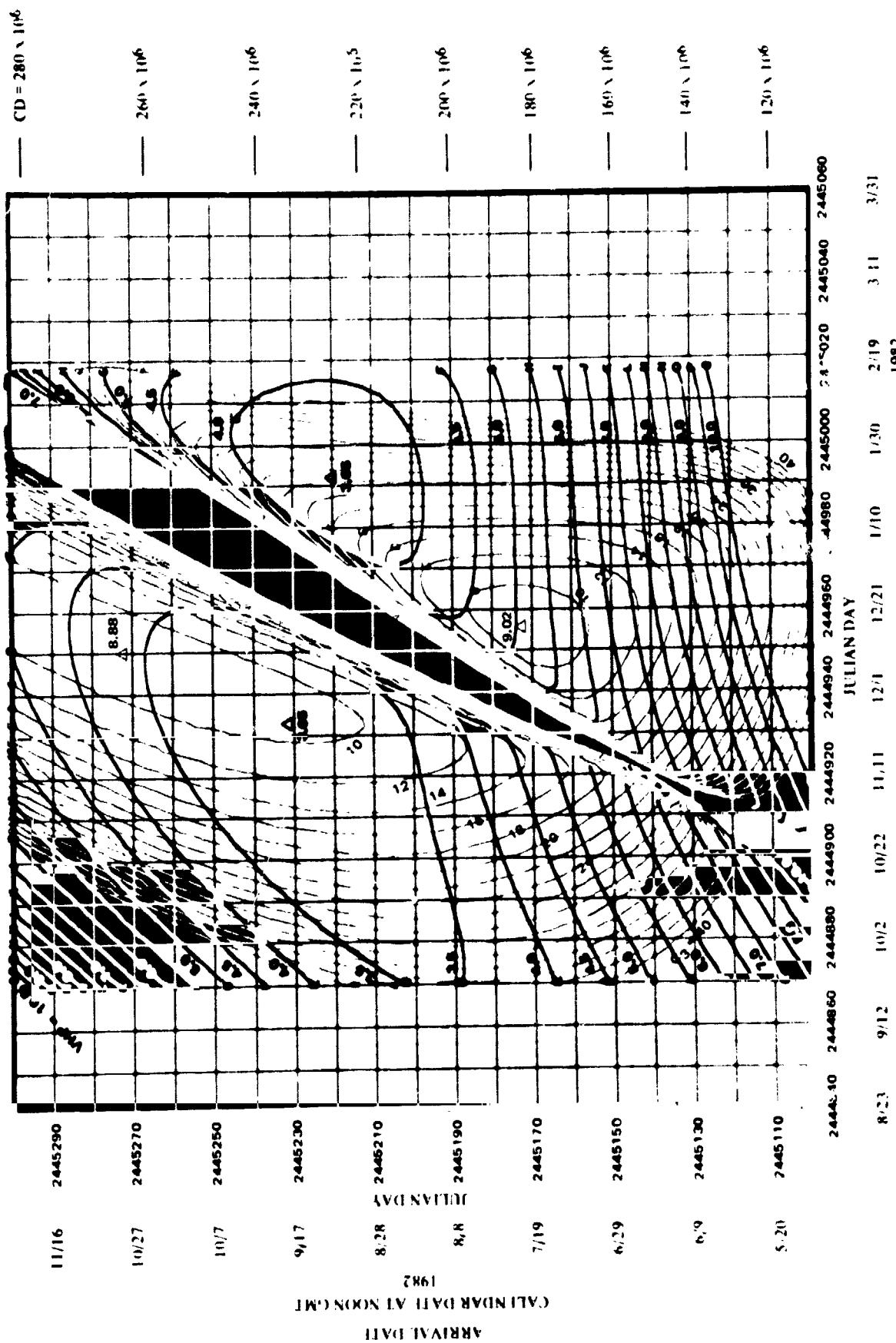
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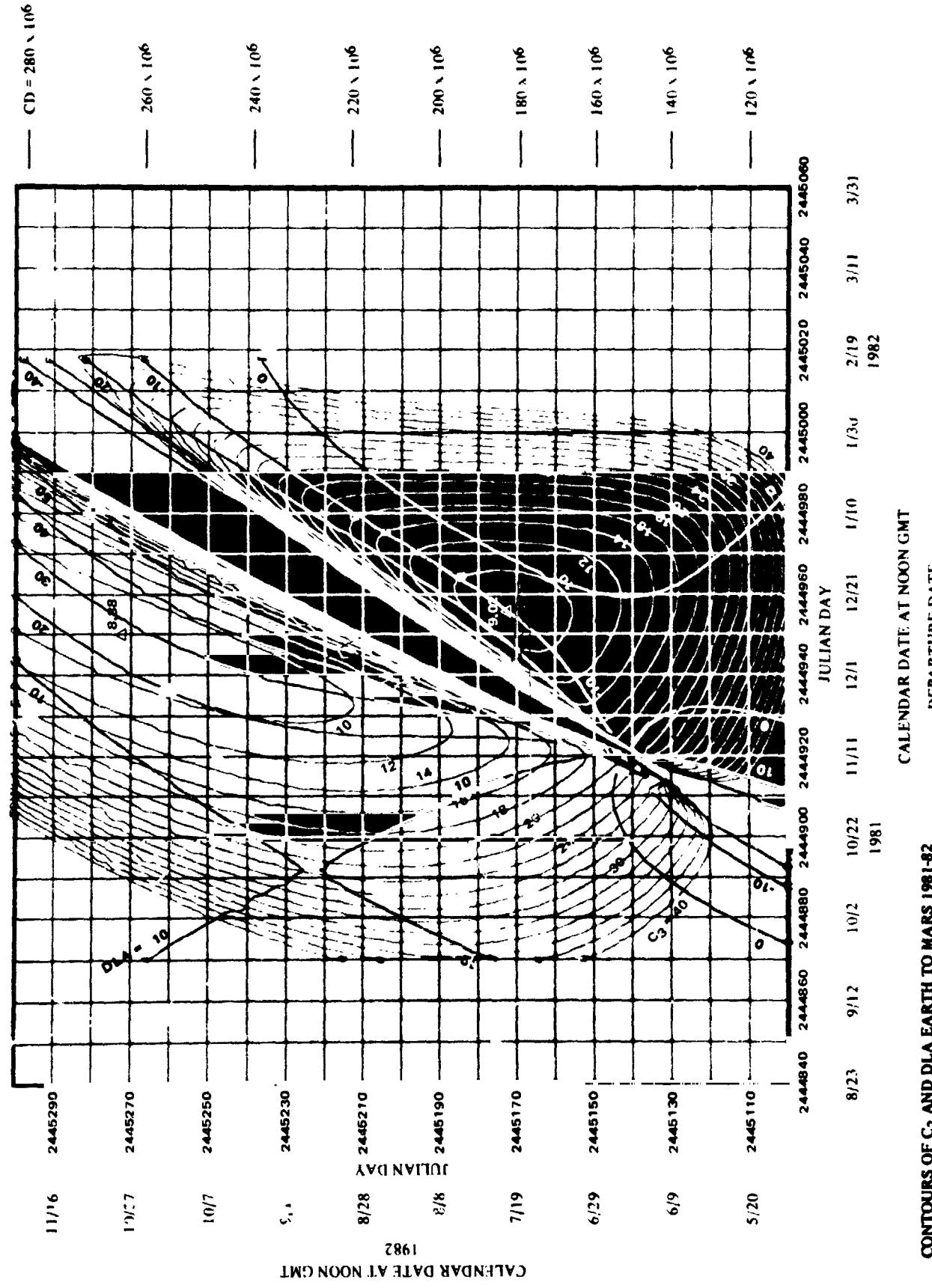
QS
1981



VHP
♂
1981



DLA
Q
1981

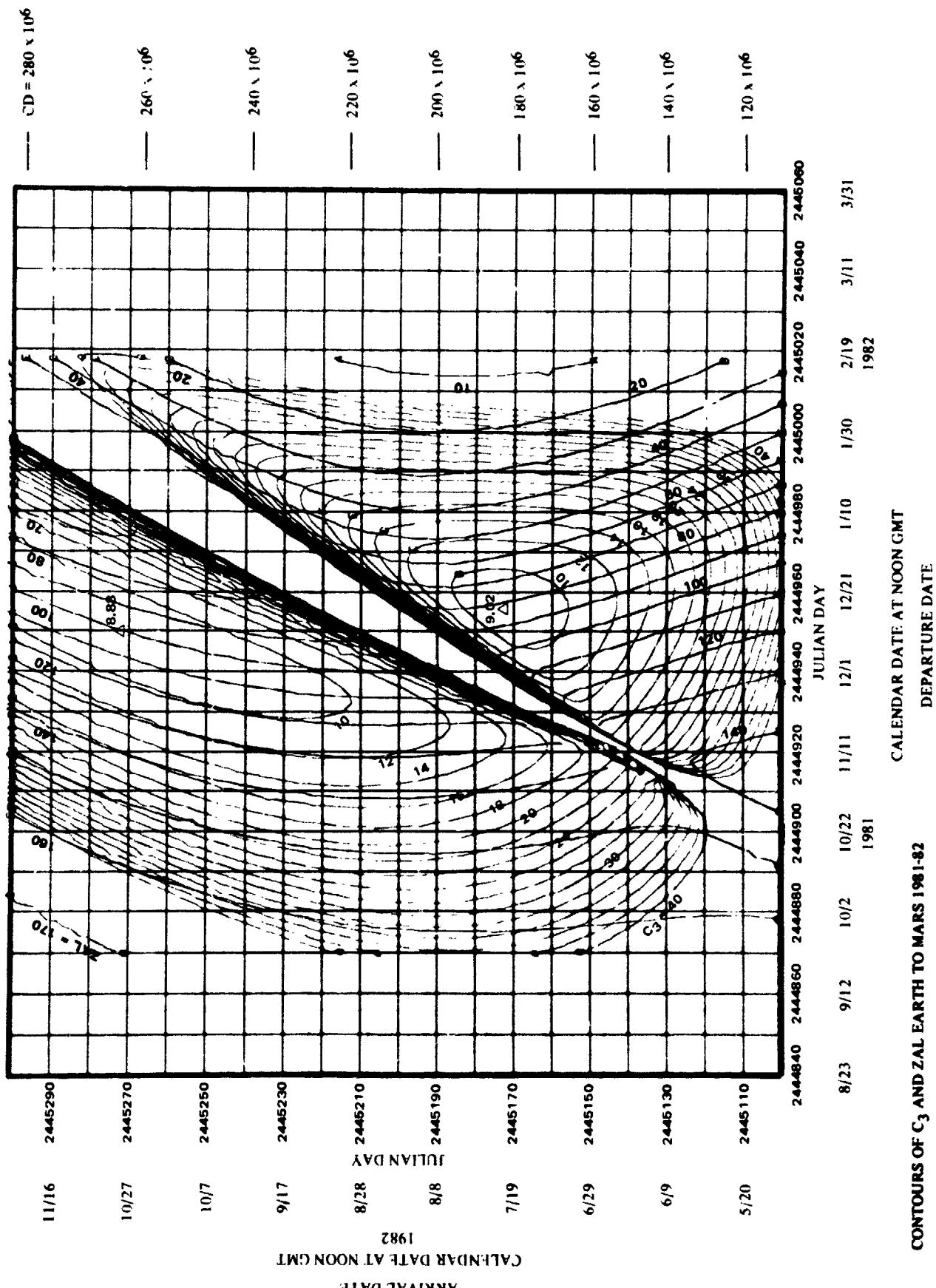


CONTOURS OF C₃ AND DLA EARTH TO MARS 1981-82

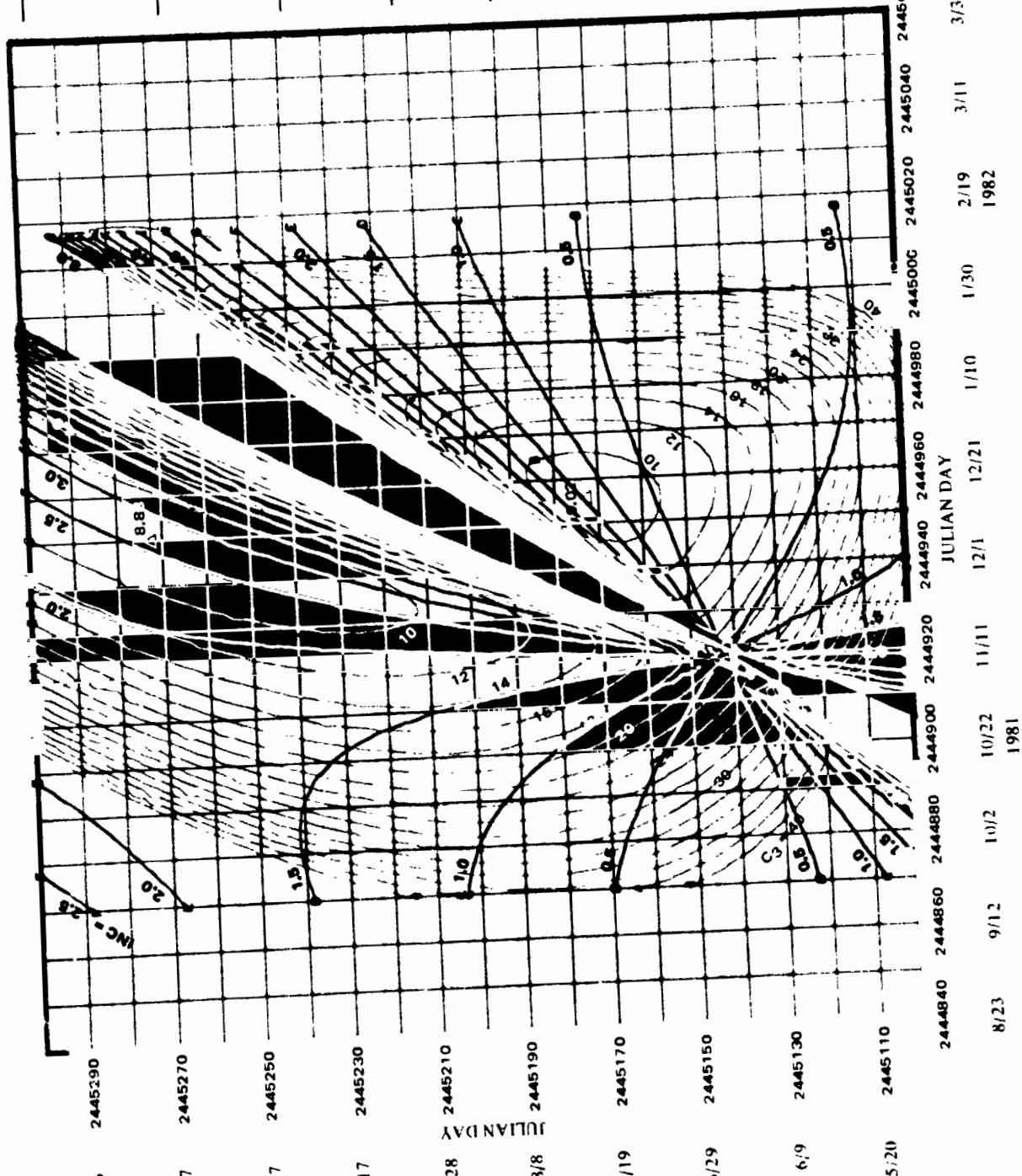
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ZAL
OA
1981



$$CD = 290 \times 10^6$$



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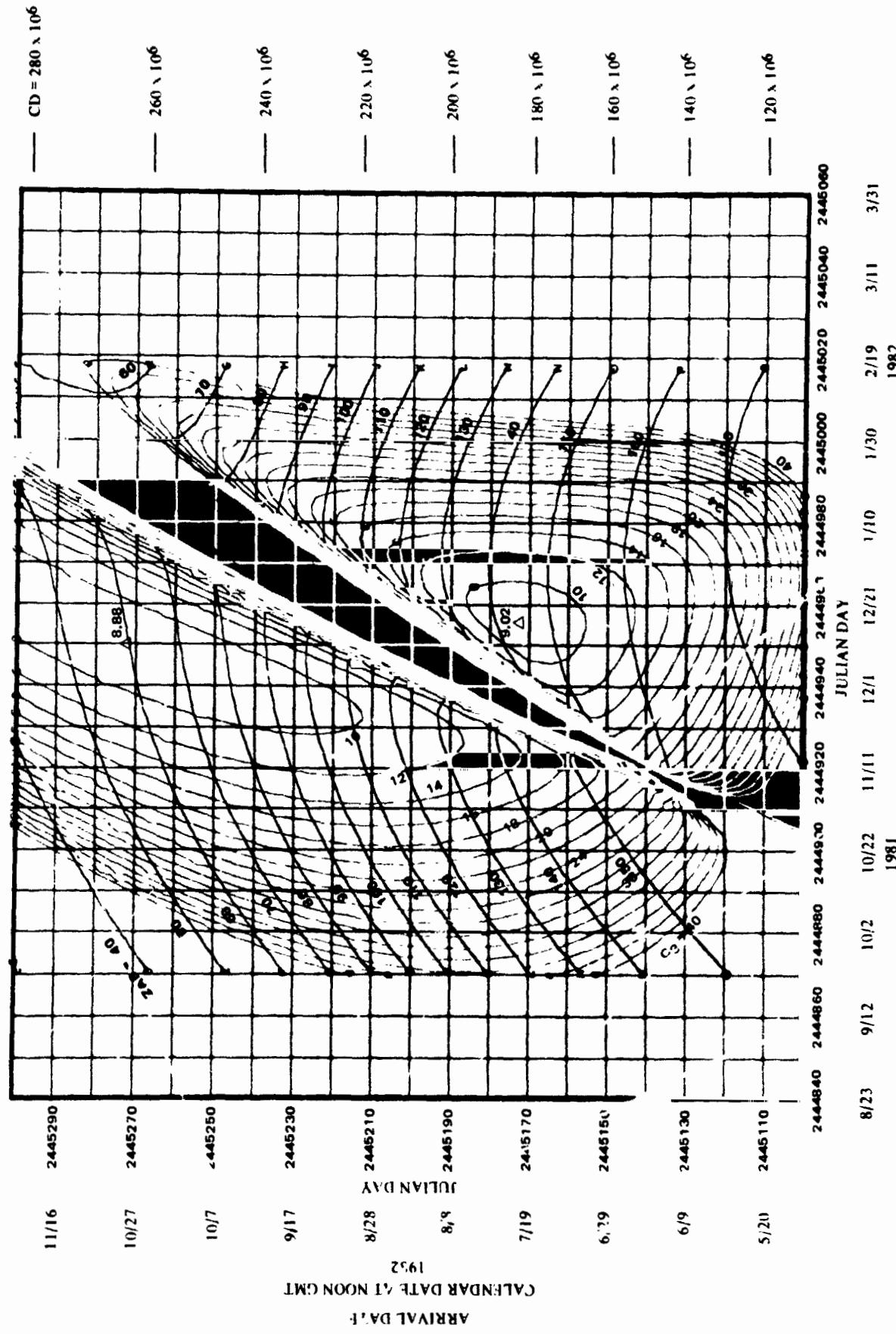
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INC
♂
1981

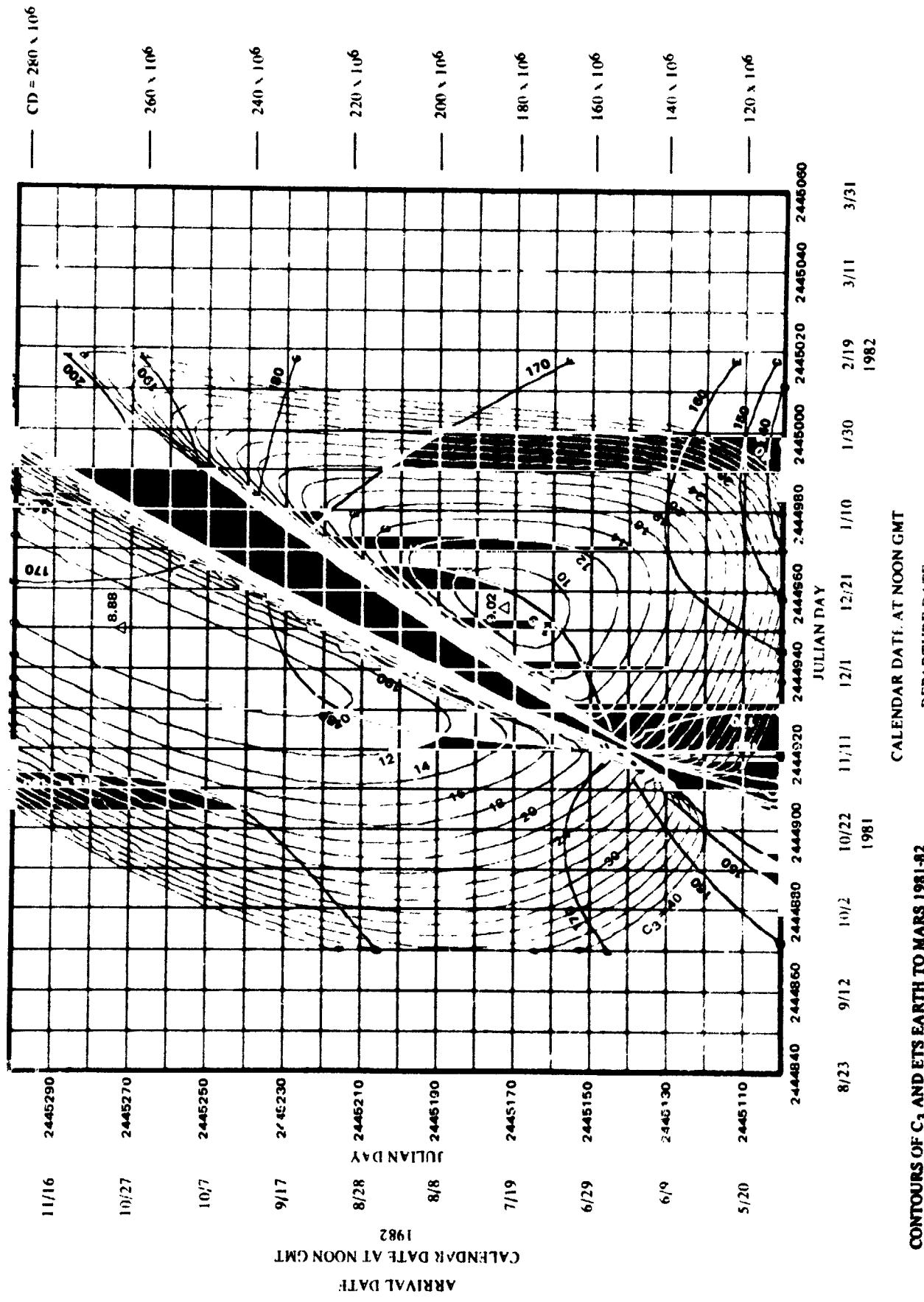
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CONTOURS OF C₃ AND INC EARTH TO MARS 1981-82

ZAP
♂
1981



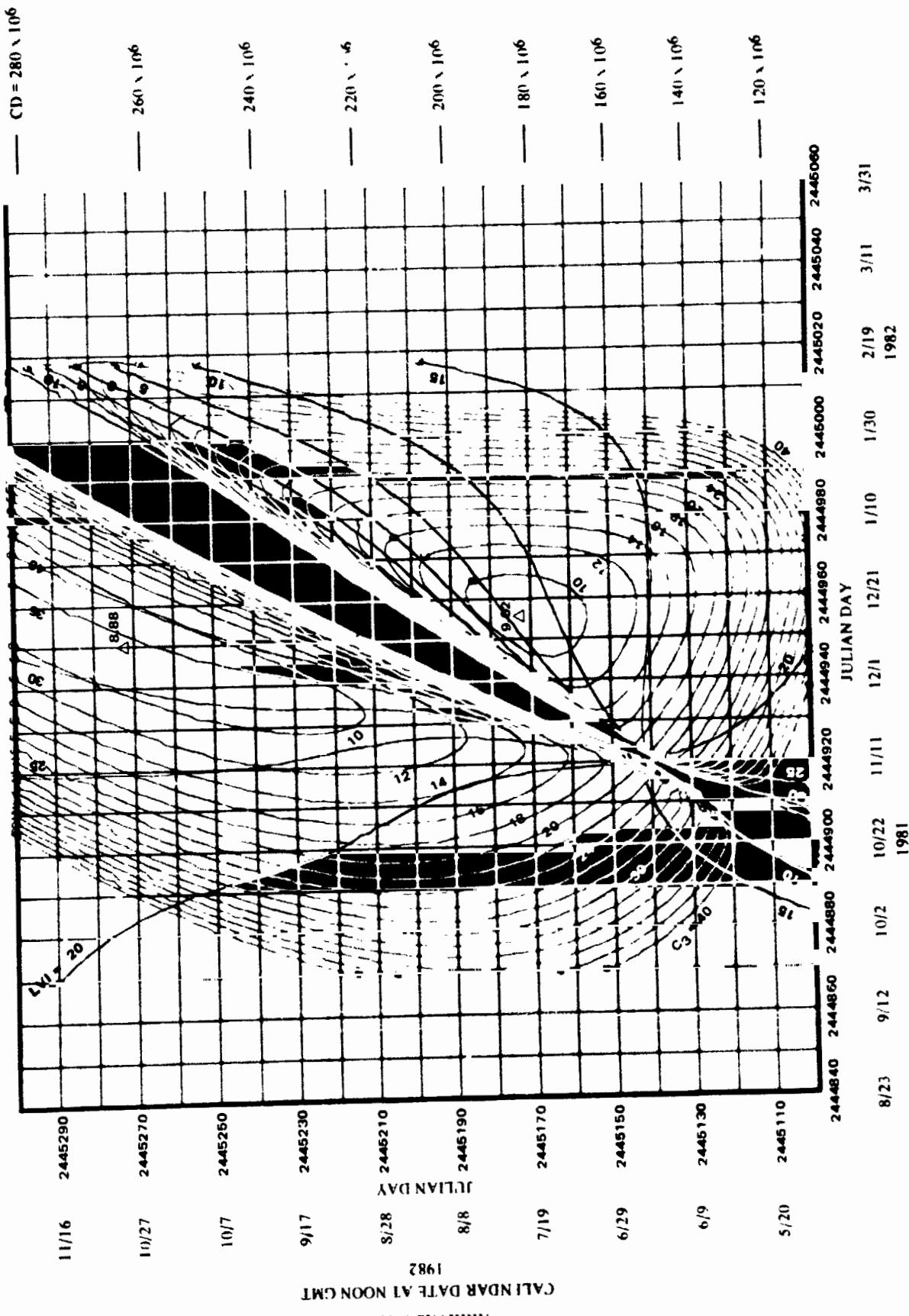
ET8
♂
1981



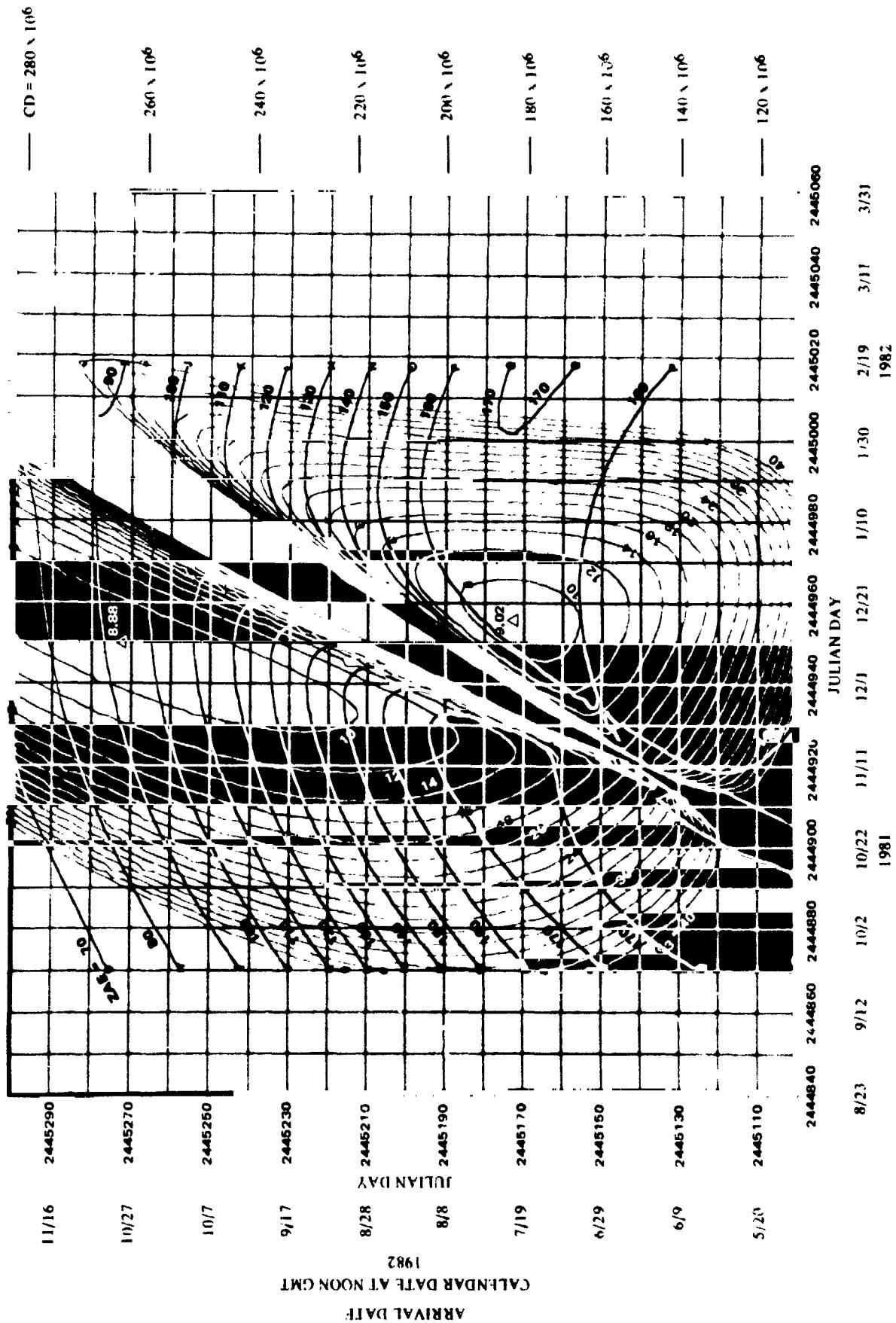
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1881
Q5

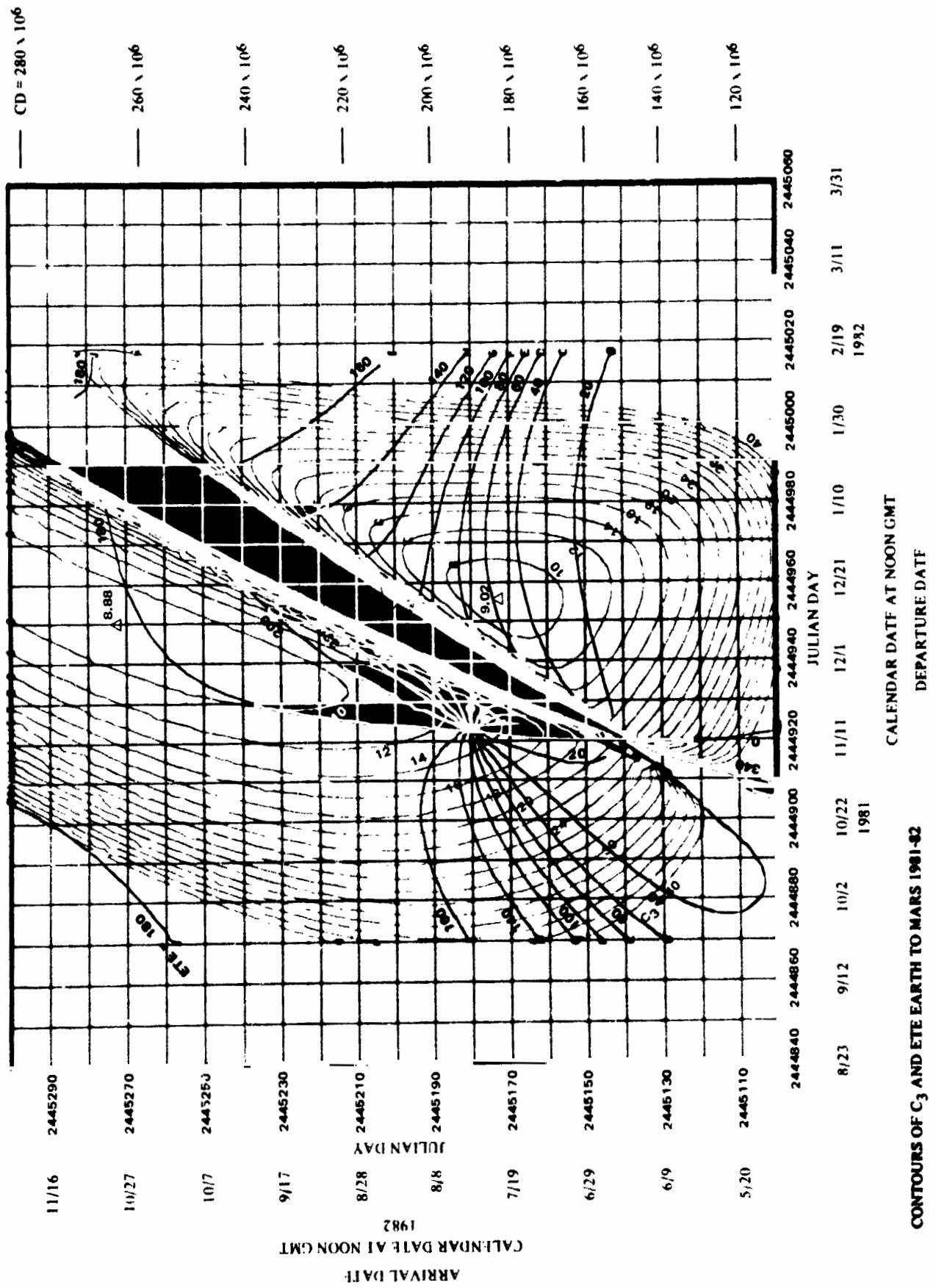
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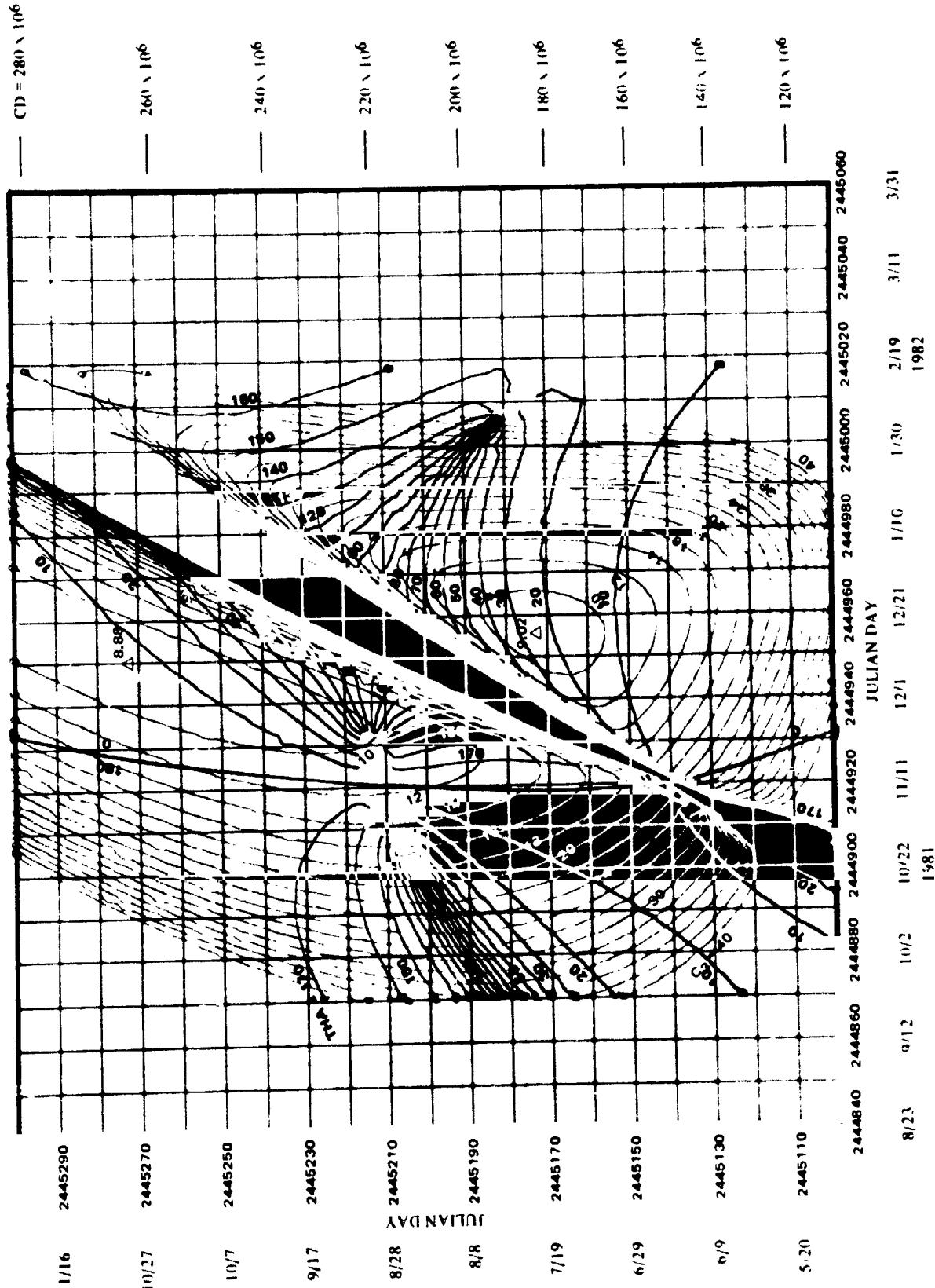
ZAE
1981



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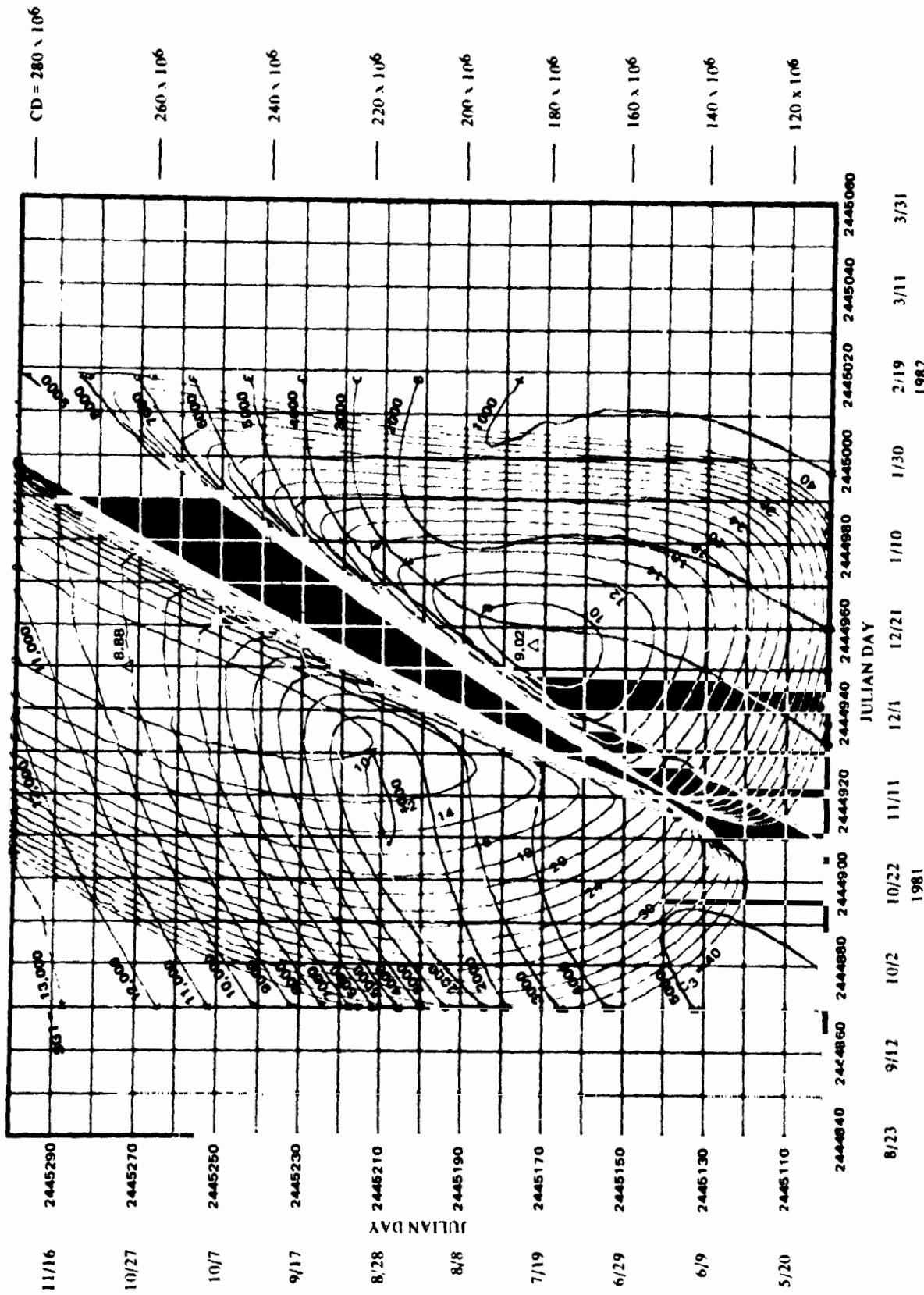


THA
Q1
1981

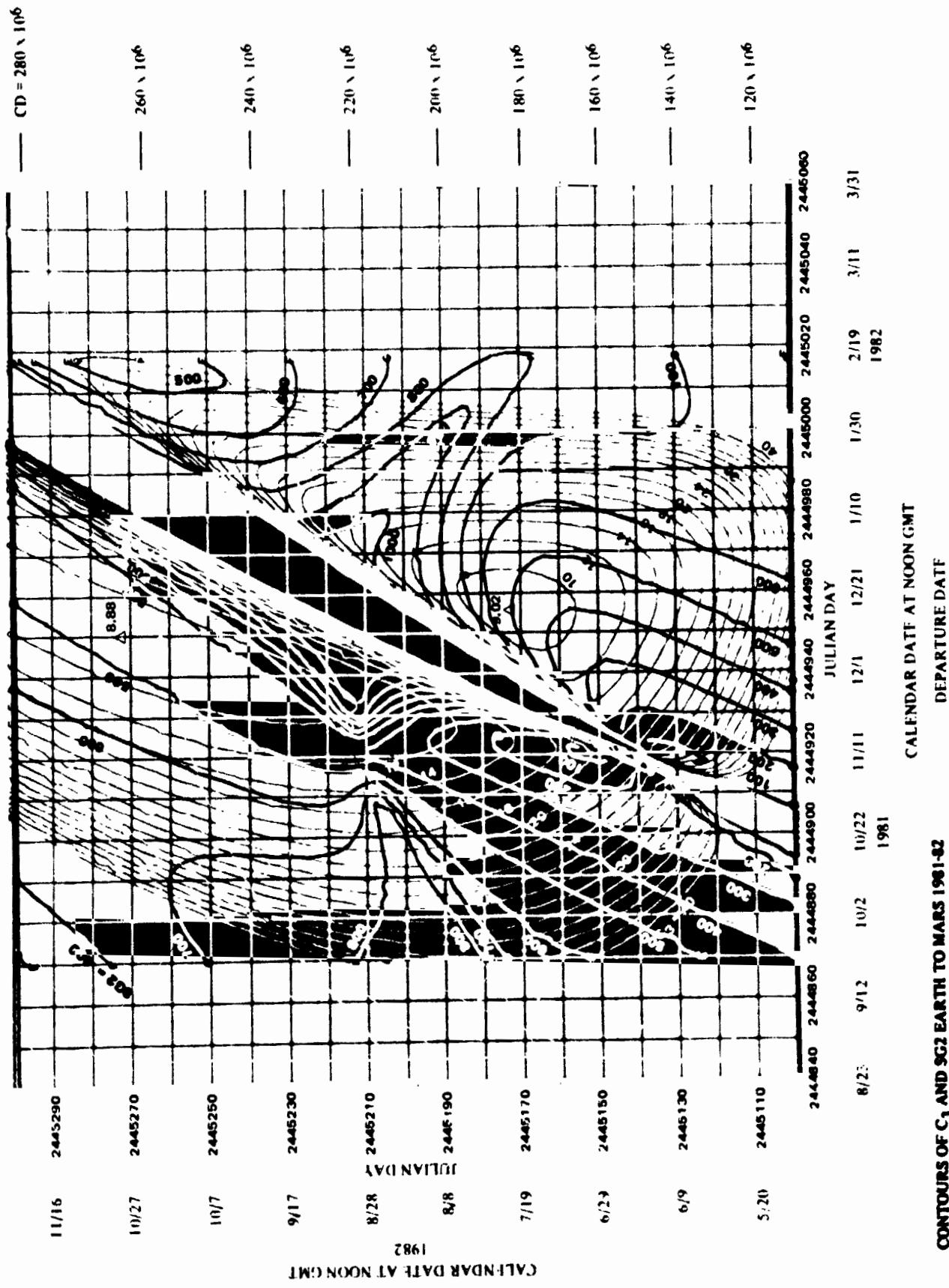


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SG1
1981

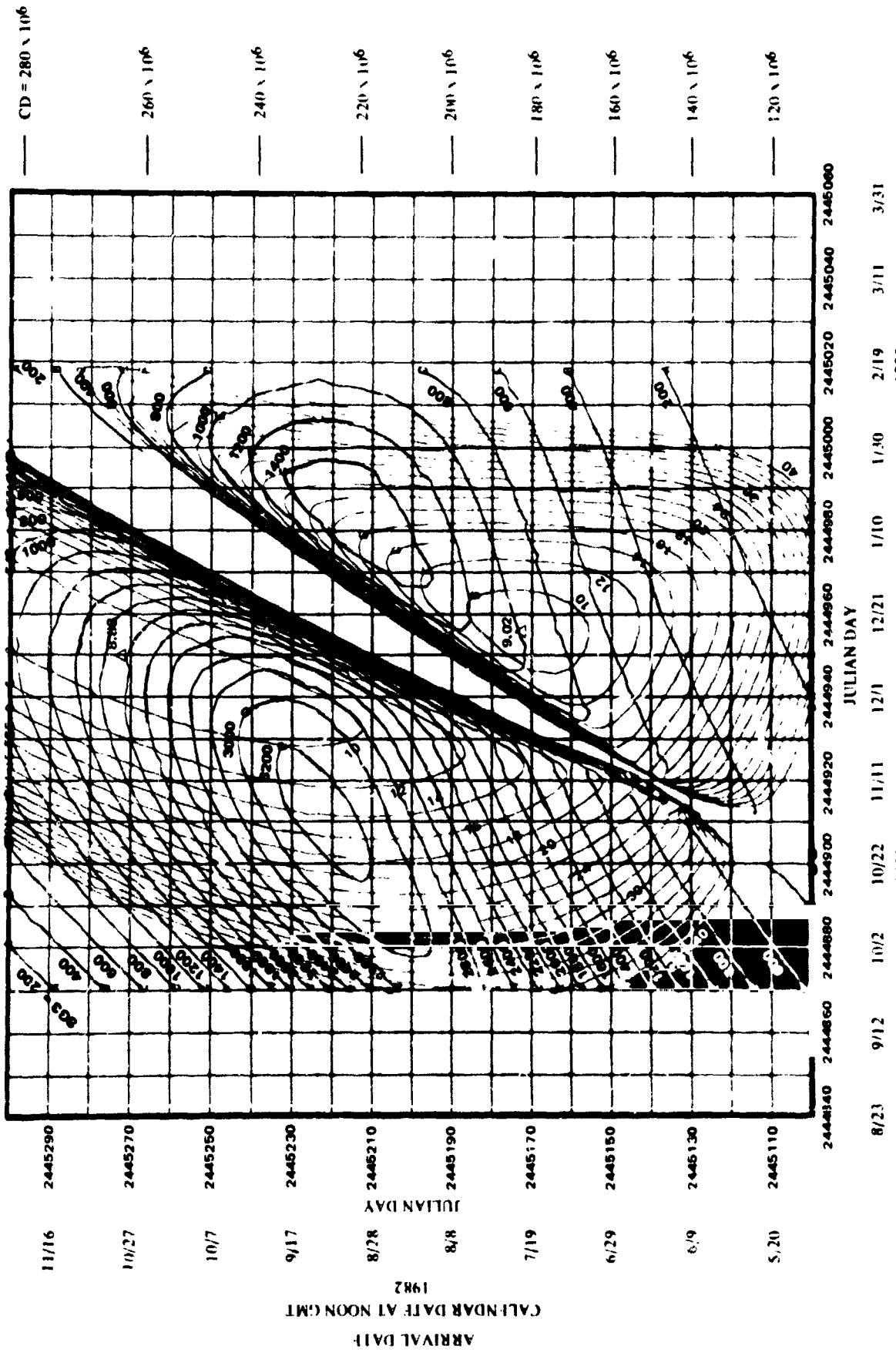


SG2
Q
1981



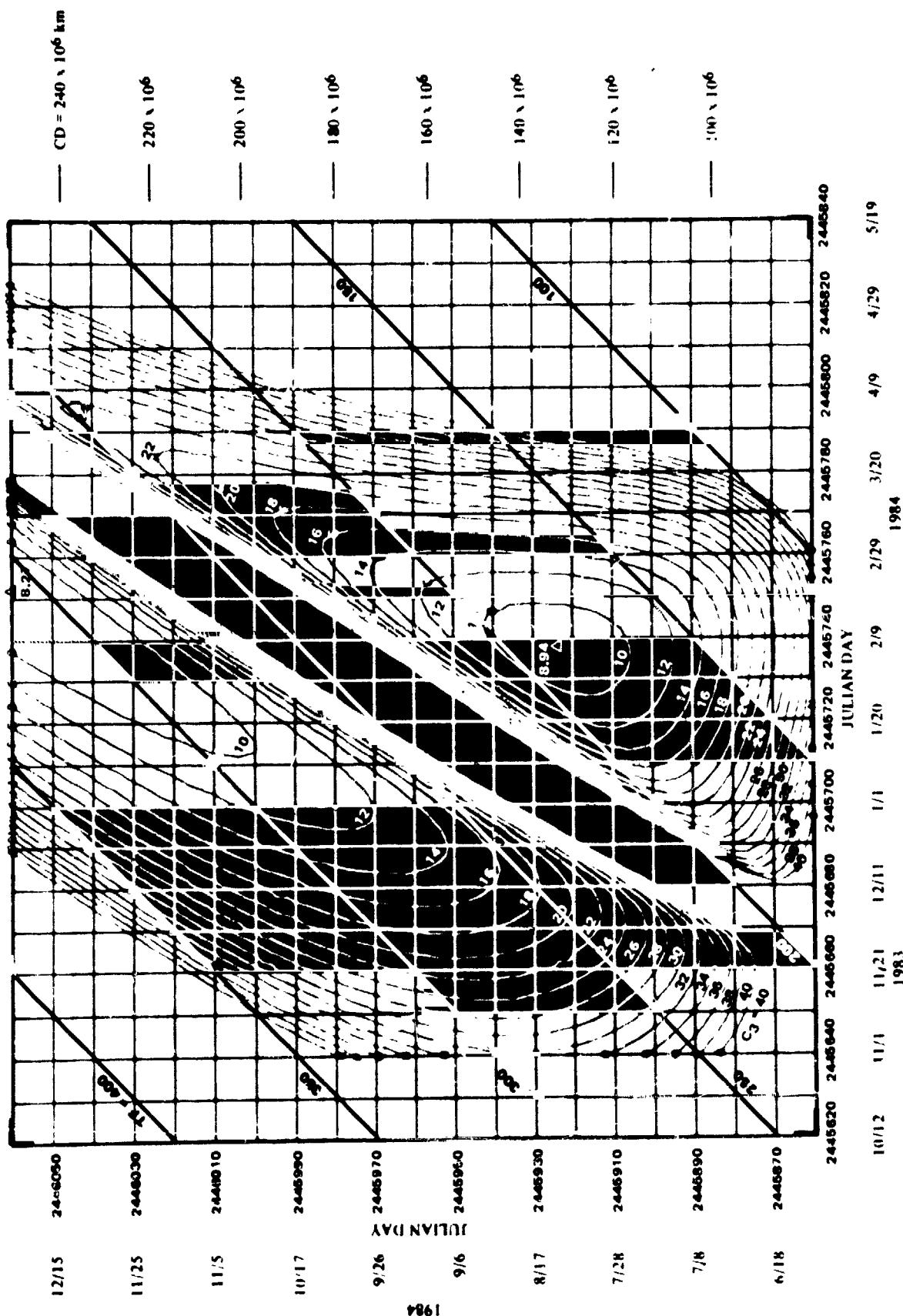
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SG3
♂
1981



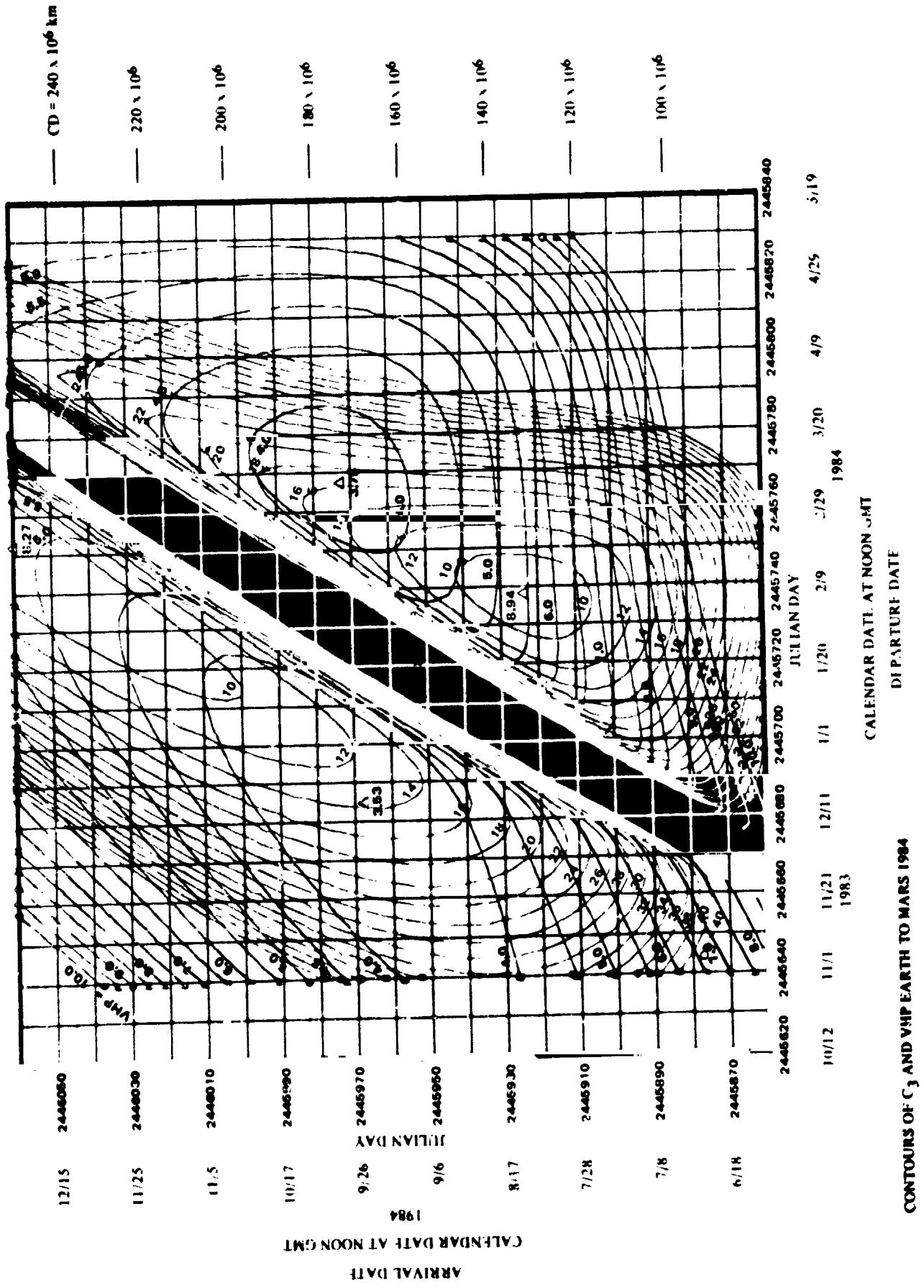
1984
C₃
Q₃

CONTOURS OF C₃ AND FLIGHT TIMES EARTH TO MARS 1984
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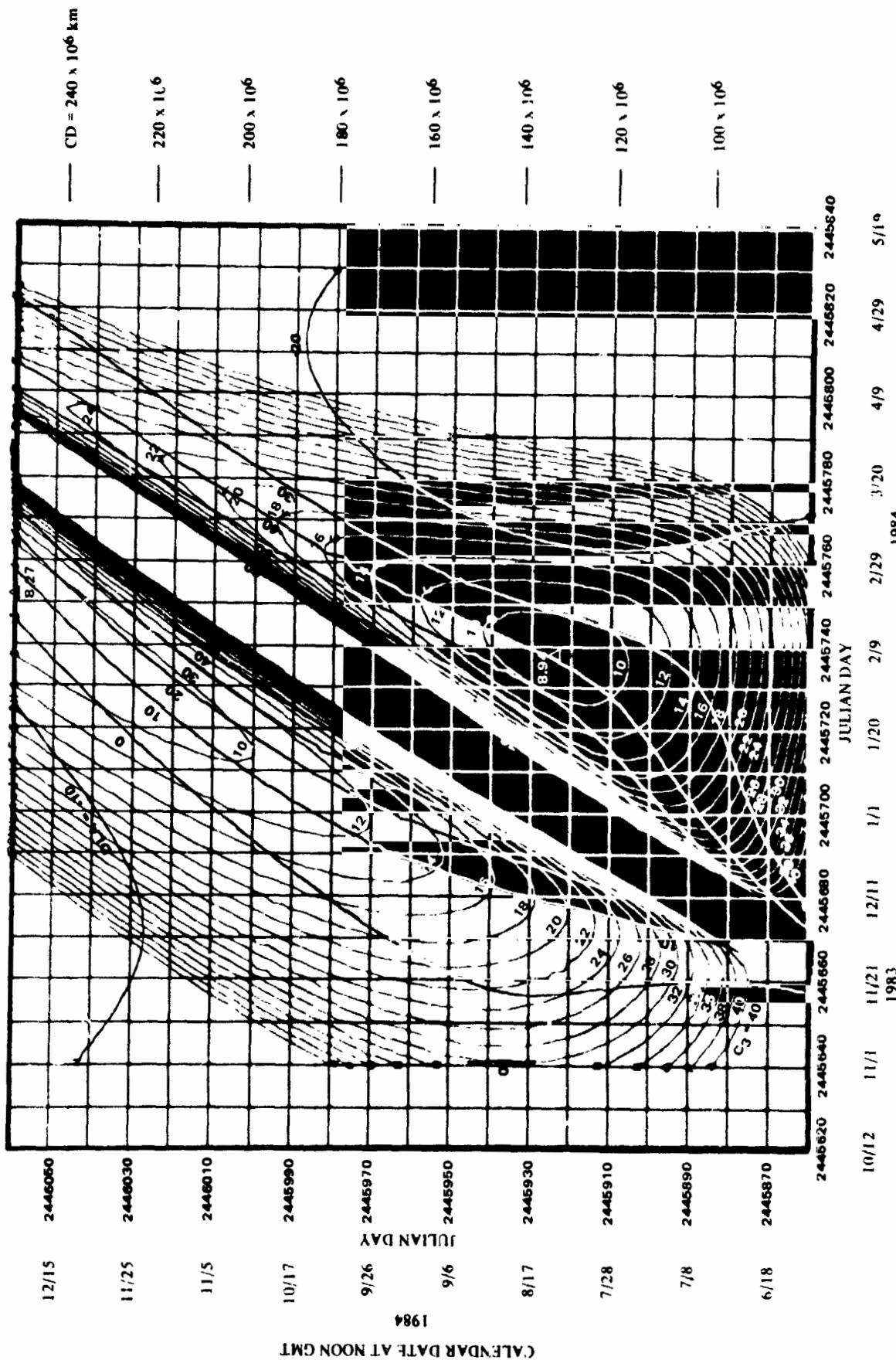


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VHP
1984



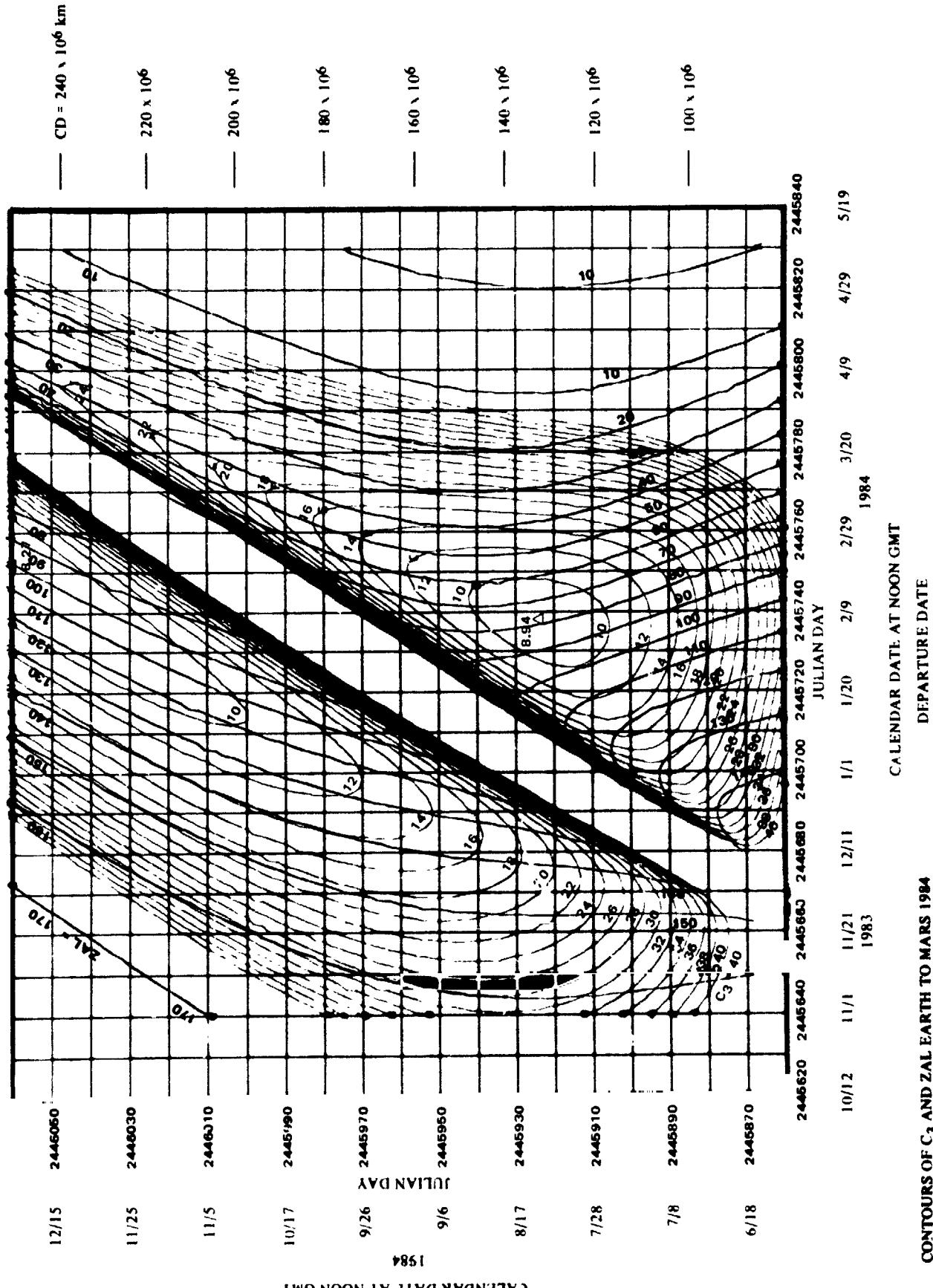
DLA
♂
1984



CONTOURS OF C₁ AND DLA EARTH TO MARS | 291

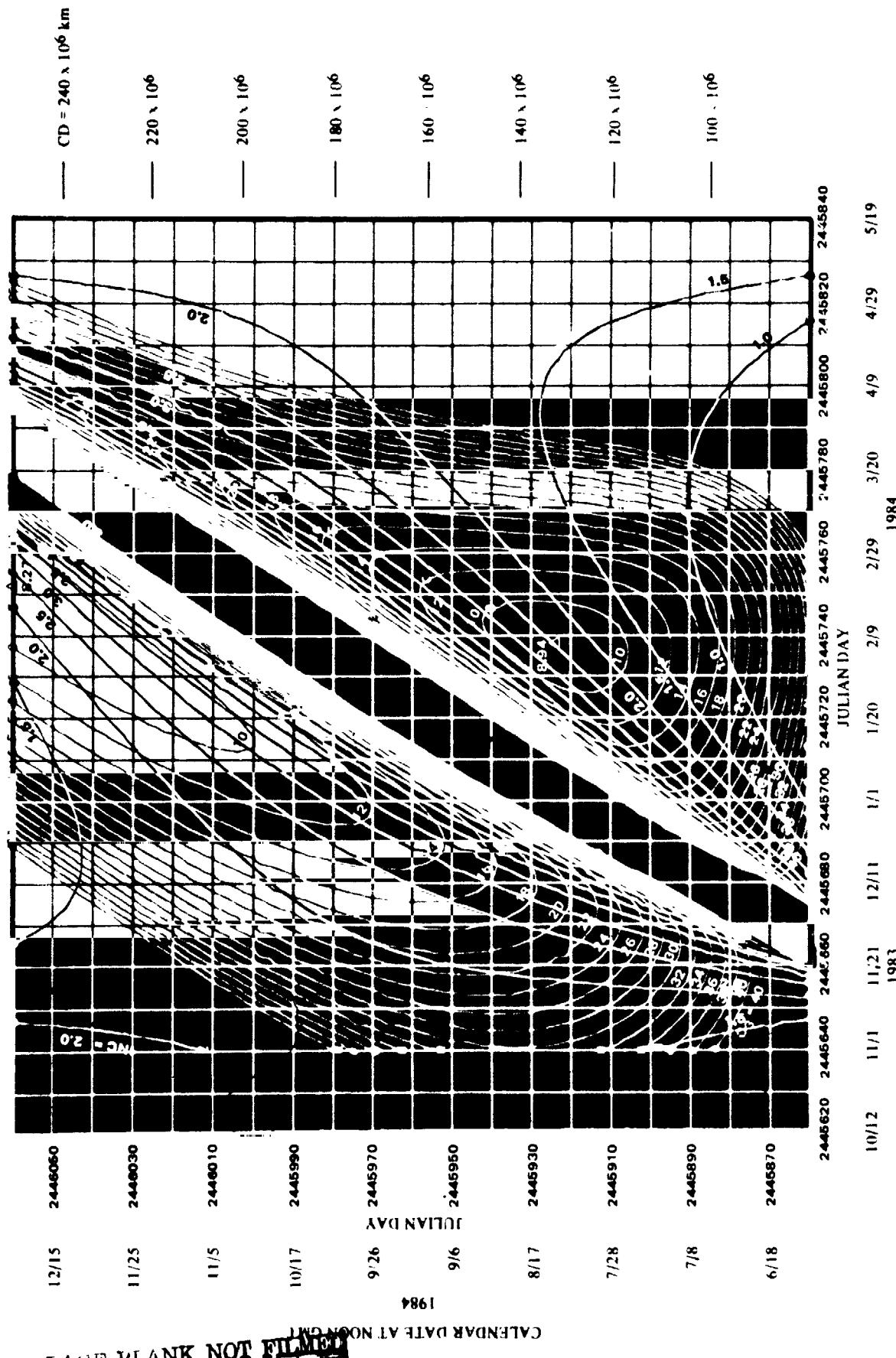
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Q1
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INC
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1984

CONTOURS OF C_3 AND INC EARTH TO MARS 1984
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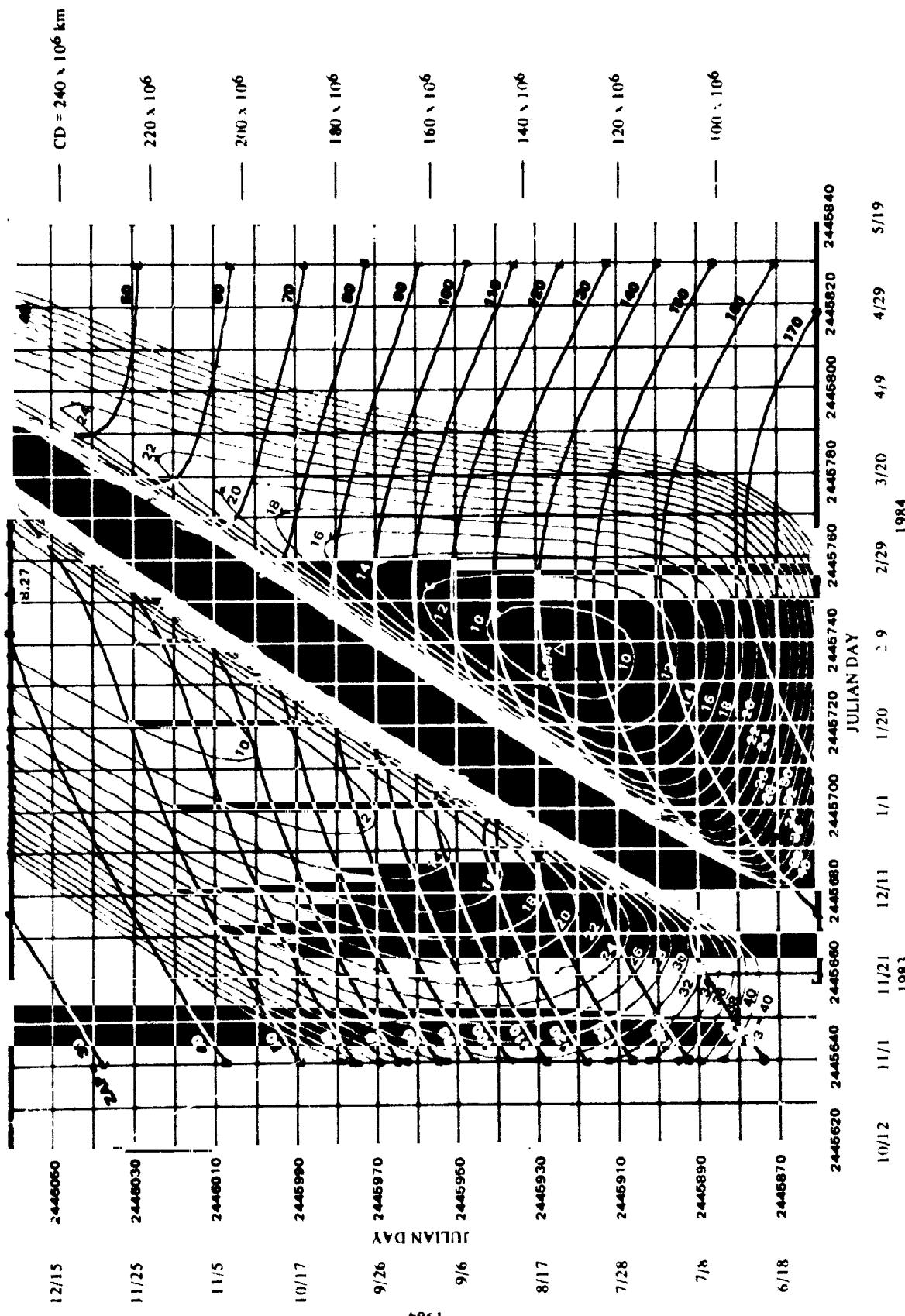


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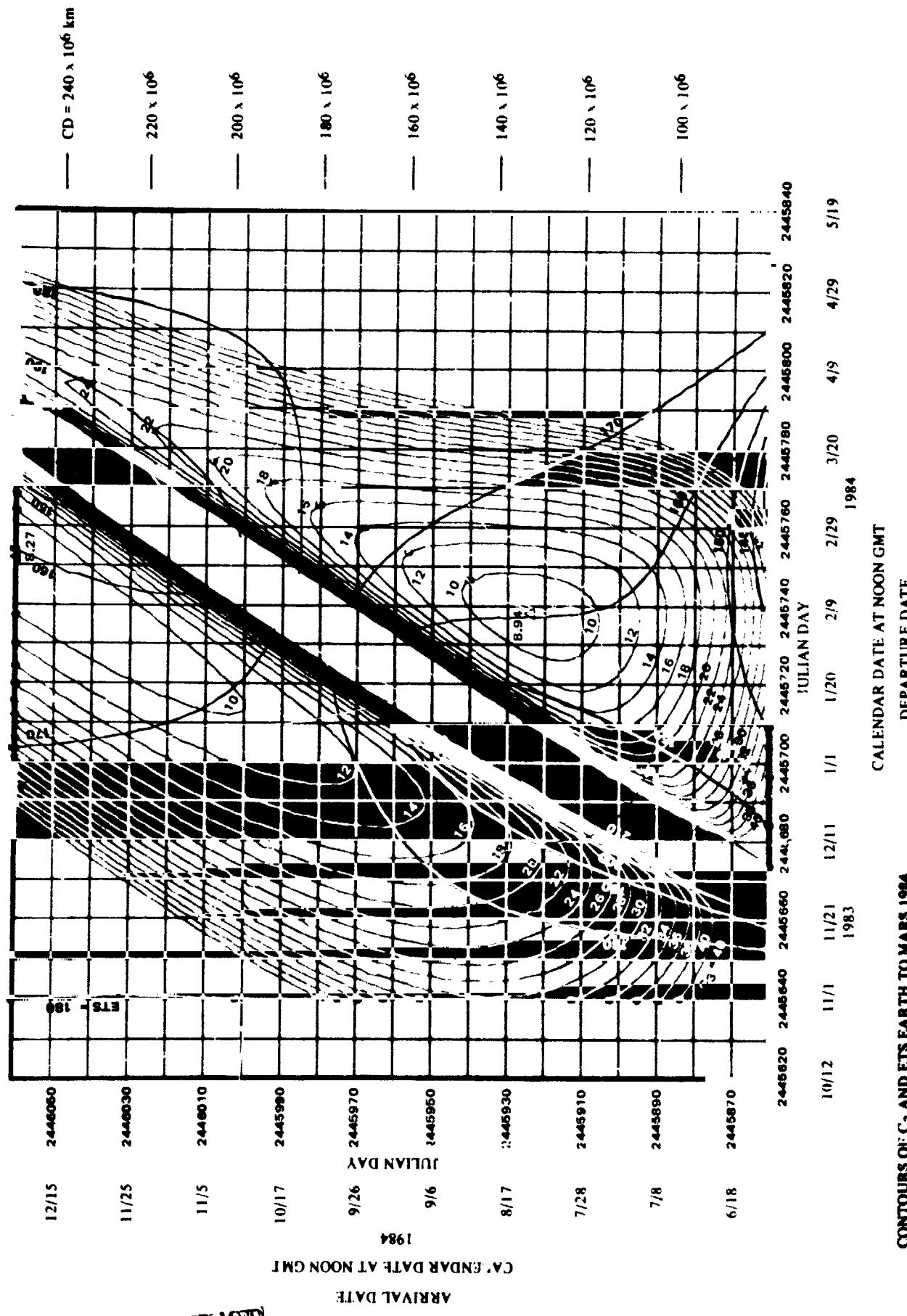
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CALENDAR DATE AT NOON GMT

DEPARTURE DATE

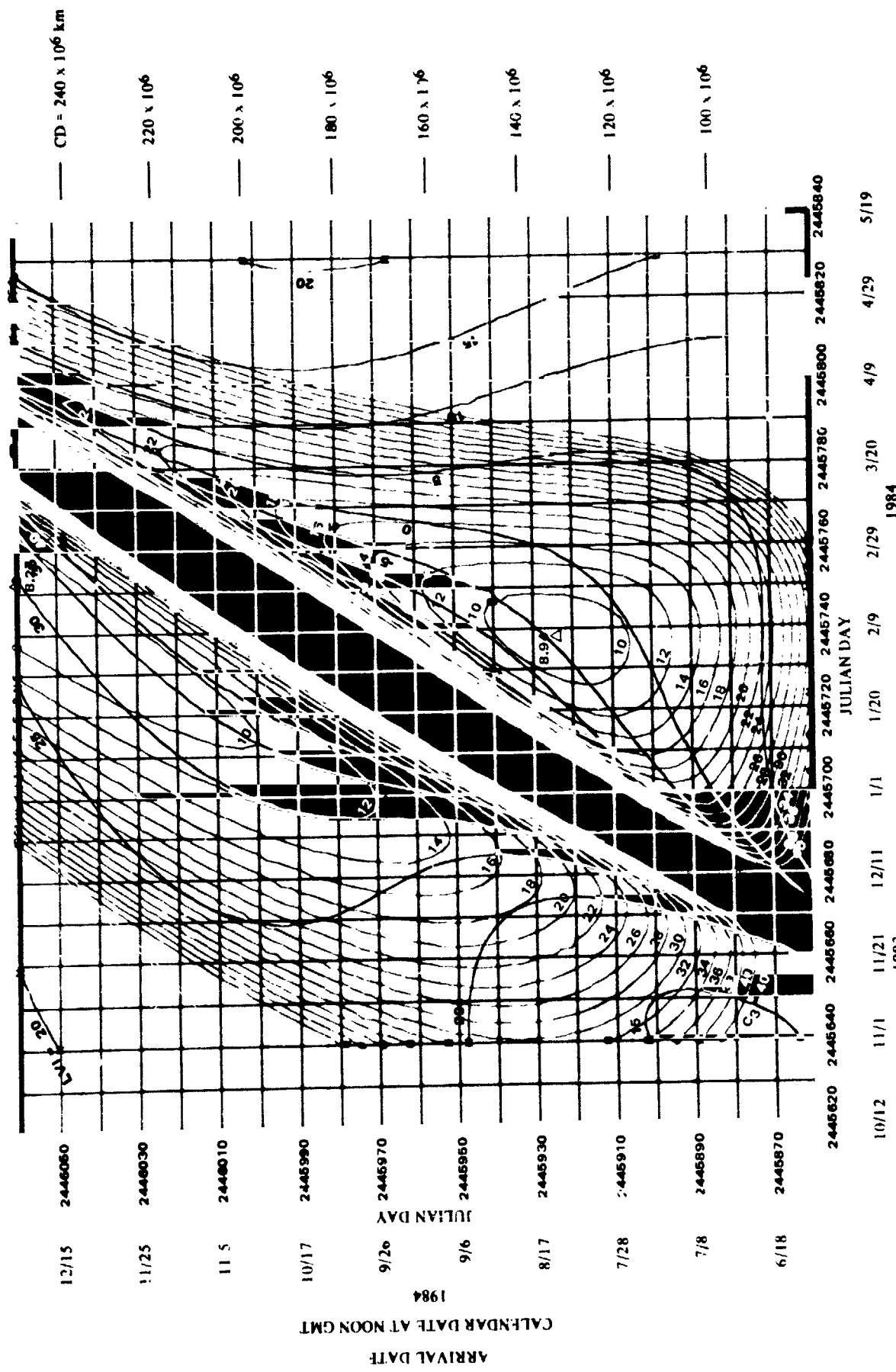


ETS
Q1
1984

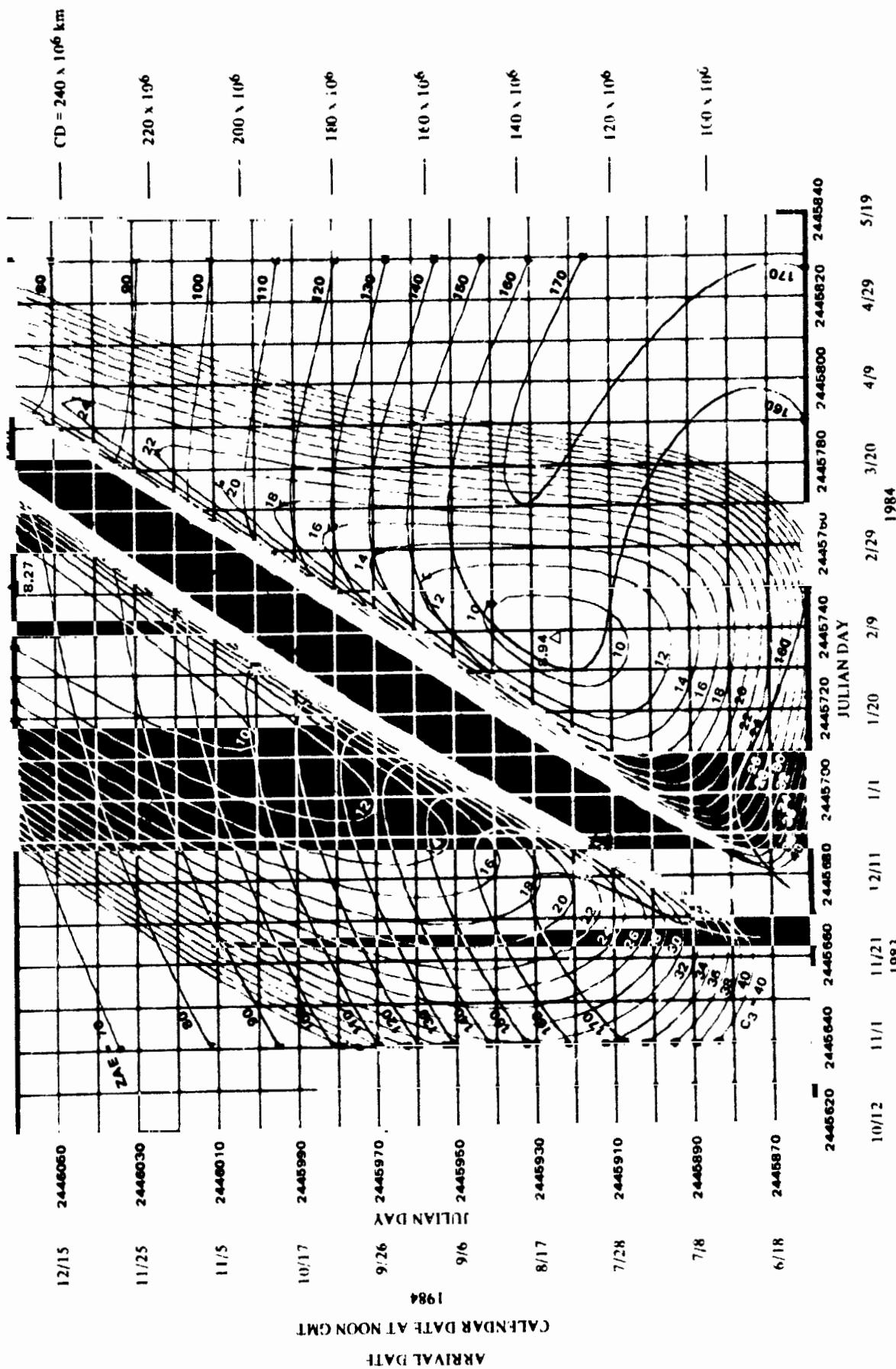


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Q₁
1984



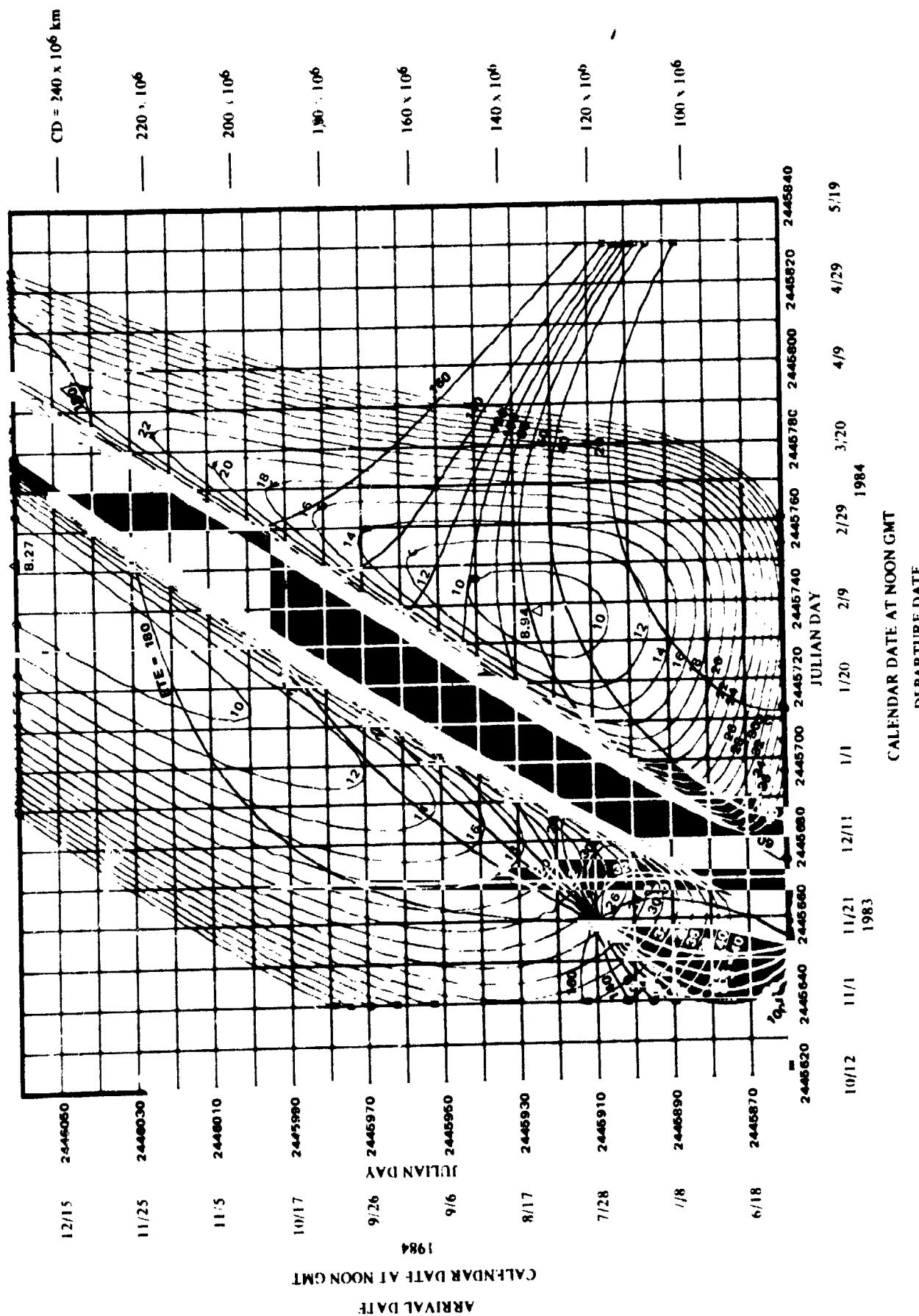
ZAE
1984



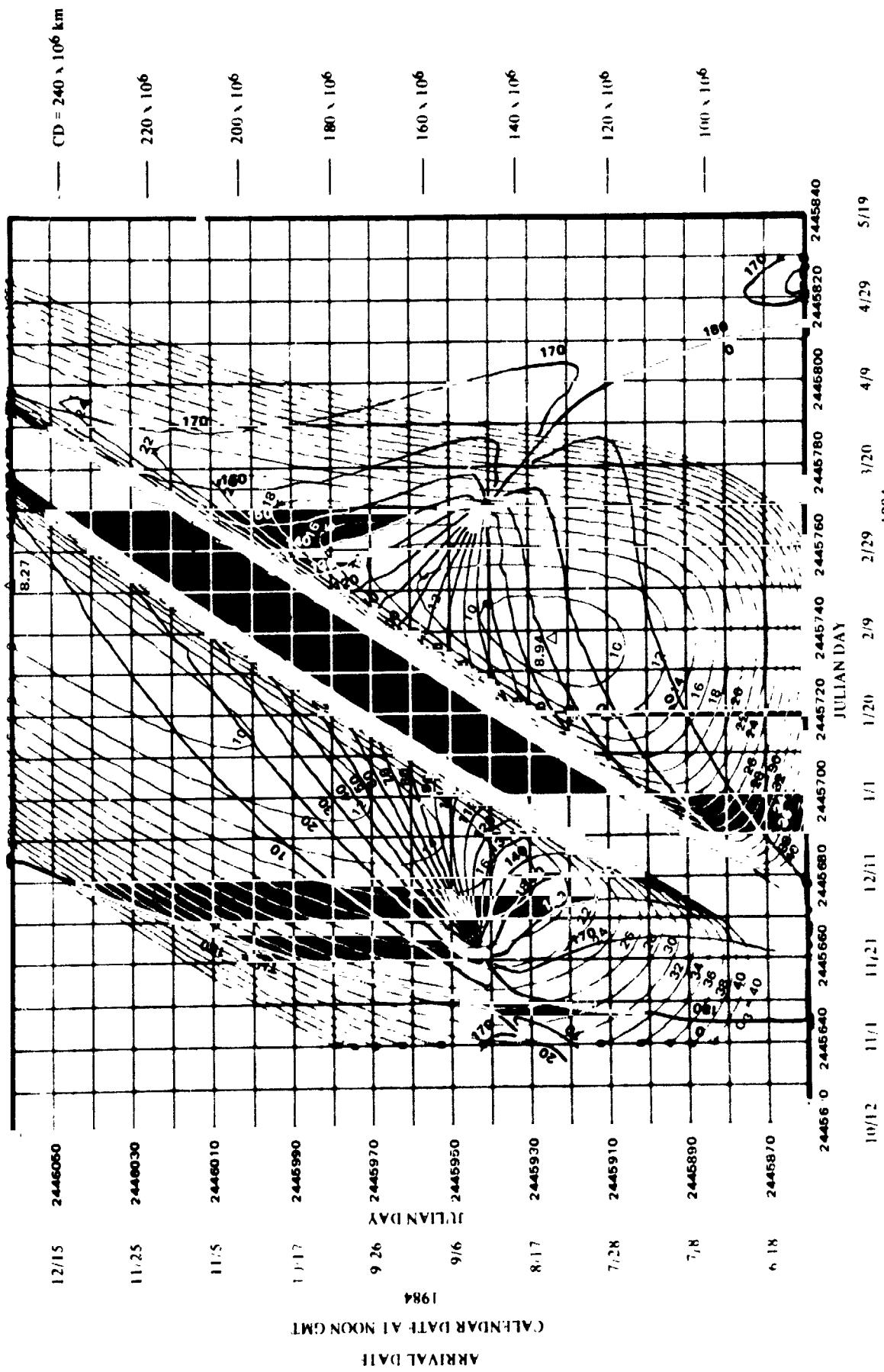
CONTOURS OF C₃ AND ZAE EARTH TO MARS 1984

CALENDAR DATE AT NOON GMT
DEPARTURE DATE

ETE
♂
1984

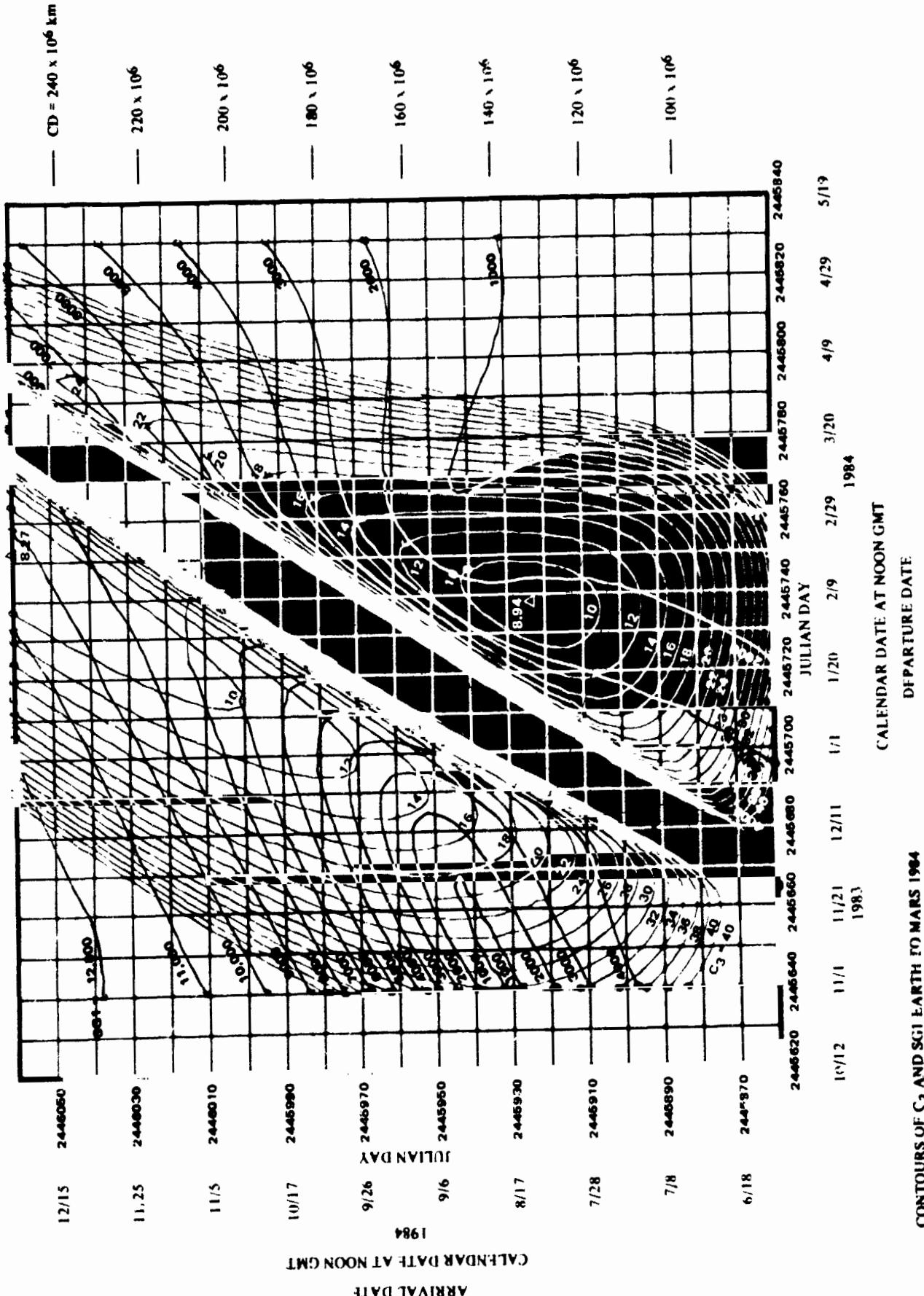


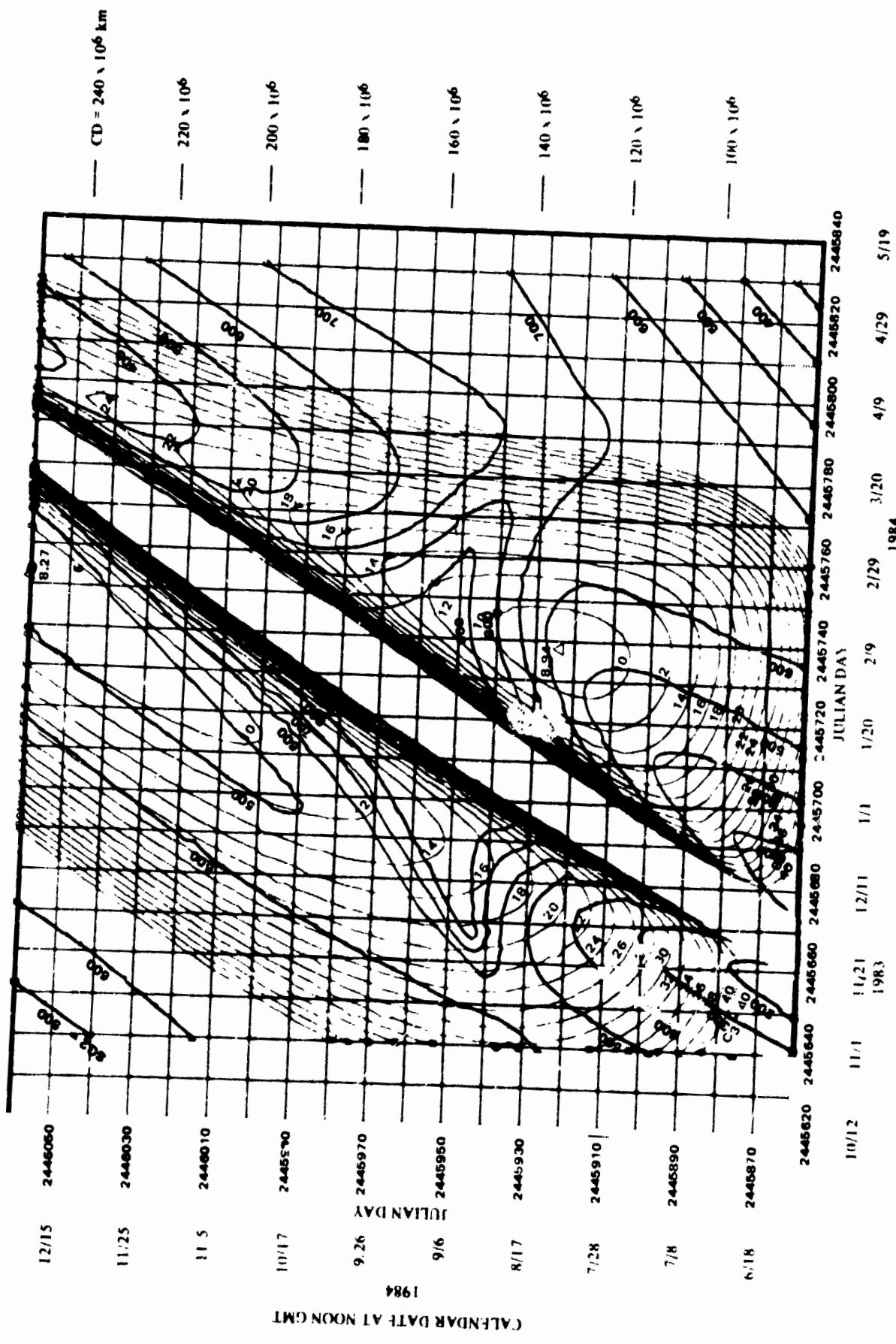
THA
1984



CONTOURS OF C_3 AND THA EARTH TO MARS 1984

CALENDAR DATE AT NOON GMT
DEPARTURE DATE





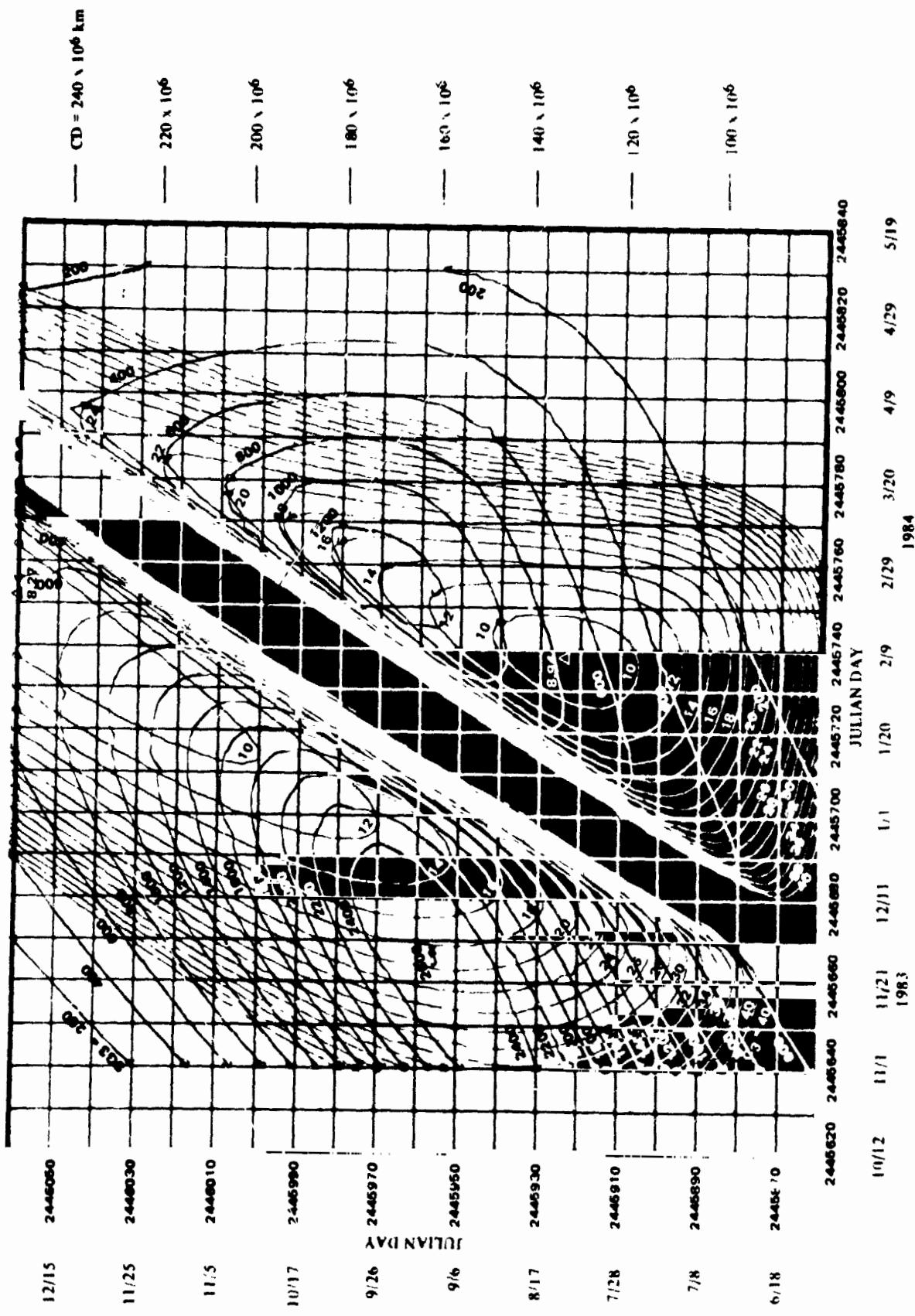
CONTOURS OF C₃ AND SG2 EARTH TO MARS 1984

JULIAN DAY
CALENDAR DATE AT NOON GMT

DEPARTURE DATE

SG2
1984 Q1

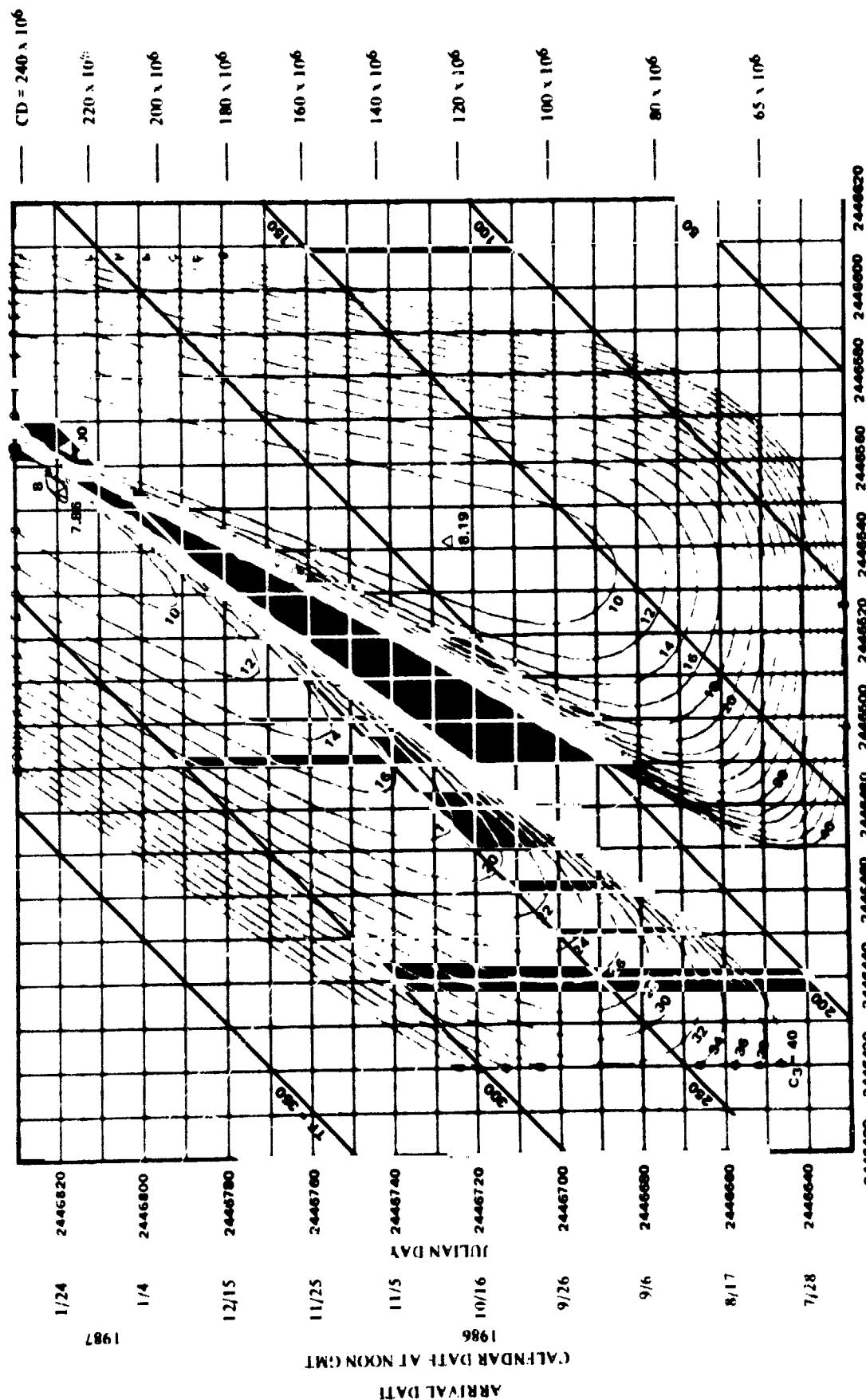
SG3
♂
1994



CONTOURS OF C_3 AND SG3 EARTH TO MARS 1984

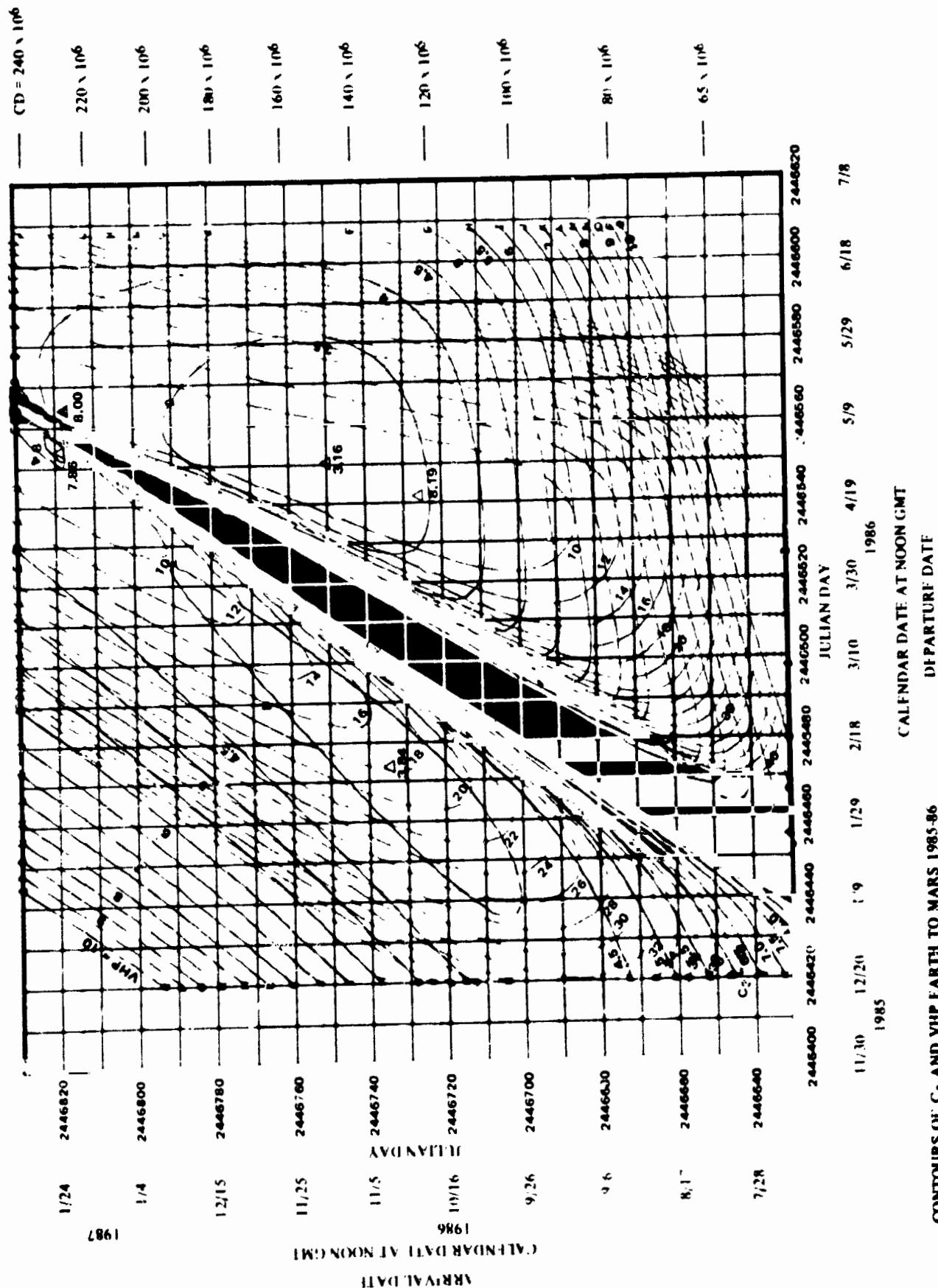
CALENDAR DATE AT NOON GMT
DEPARTURE DATE

1985 Q₃



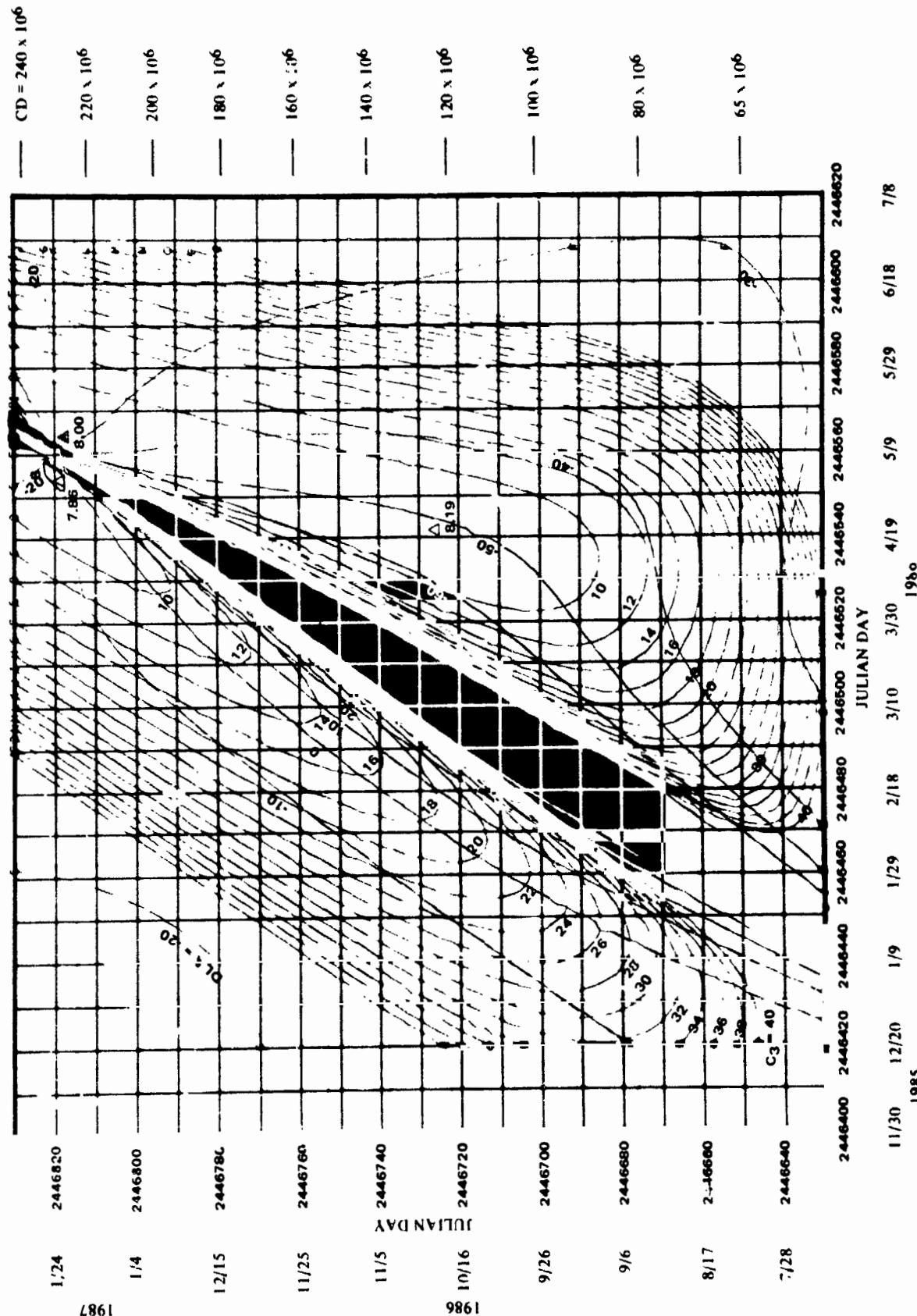
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VHP
1985



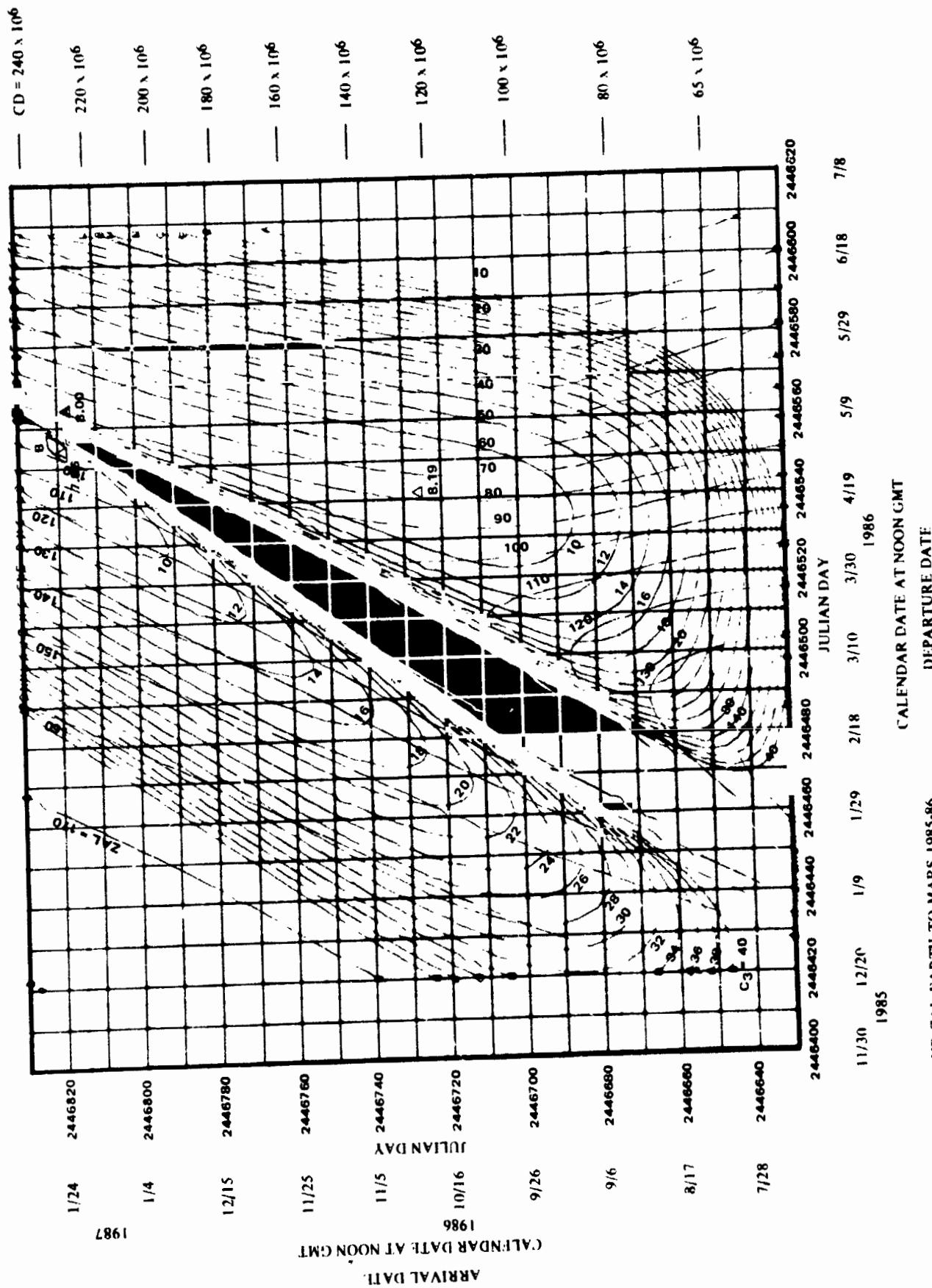
DLA
1985

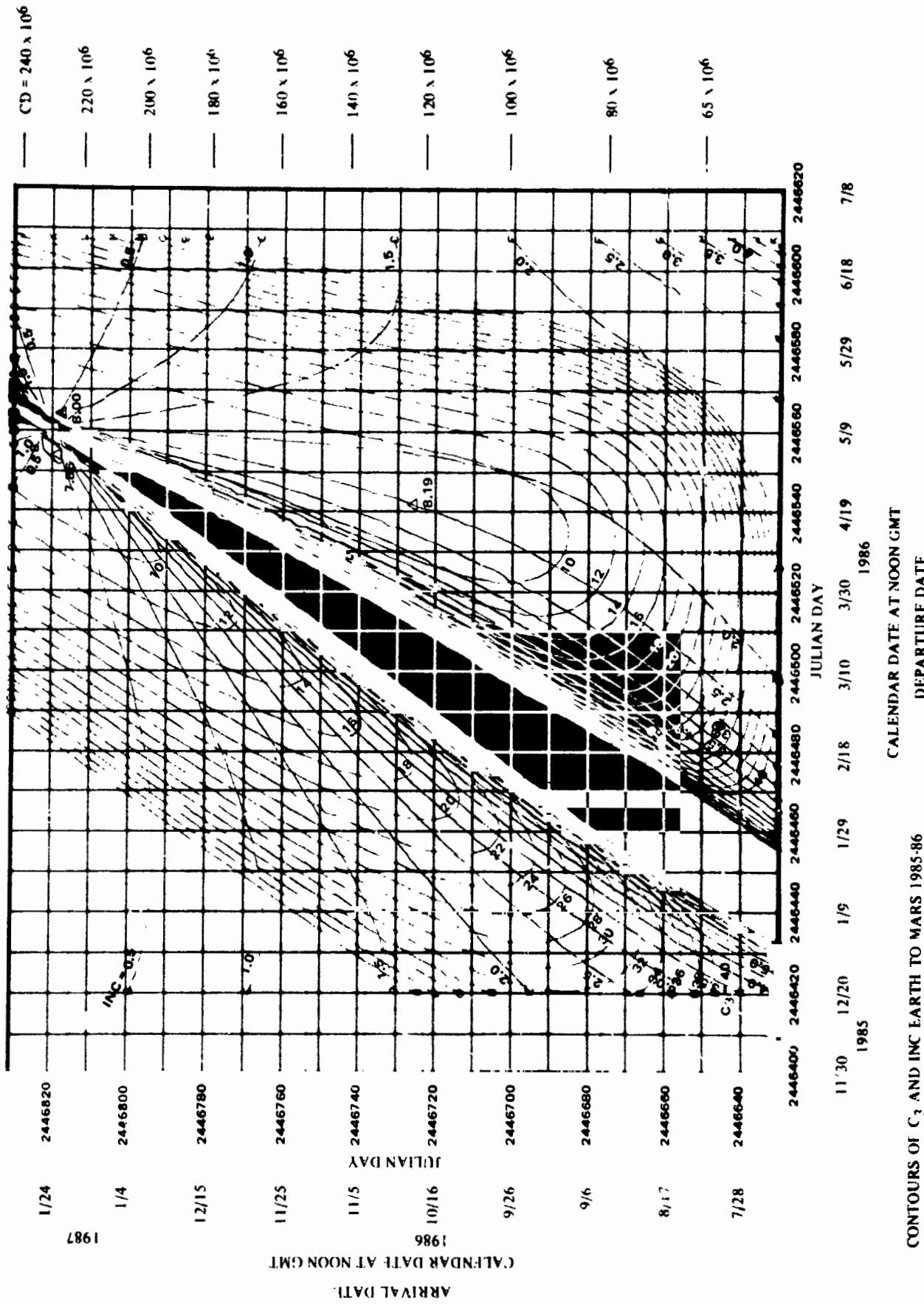
CONTOURS OF C₃ AND DLA EARTH TO MARS 1985-86
ARRIVAL DATE AT NOON GMT
CALENDAR DATE AT NOON GMT
DEPARTURE DATE



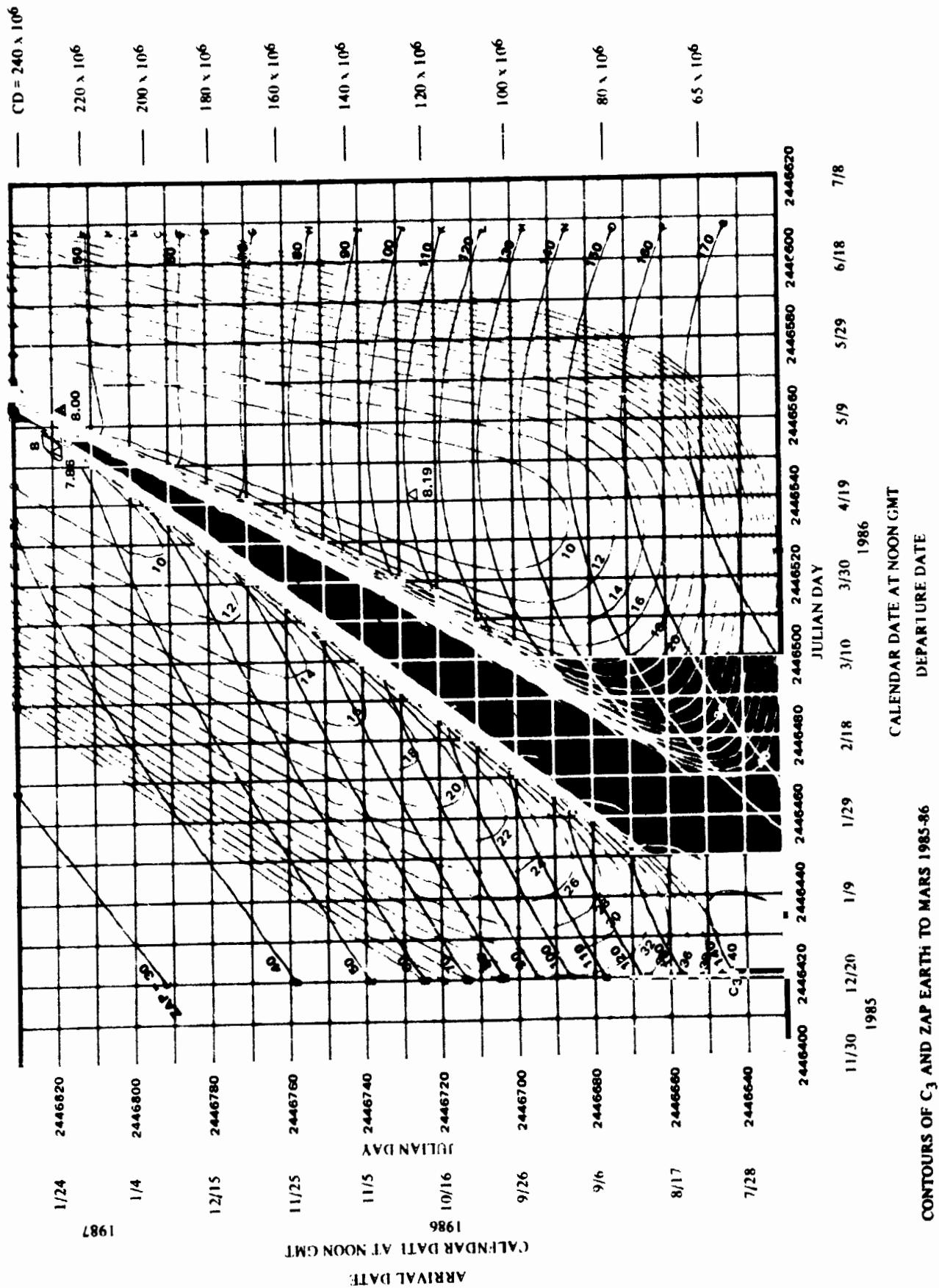
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ZAL
Q
1985

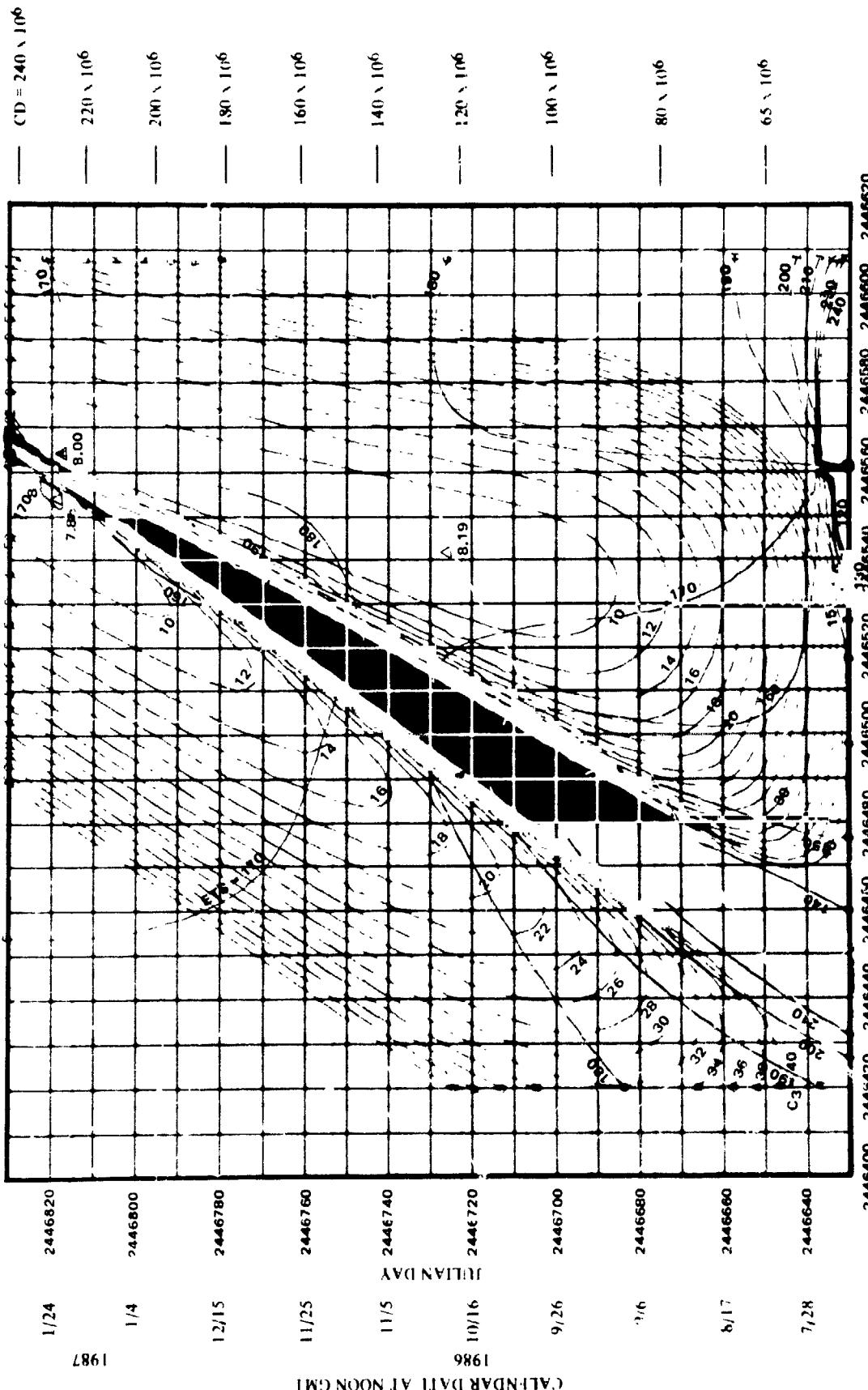




ZAP
Q
1985



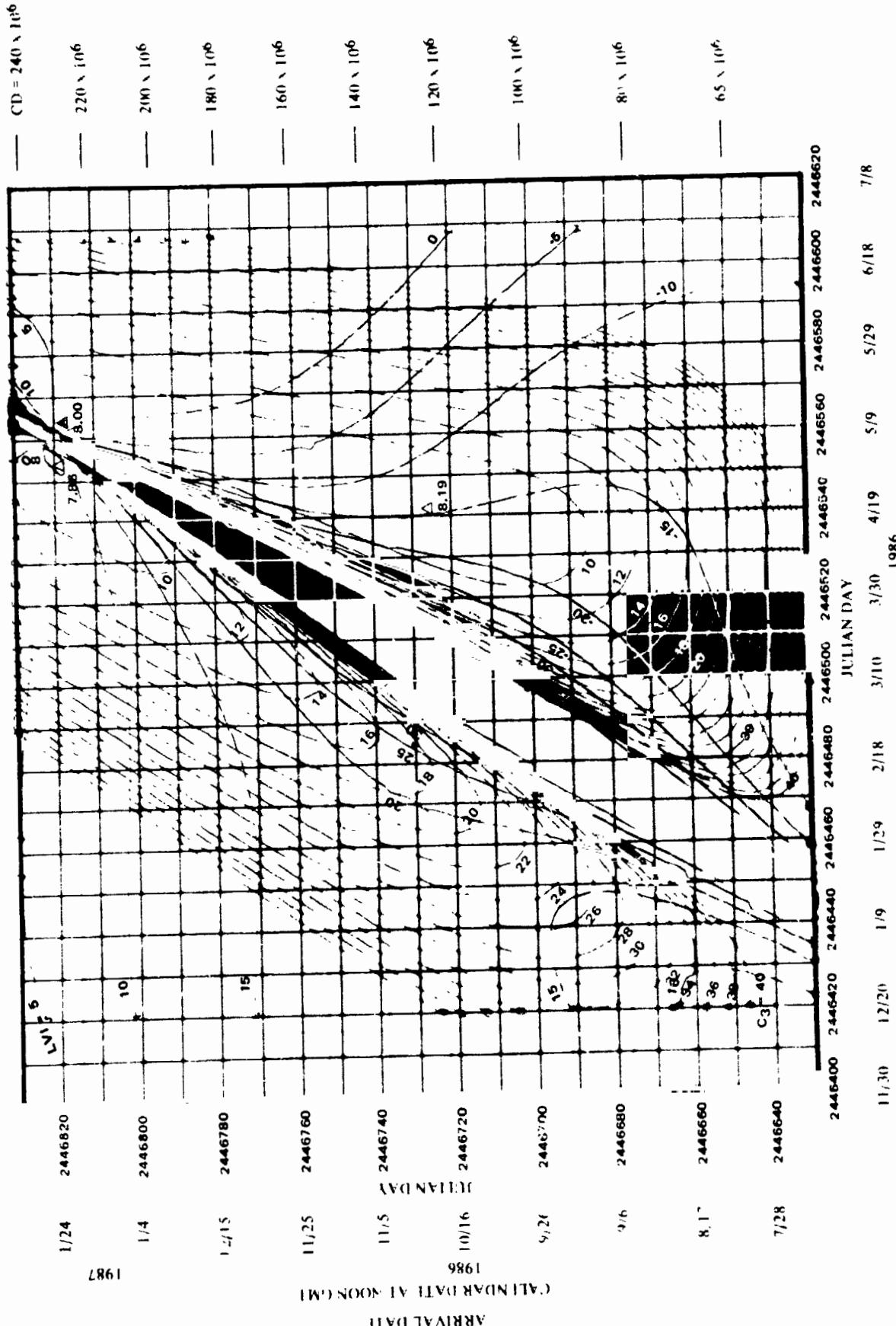
ETS
♂
1995



ARRIVAL DATE
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349

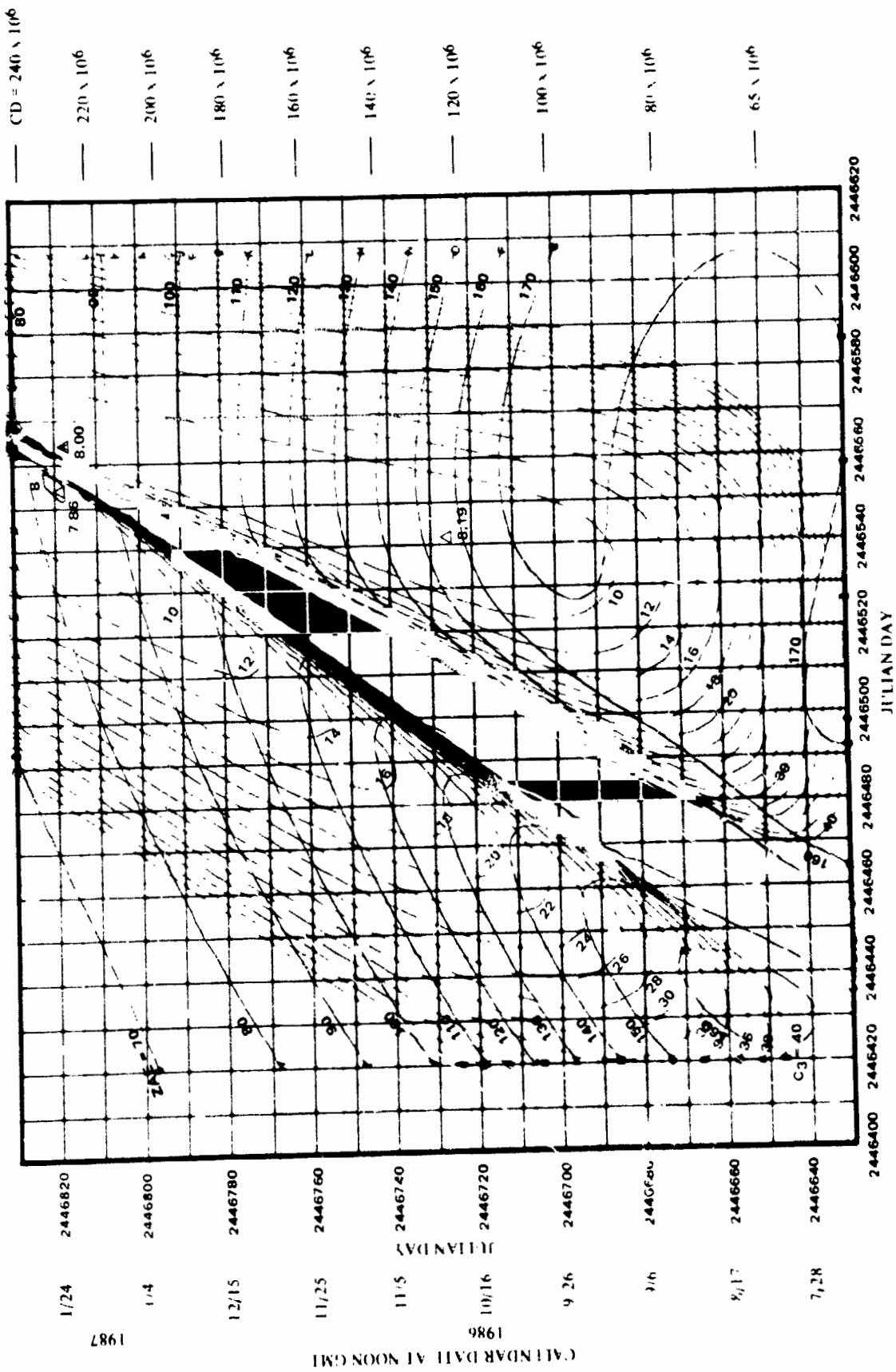
Q VI
1985



CONTOURS OF C₃ AND LVI FROM EARTH TO MARS 1985-86

CALENDAR DATE AT NOON GMT
DEPARTURE DATE

ZAE
♂
1985

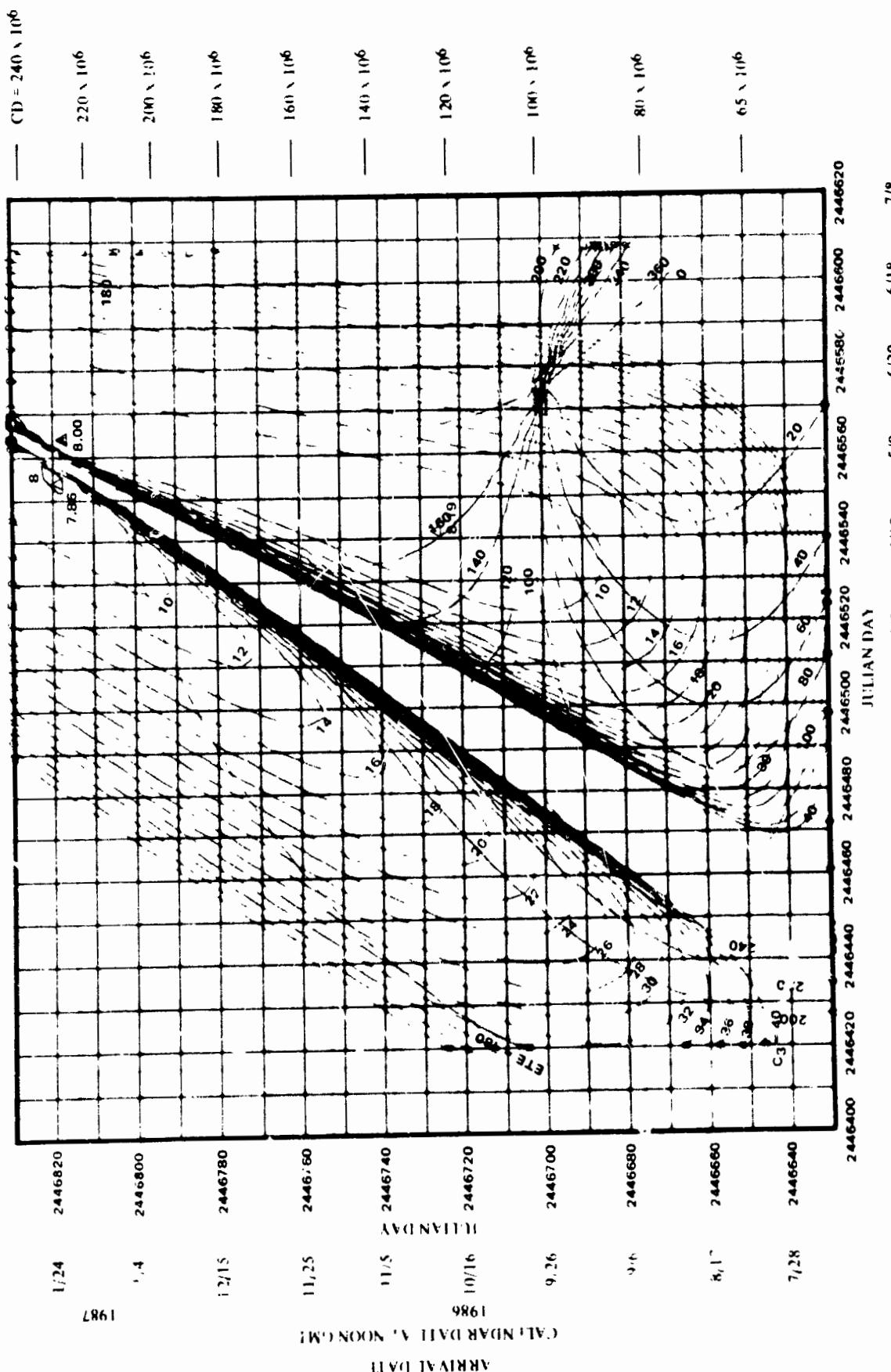


CONTOURS DE C₃ ET ZAI TERRE TO MARS 1985-86

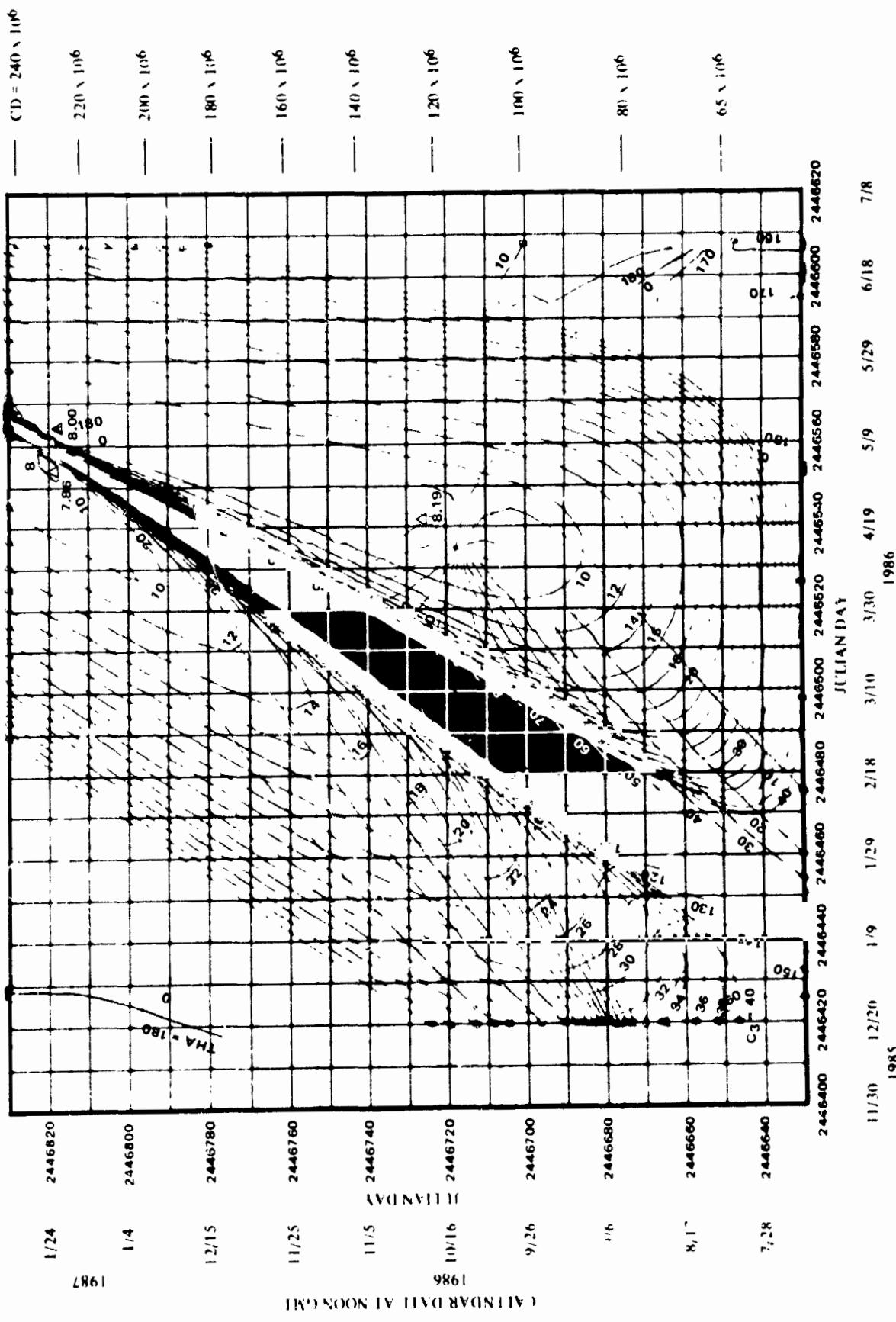
CALENDAR DATE AT NOON GMT

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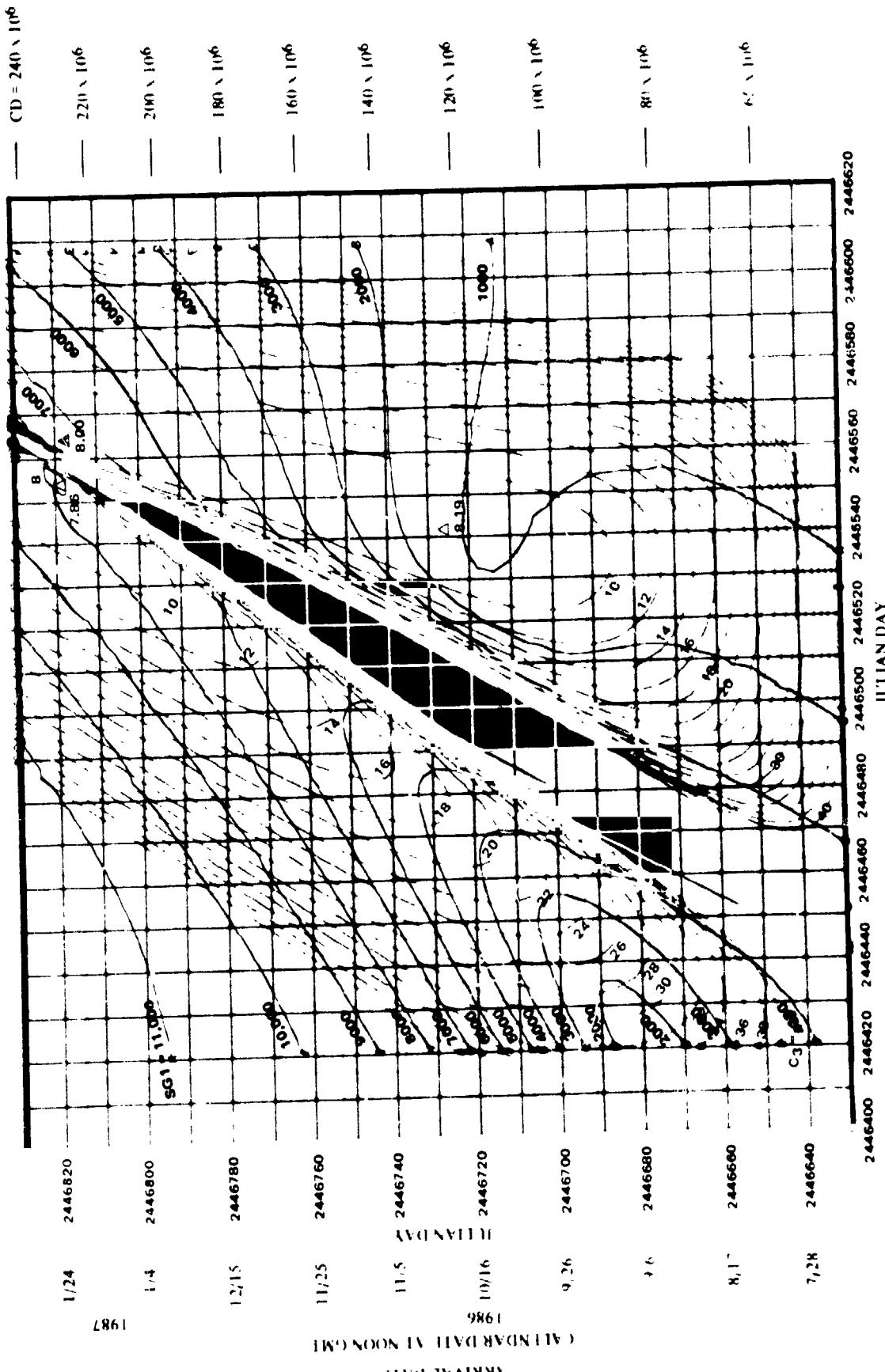
ETE
Q
1985



THA
Q
1986



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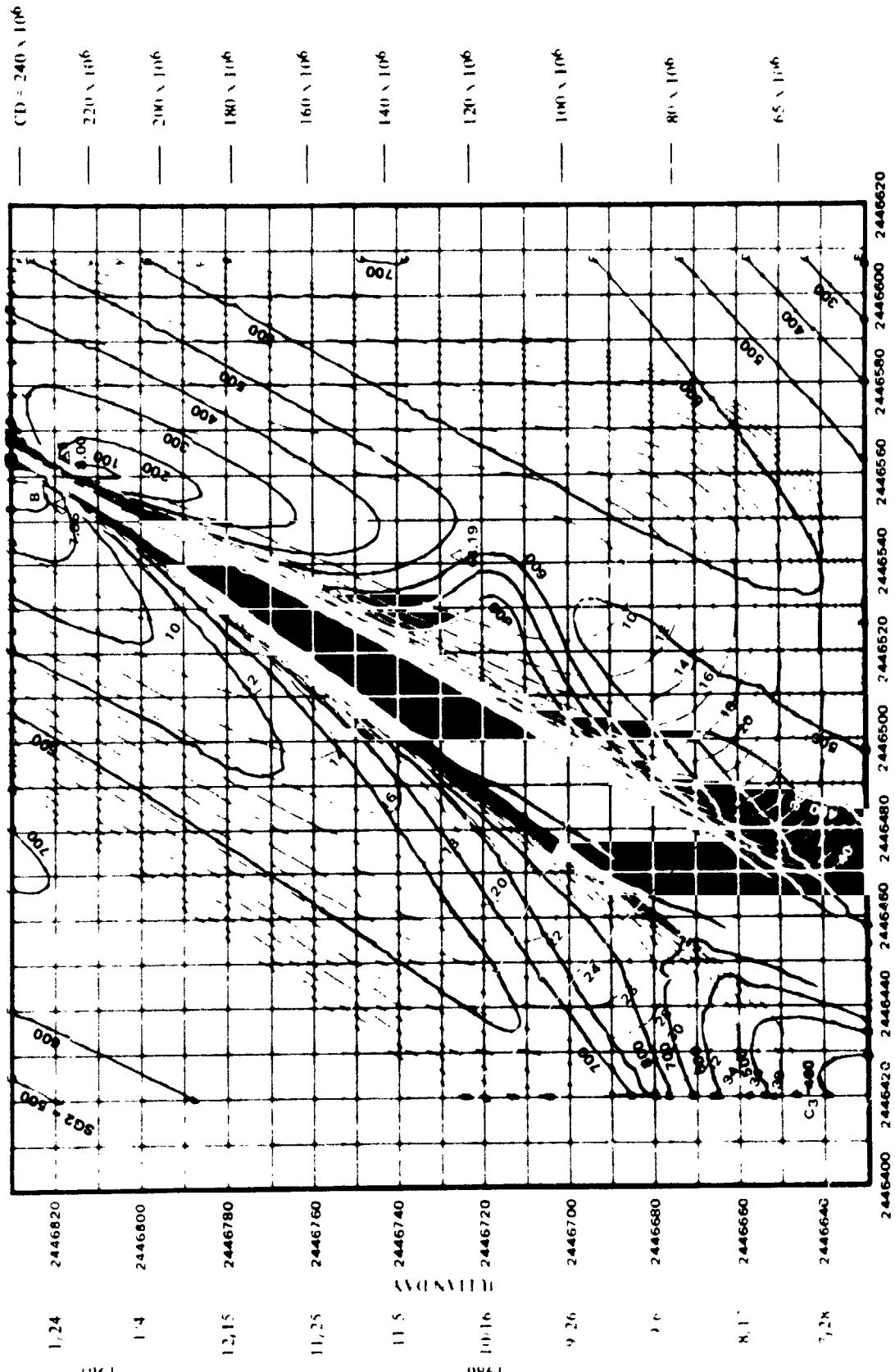


CONTOURS OF C₃ AND SG1 EARTH TO MARS 1985-86

CALENDARIATI AT NOON G.M.T.
DEPARTS 8 P.M.

SG1
♂
1985

SG2
1985

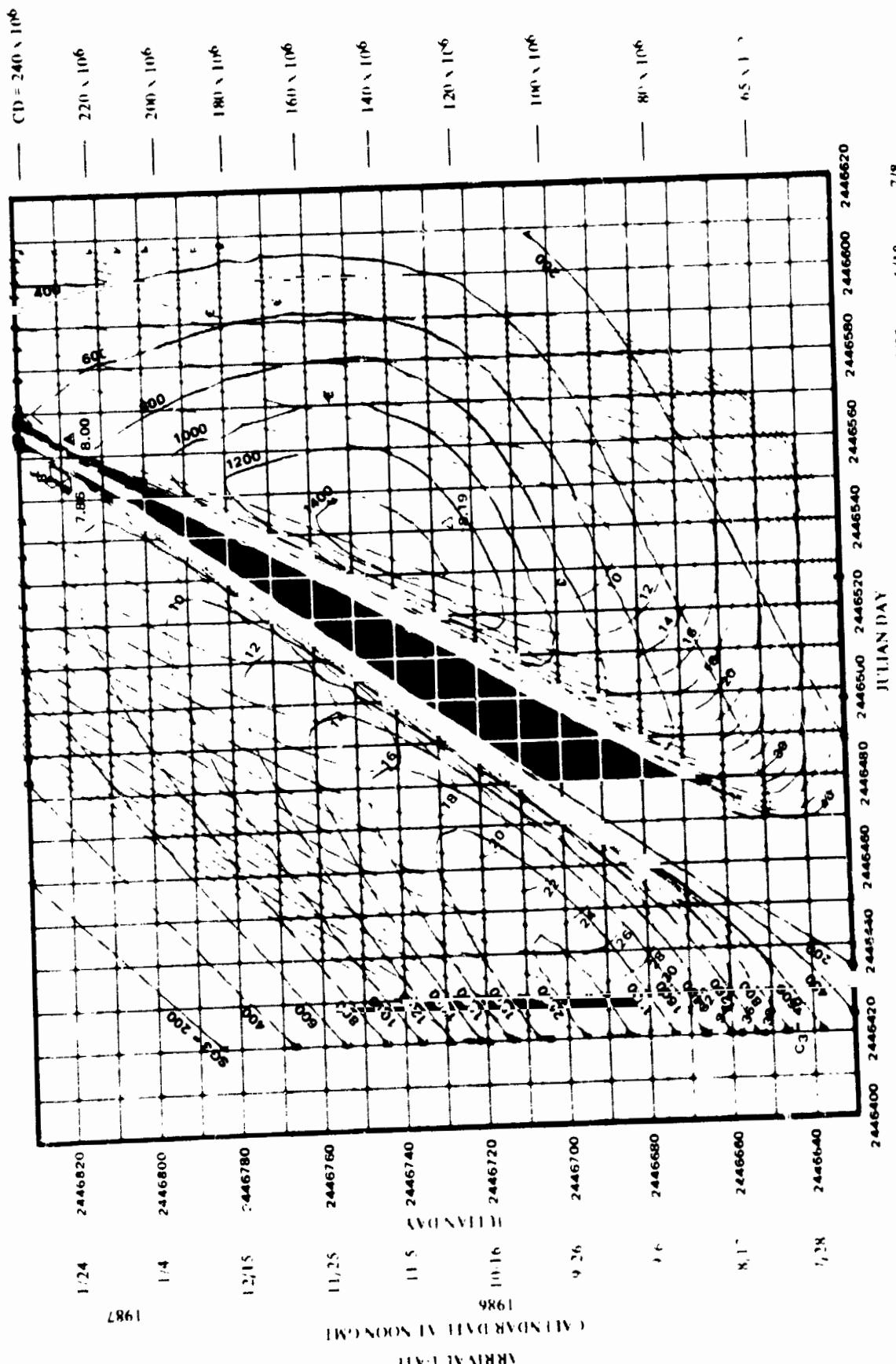


CONTOURS OF C₃ AND SG2 I. EARTH TO MARS 1985-86

CALNDAR DATE AT NOON C.MT
DEPARTURE DATE

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SG3
♂
1985

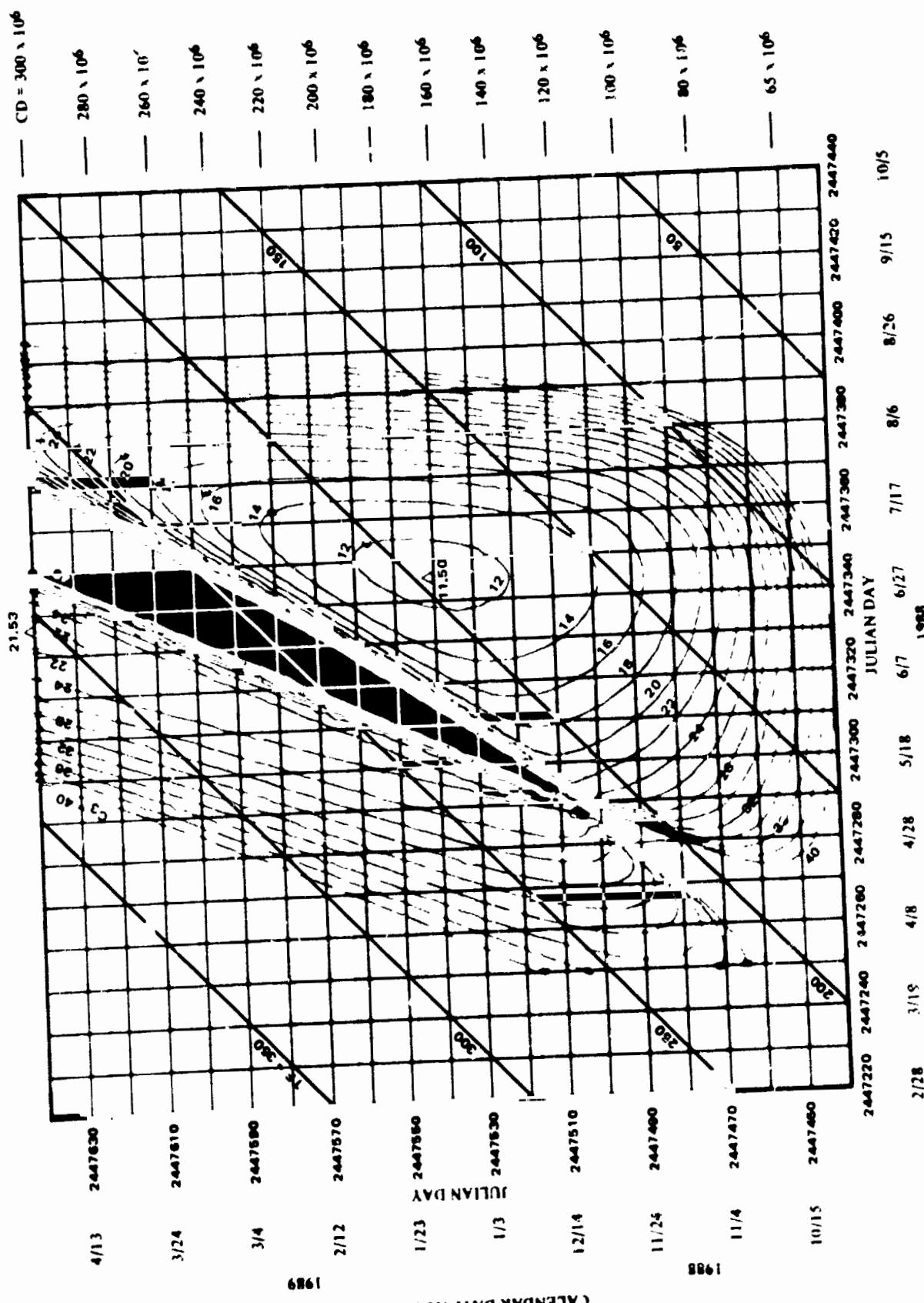


CONTOURS DE r_3 AND SURFACE EARTH TO MARS 1983 86

ELINWÆR 199

INT PARTURI DAI

C_3
1988



CALENDAR DATE AT NOON GMT

DEPARTURE DATE

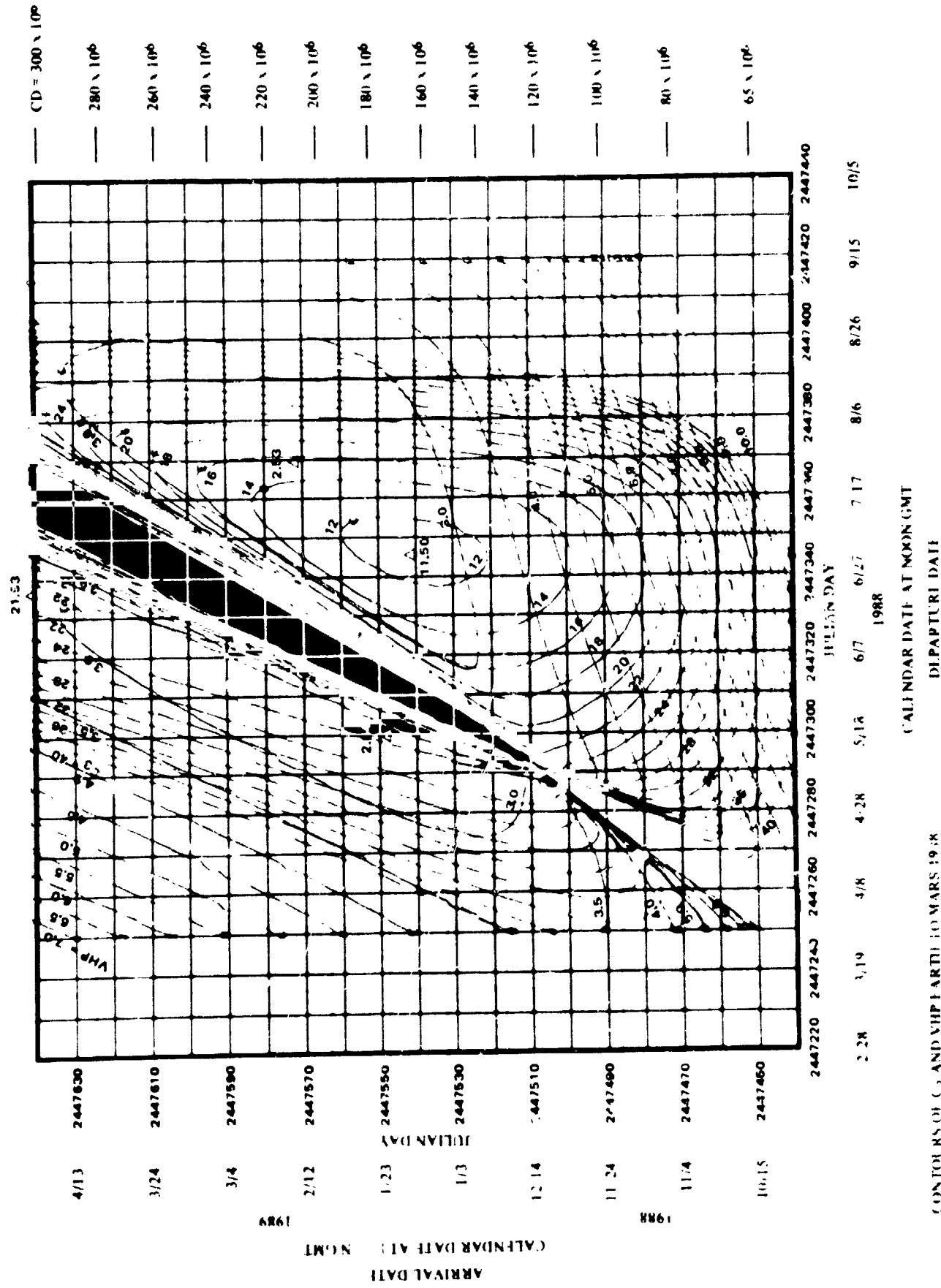
CONTOURS OF C_3 AND FLIGHT TIMES EARTH TO MARS 1988

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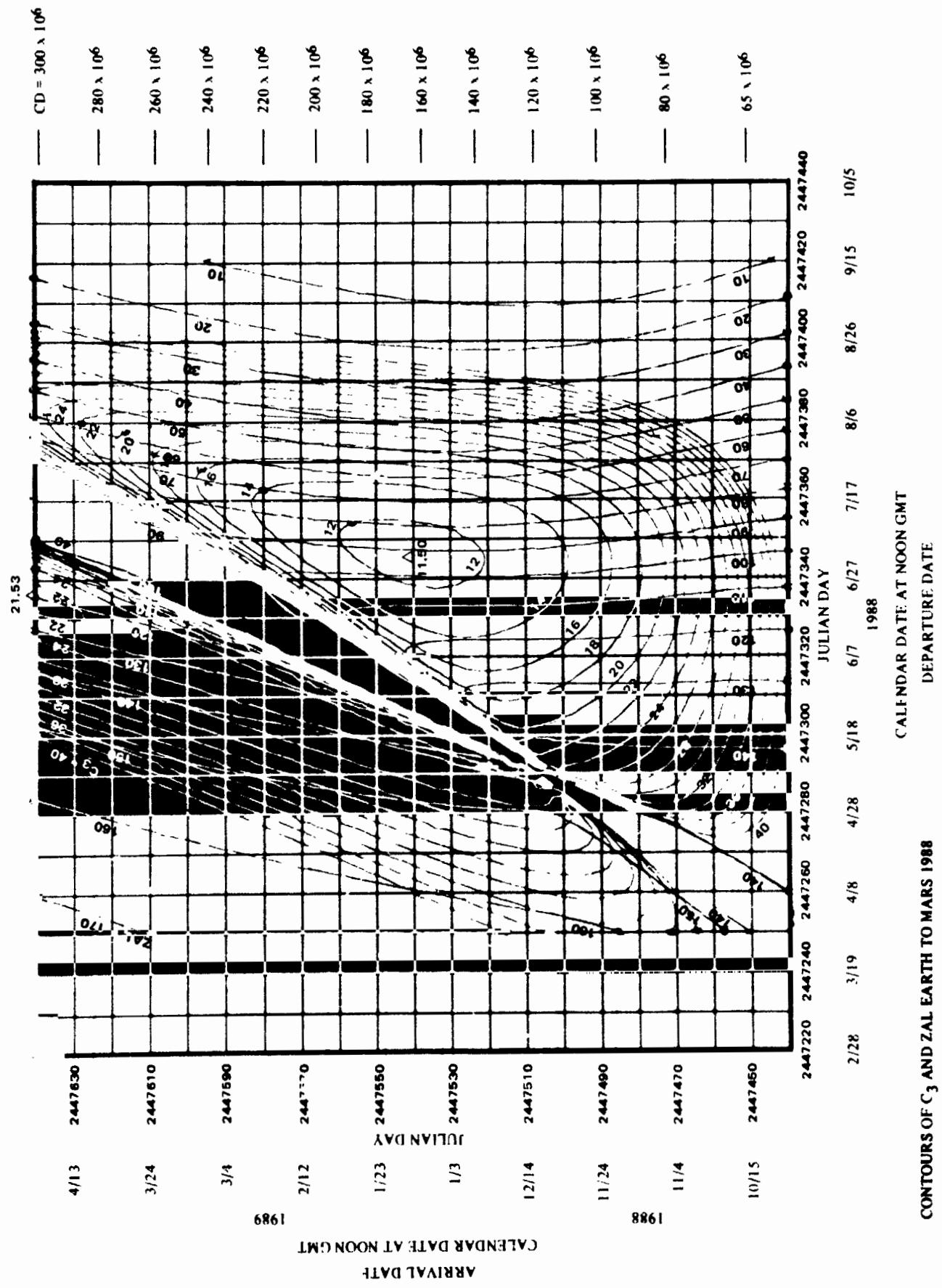
ARRIVAL DATE

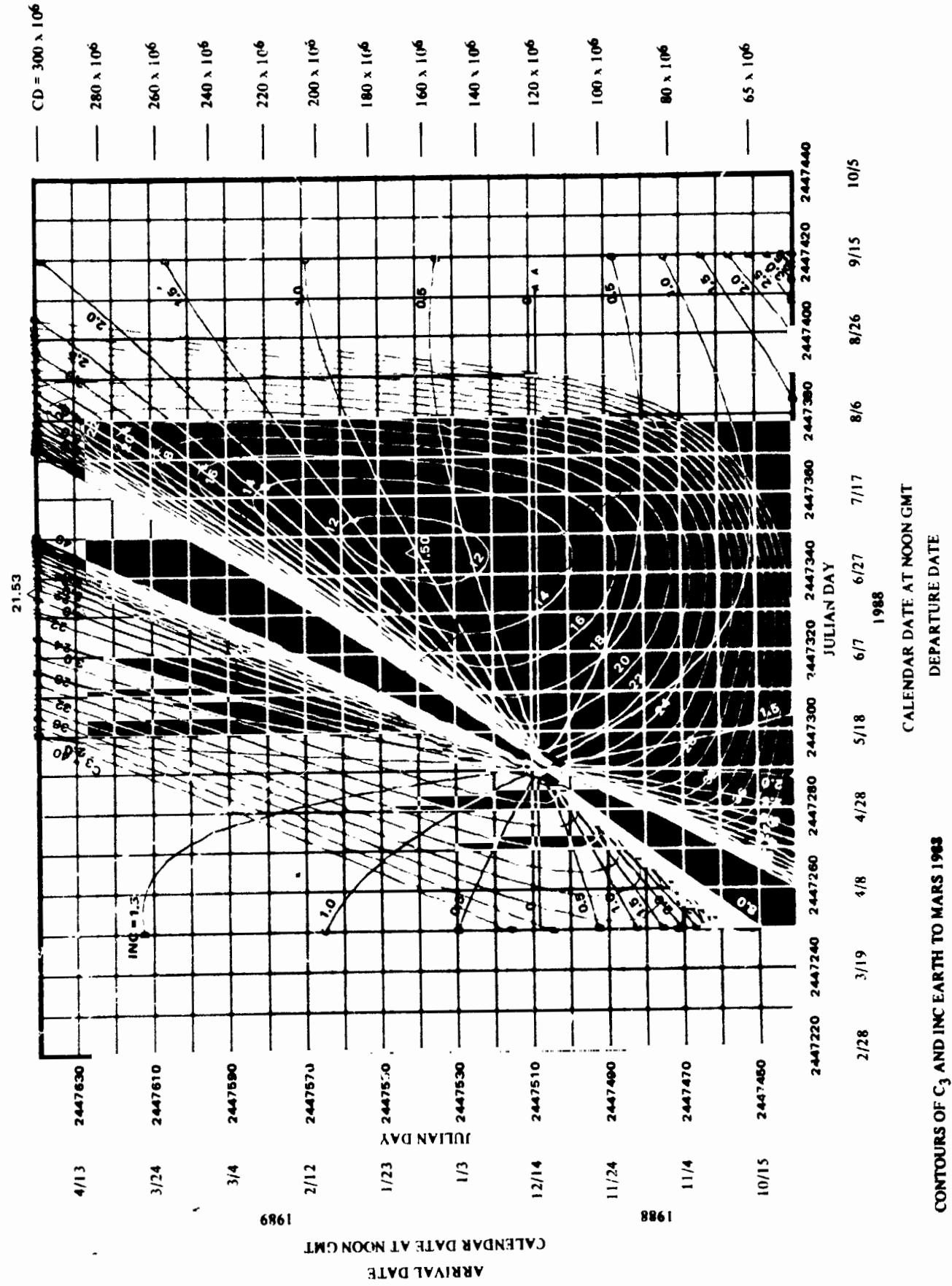
CALENDAR DATE AT NOON GMT

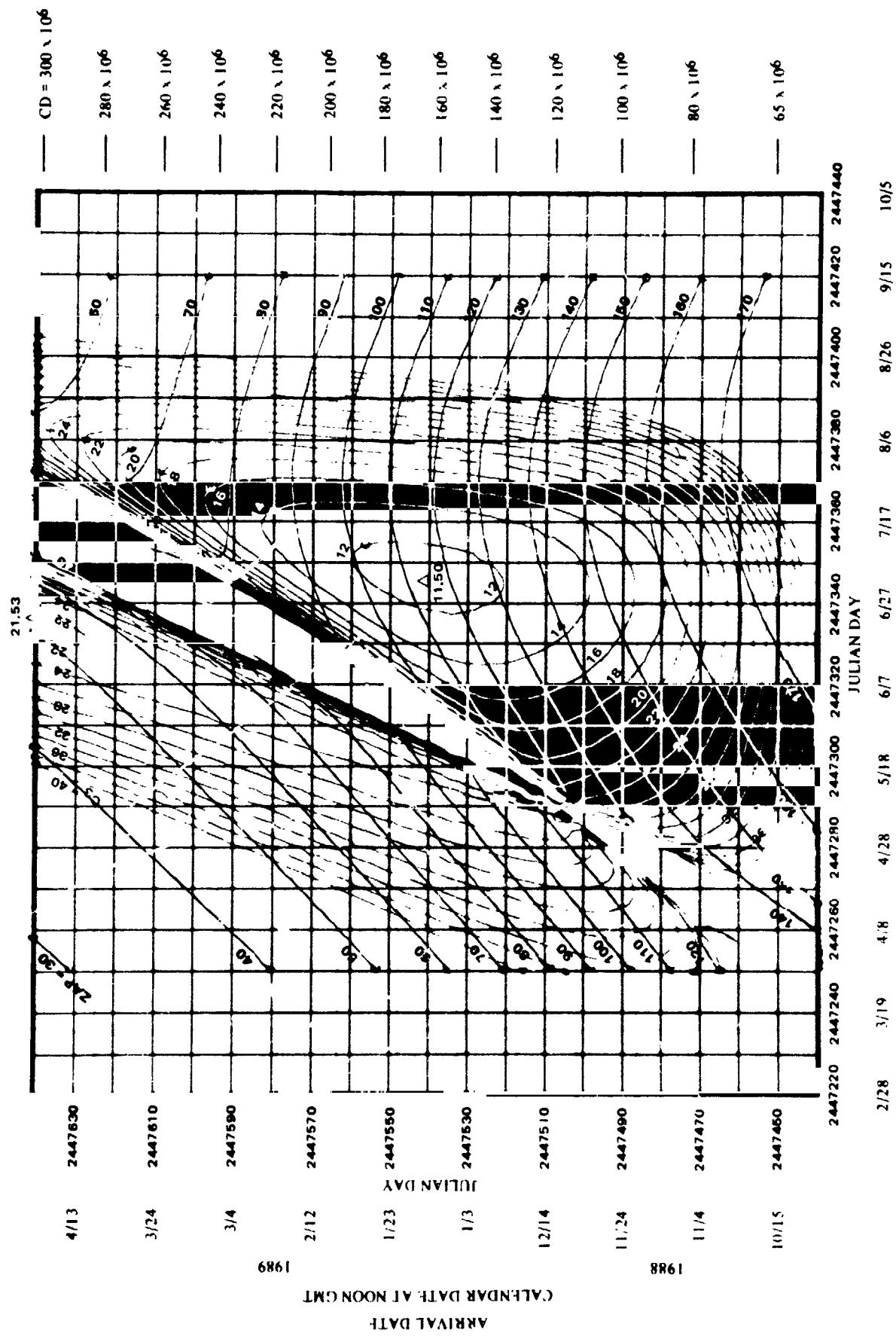
VHP
Q
1388



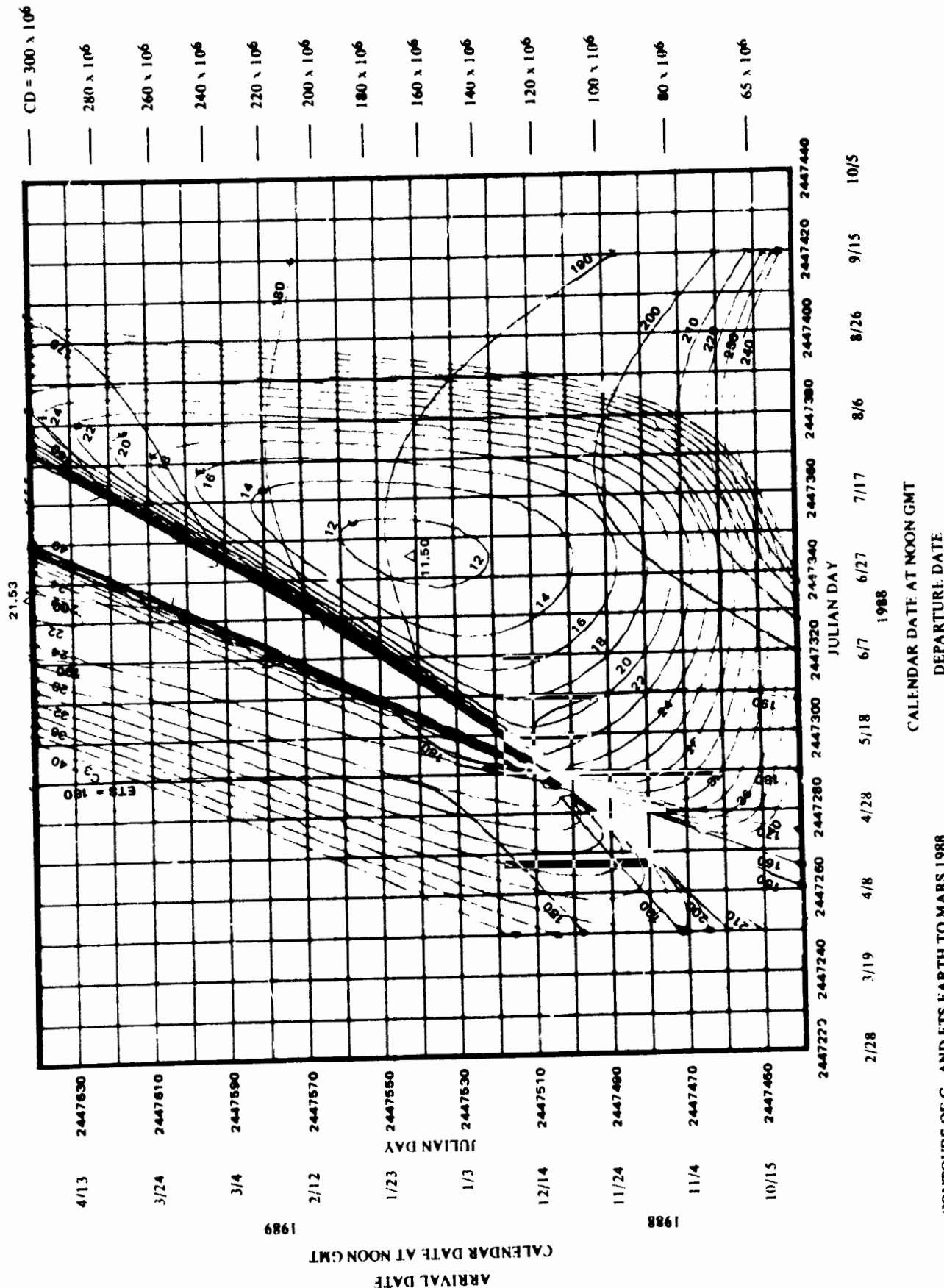
ZAL
QA
1988



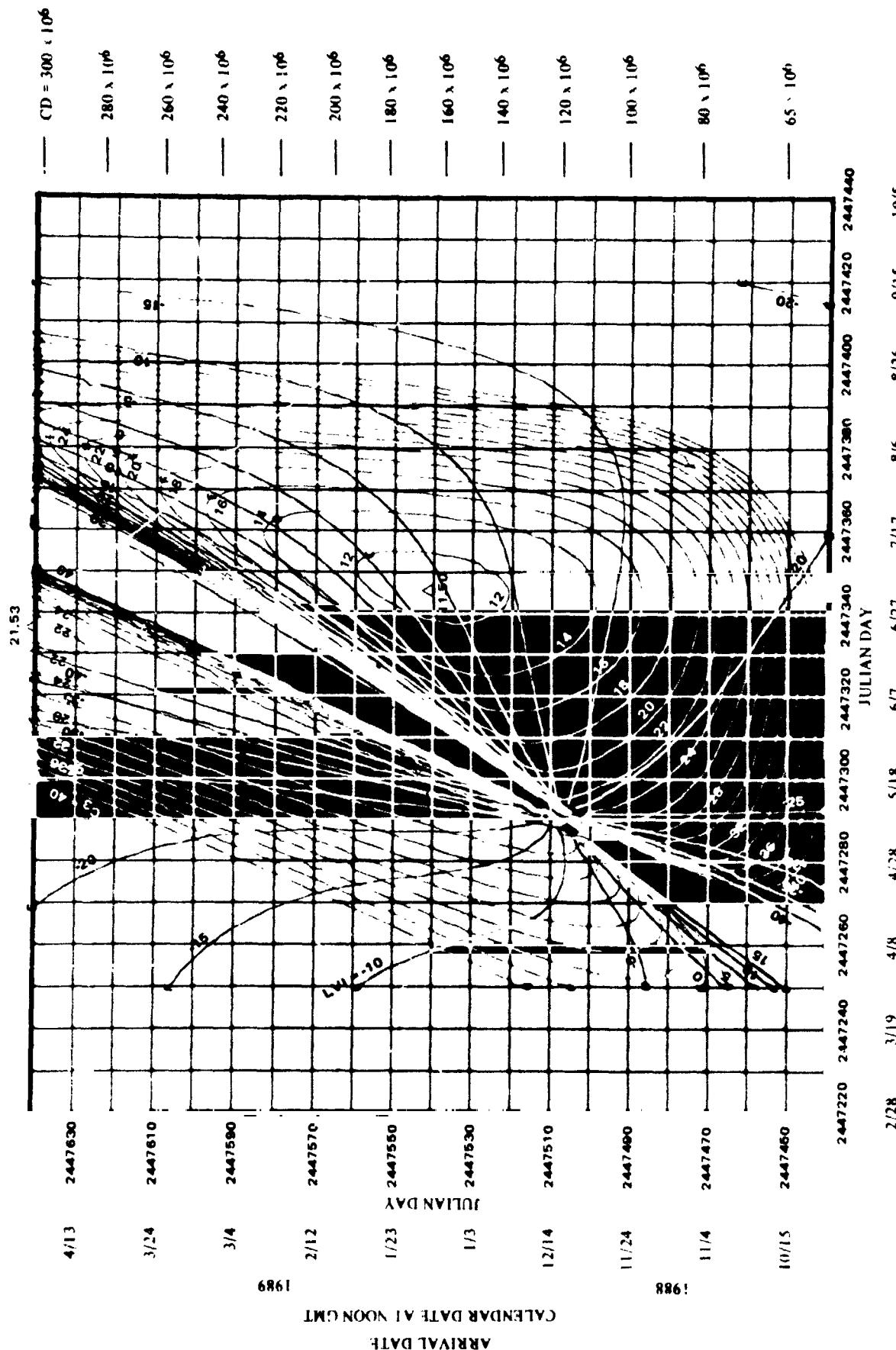




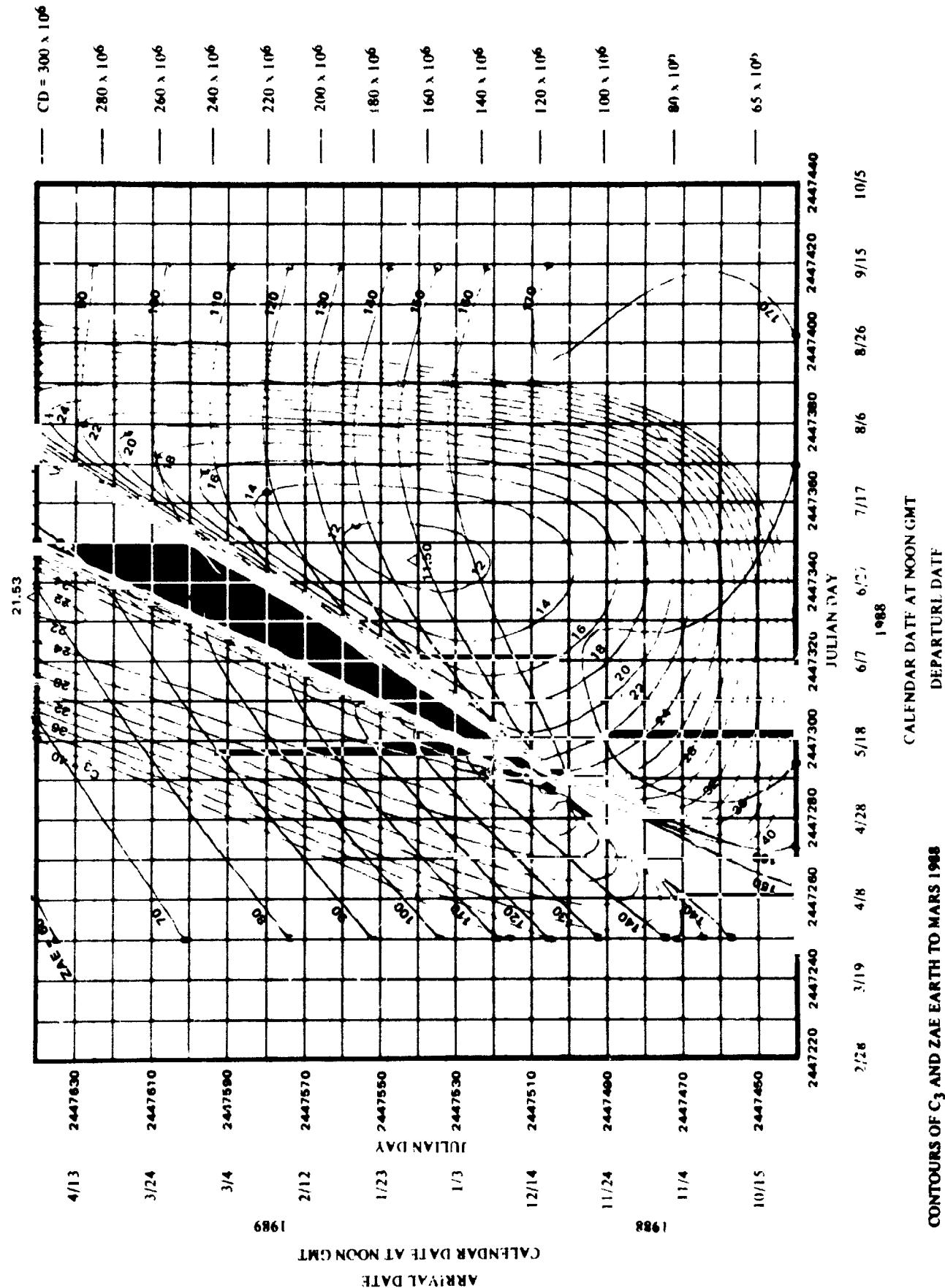
ETS
♂
1968



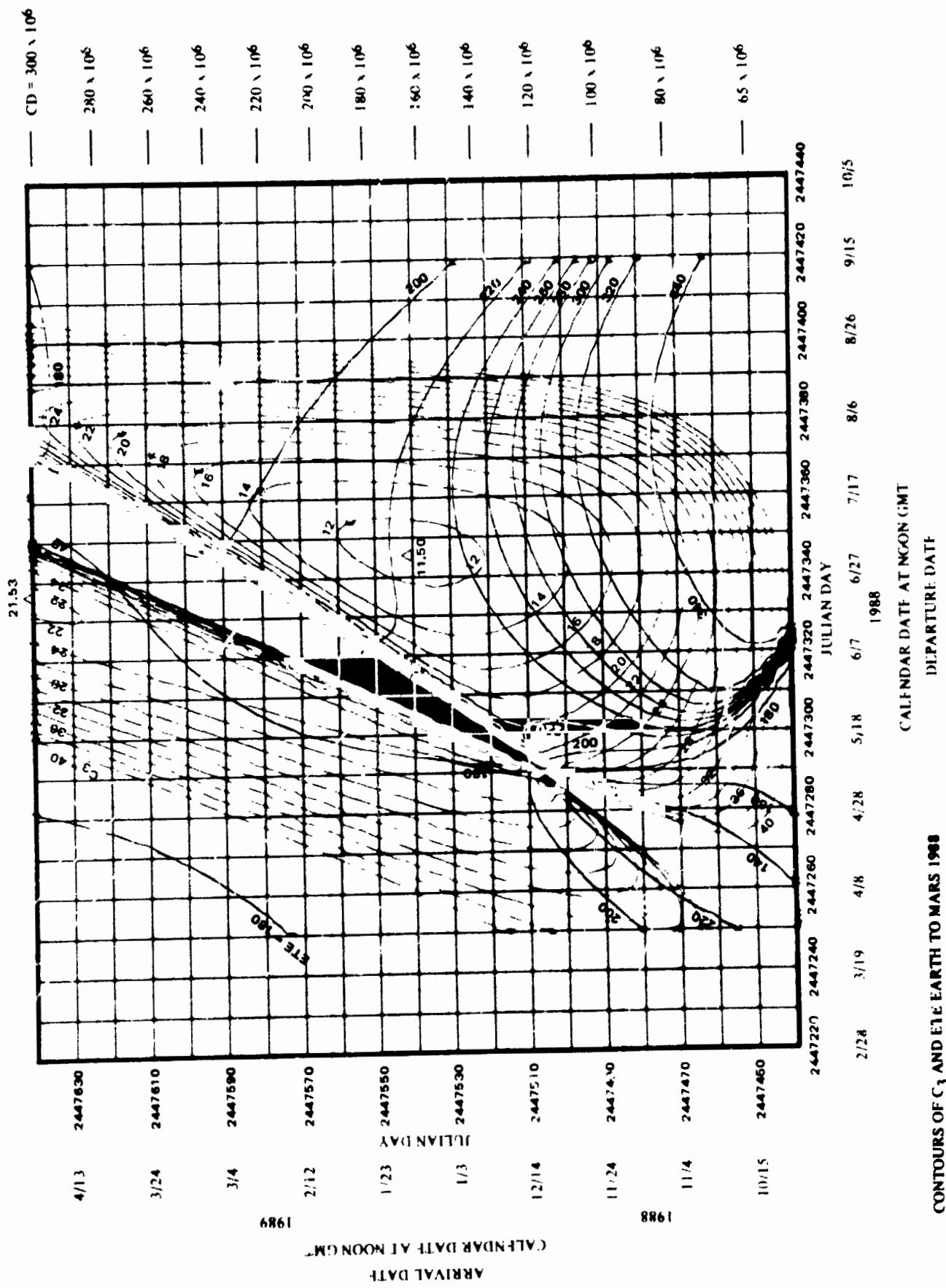
LV
Q
1988



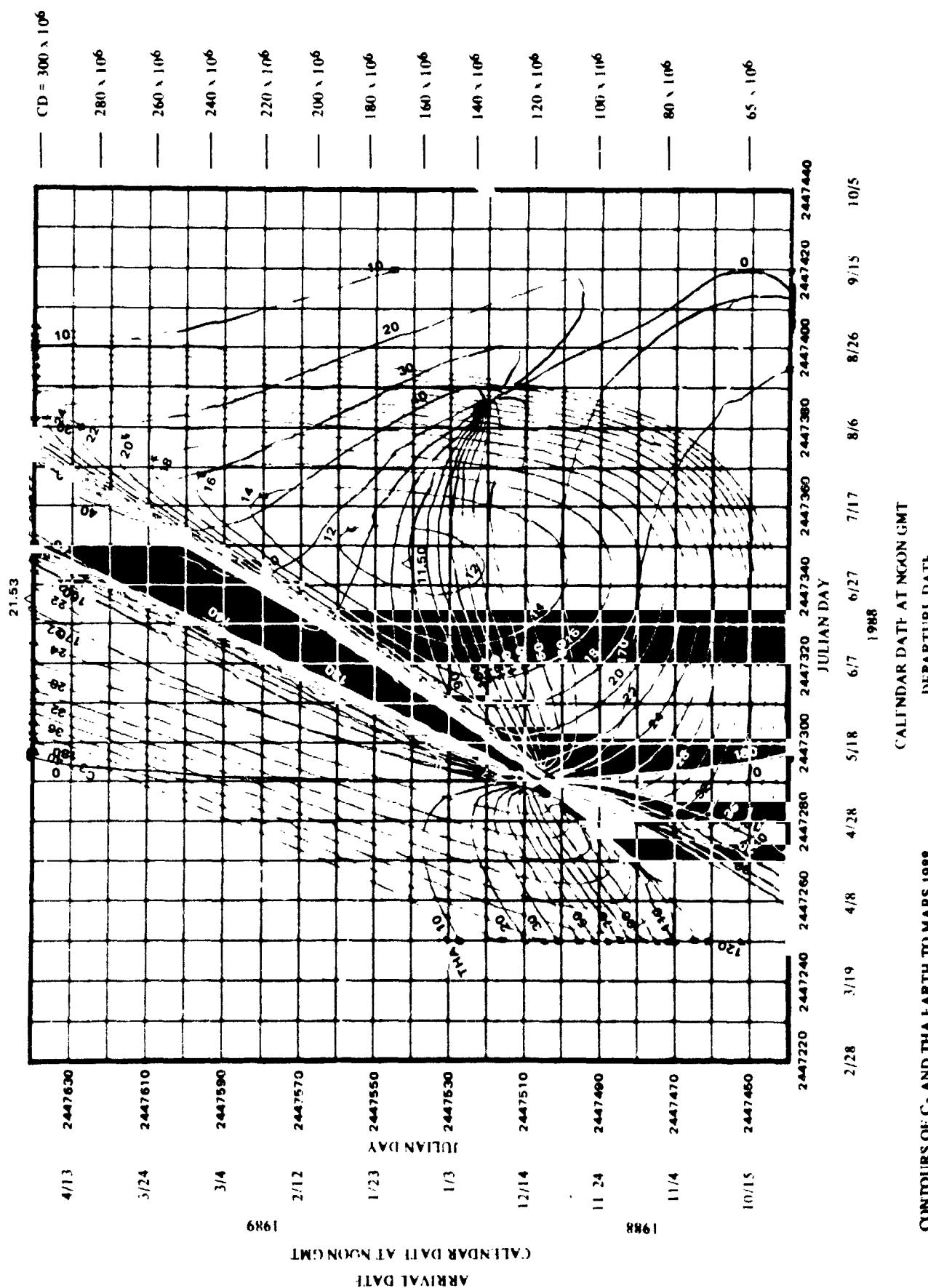
ZAE
♂
1988



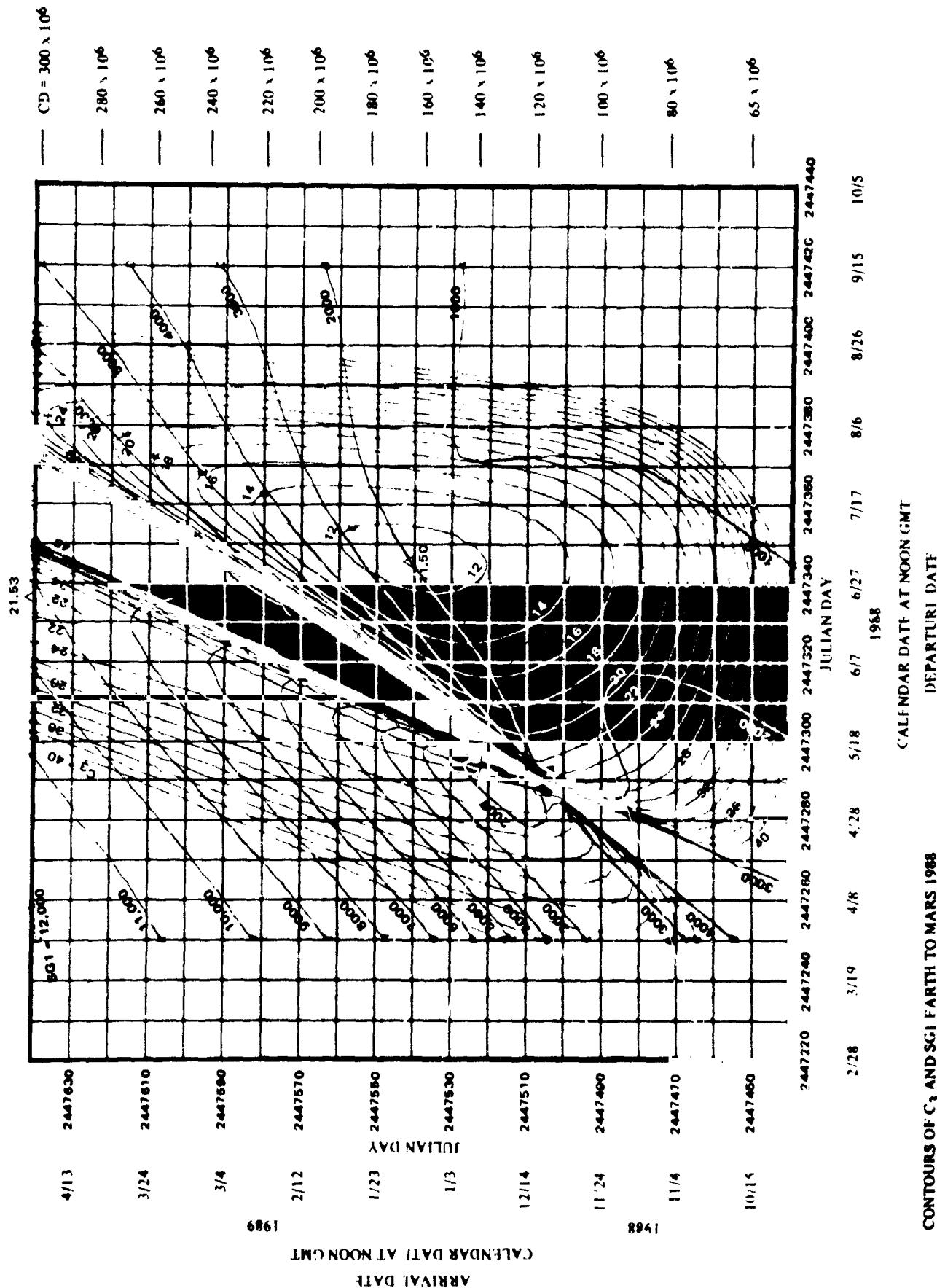
Q ETE
1988



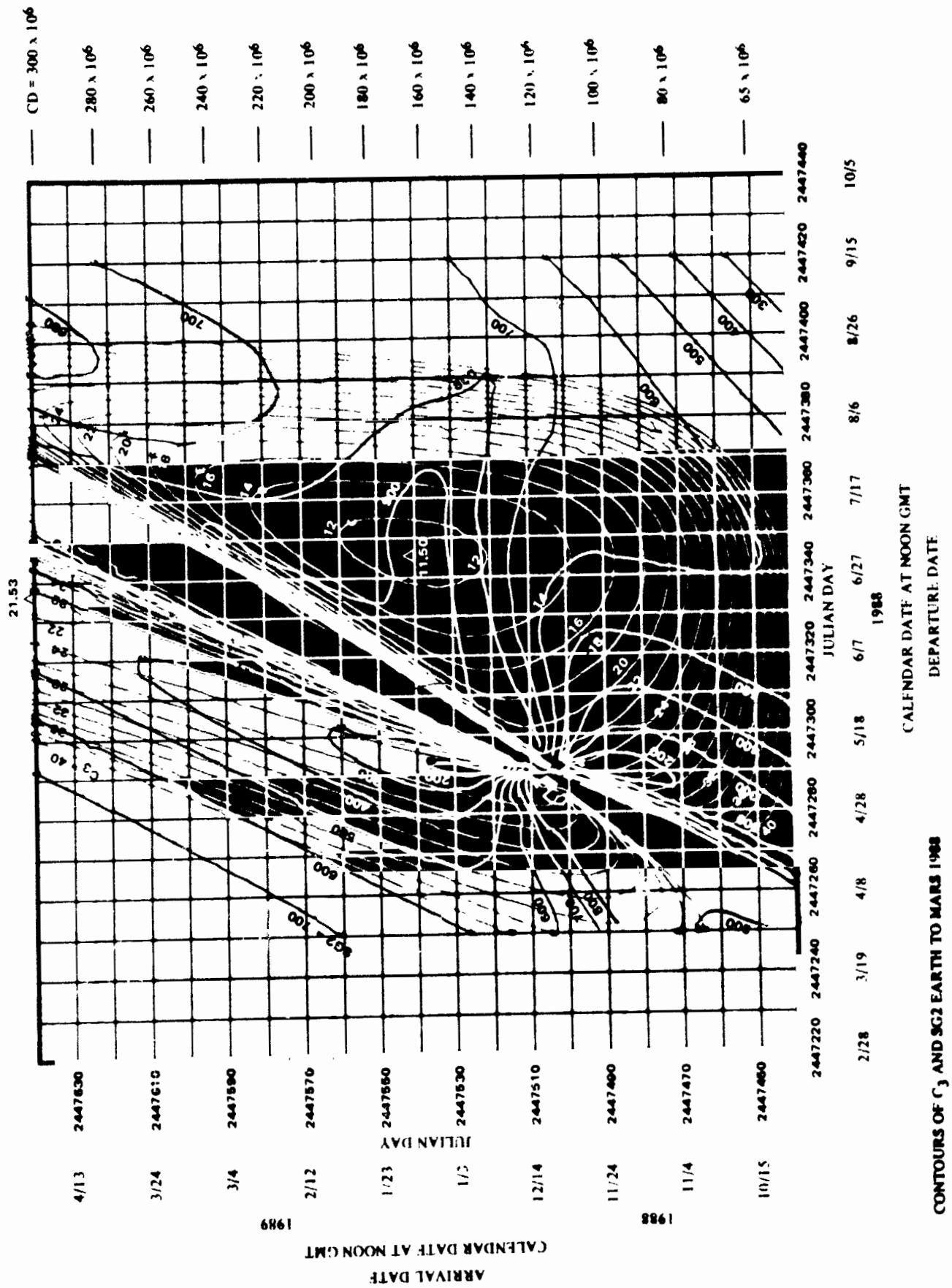
THA
Q
1988



SG1
Q1
1968



SG2
♂
1998

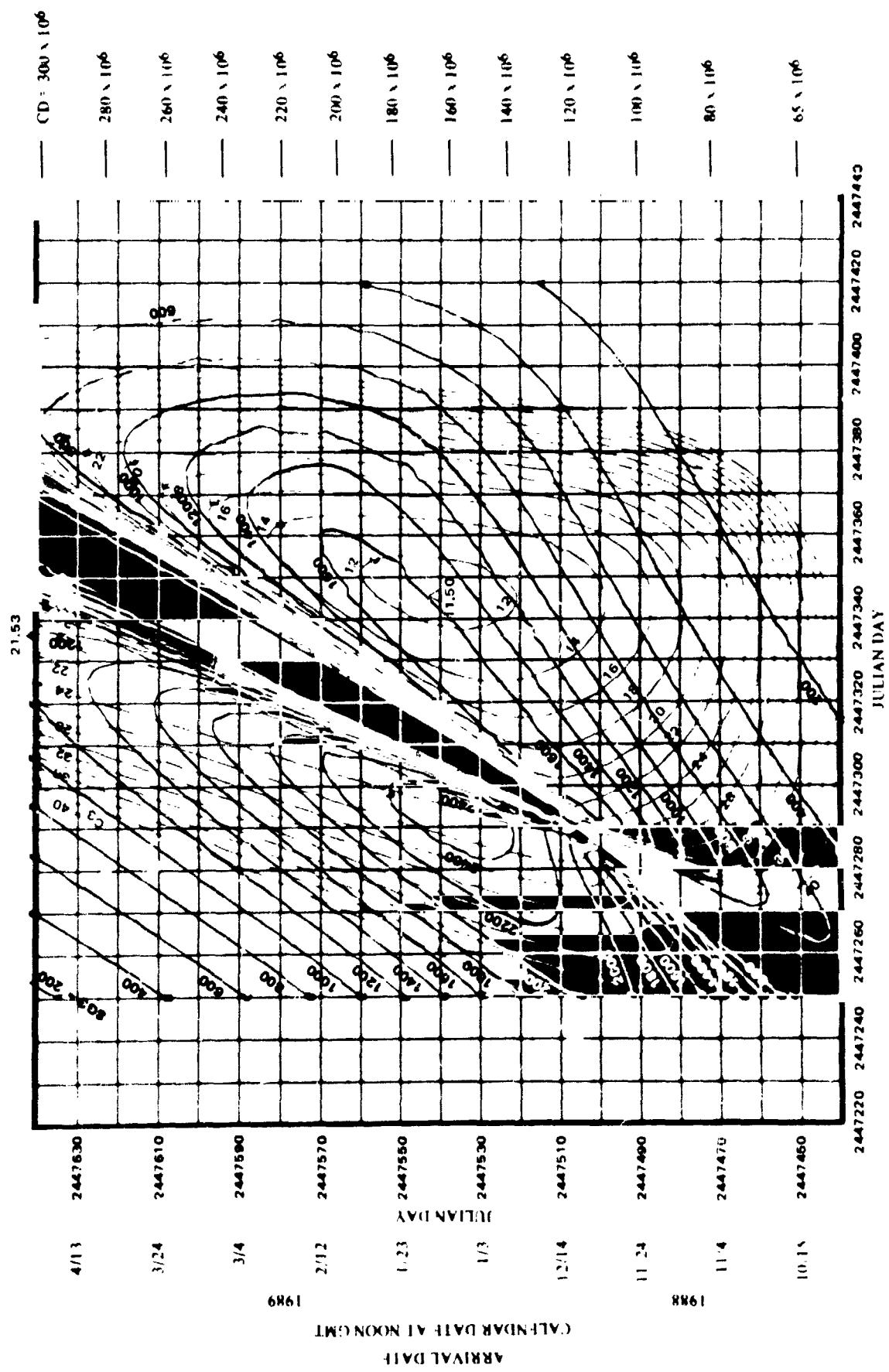


CONTOURS OF C₁ AND SC2 EARTH TO MARS 1963

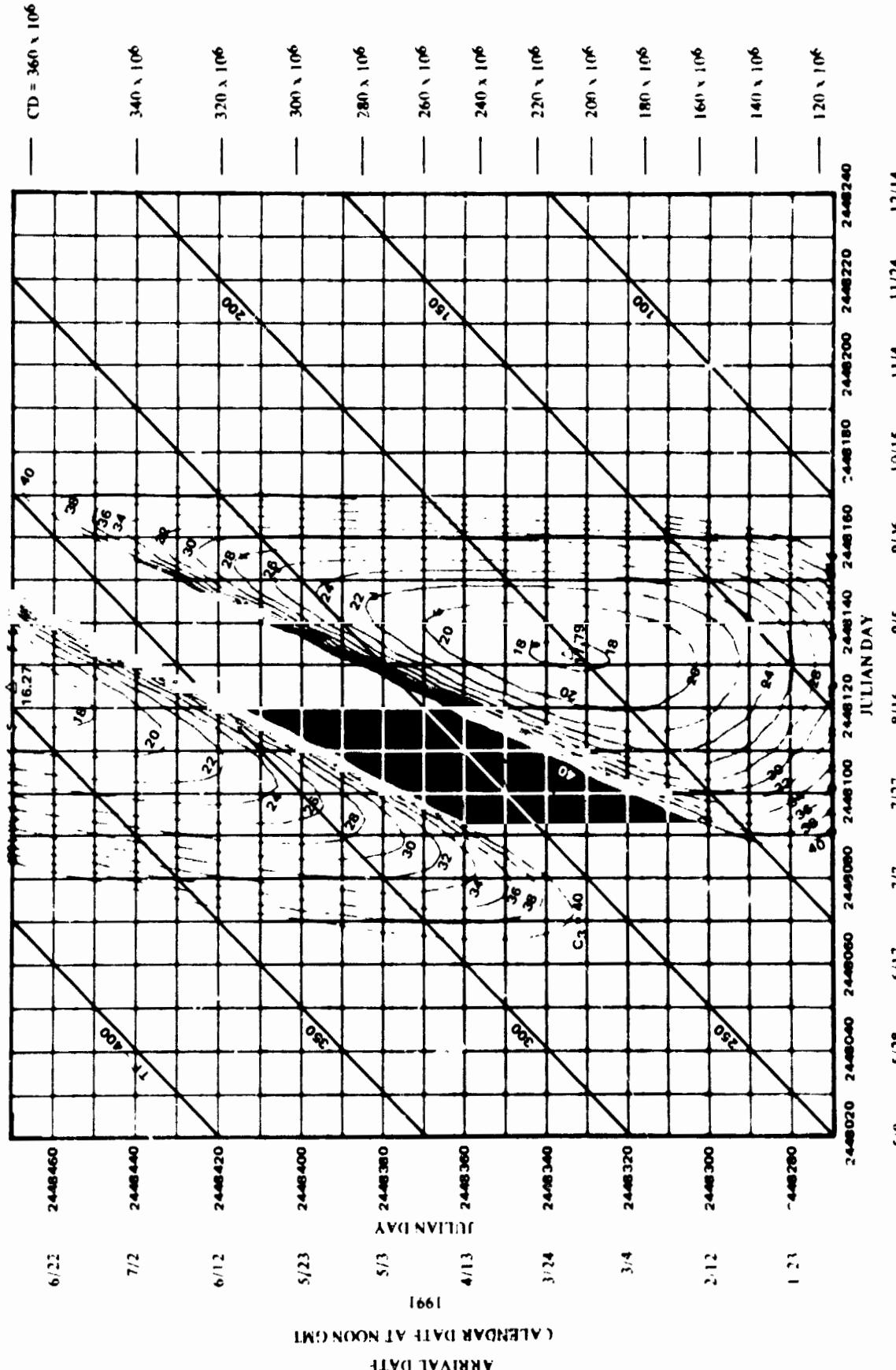
DEPARTURE DATA

CALENDAR DATE AT NOON GMT

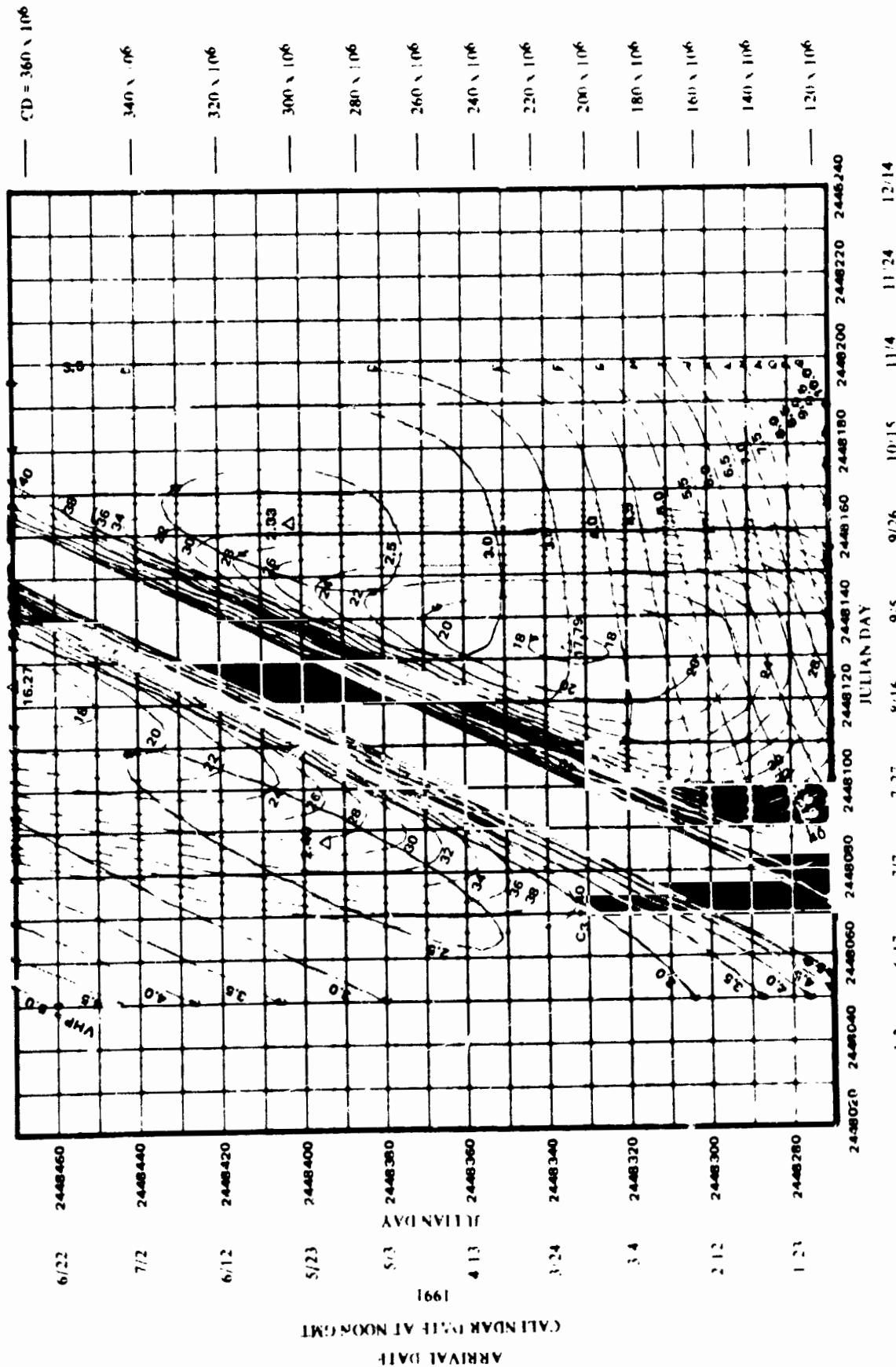
SG3
1000



Q₃
1990



VHP
♂
1990



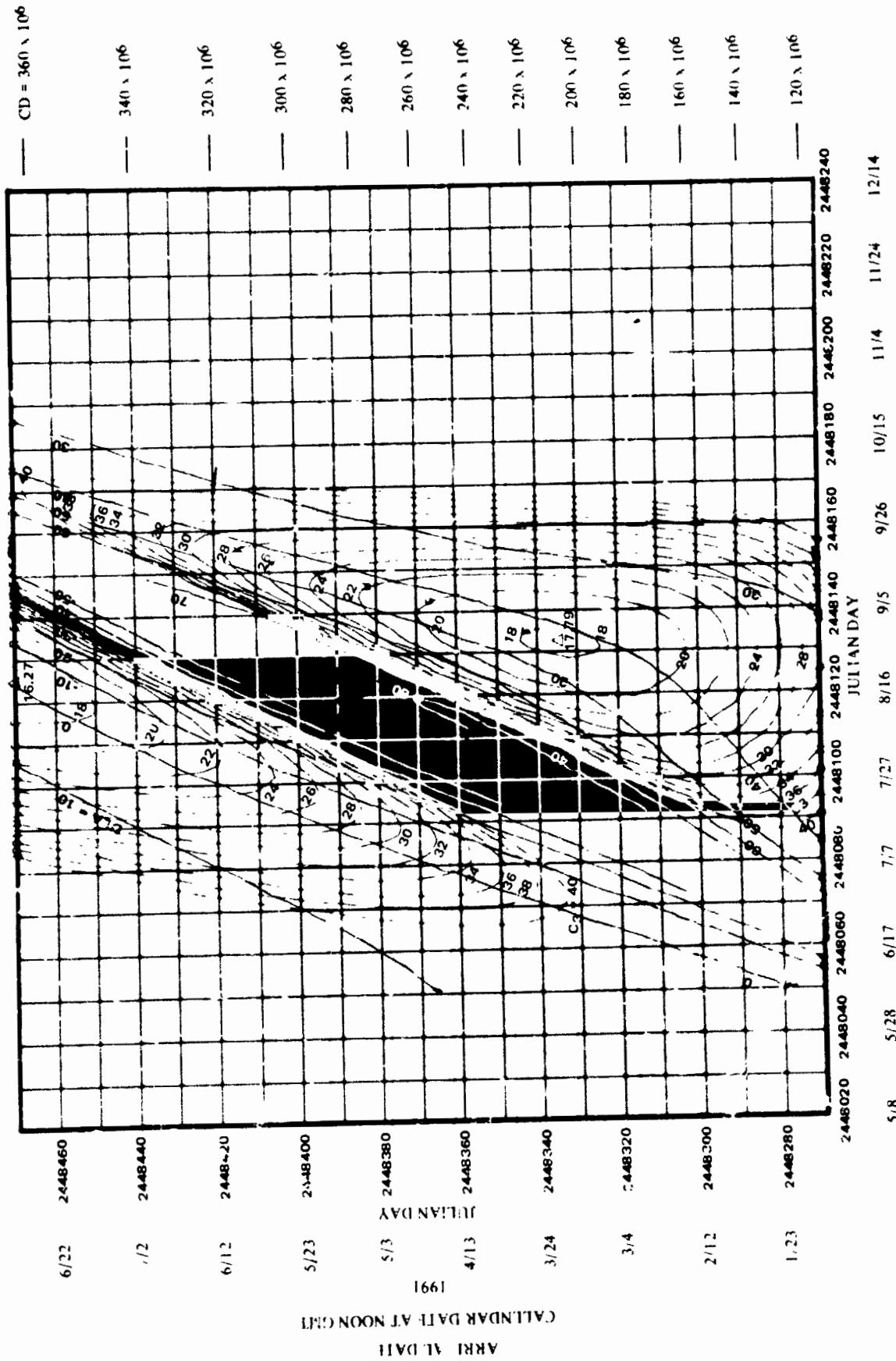
CONTOURS DE C₁ ET VITESSE D'ARRIVÉE SUR MARS 1990

DEPARTURE DATA

(ALENDAHAI AL MAJLIS)

卷之三

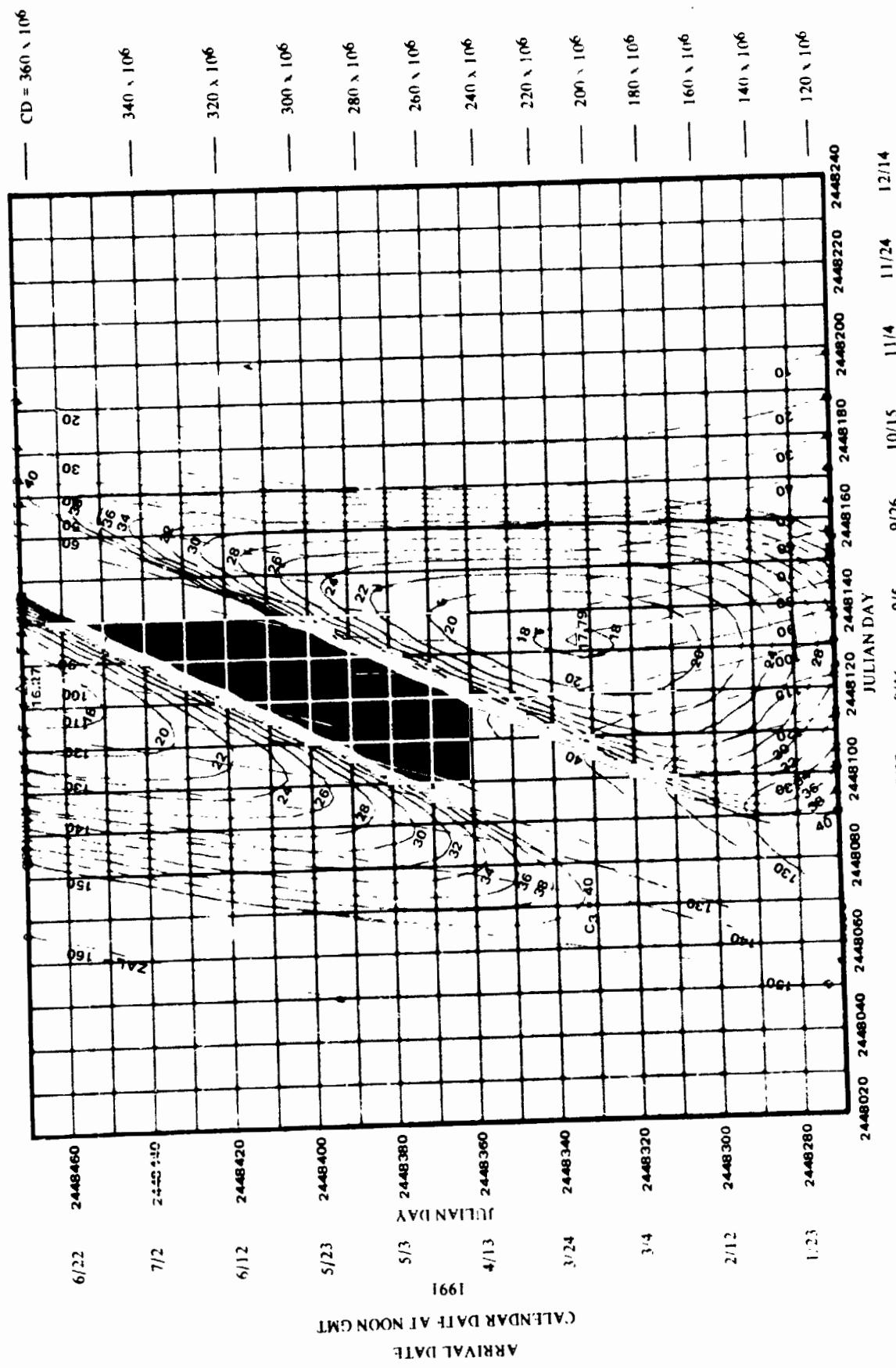
DLA
O
1990



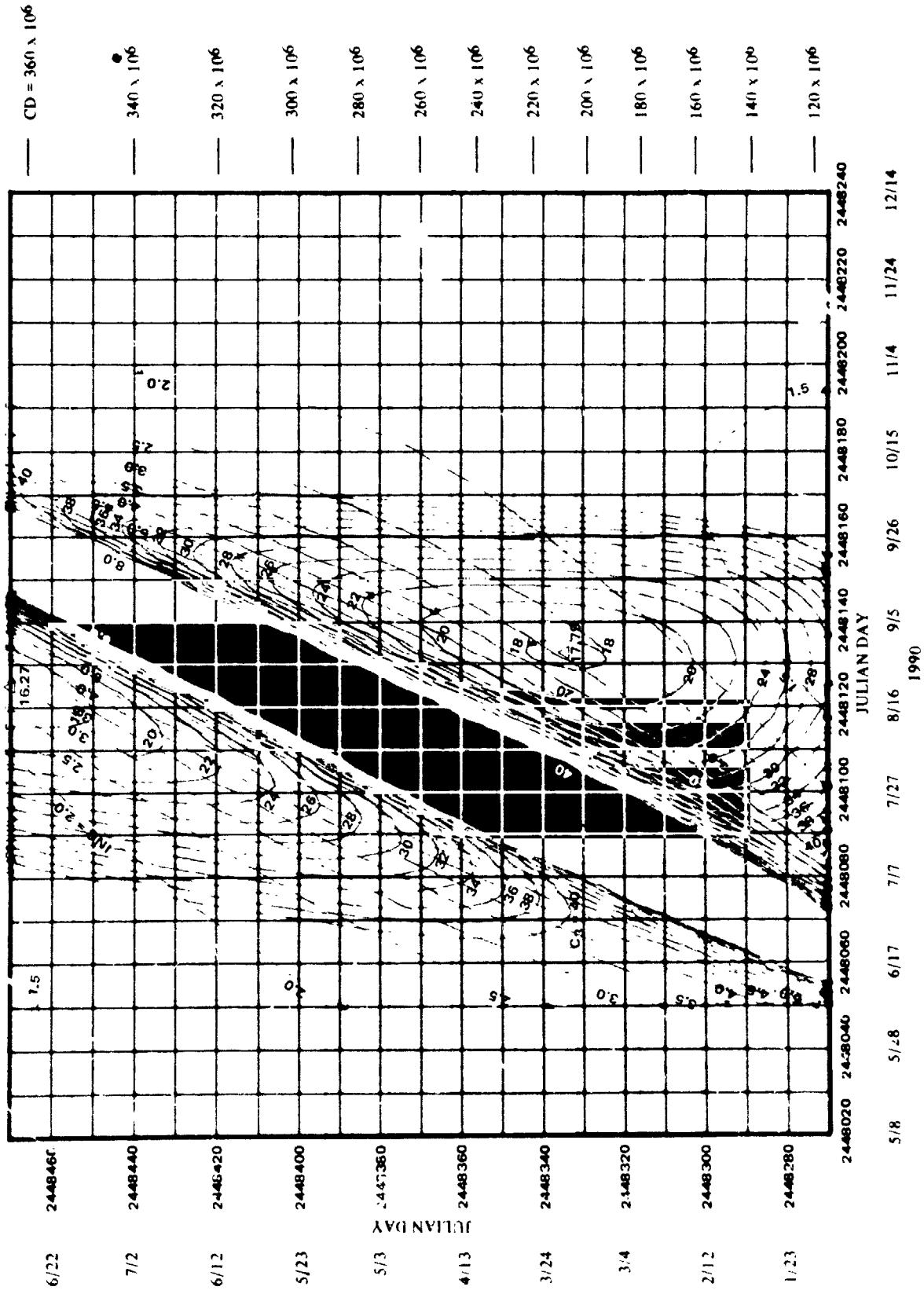
CONTOURS OF C_3 AND DLA EARTH TO MARS 1990

CALCNDAR DATA AT NOON GMT
DEPARTURE DATE

ZAL
Q
1990

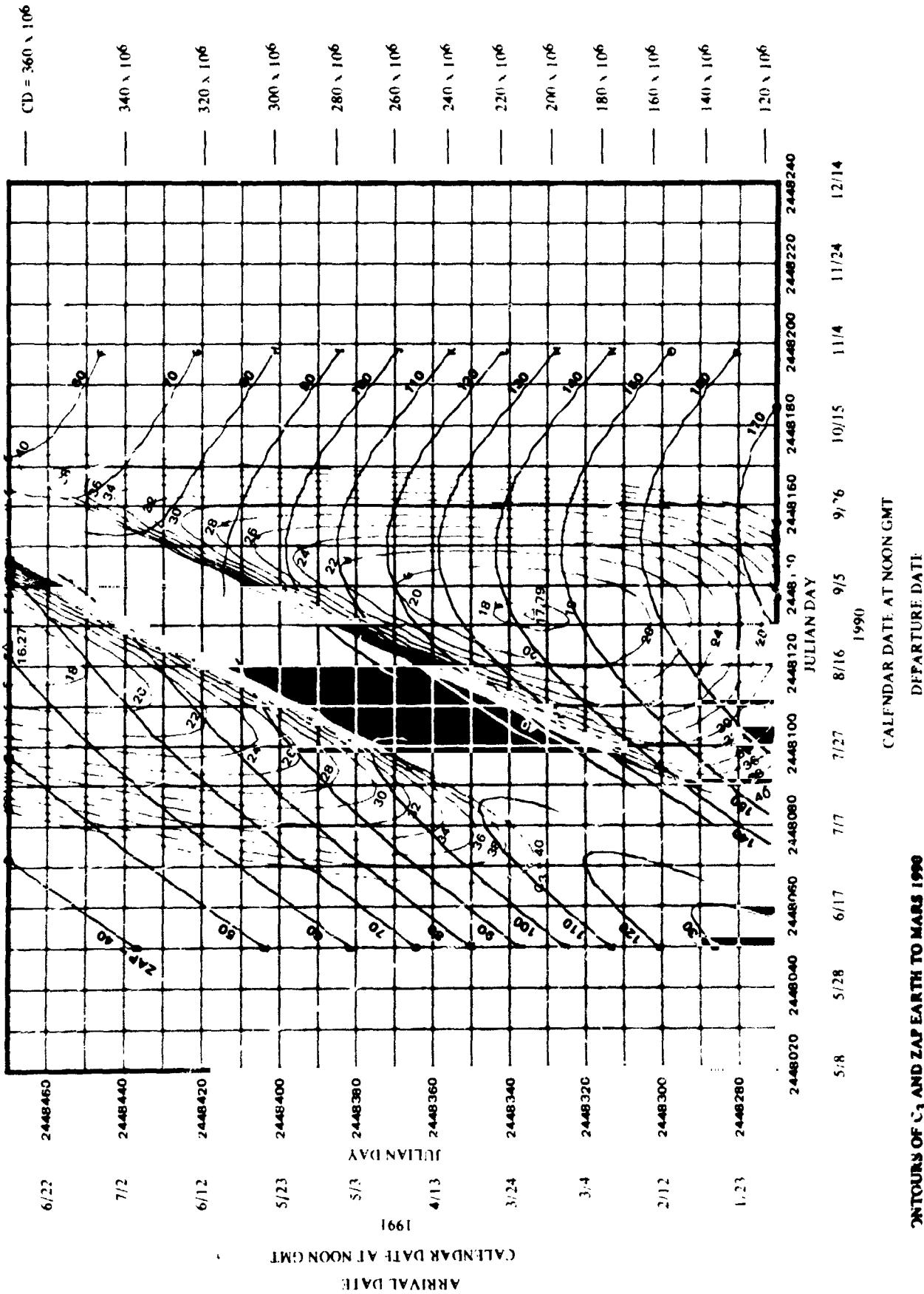


INC
♂
1990

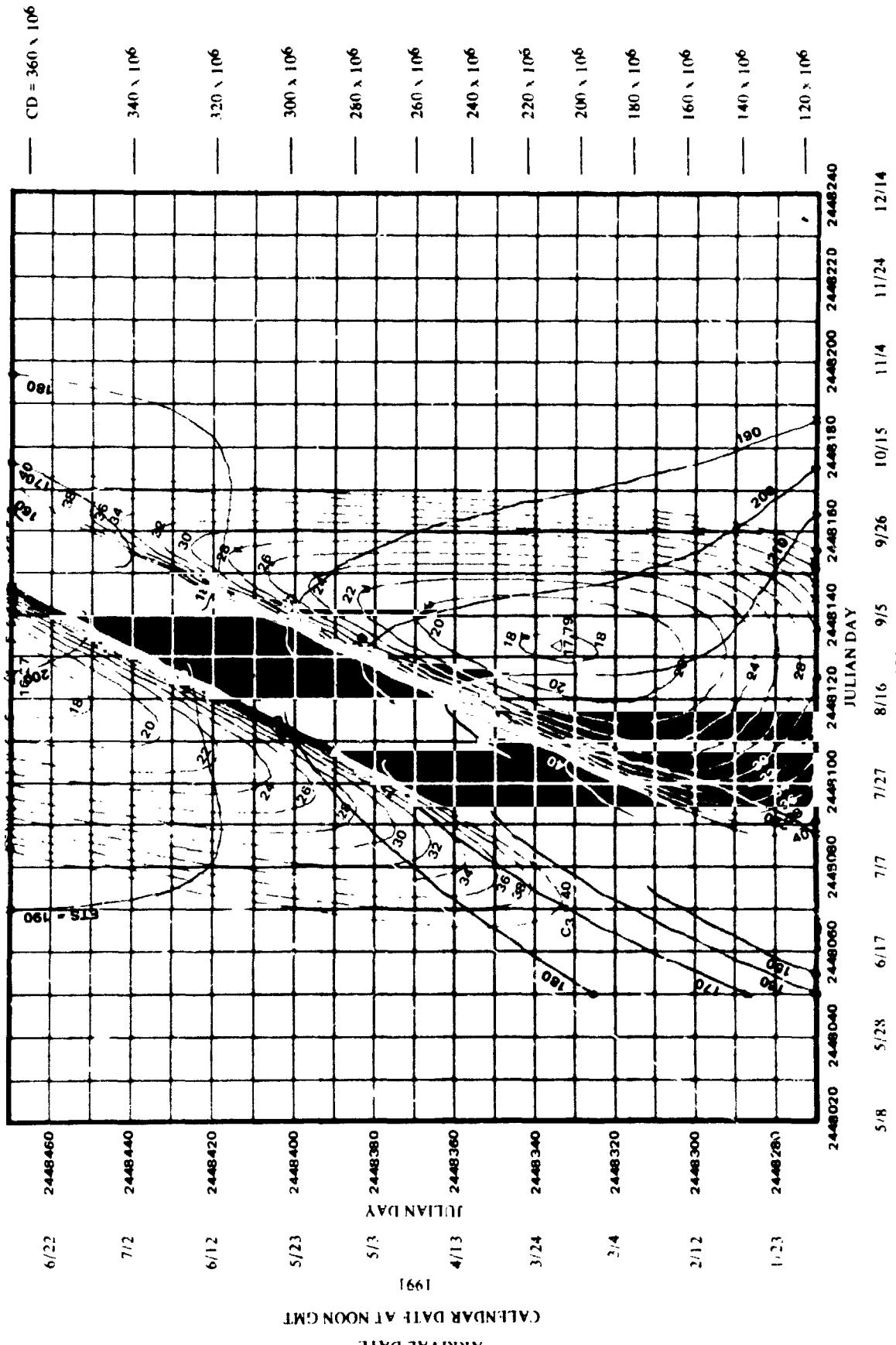


CONTOURS DE C₁ ET INCARCÉRATION DE MARS (1991)

DEPARTURE DATA



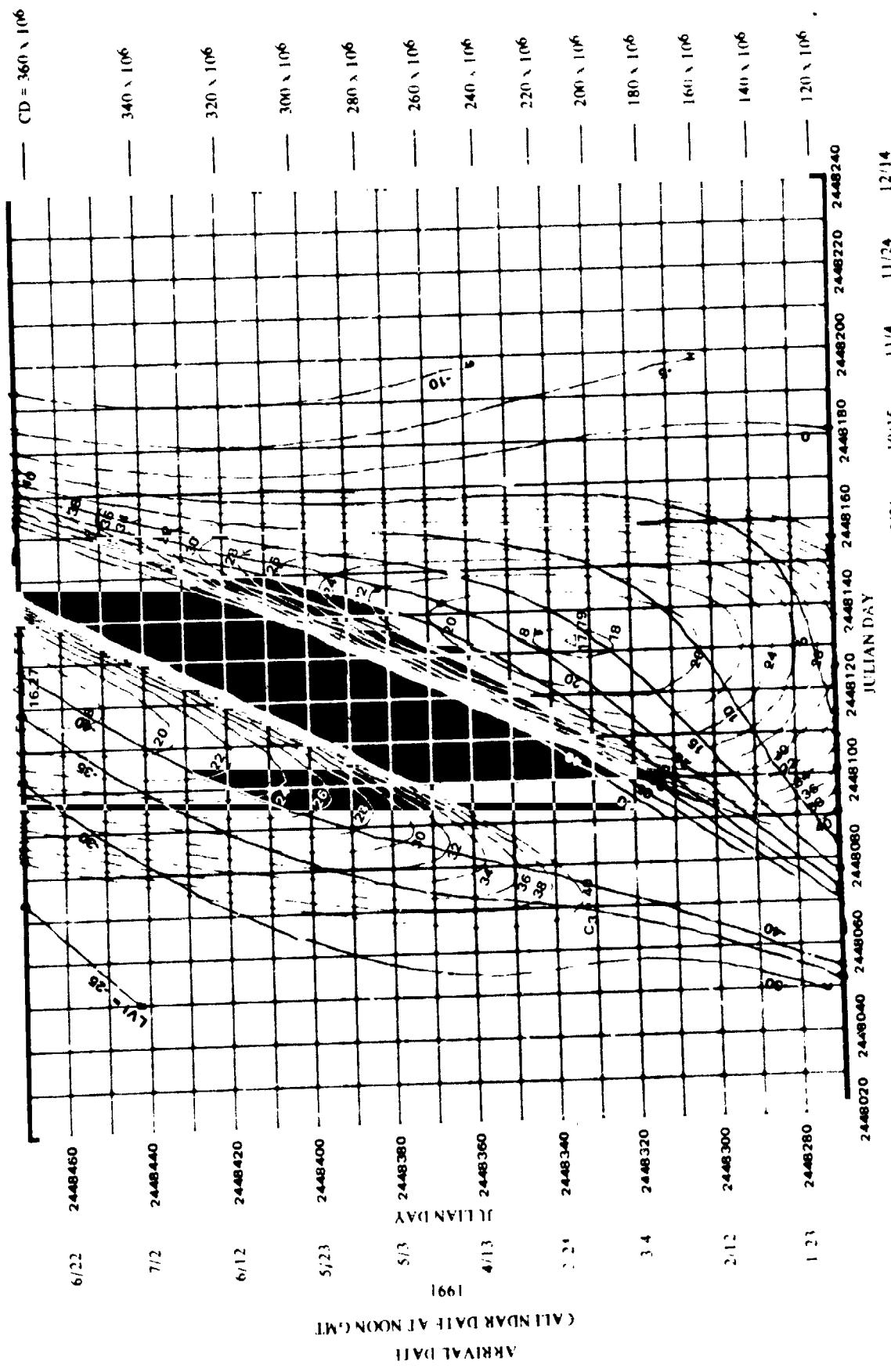
ETS
♂
1990



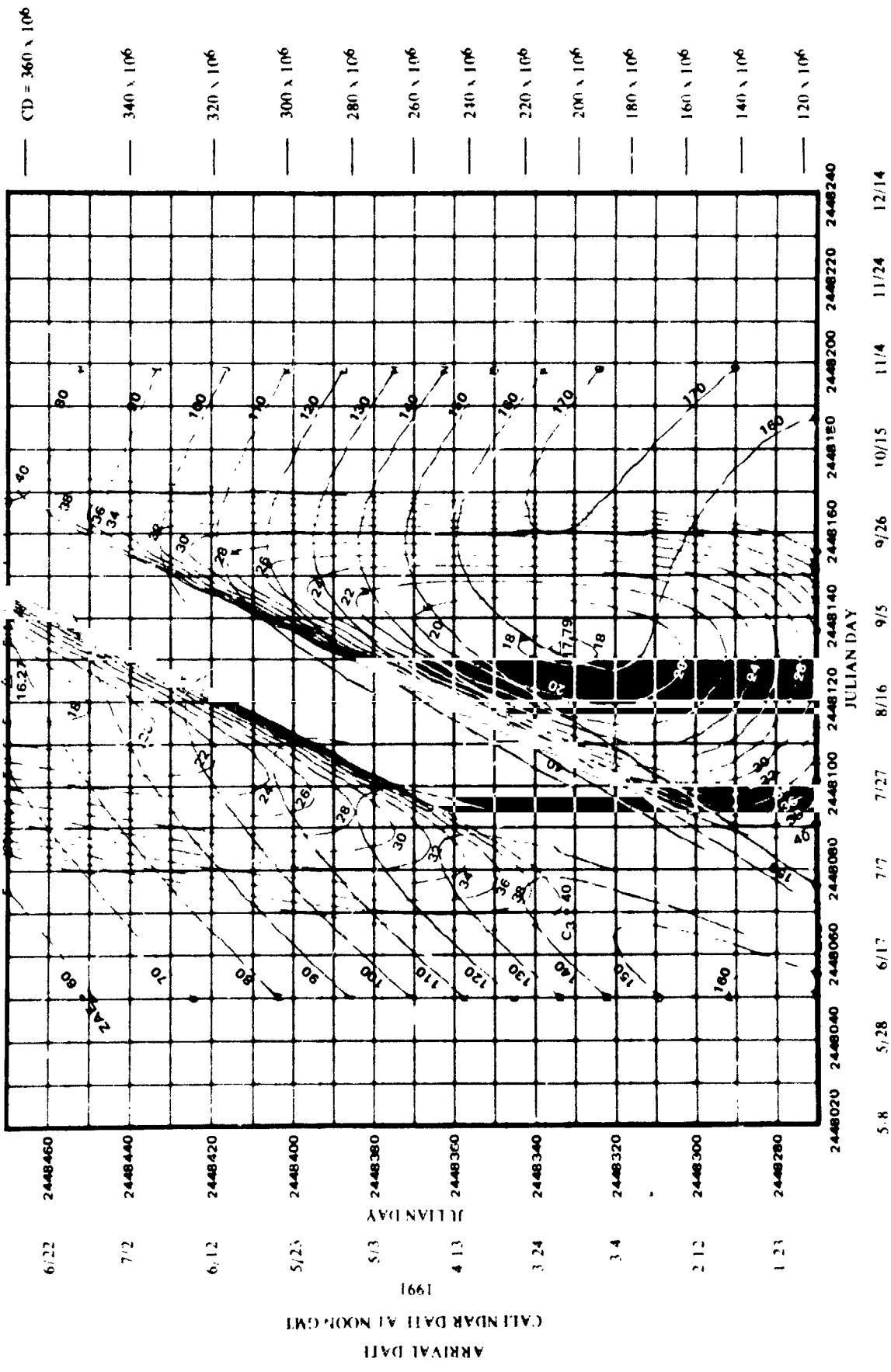
CONTOURS OF C₃ AND ET'S EARTH TO MARS 1990

CALNDAR DATE AT NOON GMT
DEPARTURE DATE

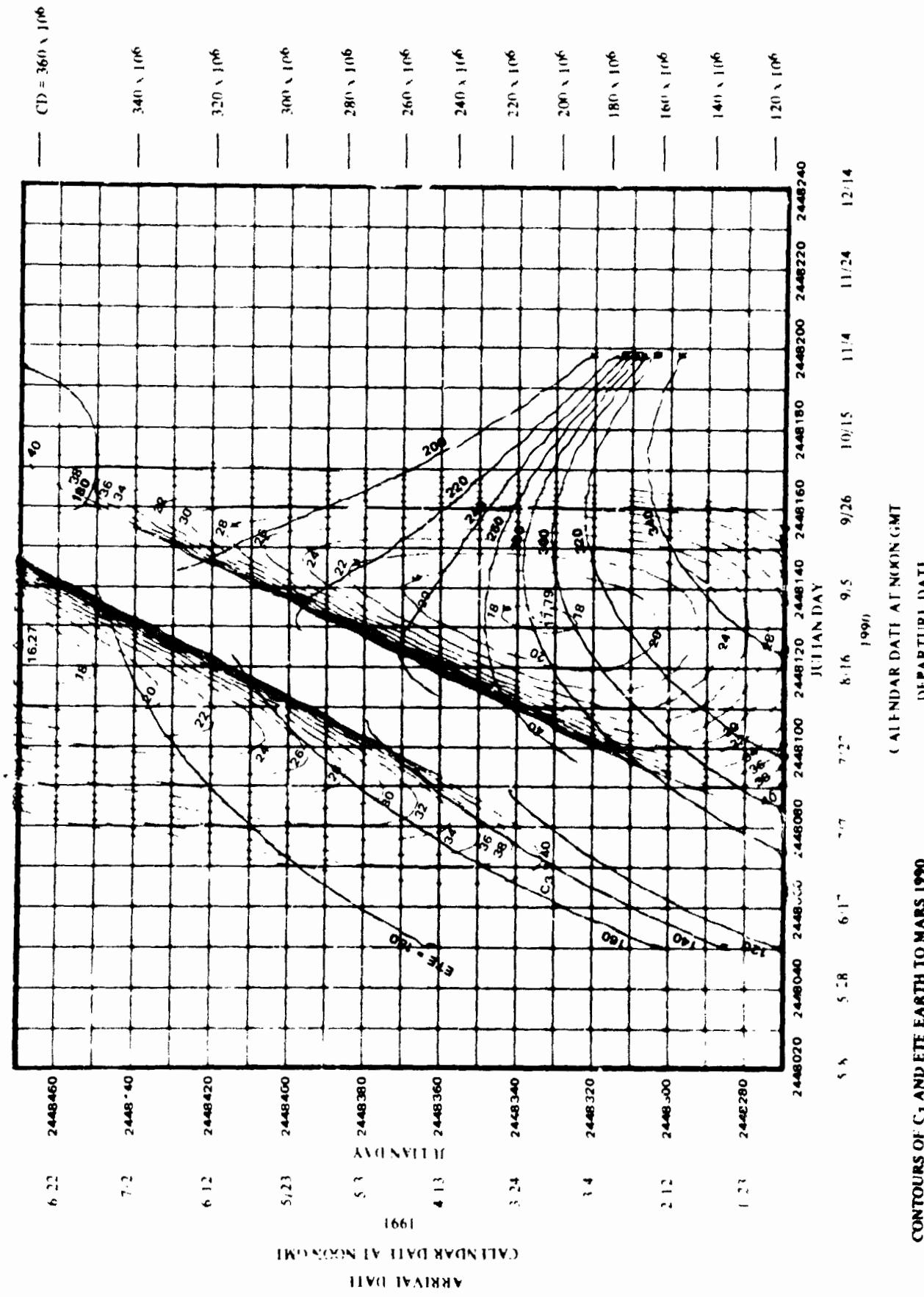
Q E
1990



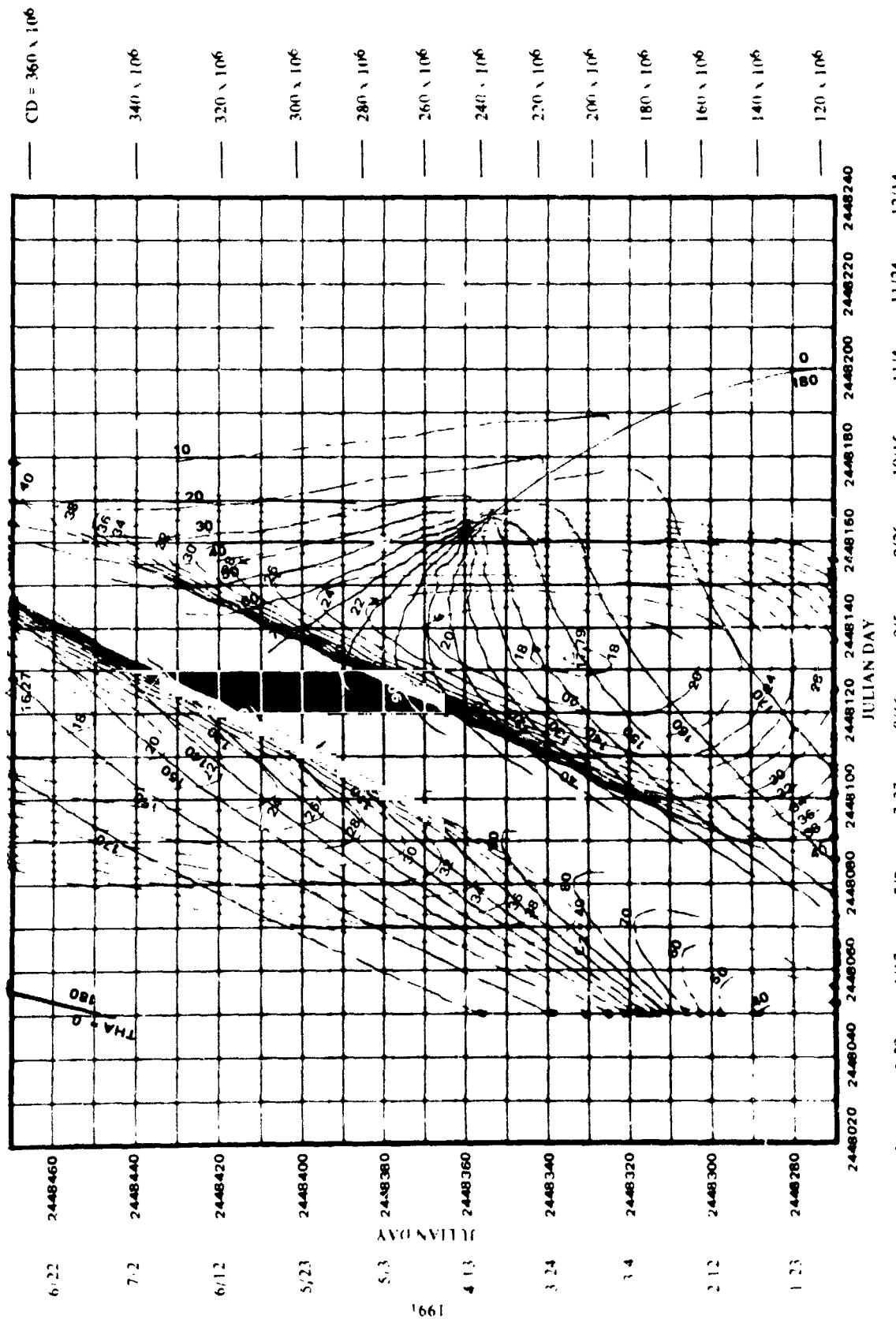
CONTOURS OF C₃ AND LVI EARTH TO MARS 1990



ETE
♂
1990



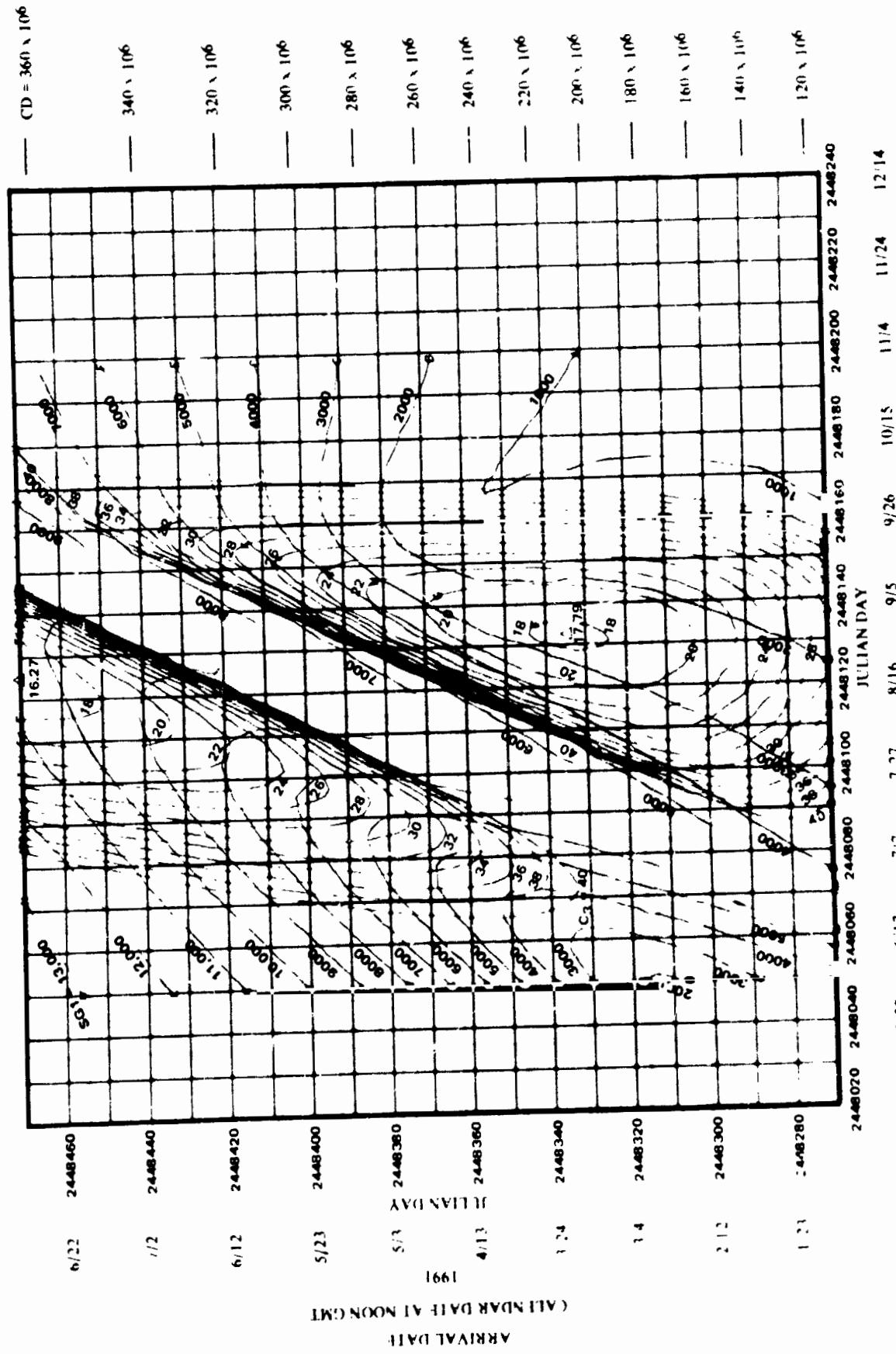
THA
Q
1990



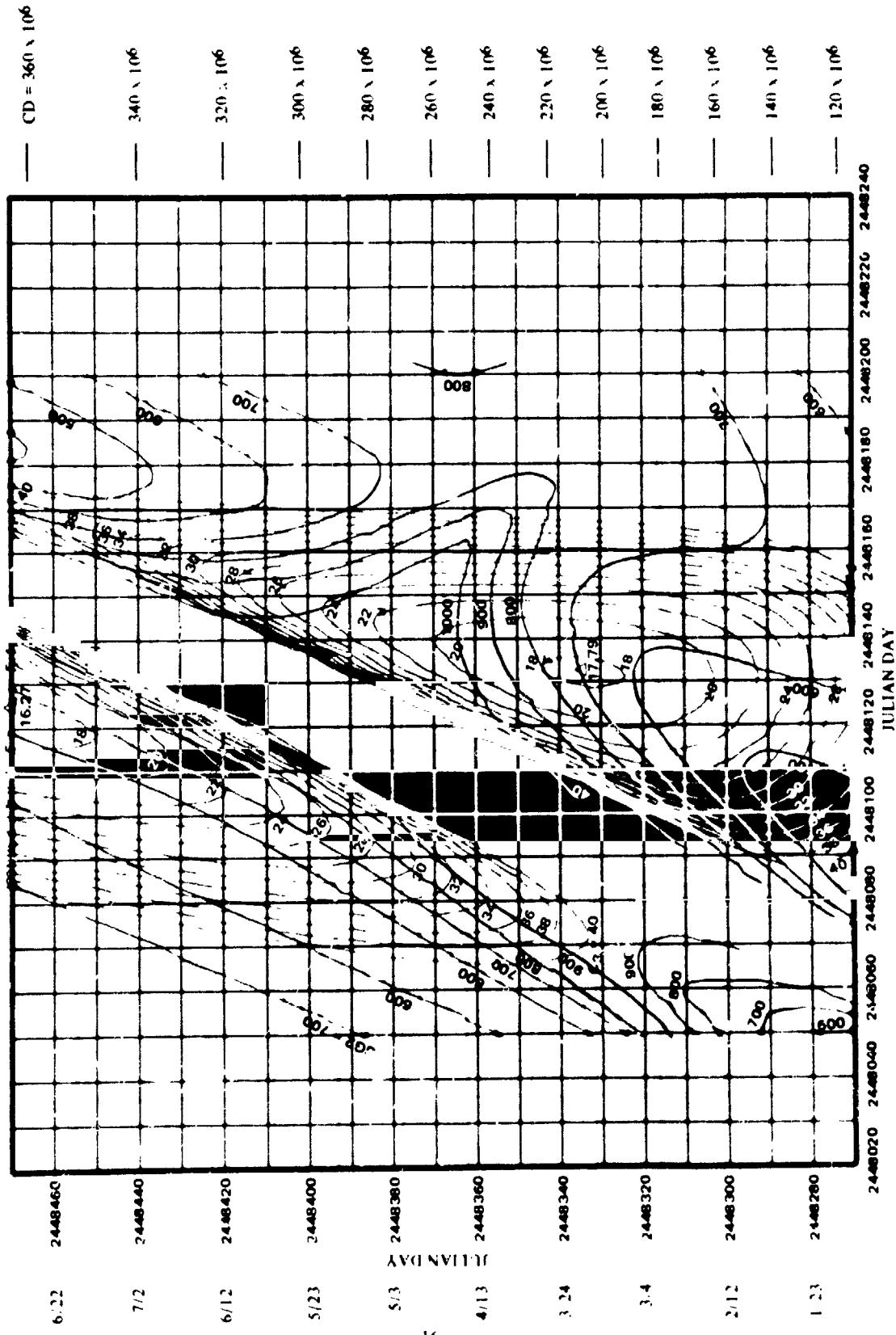
CONTOURS OF C₃ AND THA EARTH TO MARS 1990

CALNDAR DATE AT NOON GMT
DEPARTURE DATE

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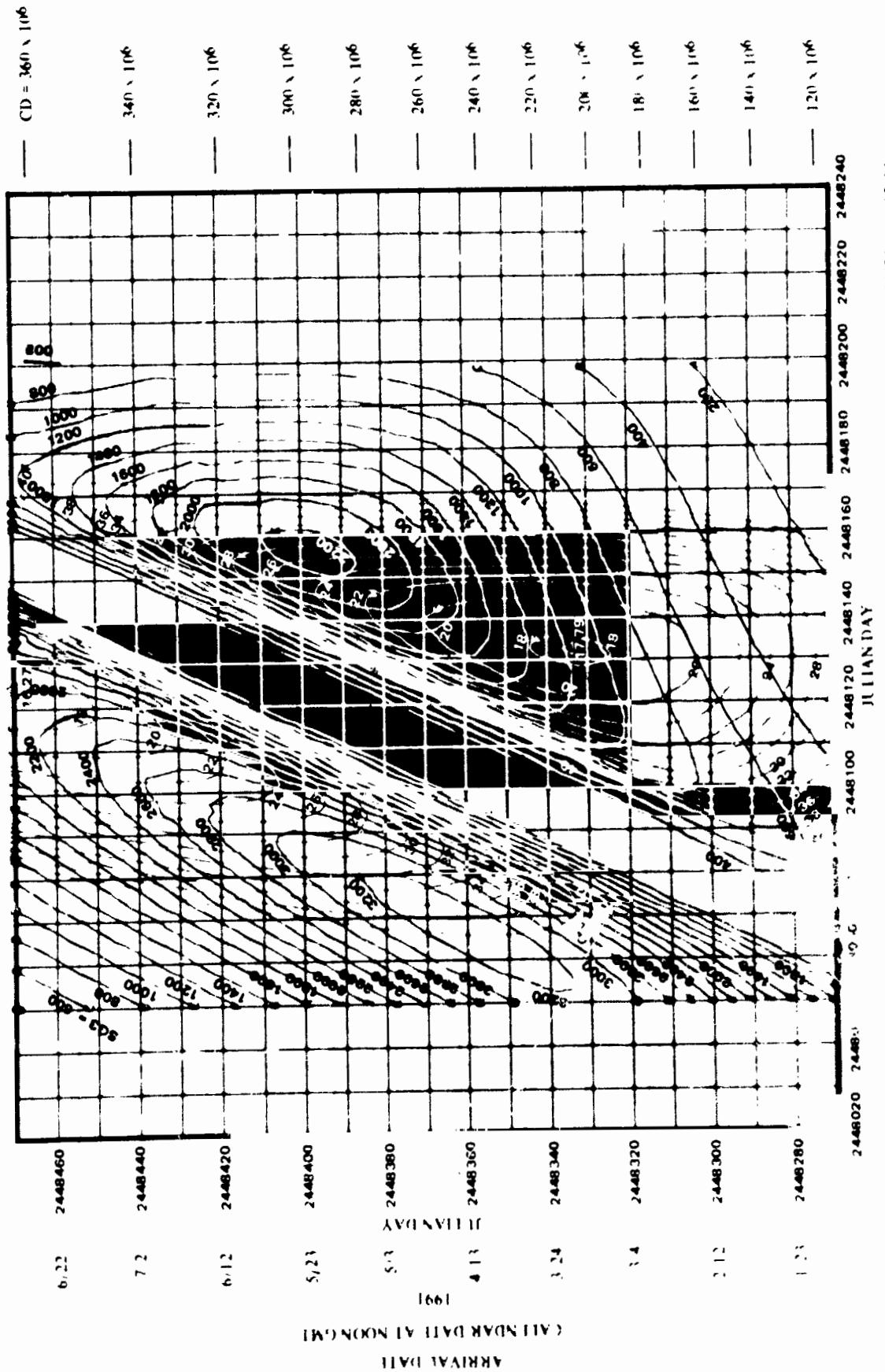
SG2
Q
1990



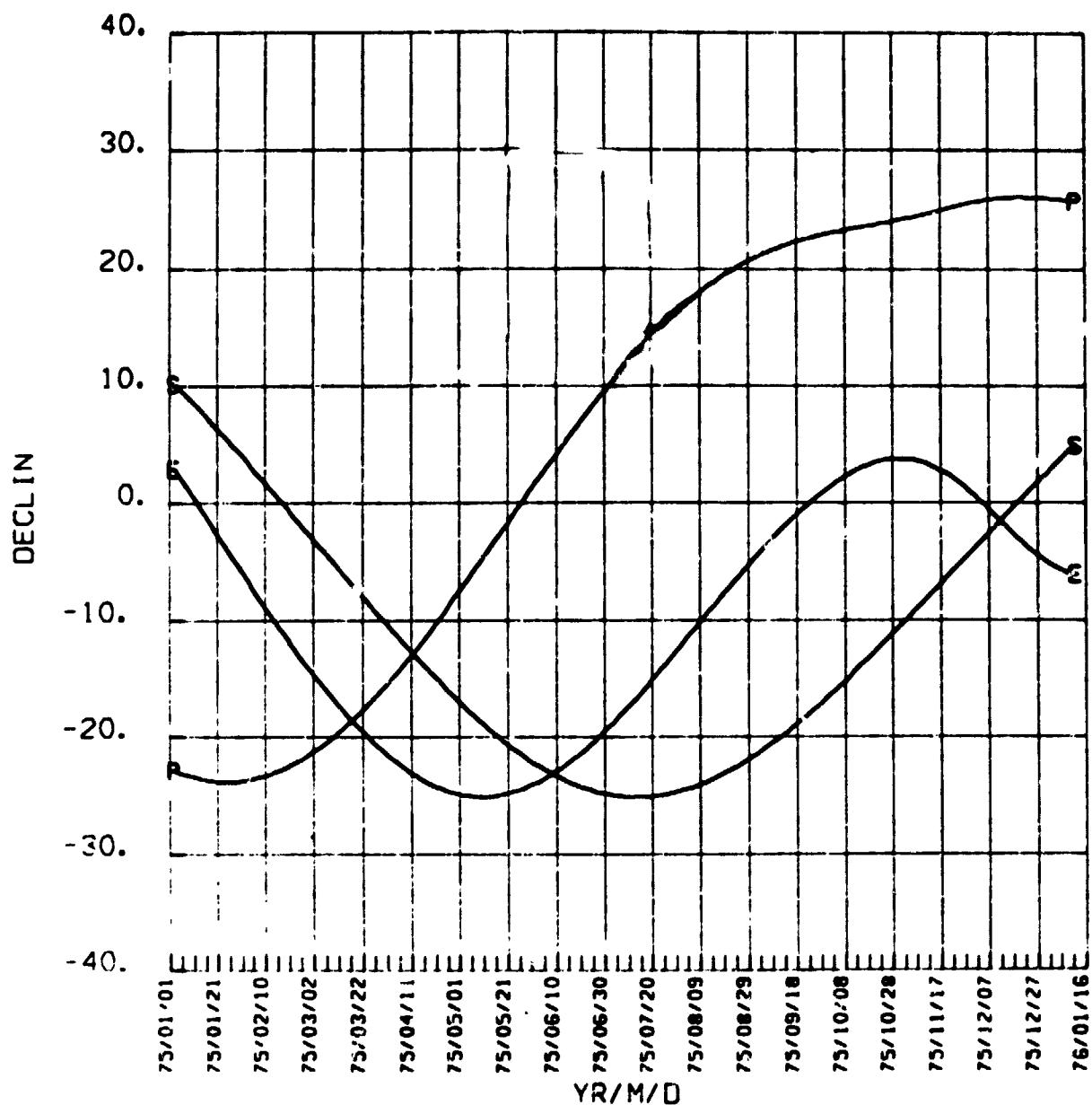
CONT'D: SG1 AND SG2 FROM EARTH TO MARS 1990

CALNDAR DATE AT NOON GMT
DEPARTURE DATE

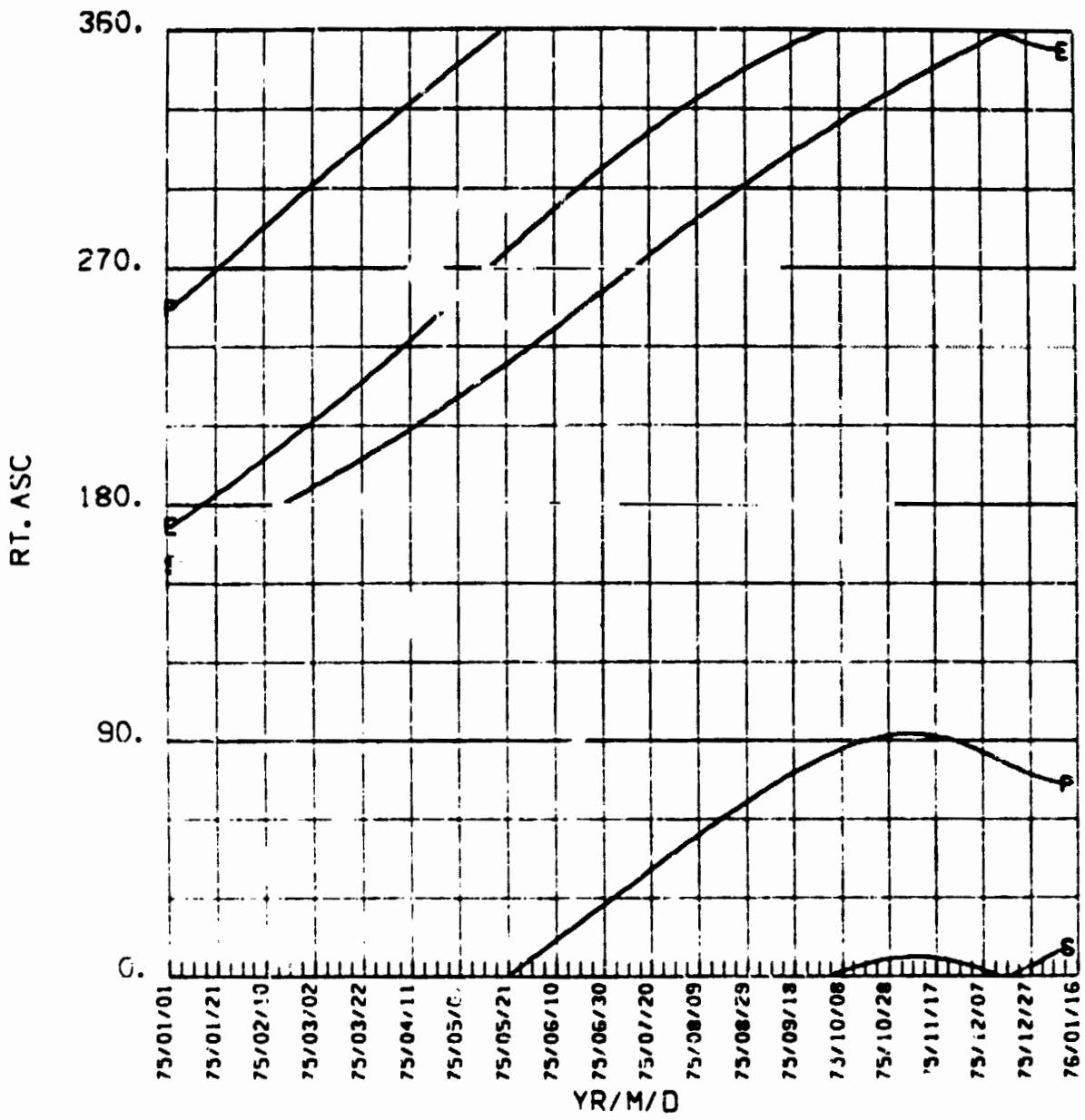
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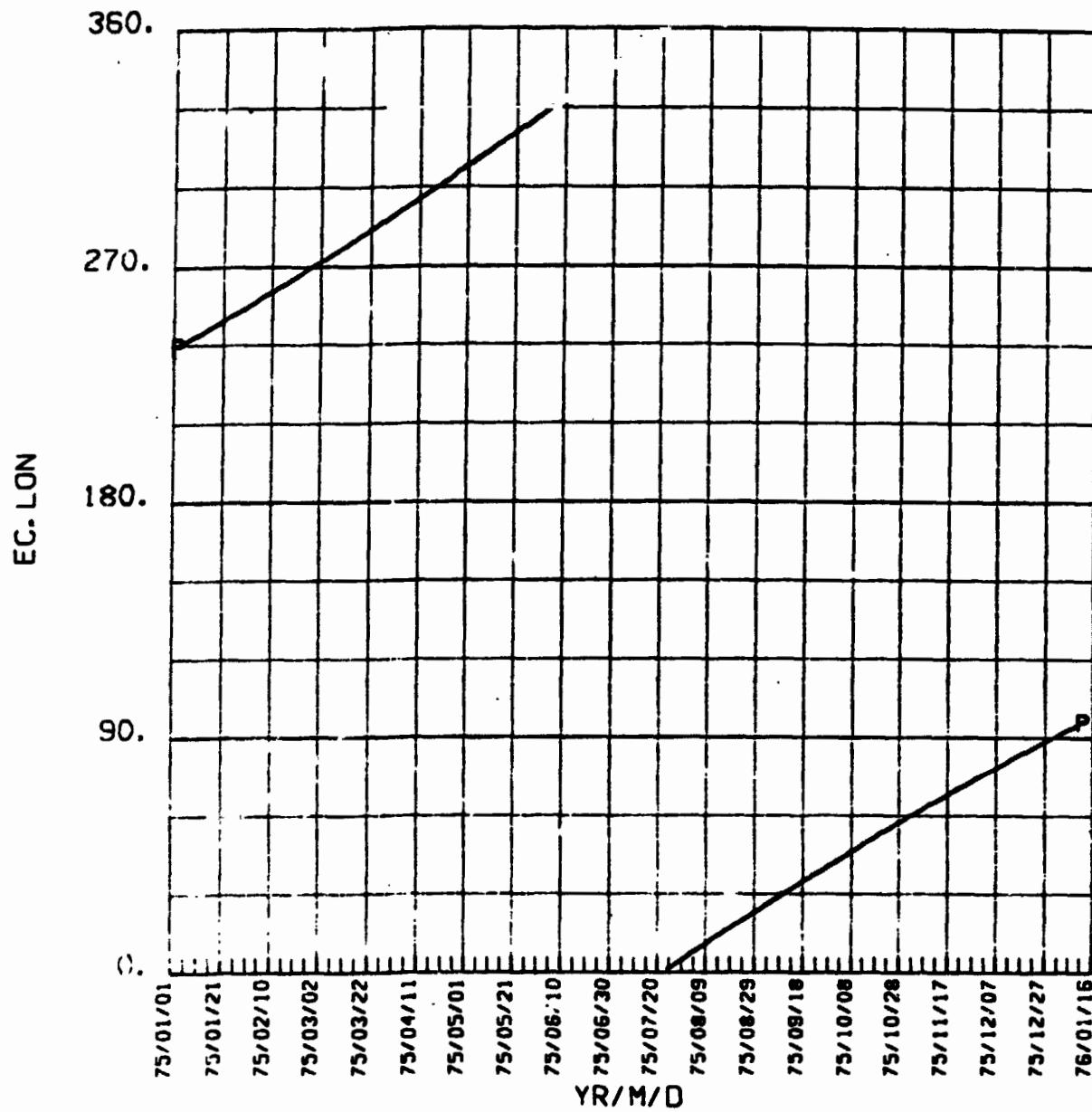
MARS 1975



MARS 1975



MARS 1975

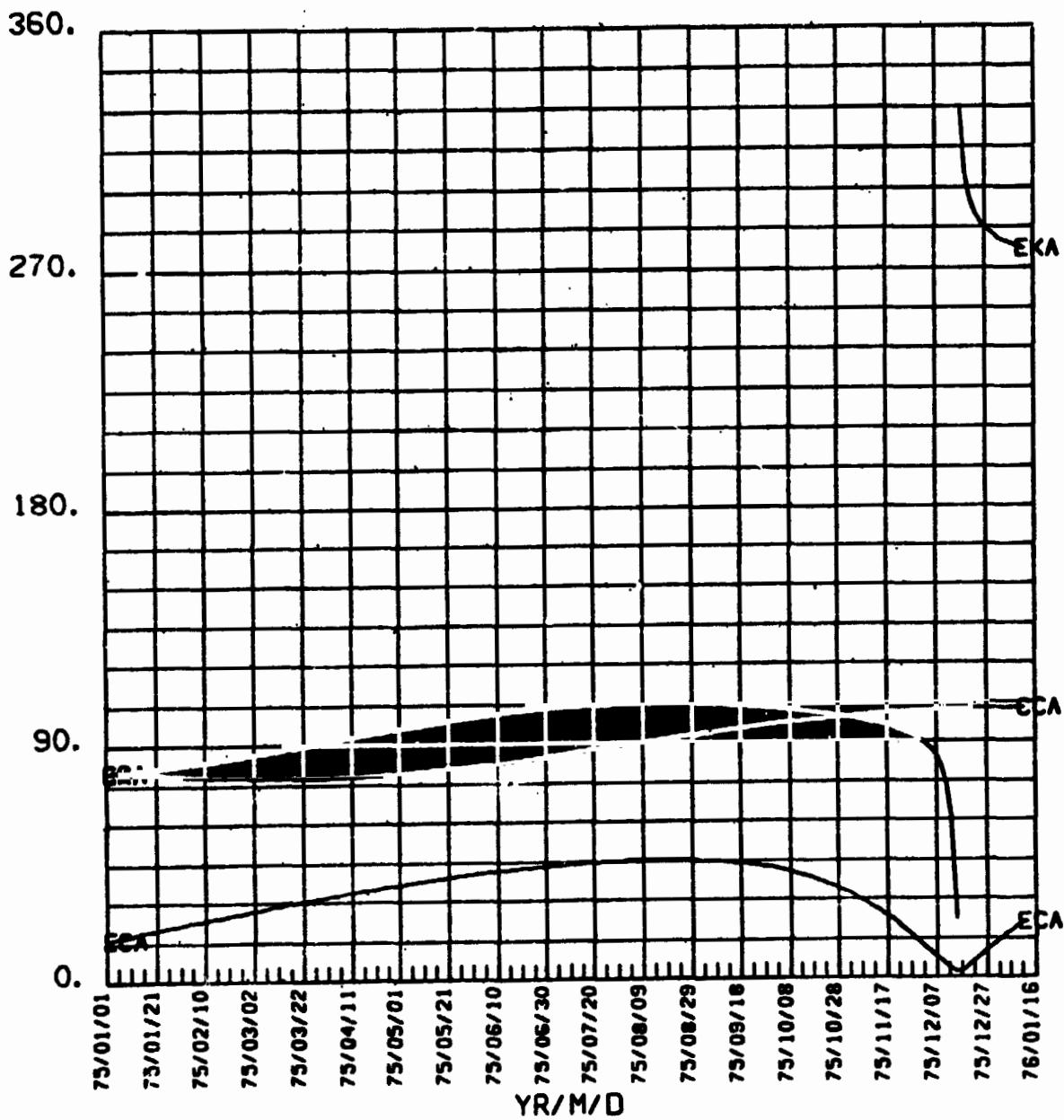


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C - 2

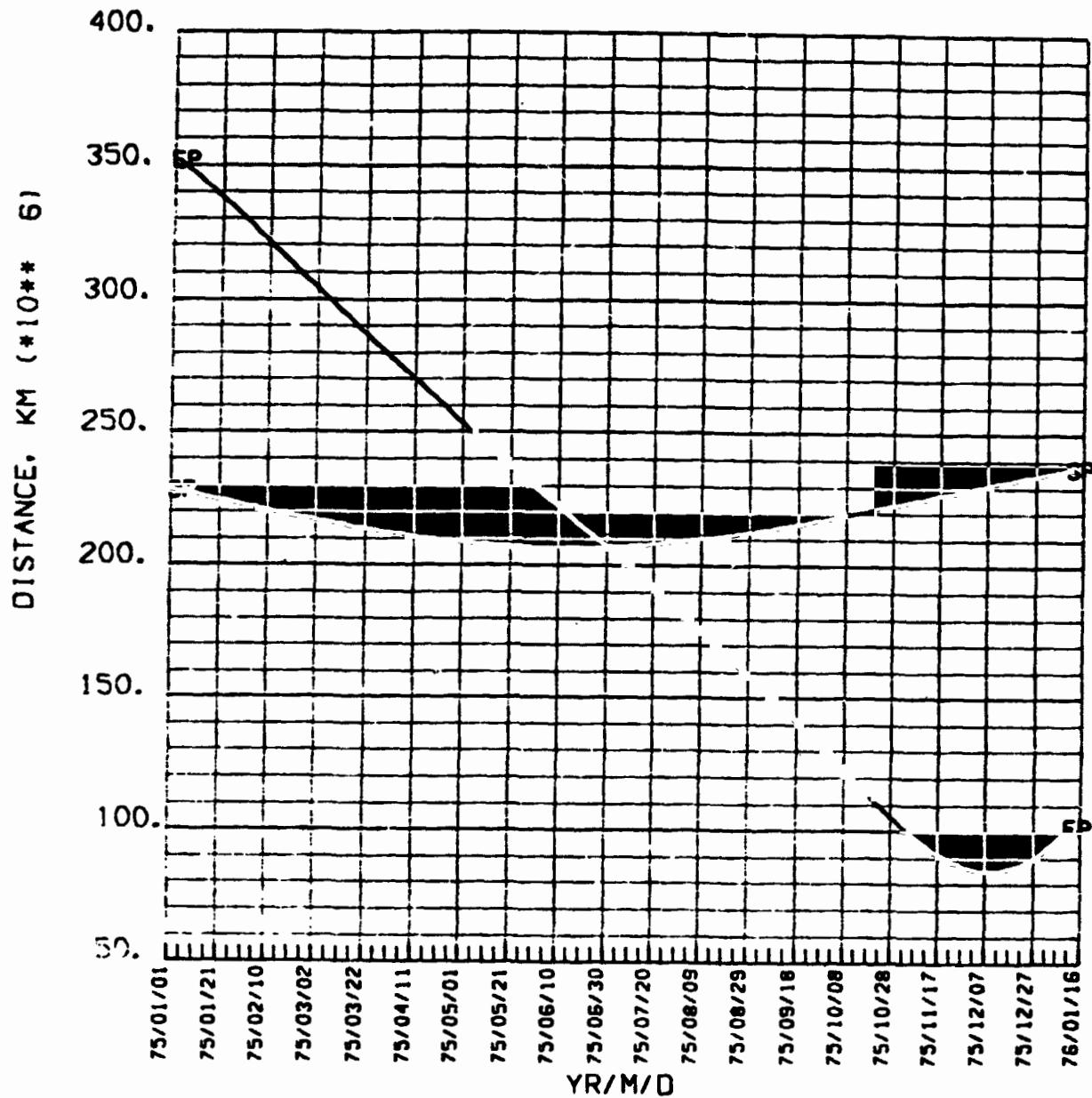
CA, KA OF EARTH, CA CANOP

MARS 1975

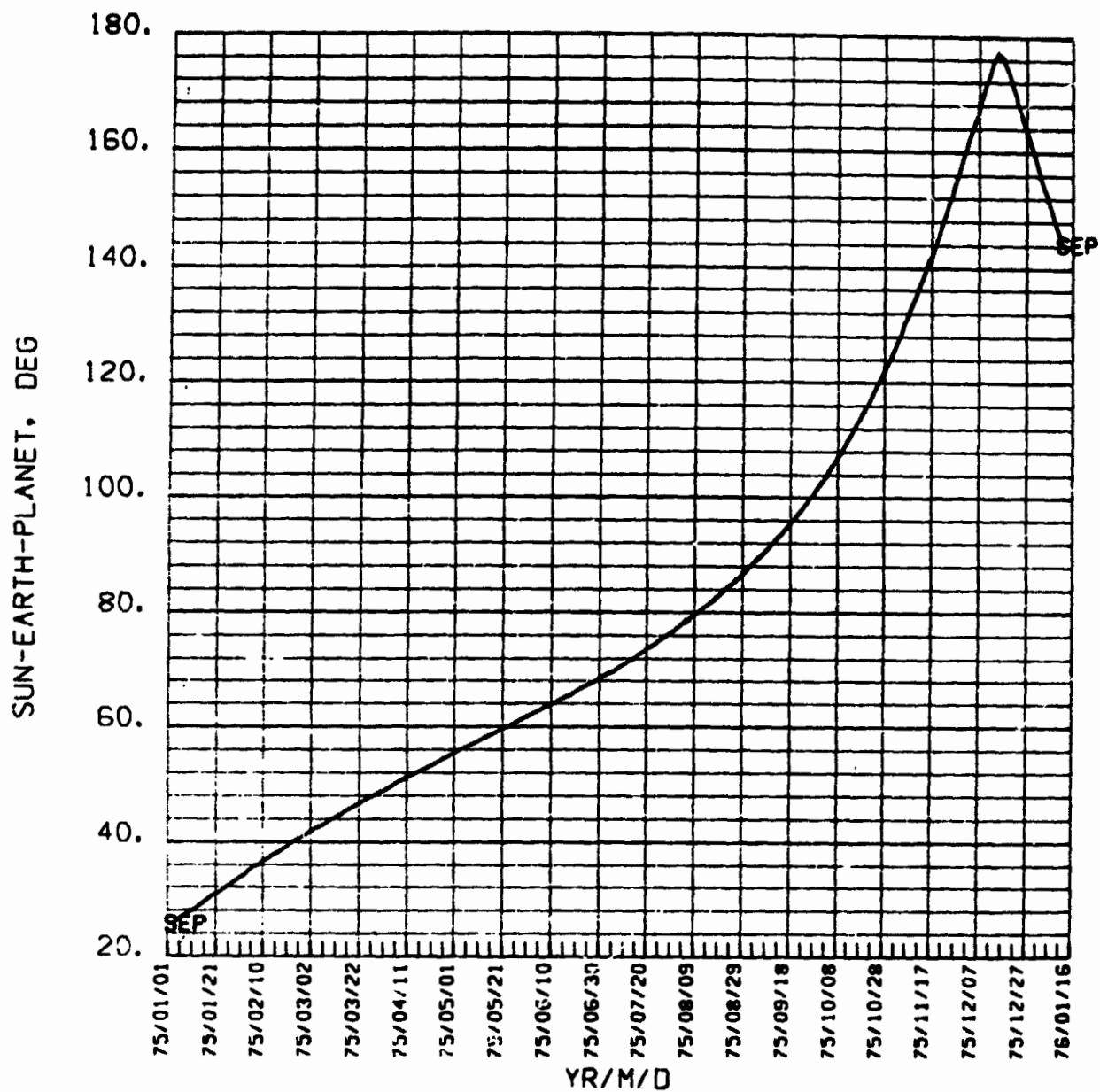


MARS

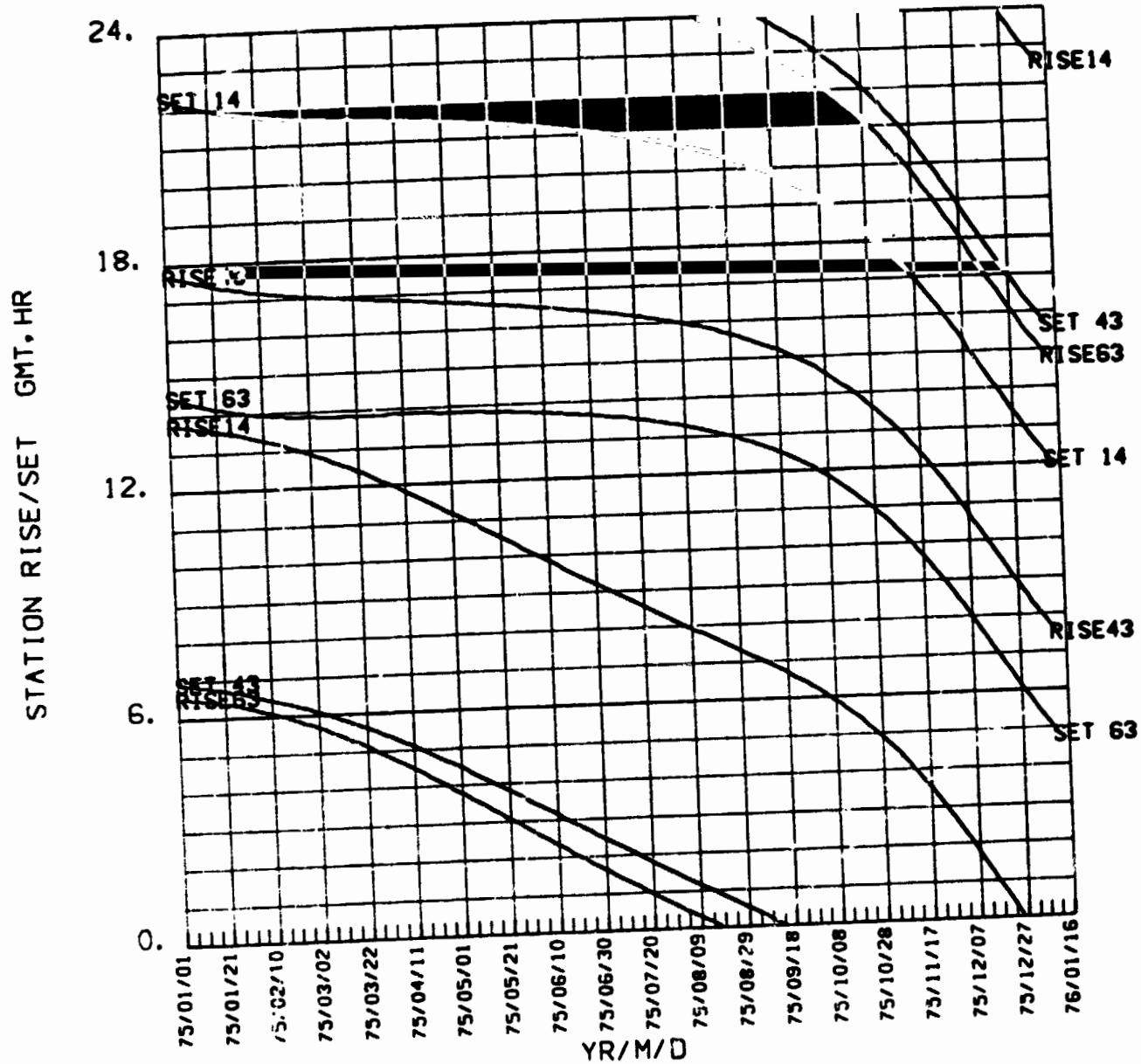
1975



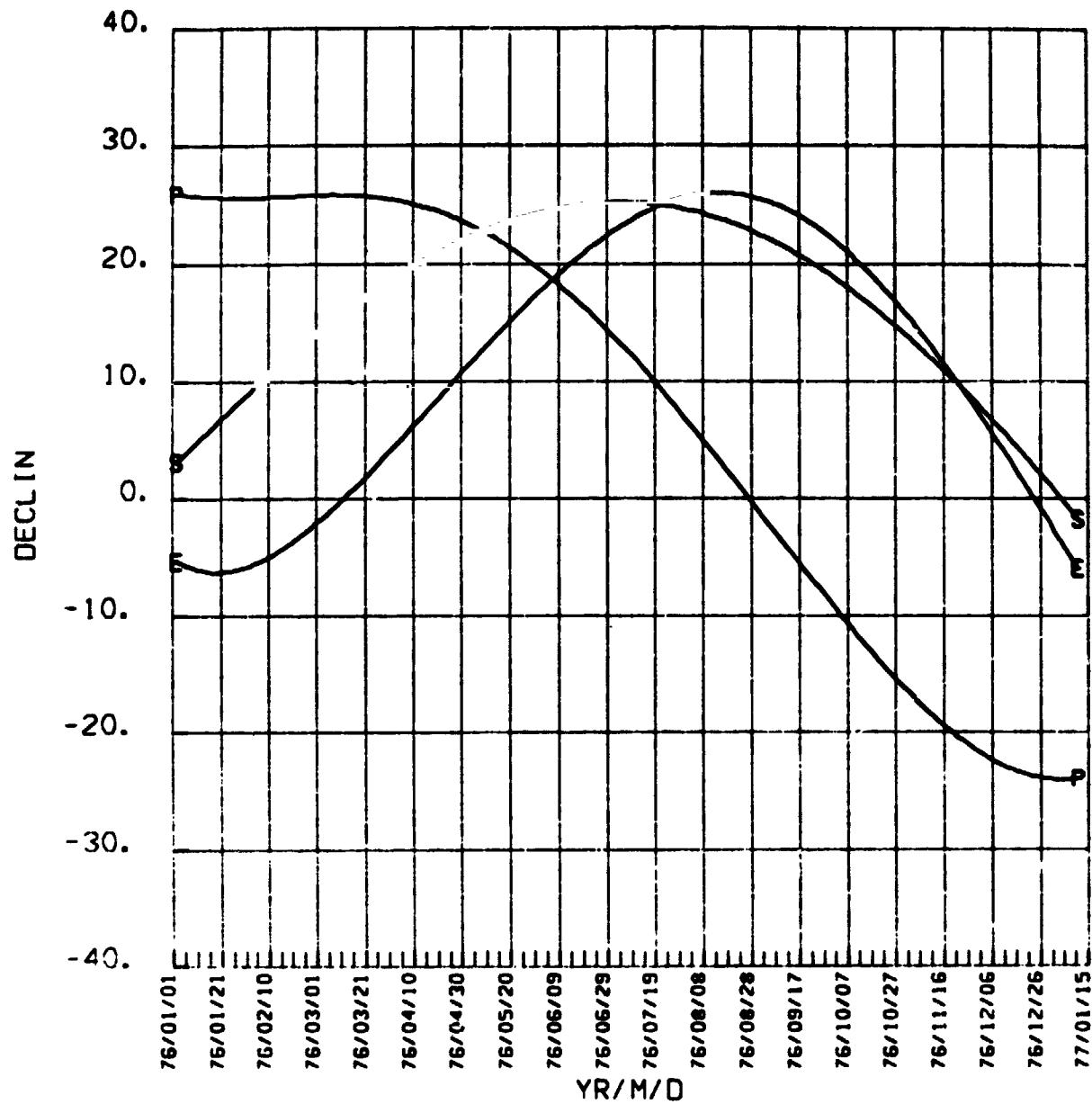
MARS 1975



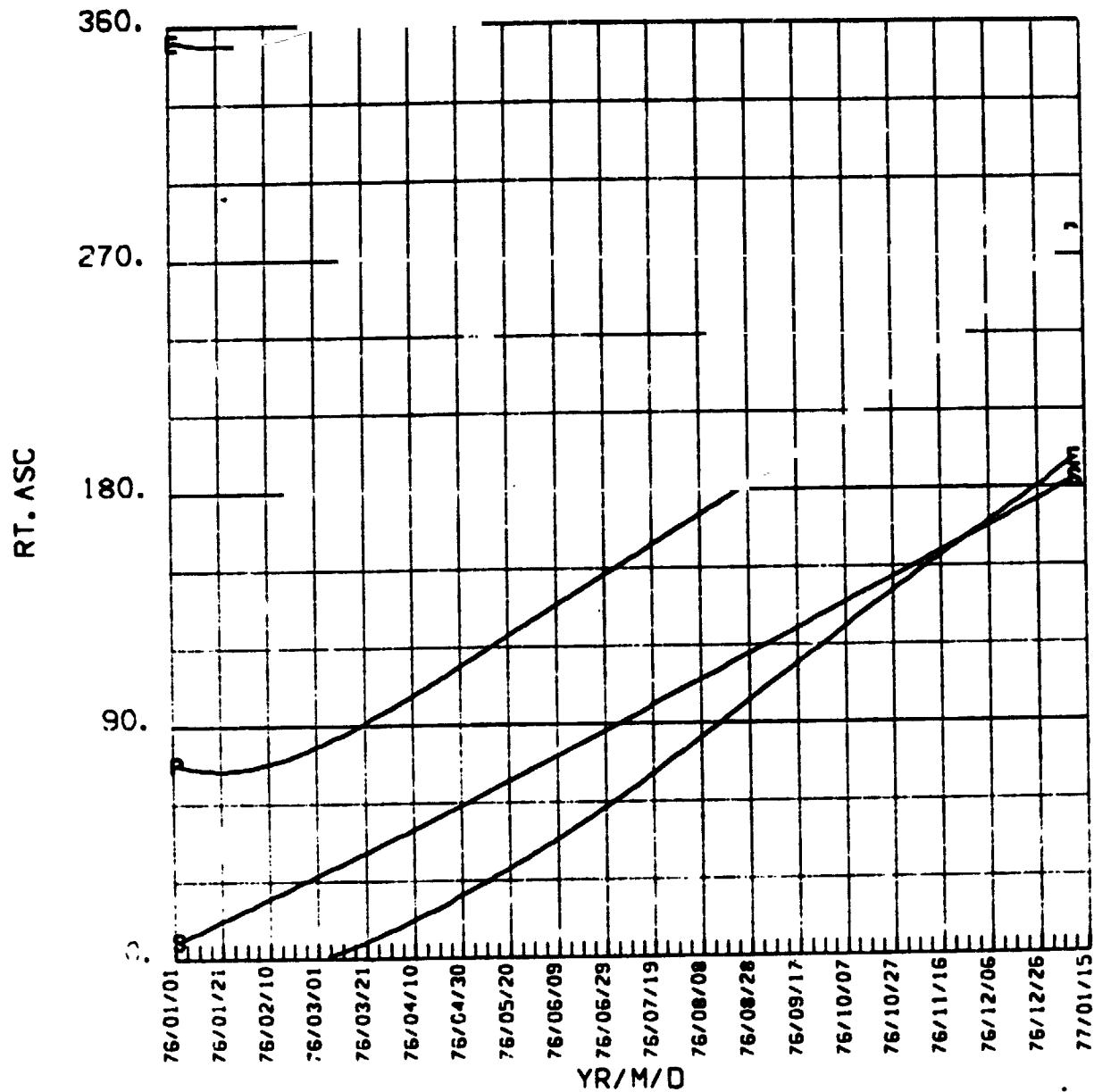
MARS 1975



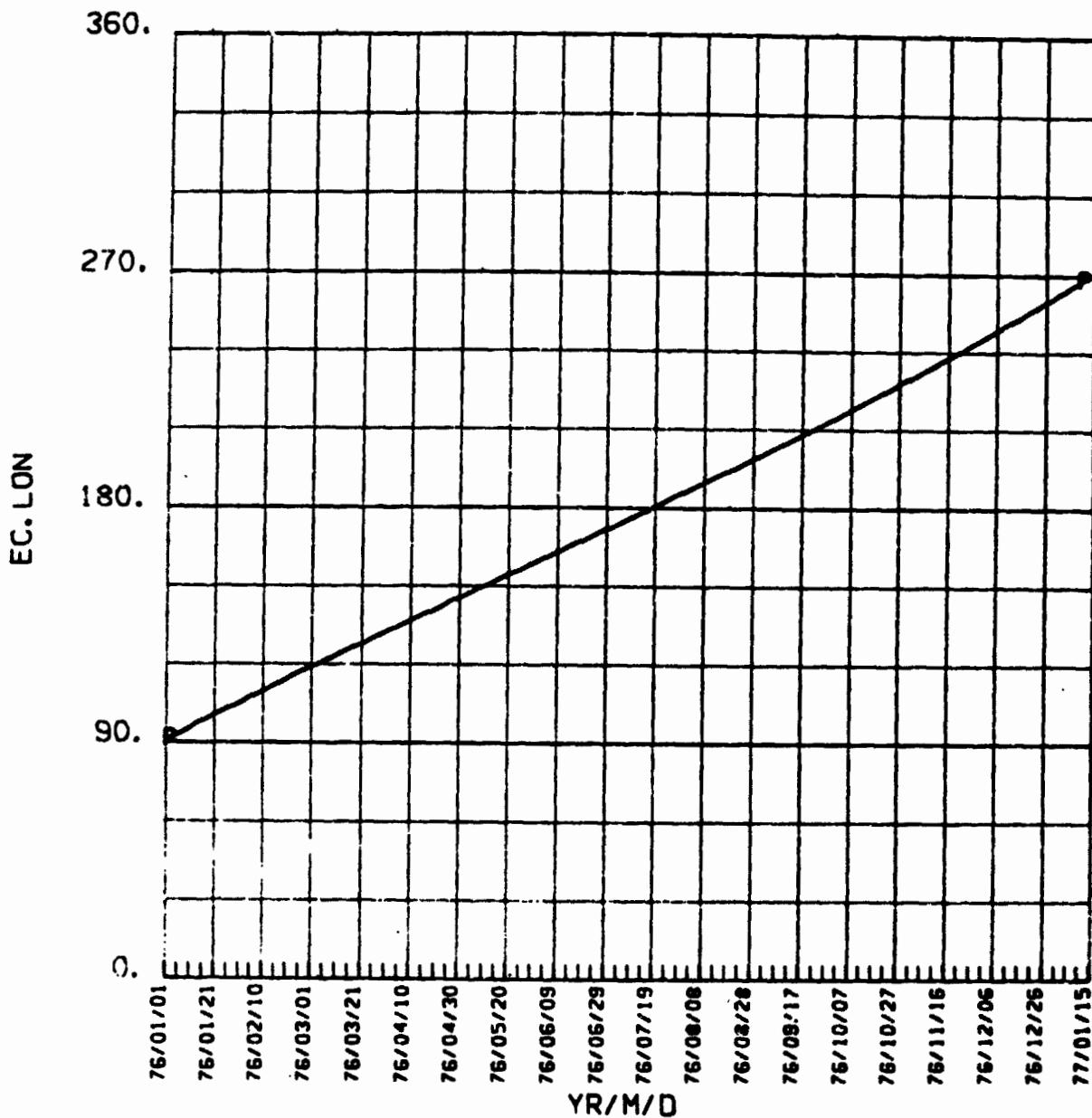
MARS 1976



MARS 1976

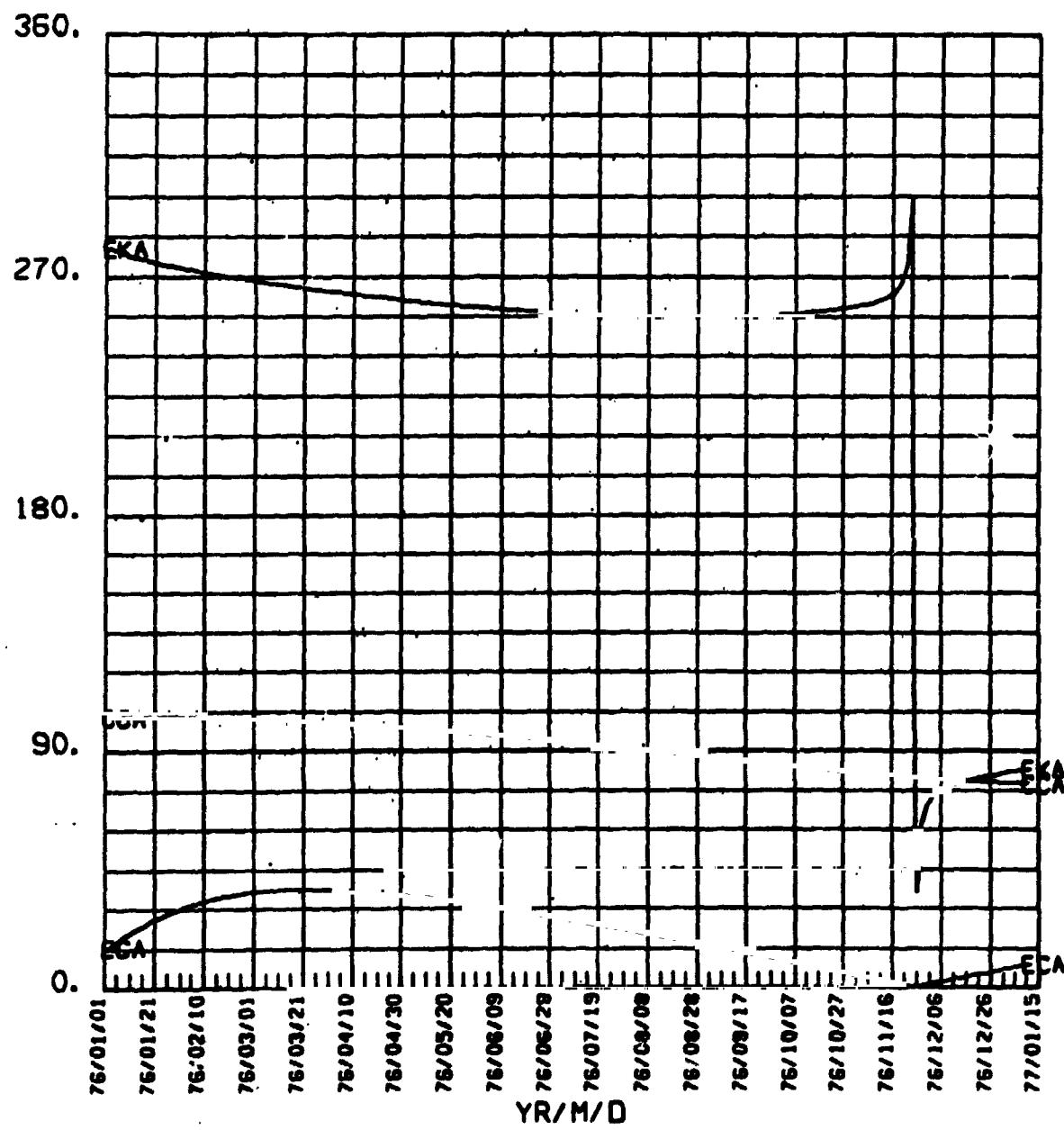


MARS 1976



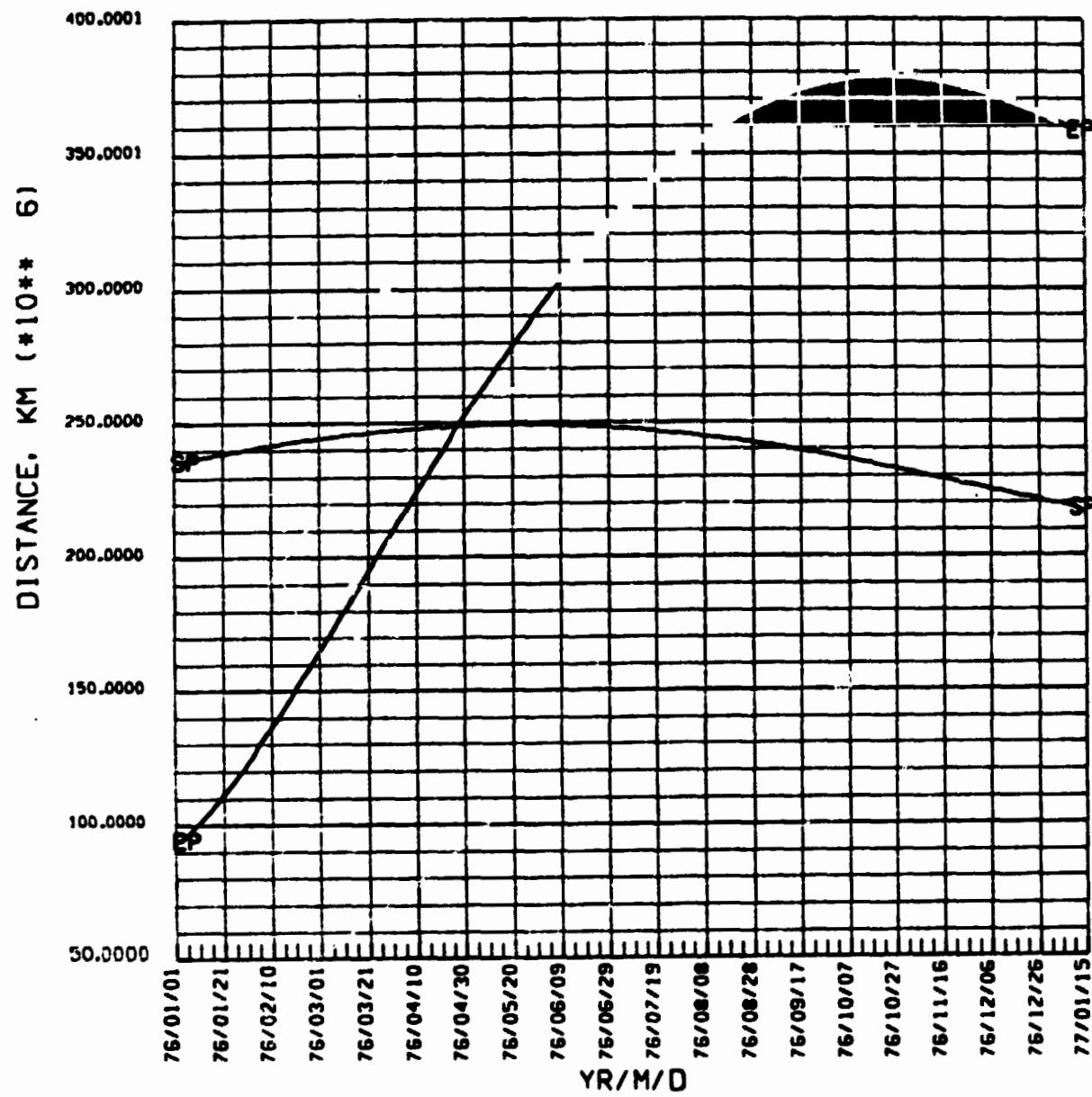
MARS 1976

CA, KA OF EARTH, CA CANOP

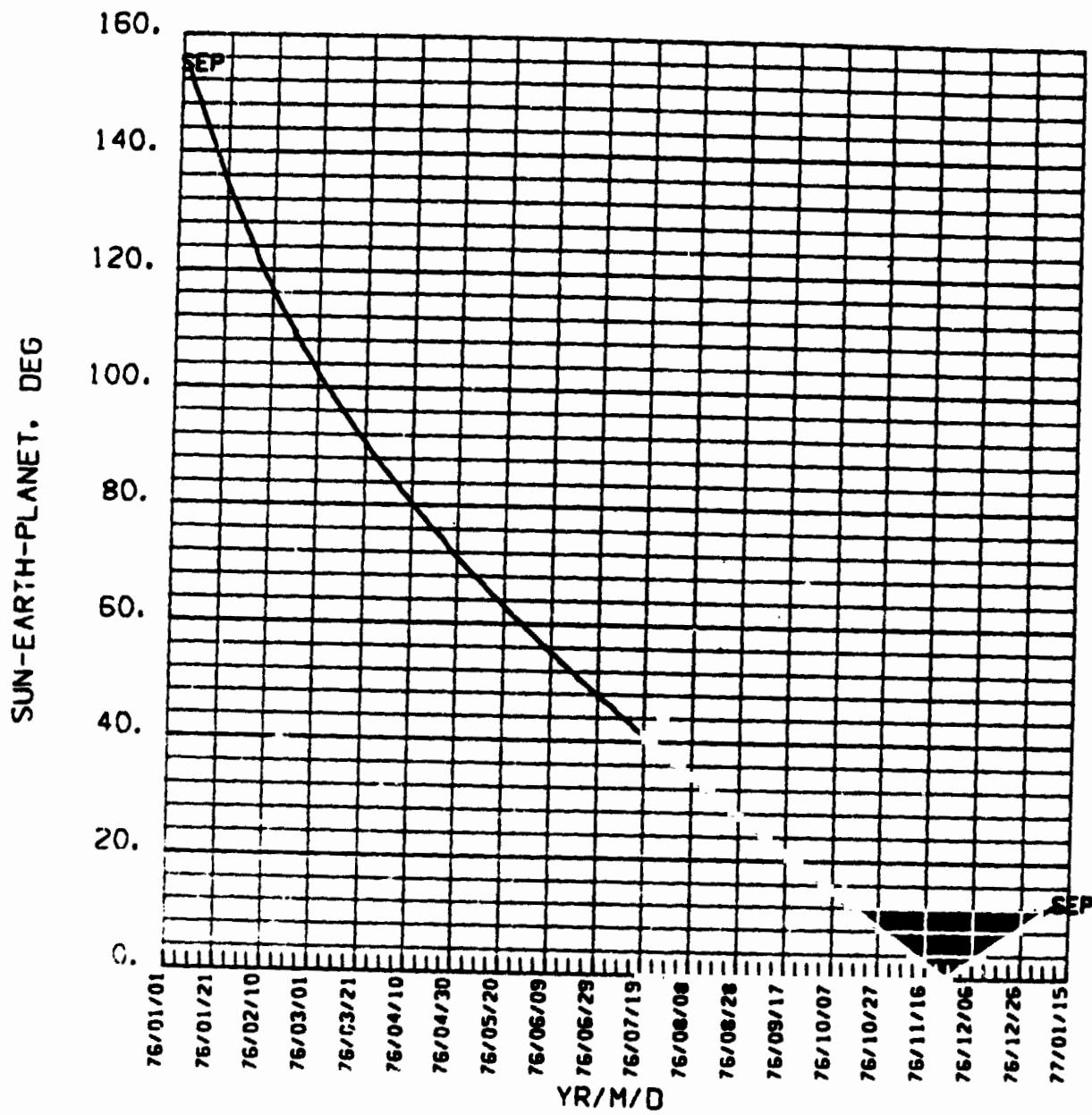


MARS

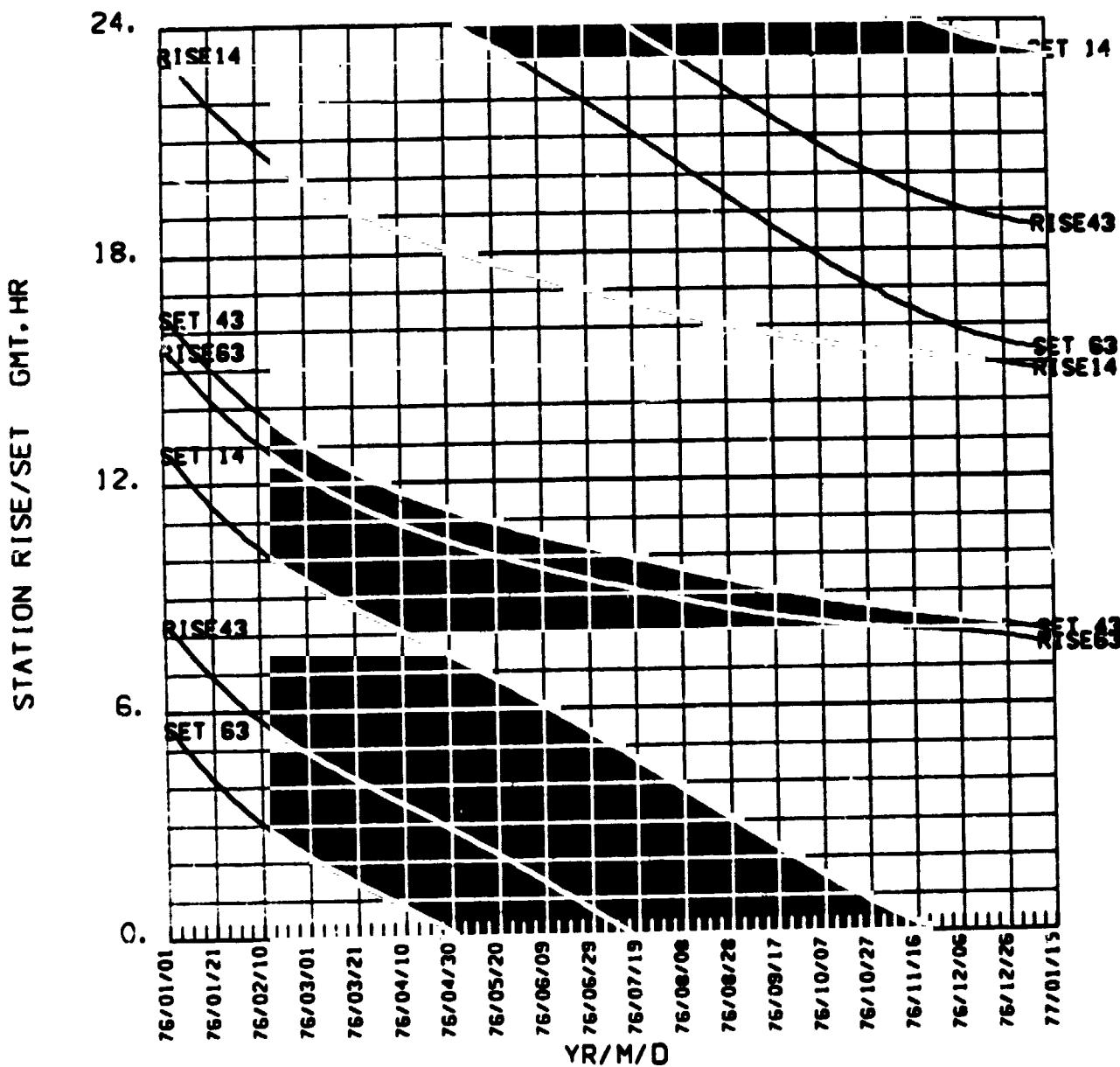
1976



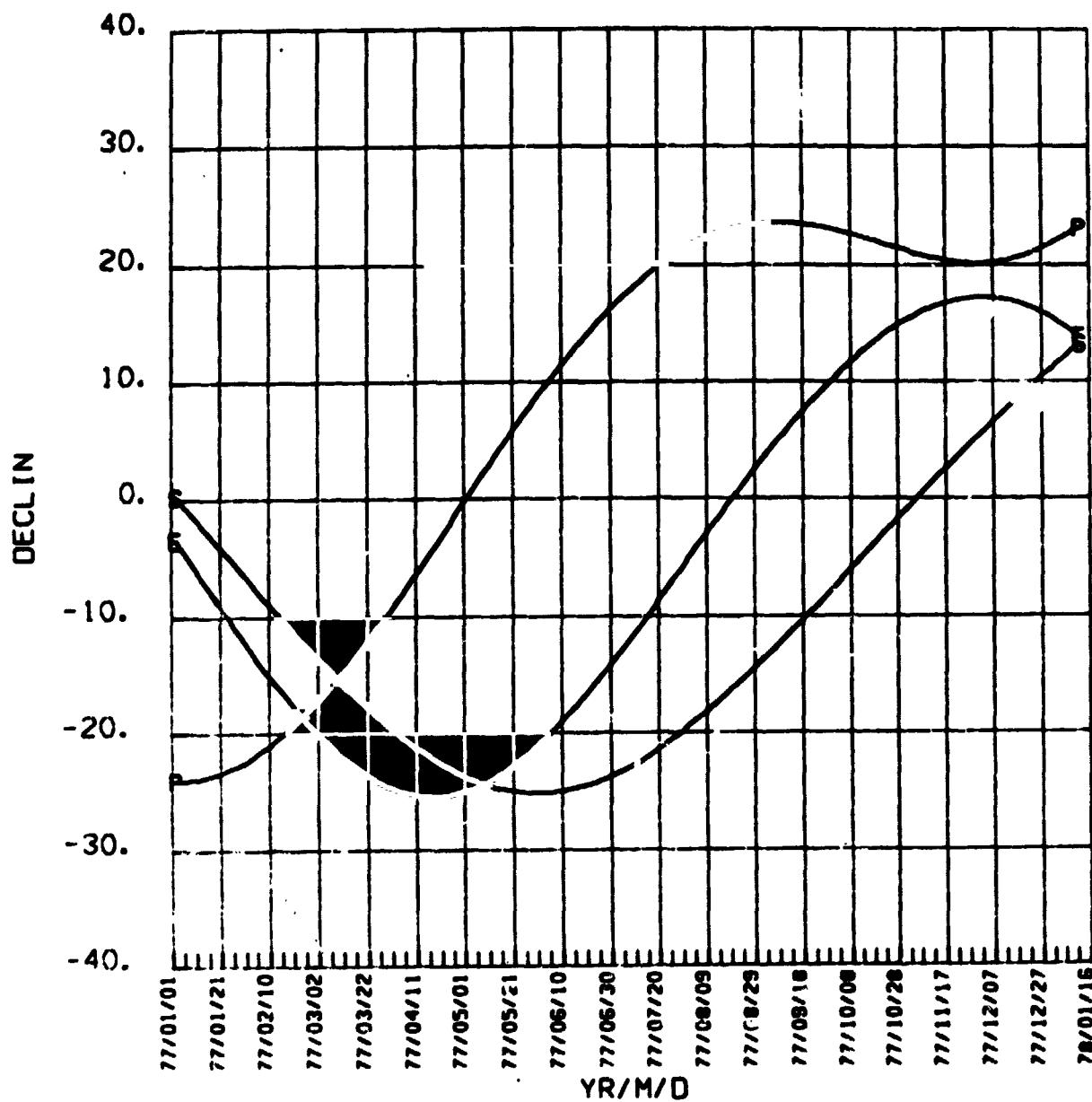
MARS 1976



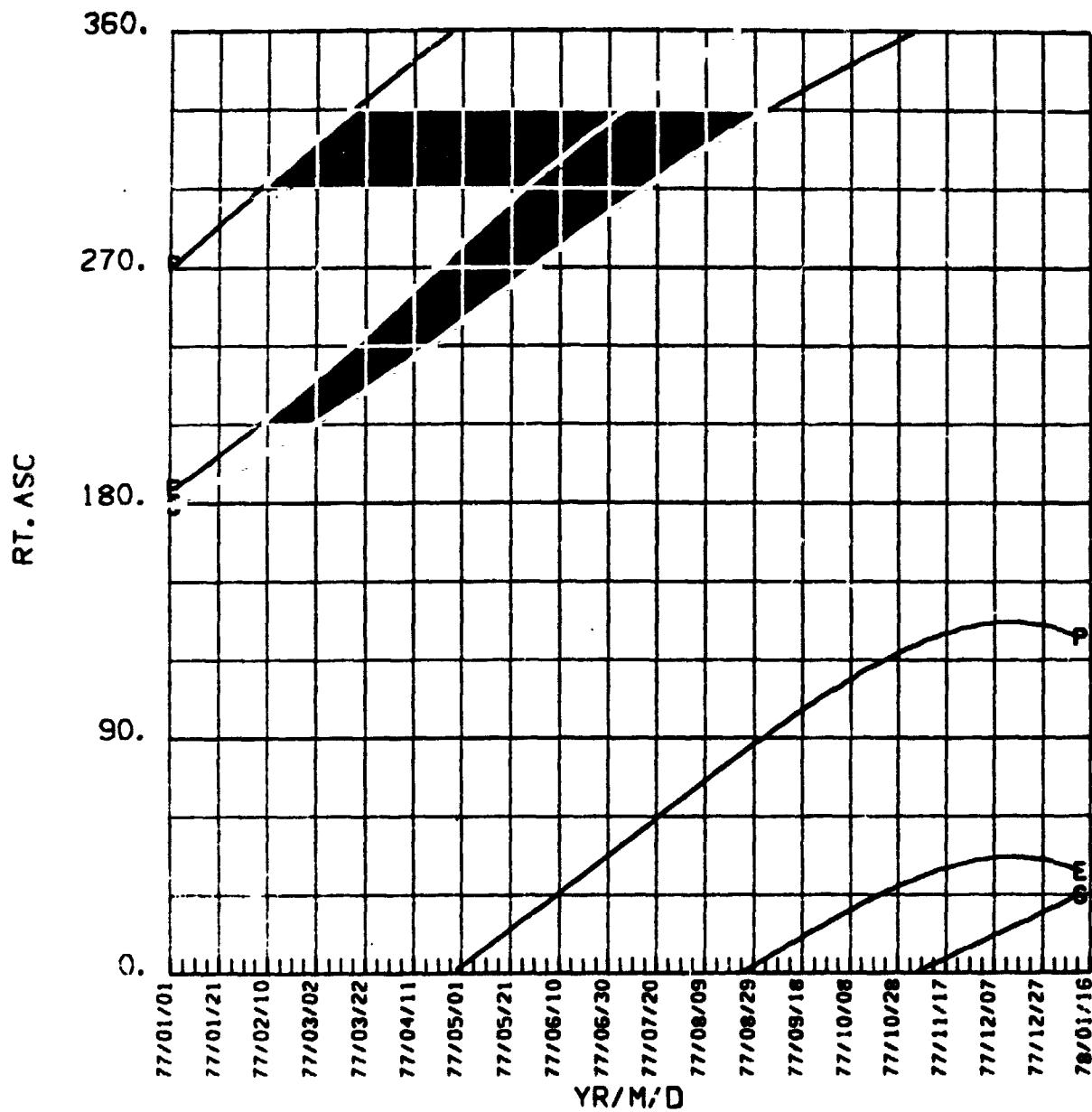
MARS 1976



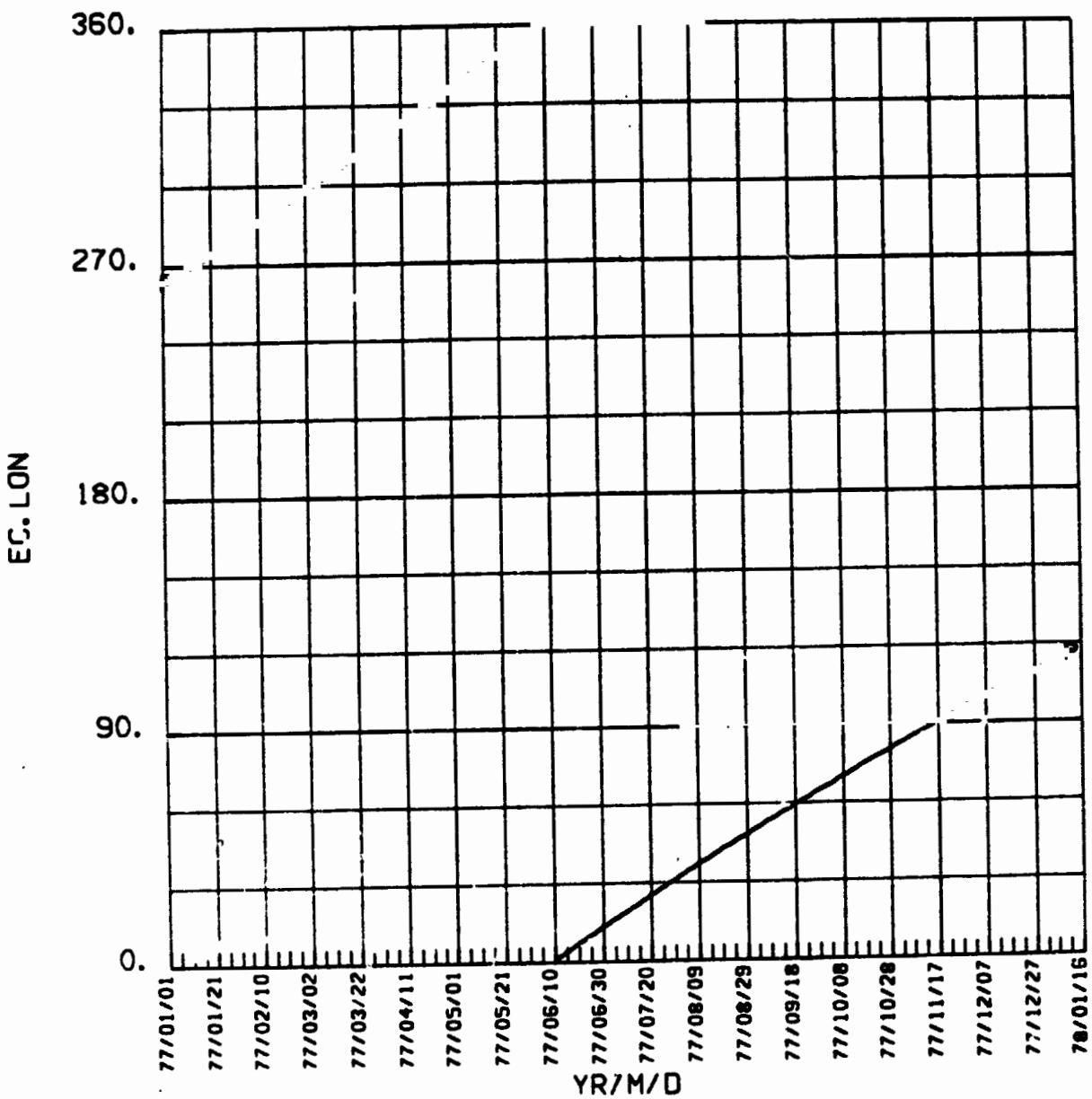
MARS 1977



MARS 1977

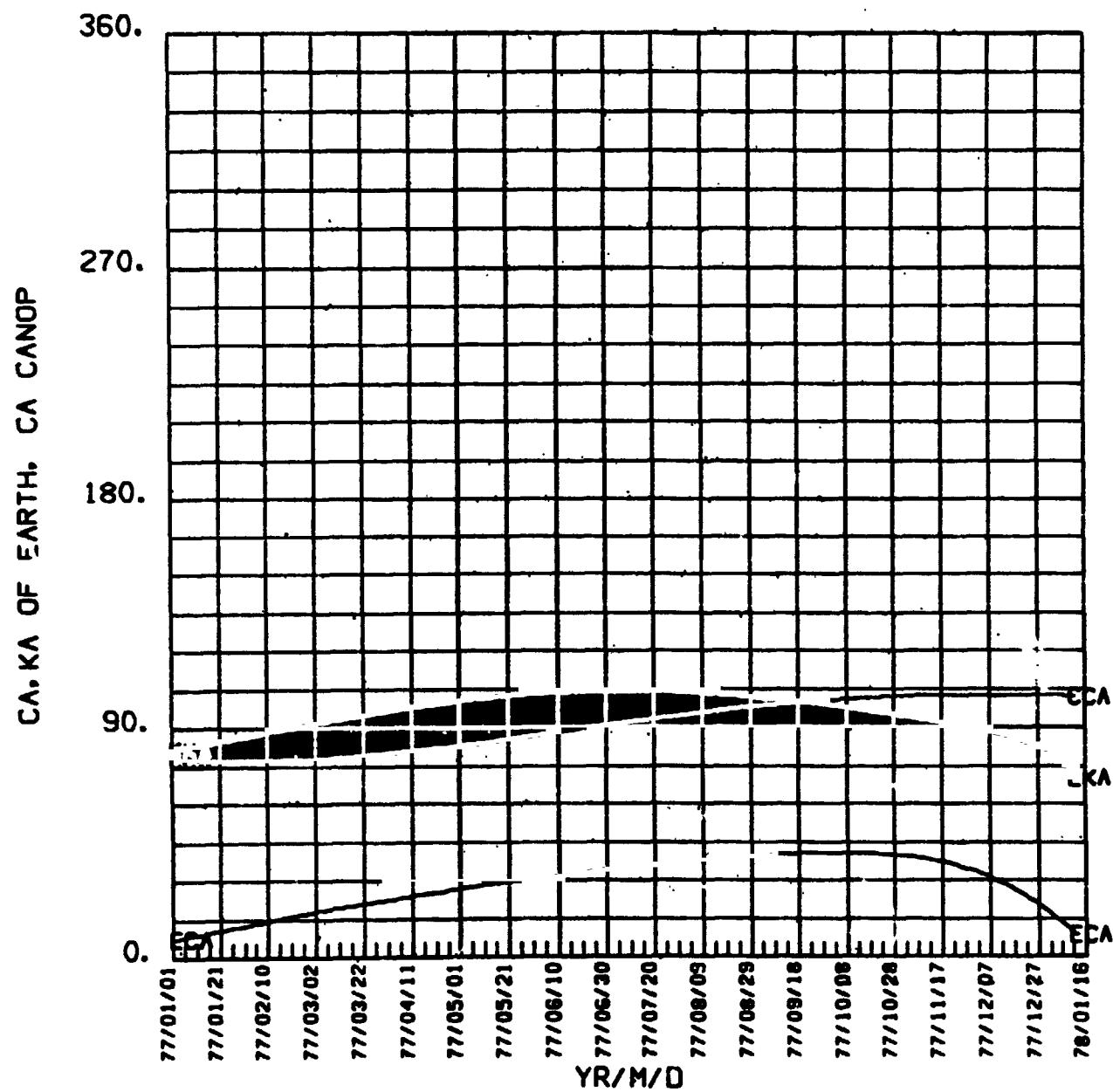


MARS 1977

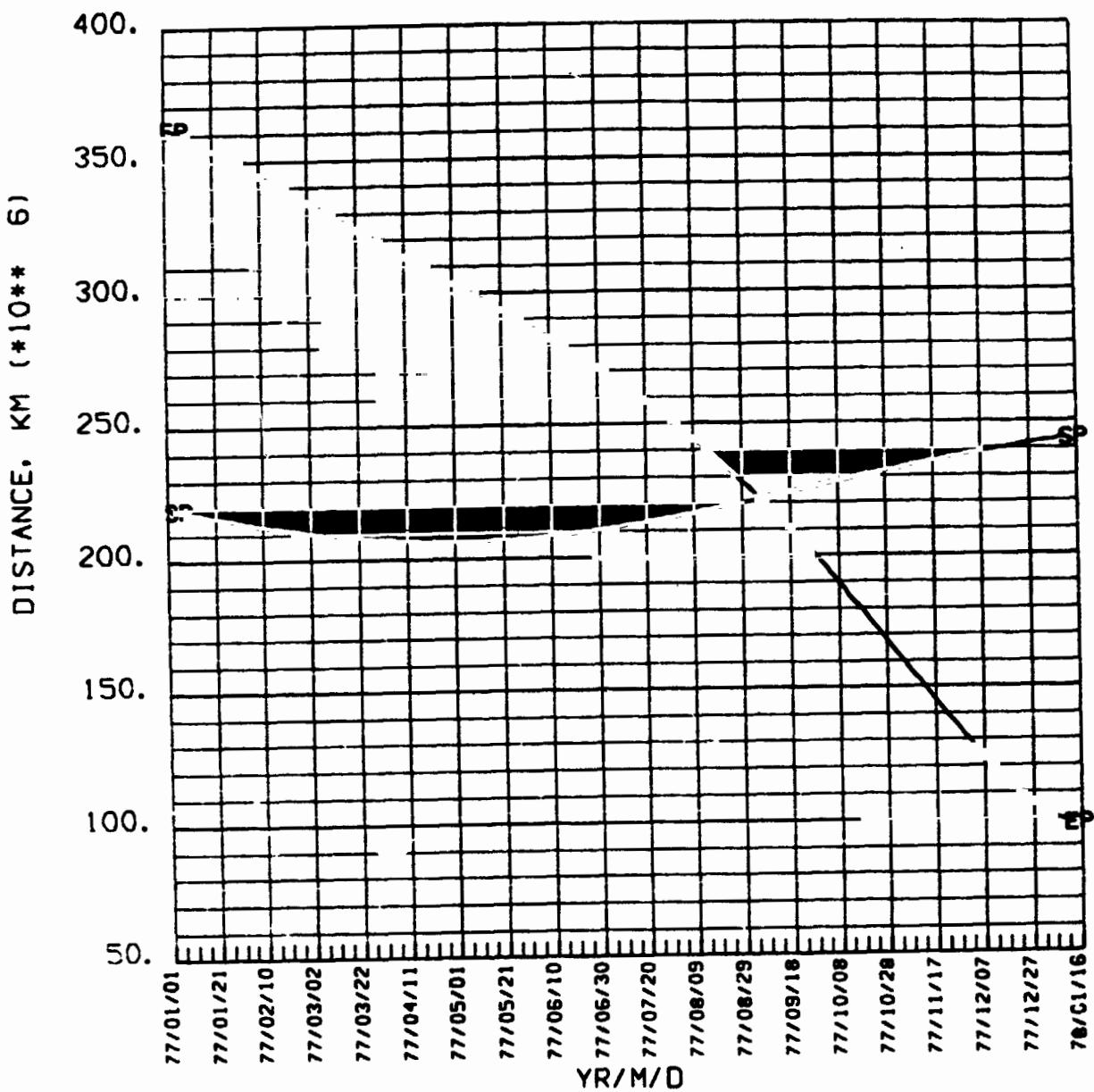


MARS

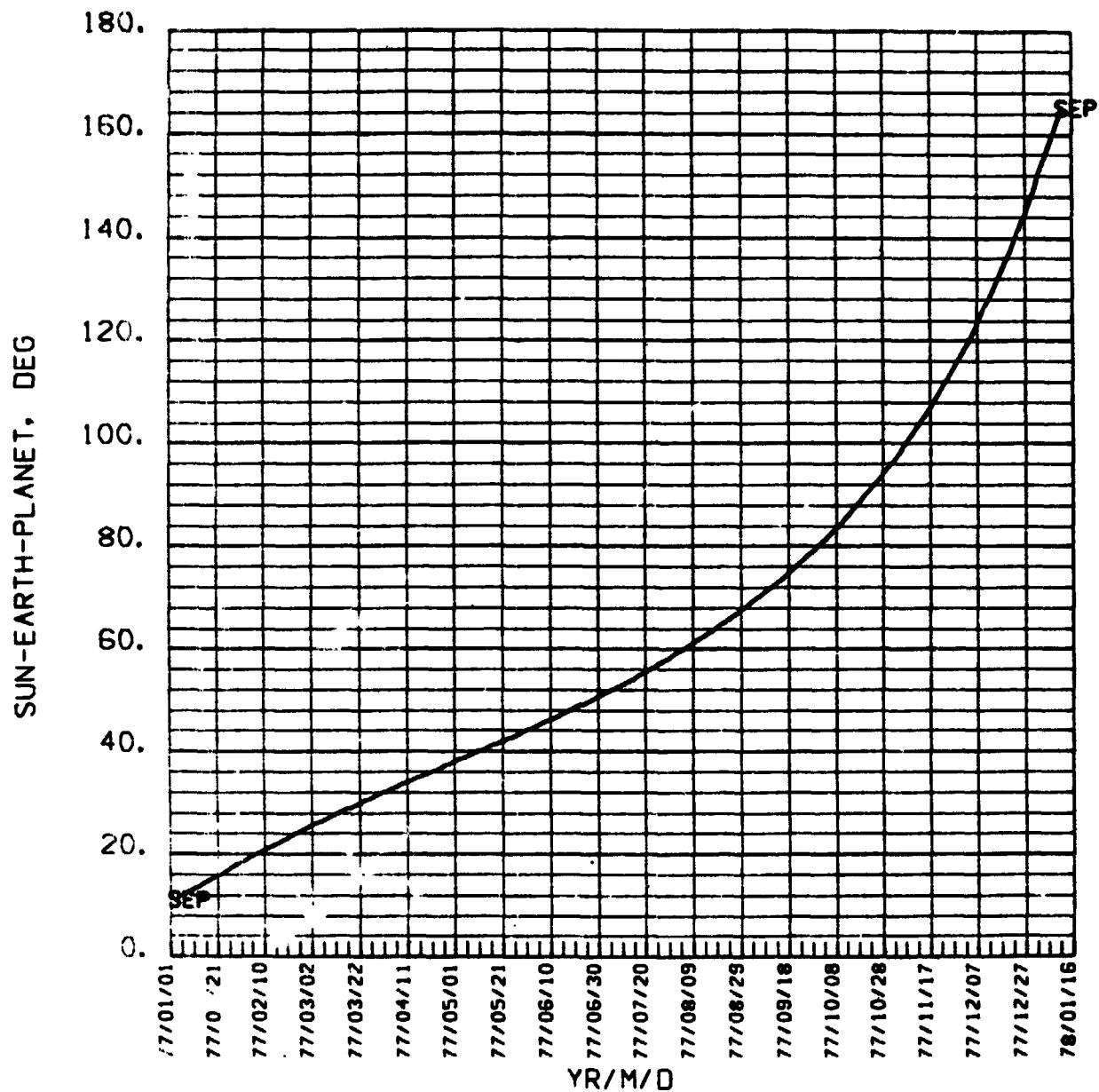
1977



MARS 1977



MARS 1977



MARS 1977

STATION RISE/SET GMT. HR

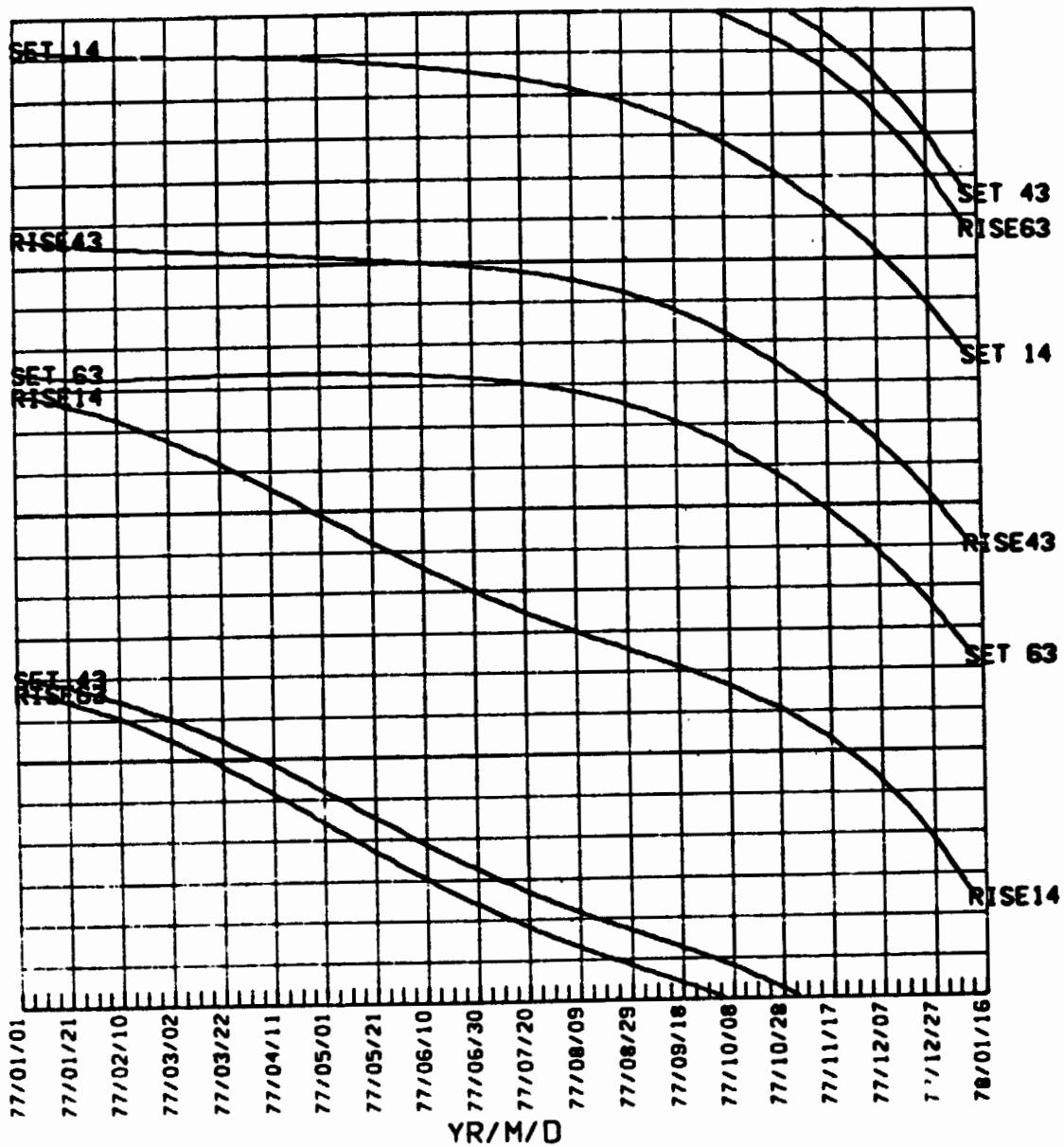
24.

18.

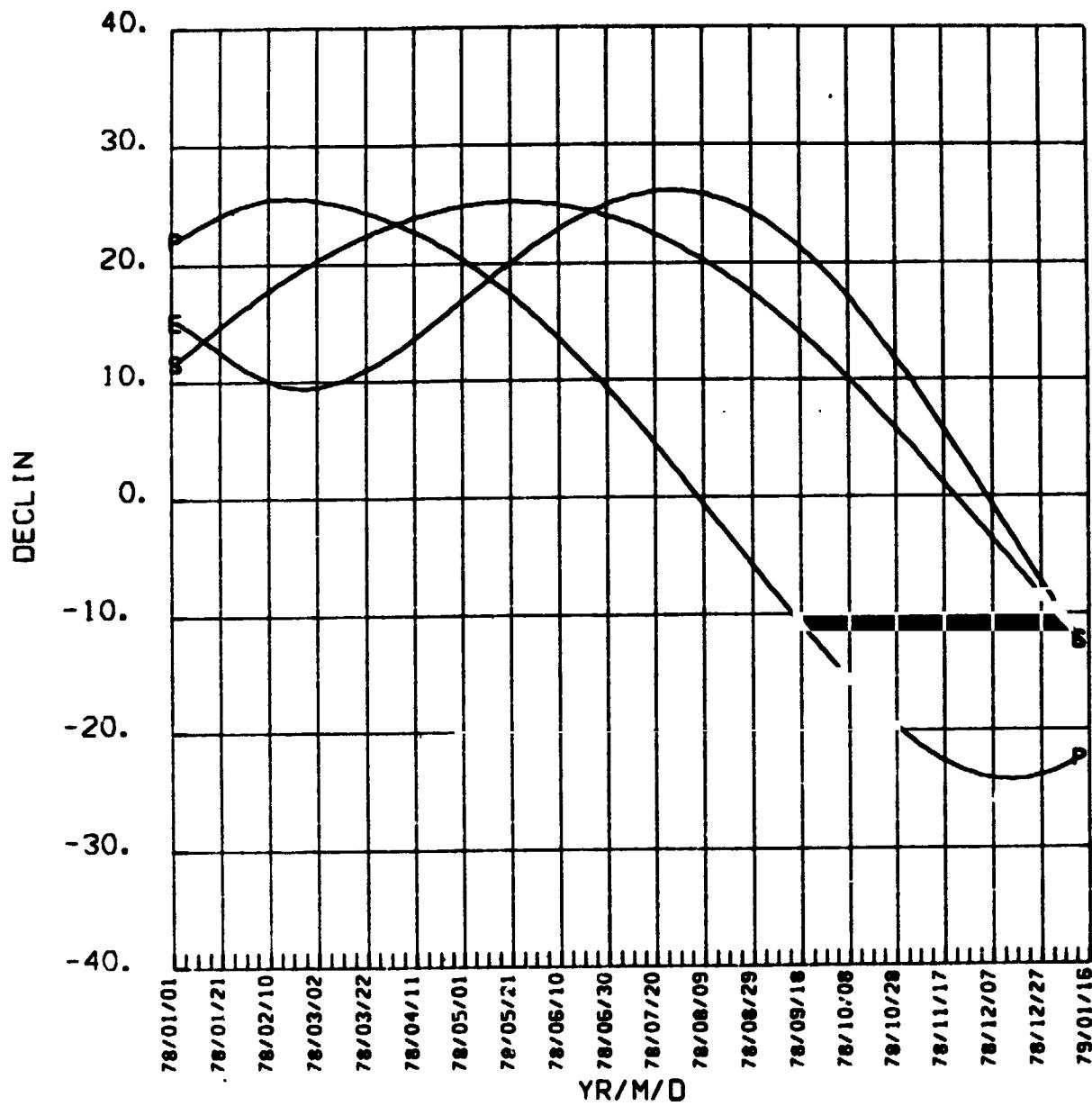
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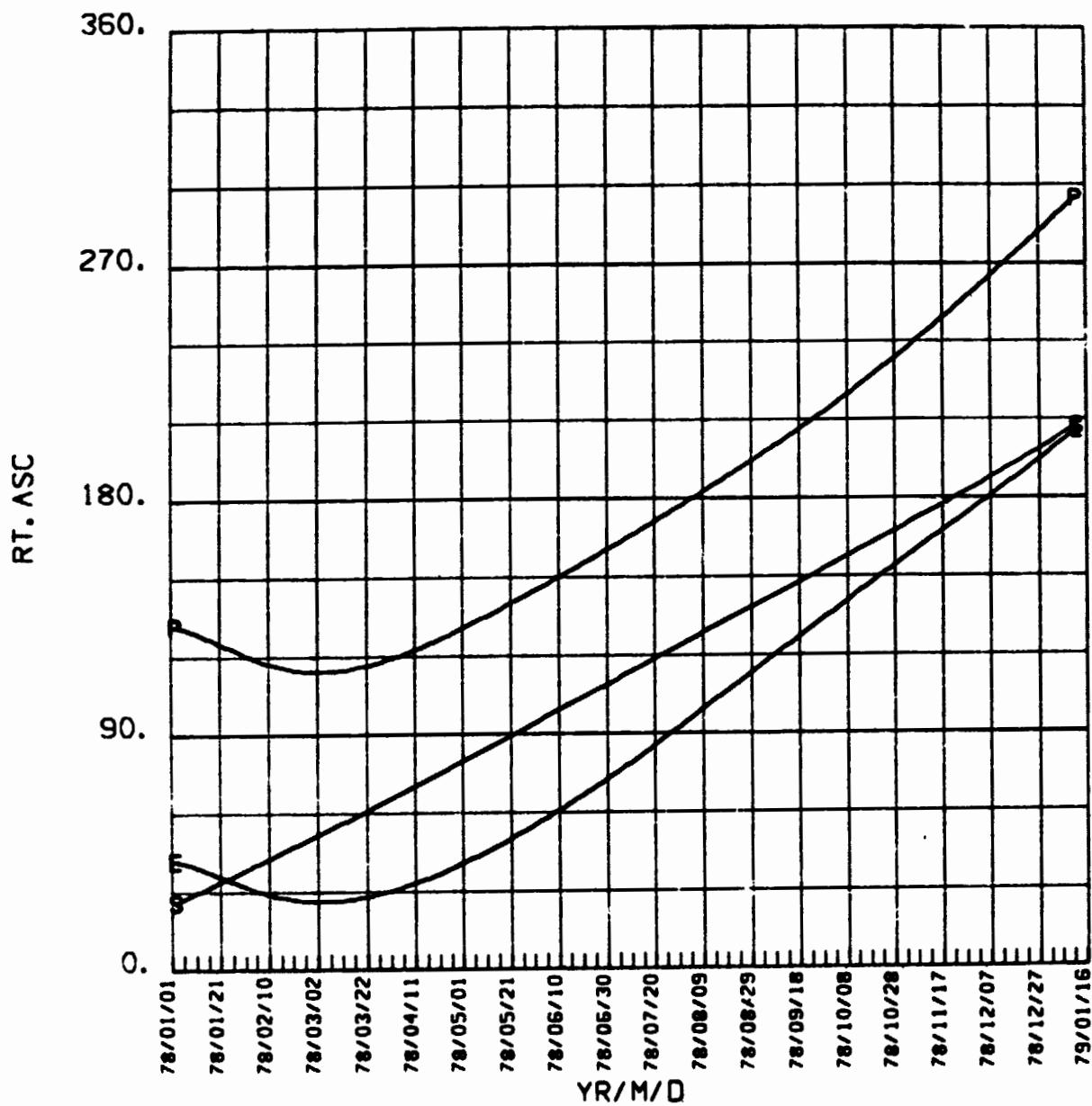
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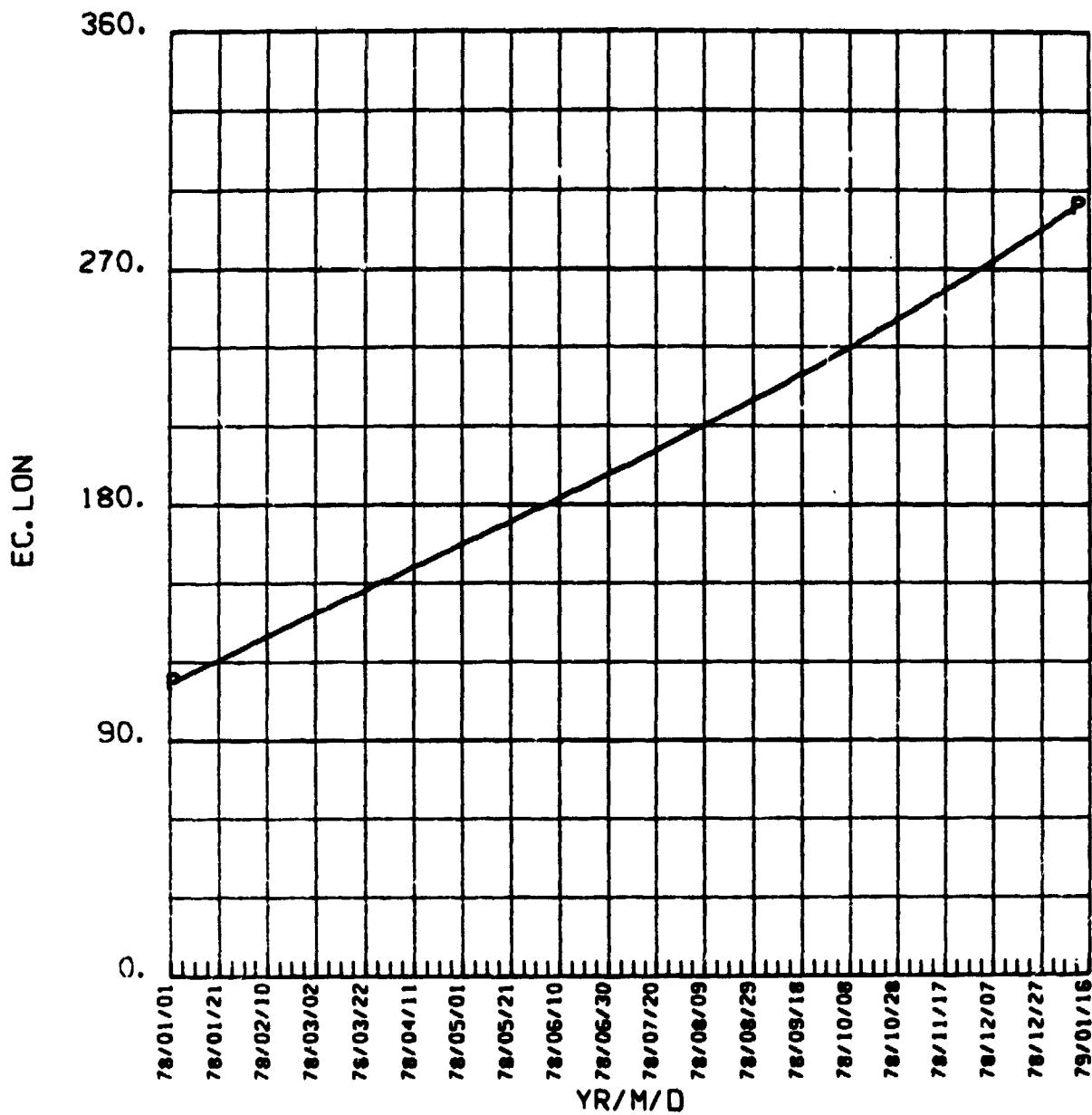
MARS 1978



MARS 1978

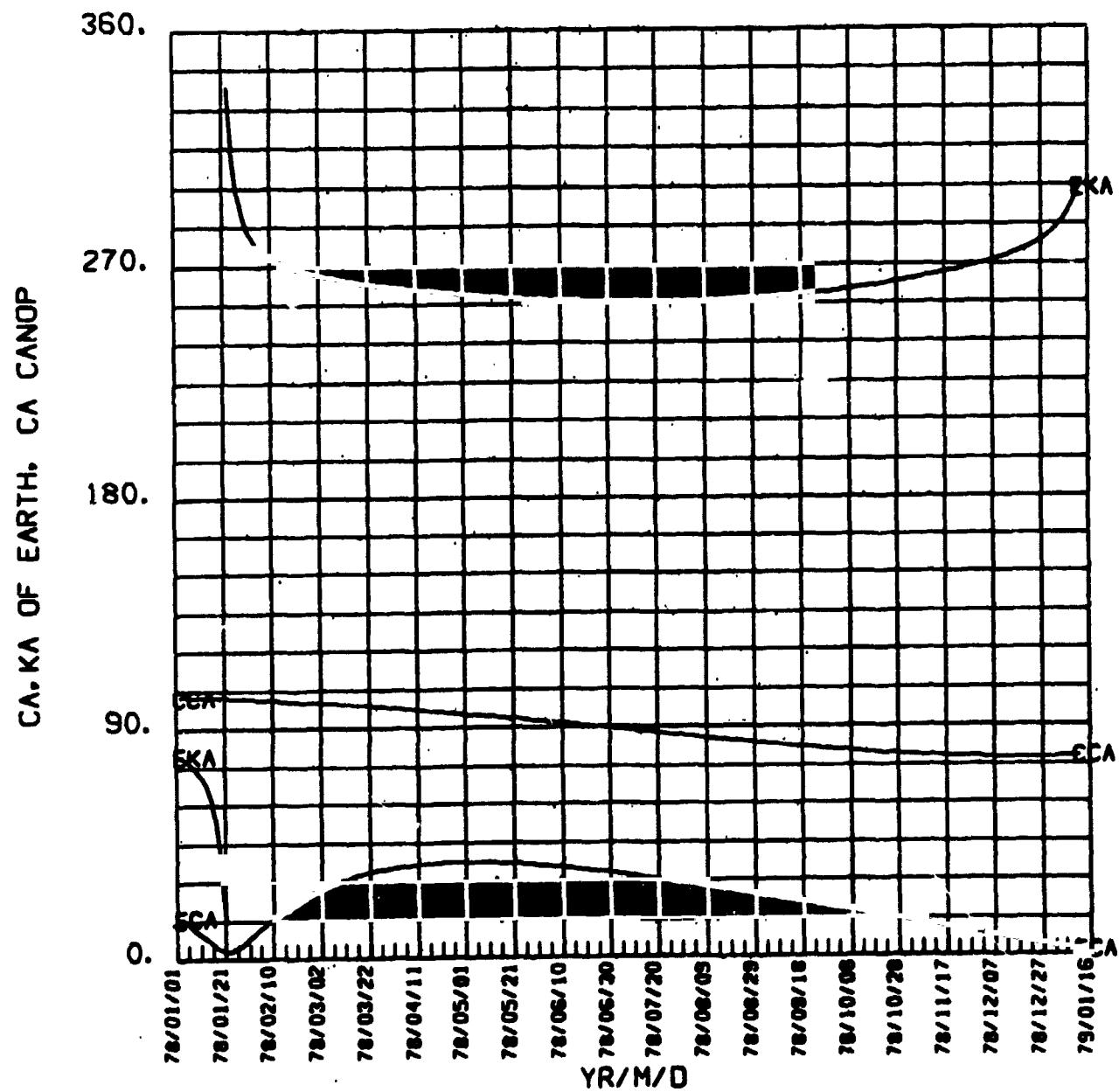


MARS 1978

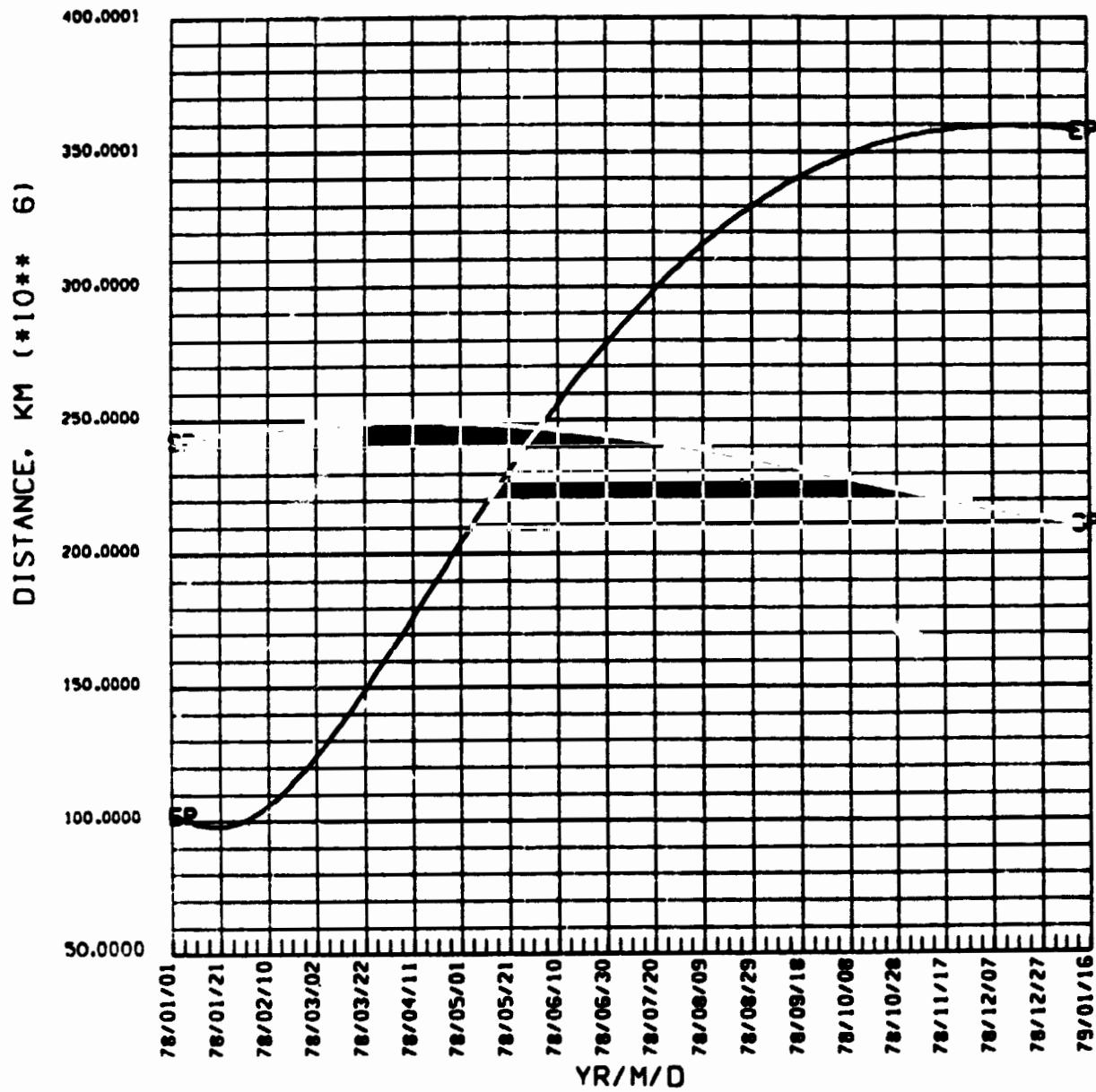


MARS

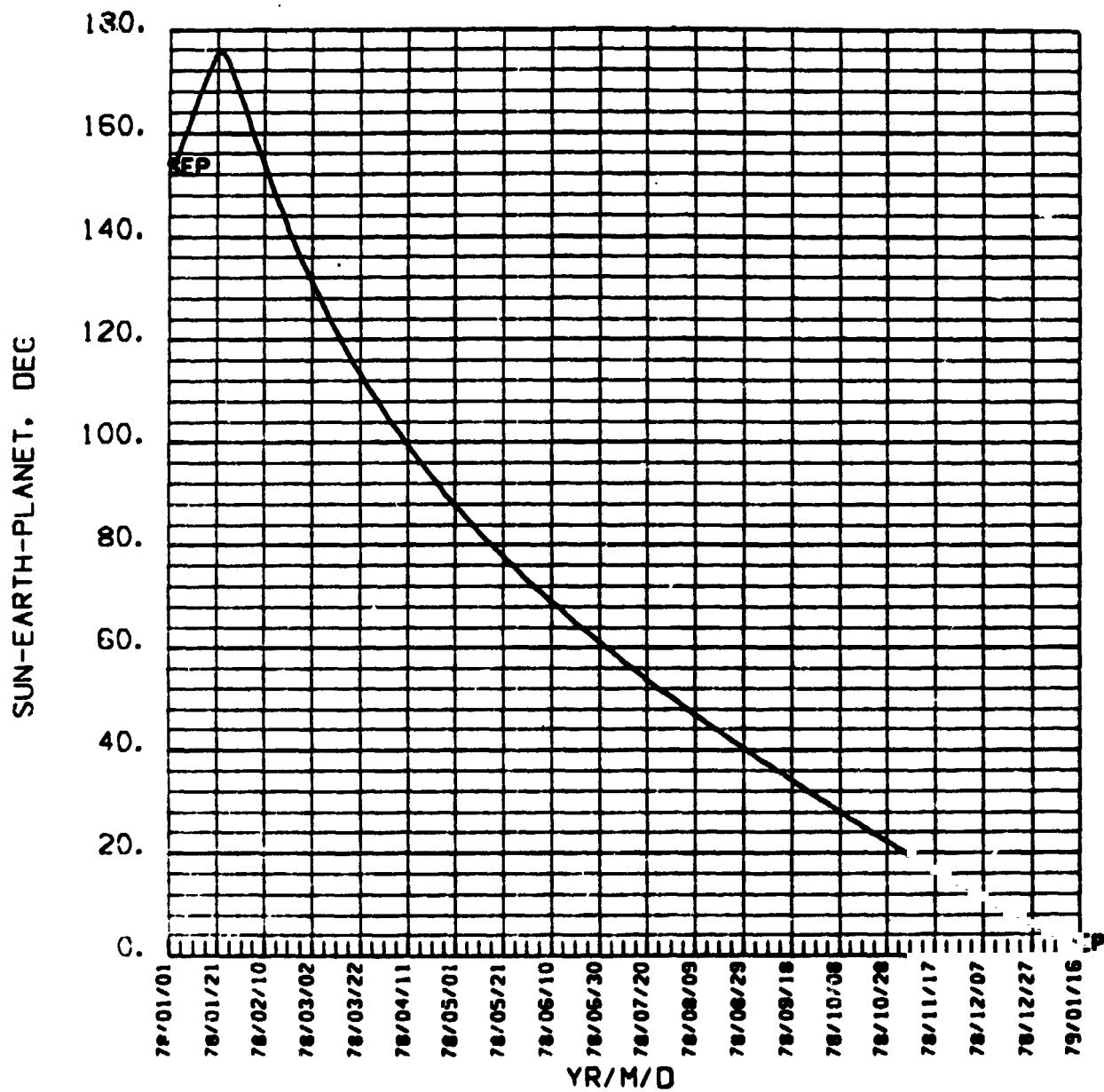
1978



MARS 1978

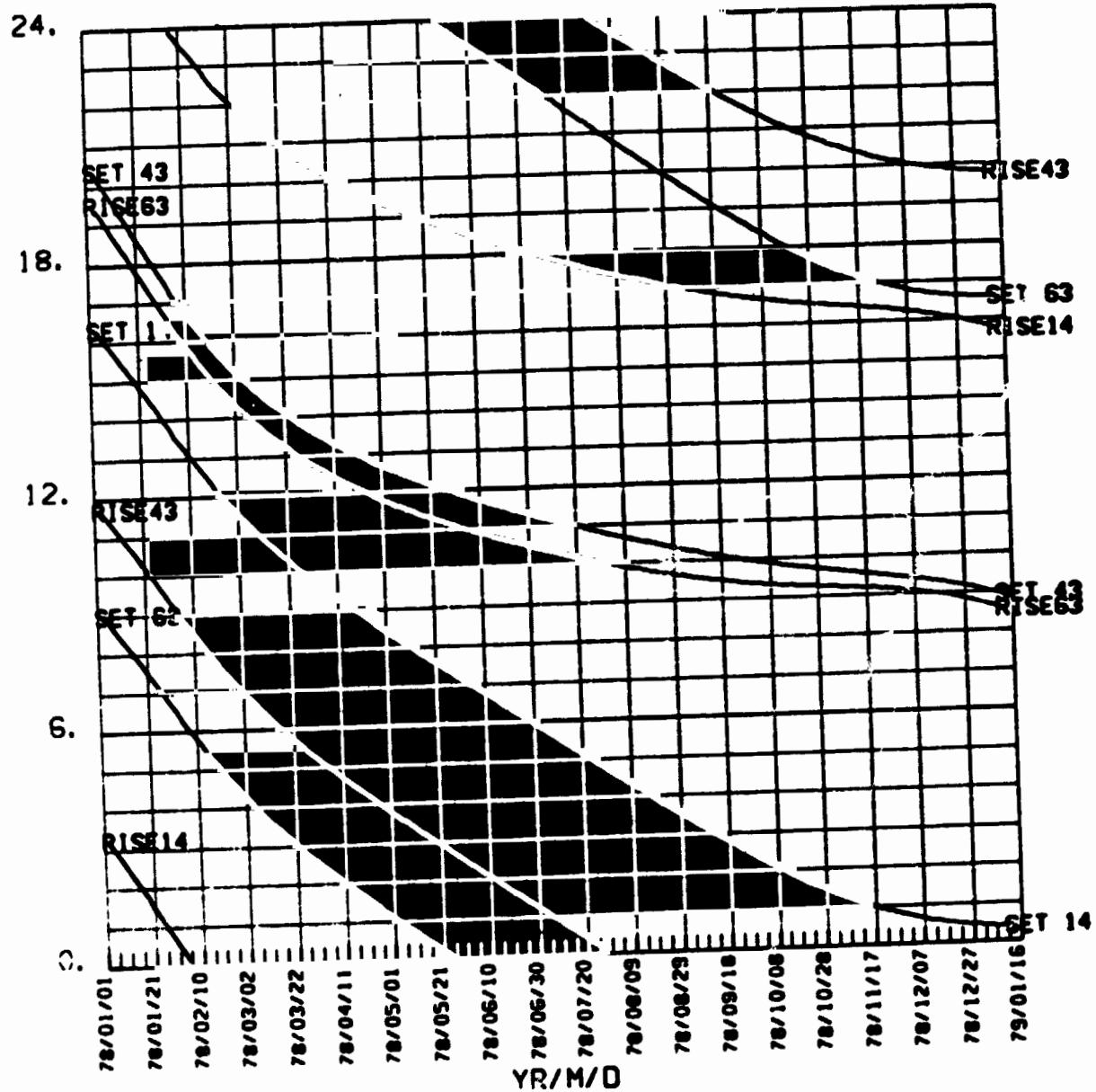


MARS 1978

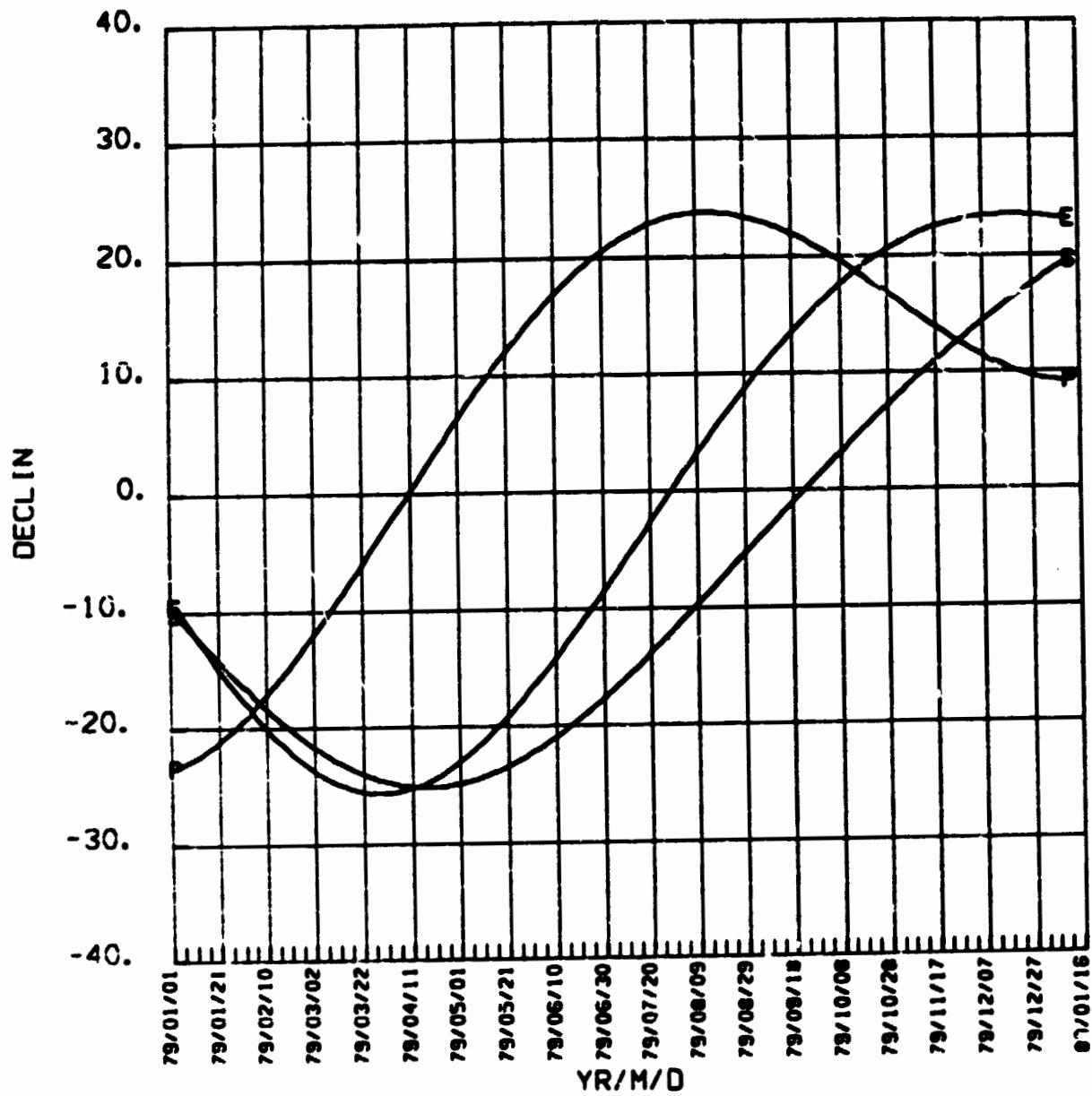


STATION RISE/SET GMT. HR

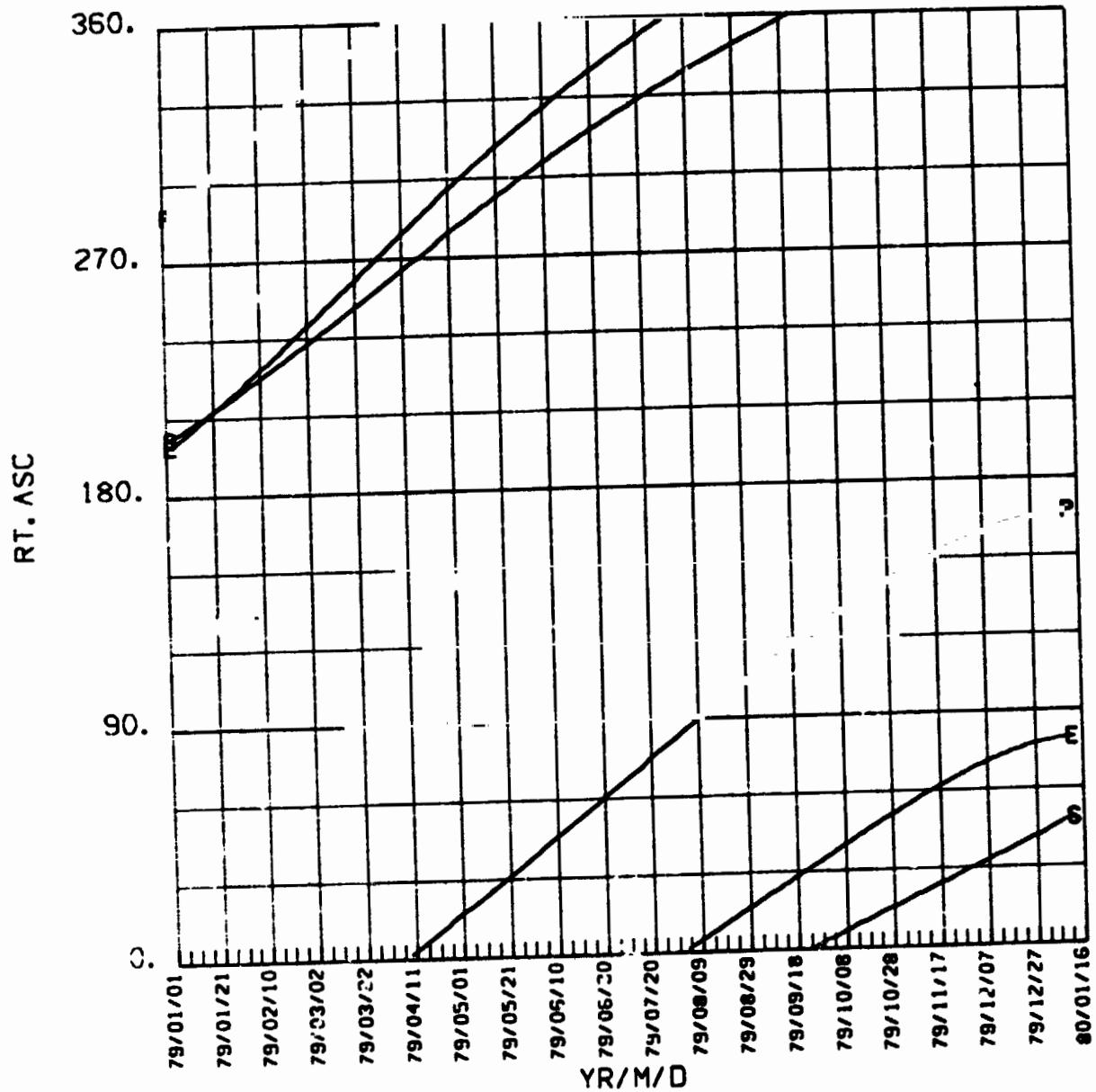
MARS 1978



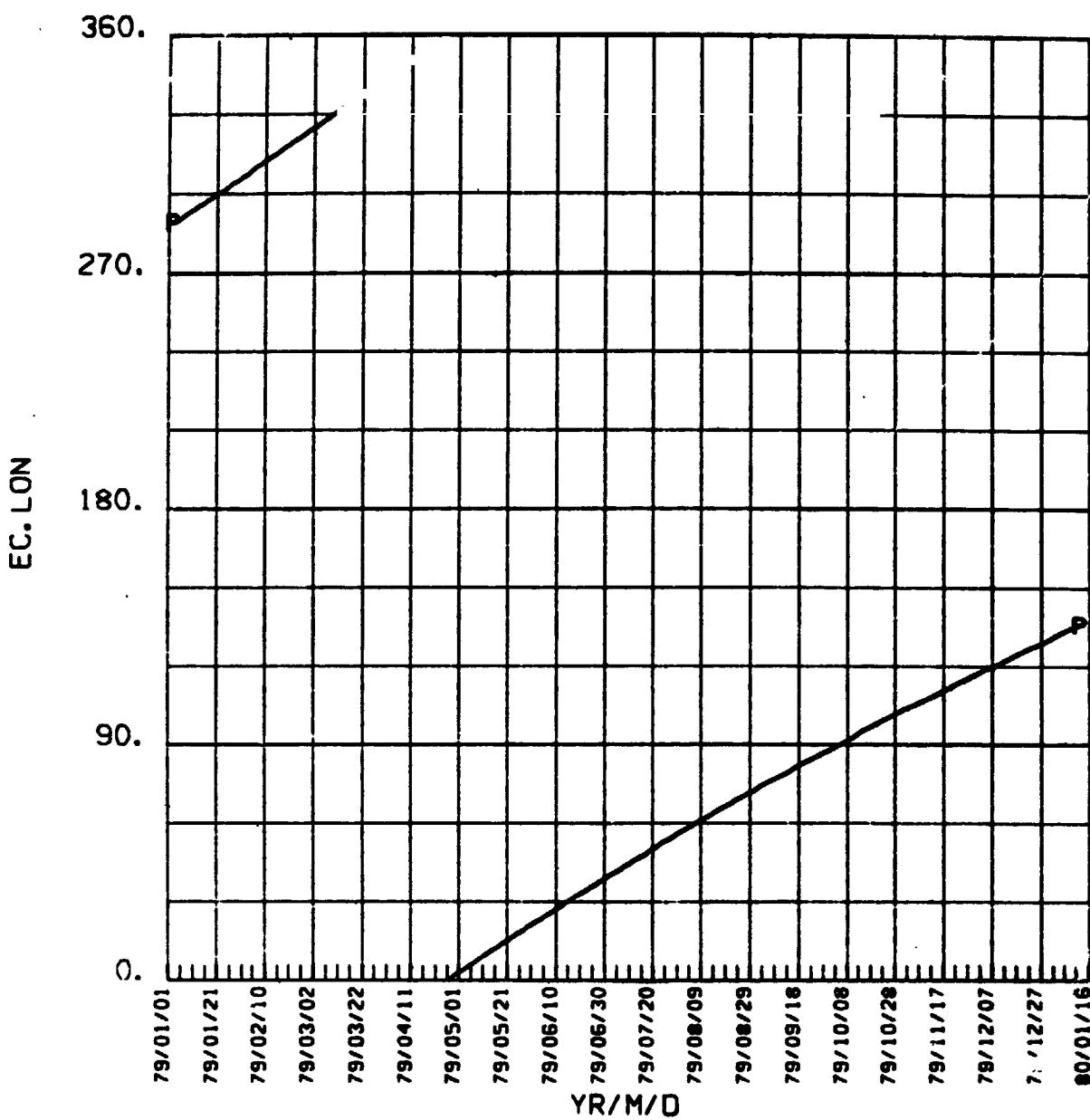
MARS 1979



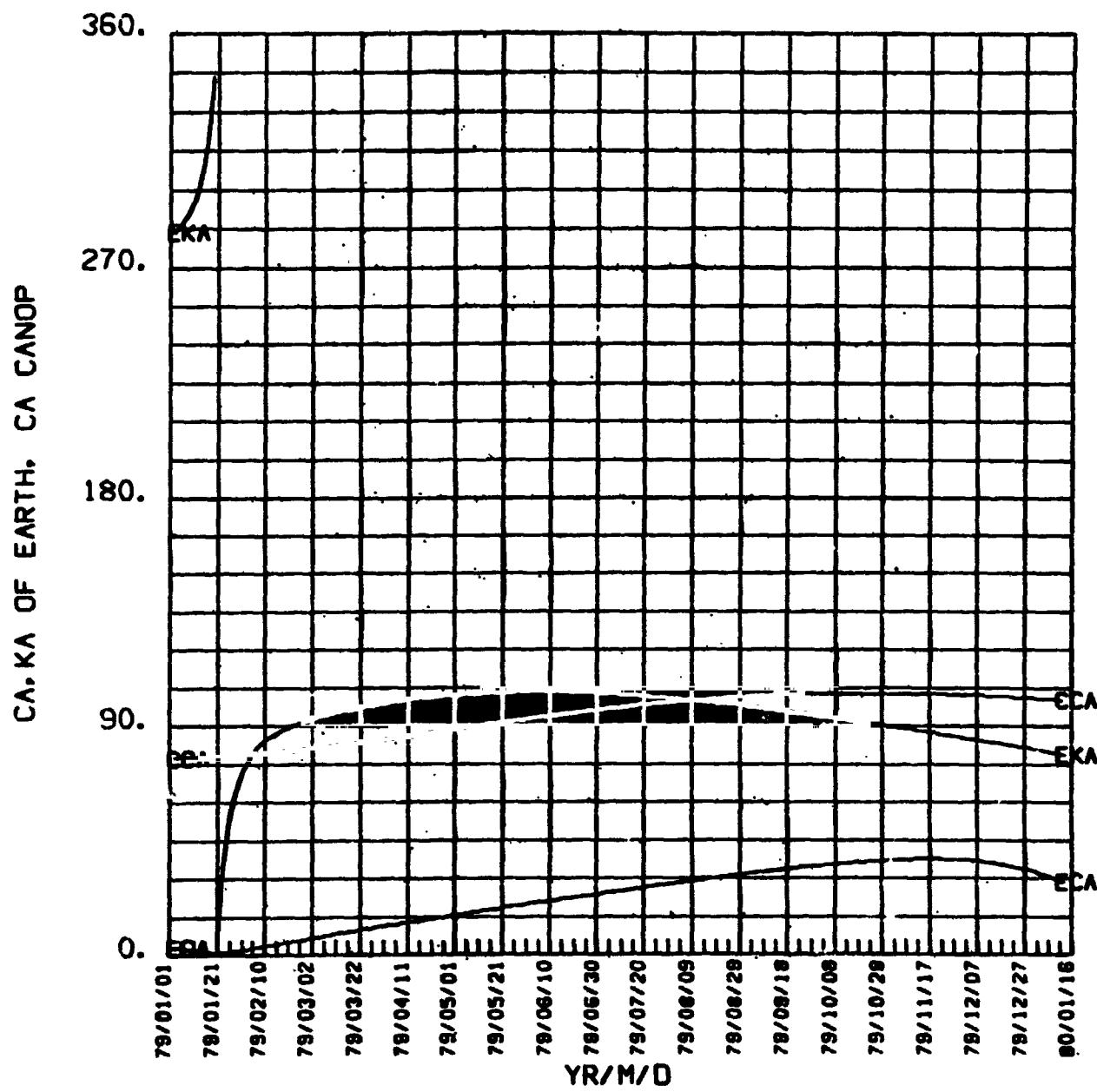
MARS 1979



MARS 1979

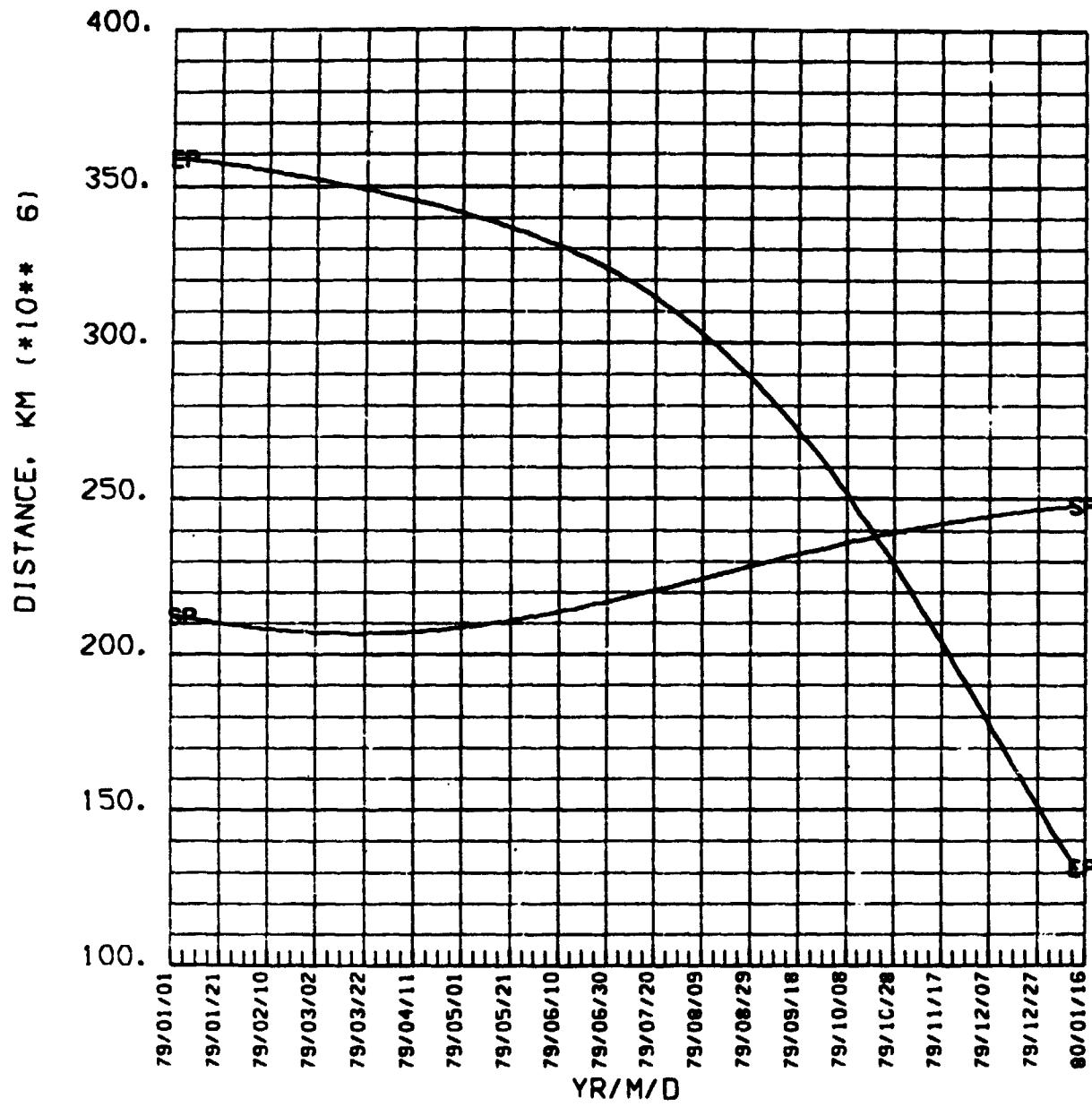


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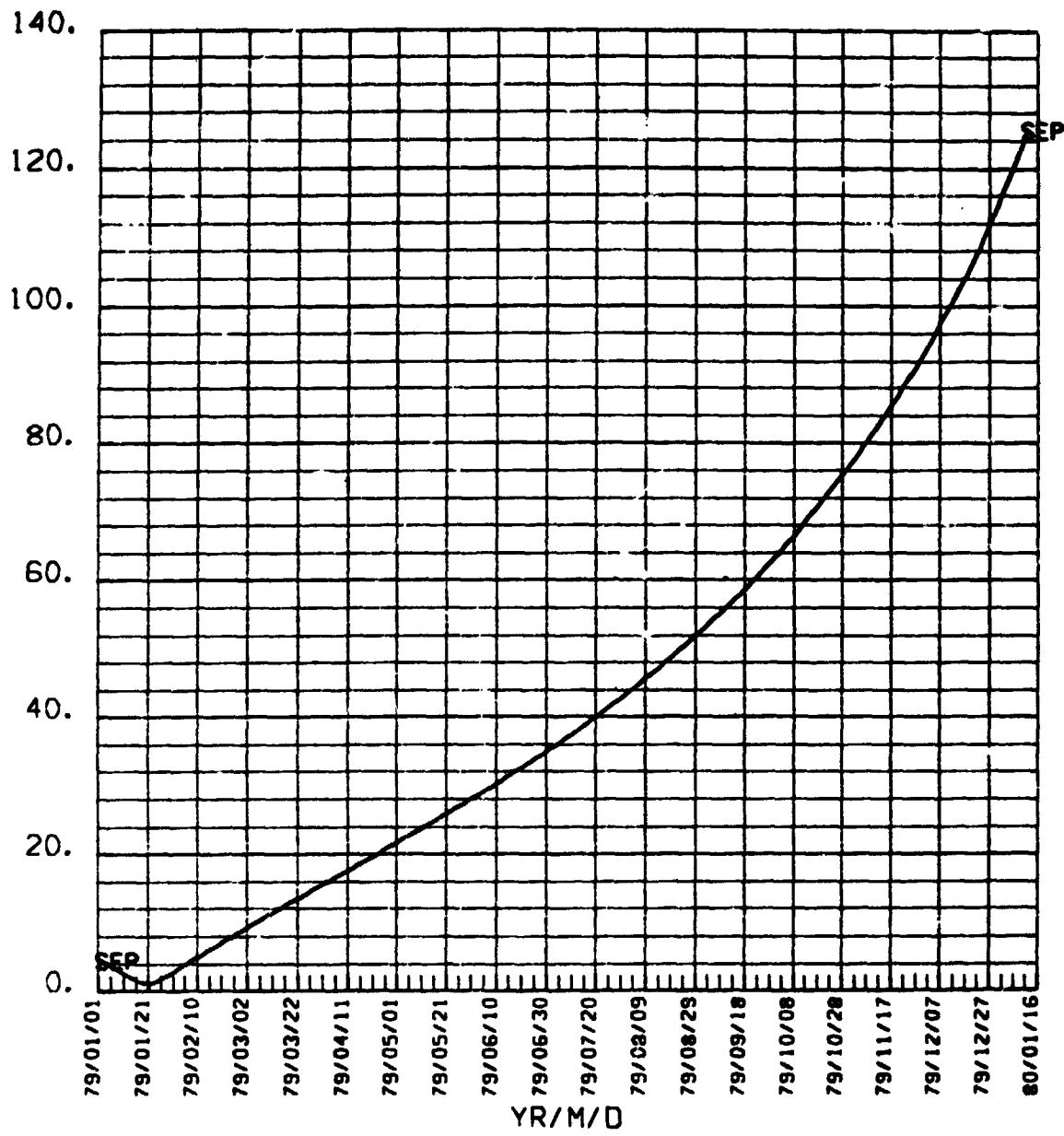
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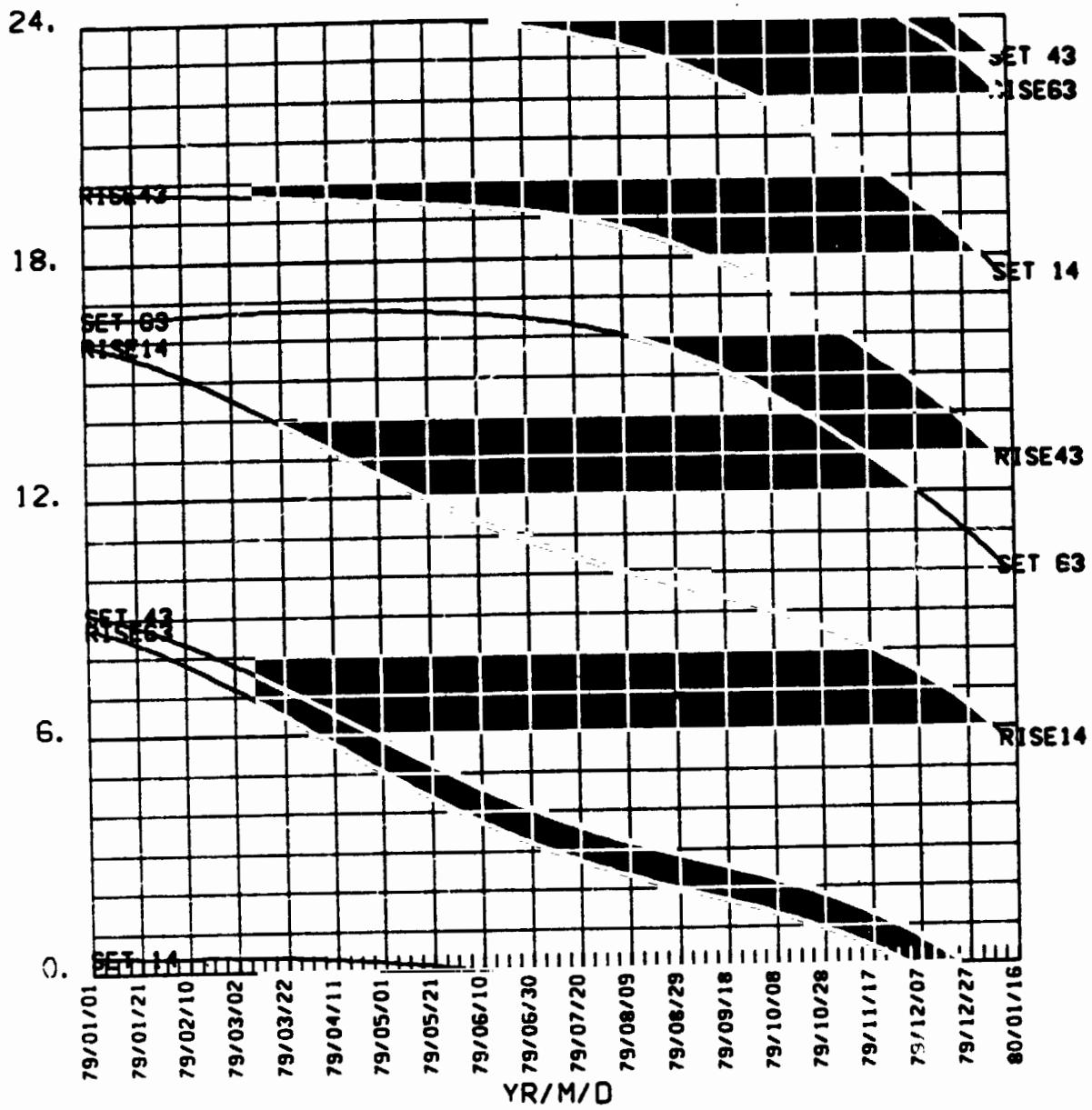
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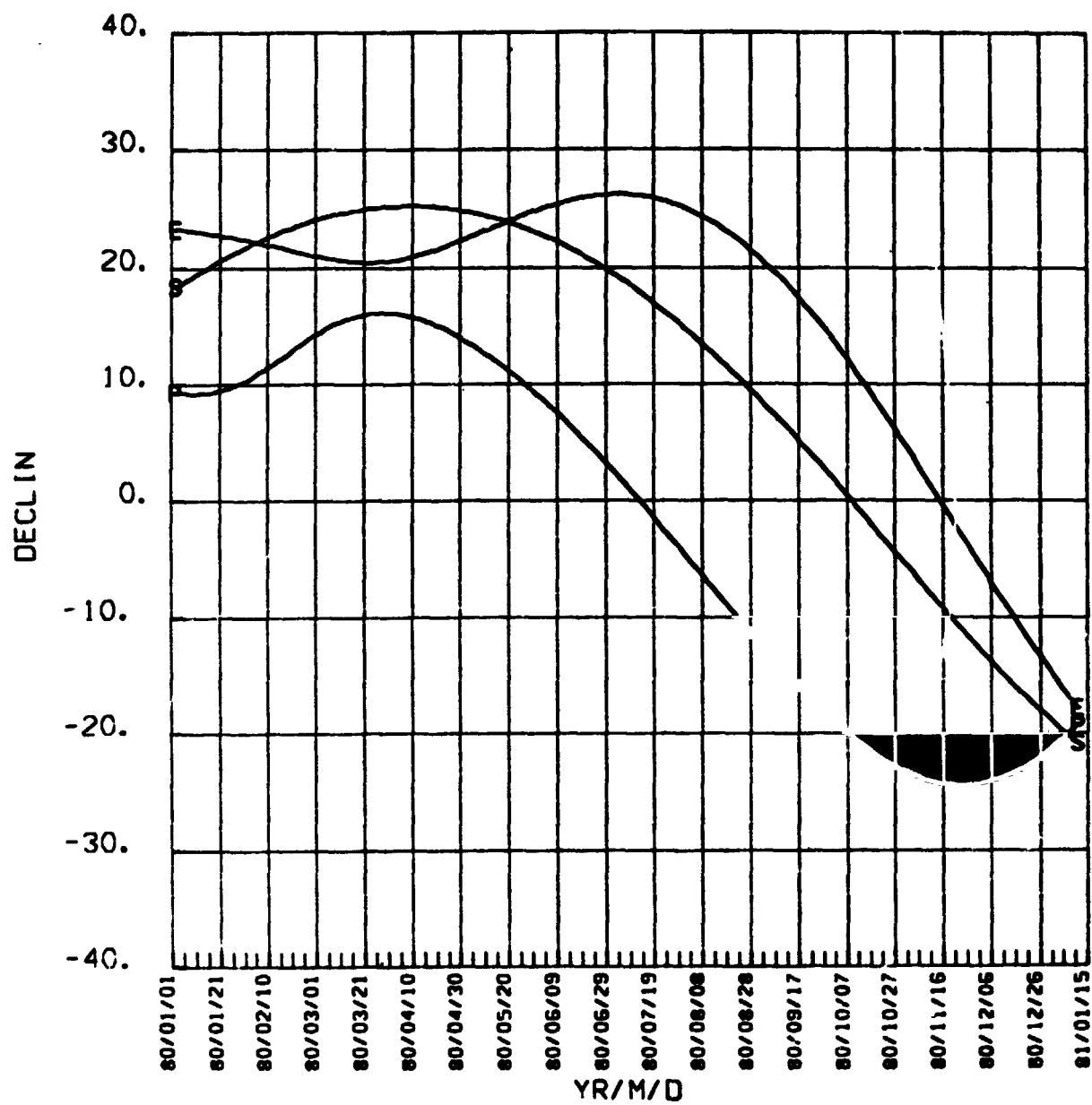


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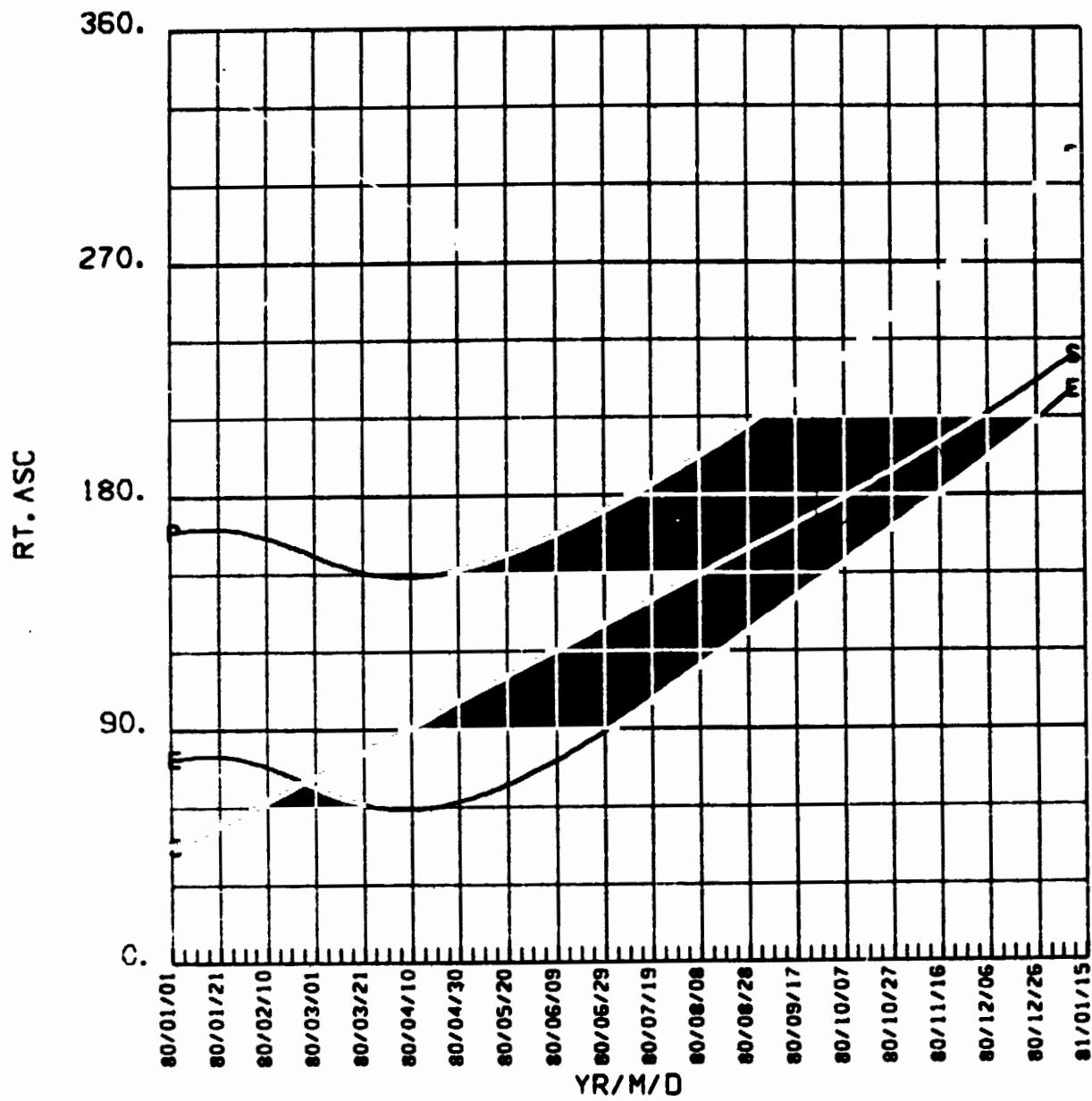
STATION RISE/SET GMT. HR



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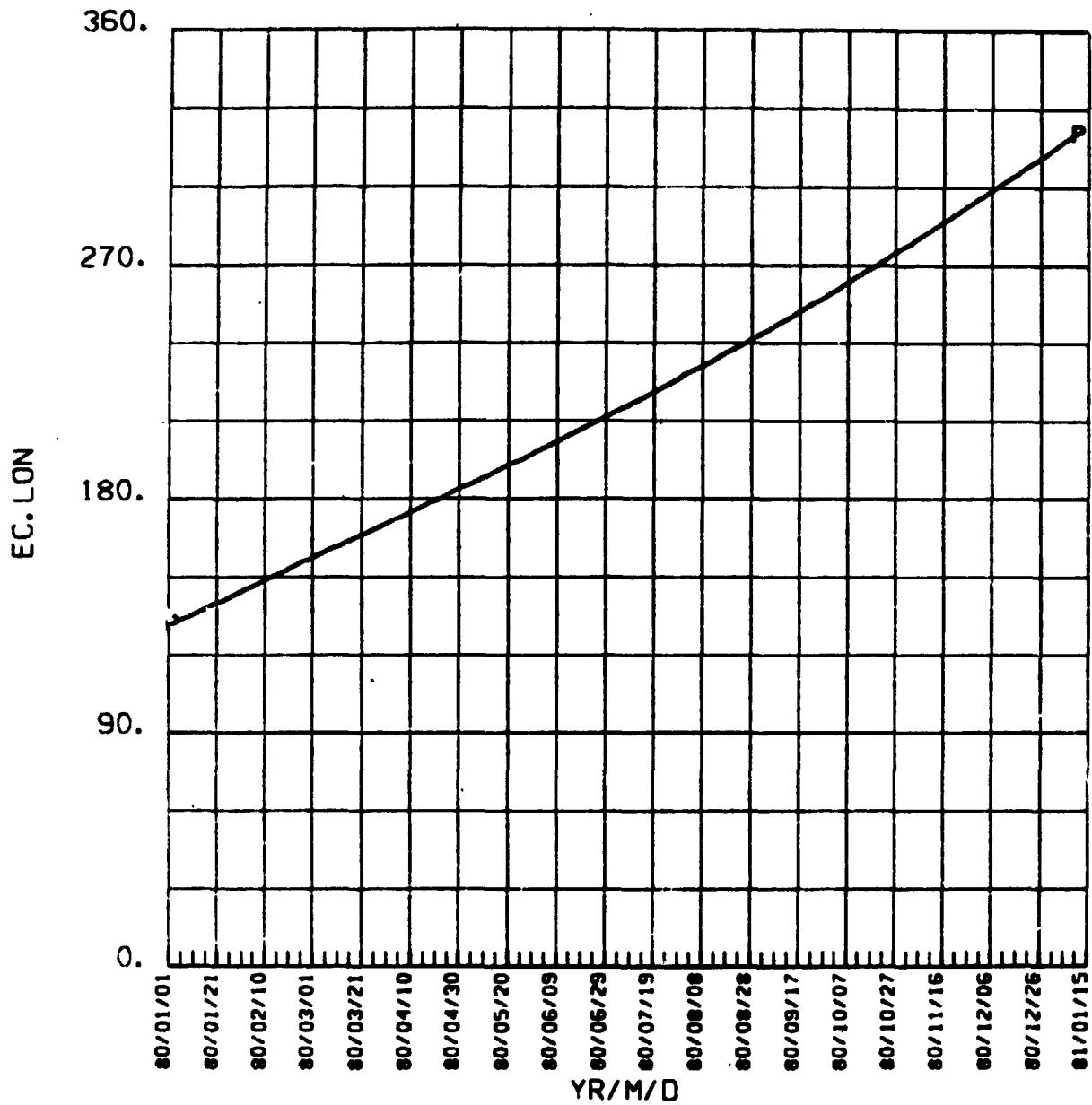


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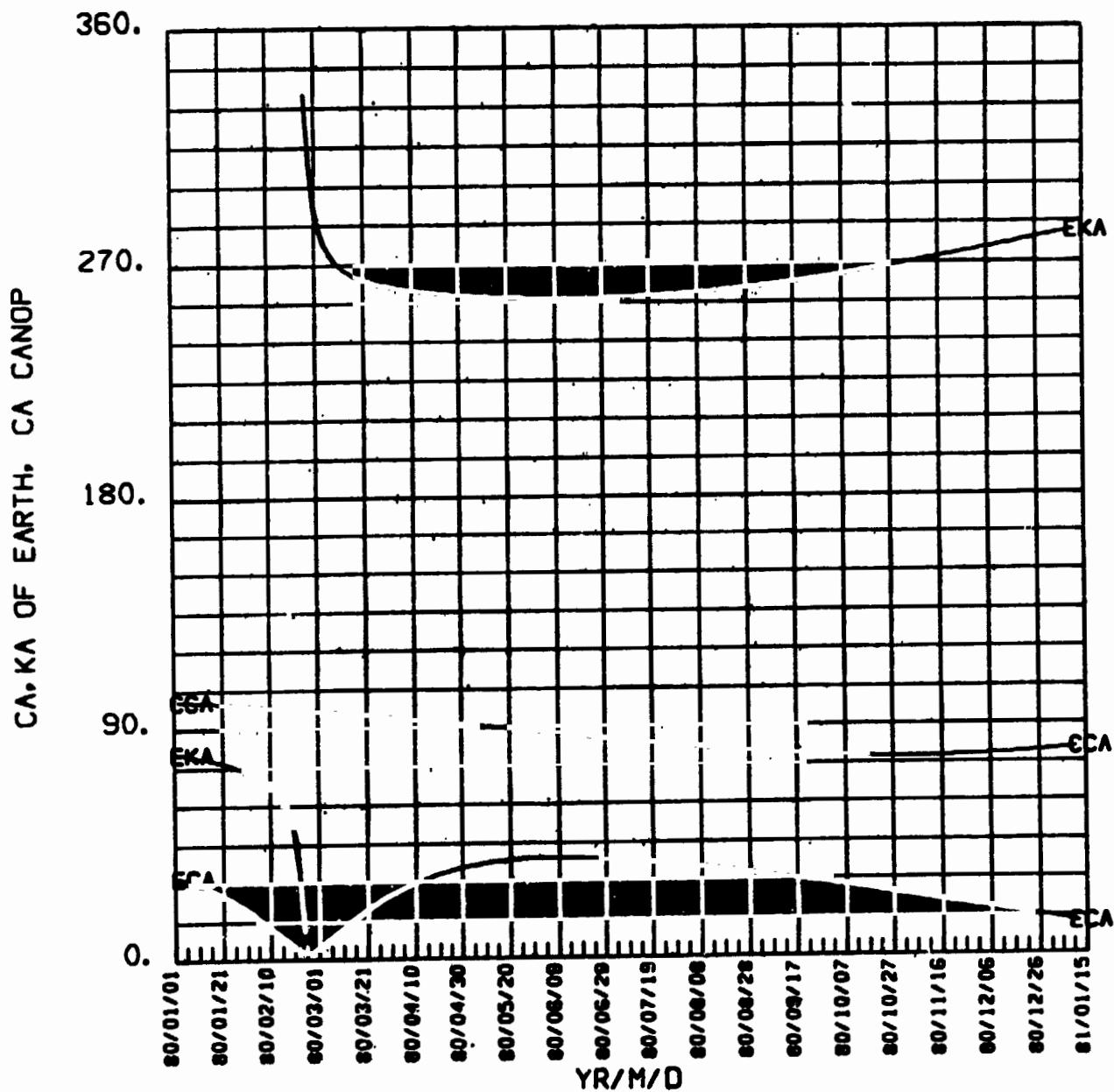
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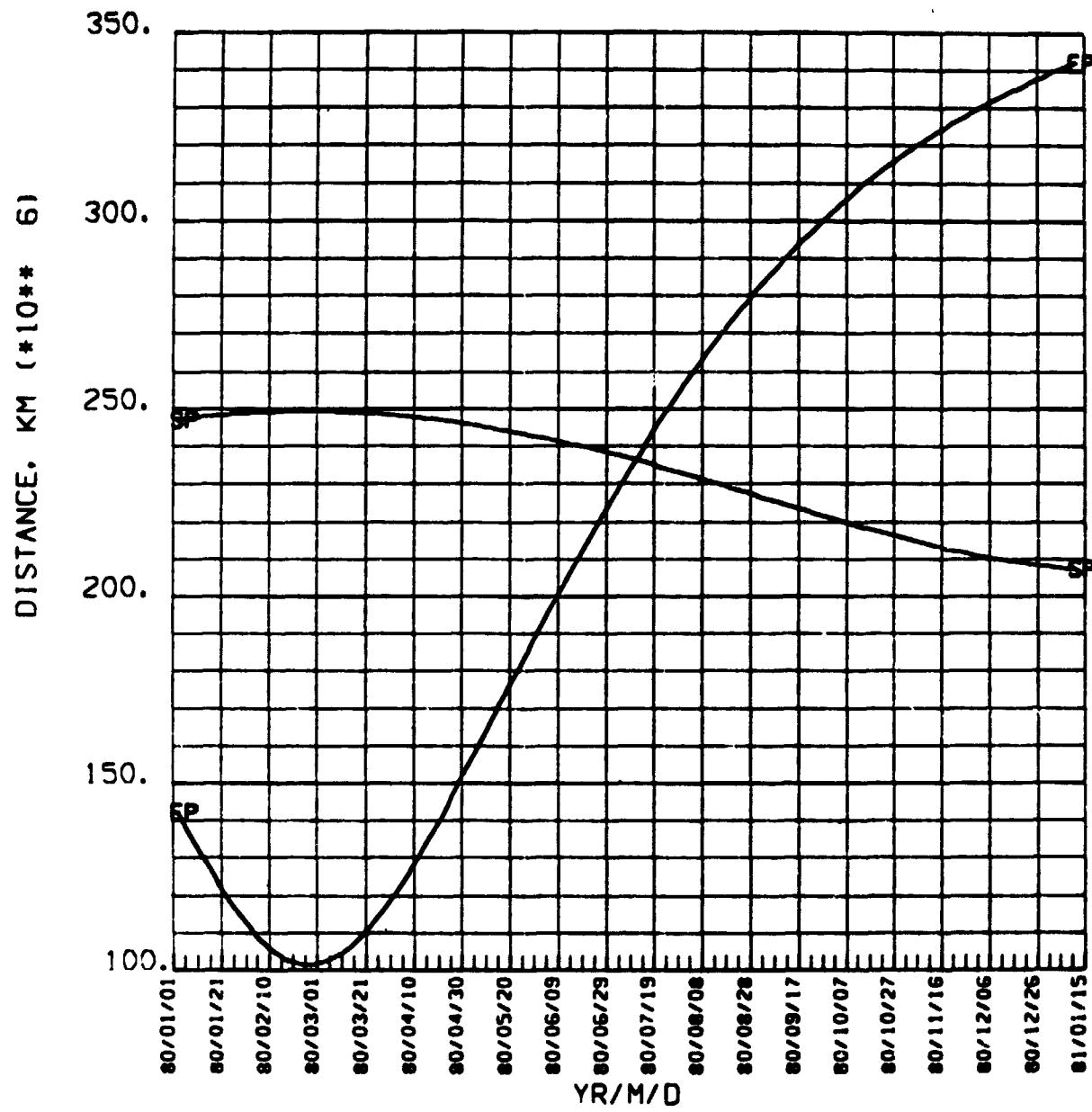


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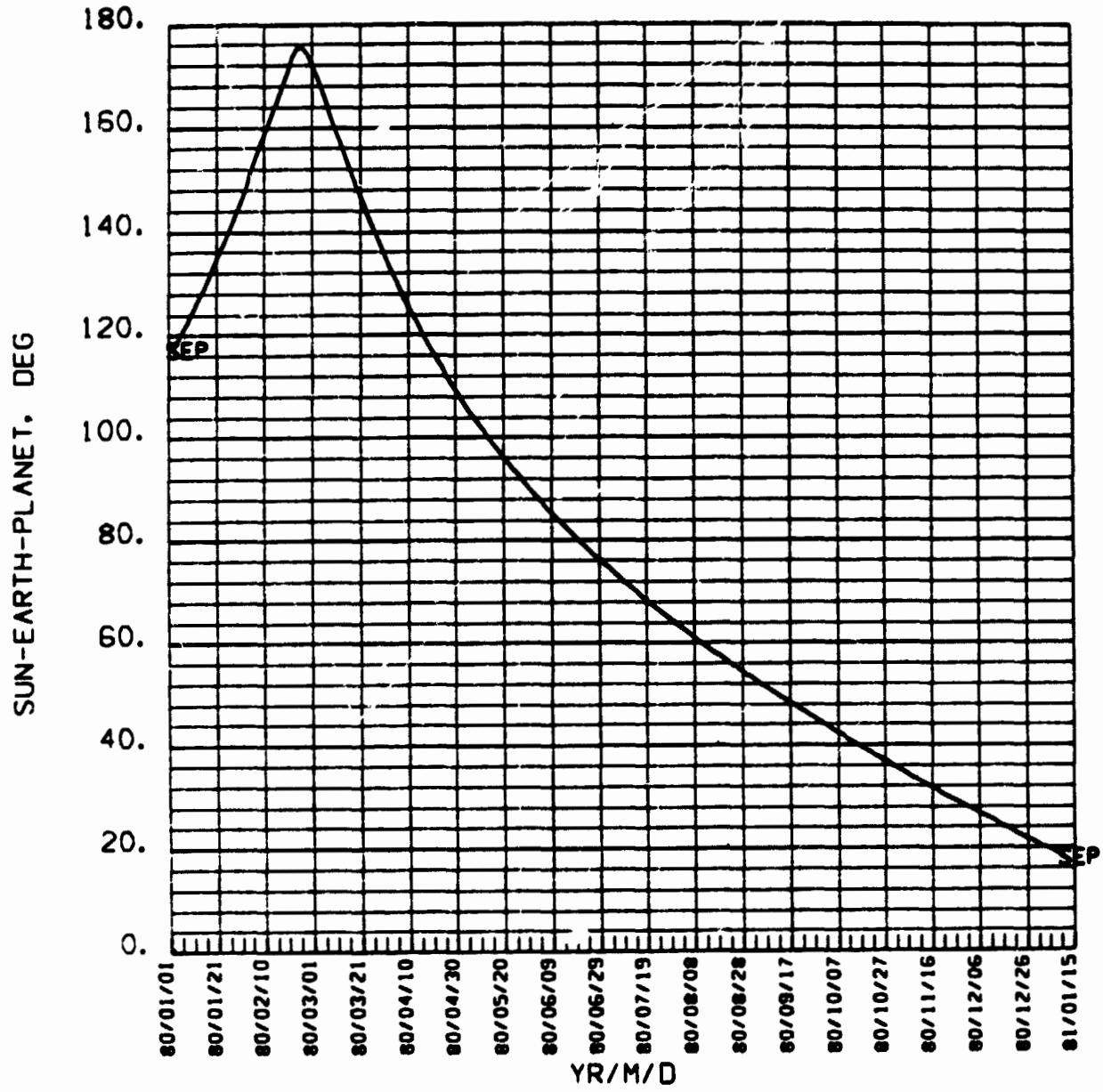
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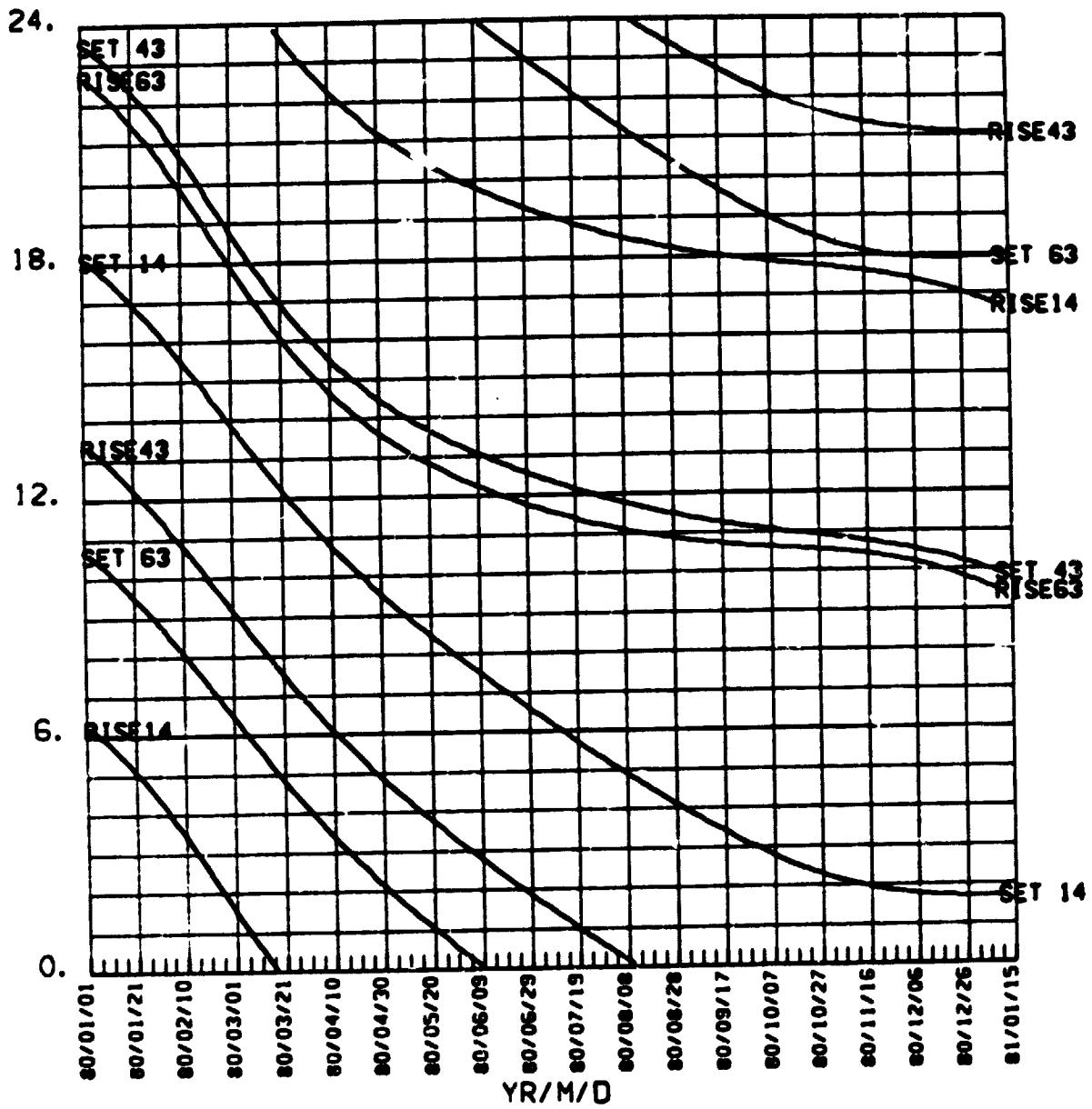


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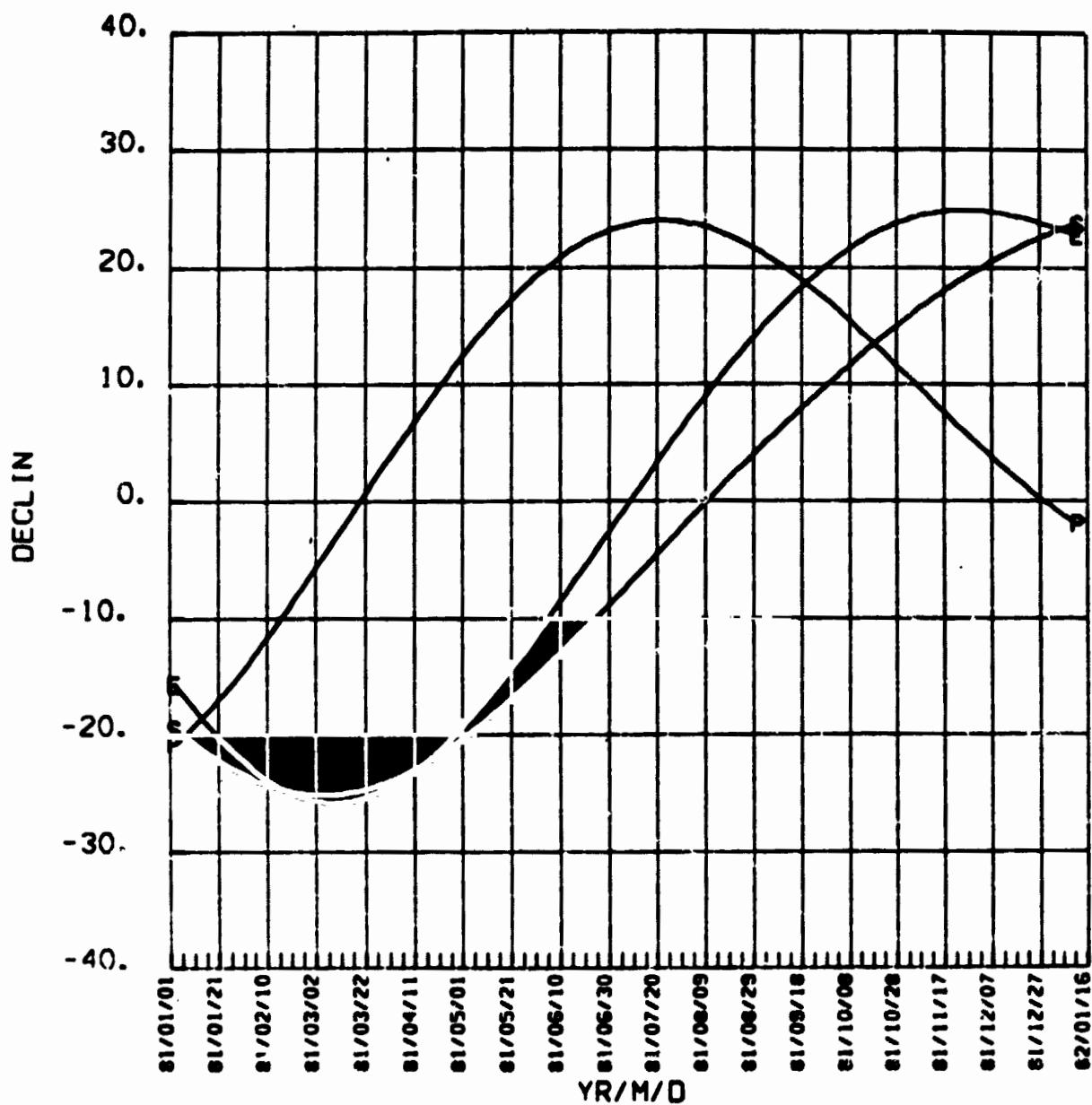


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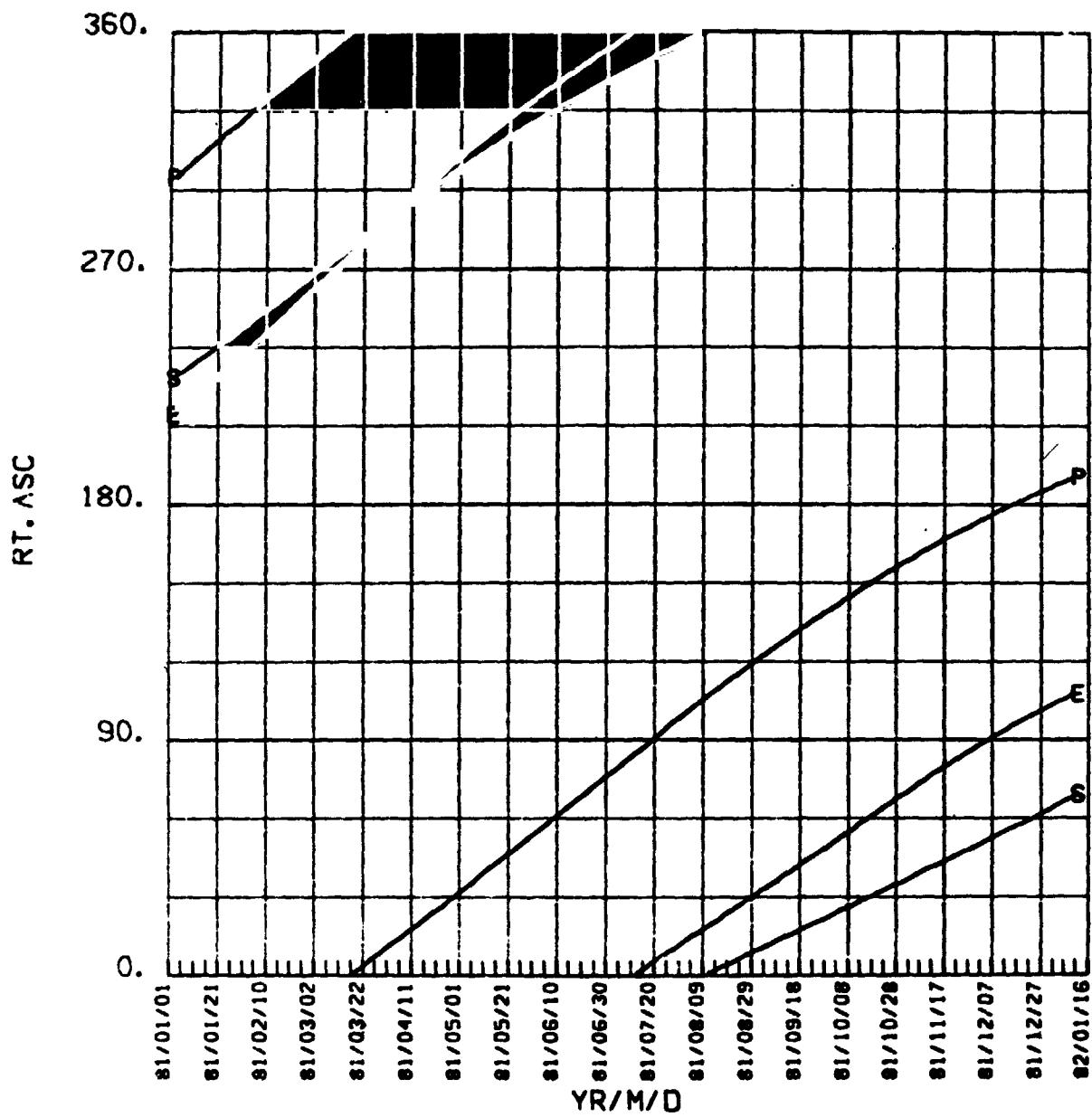
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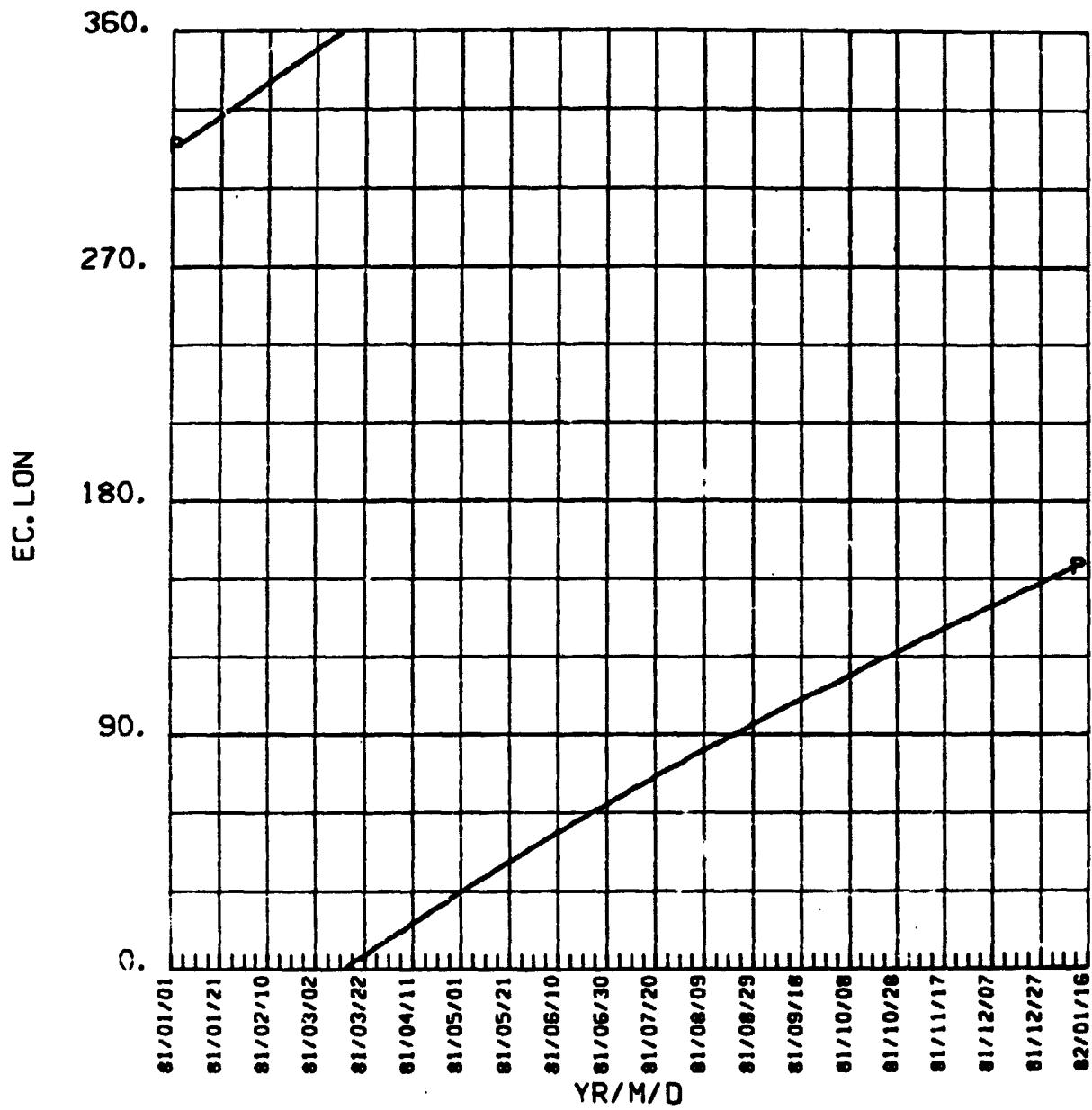
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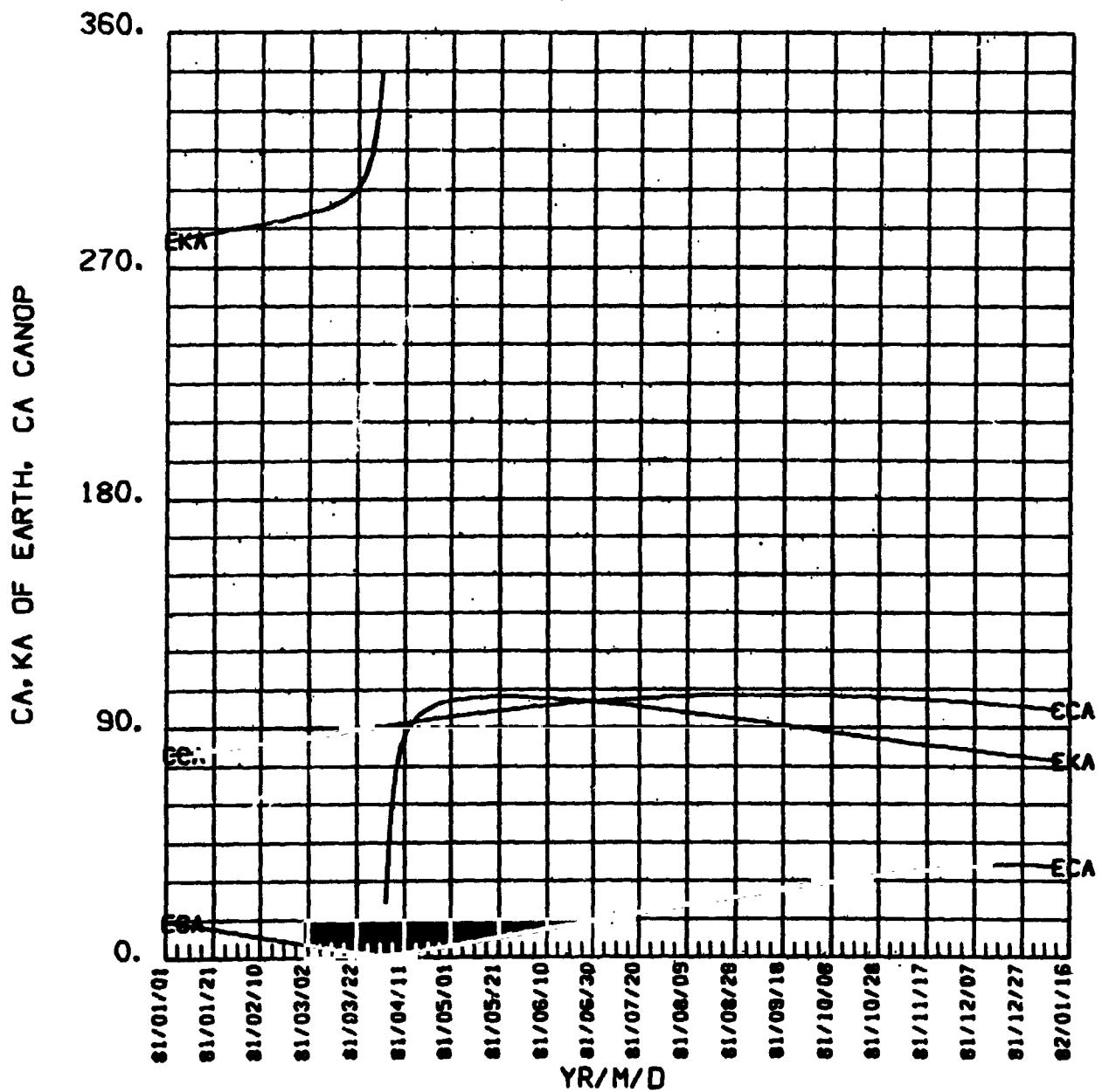
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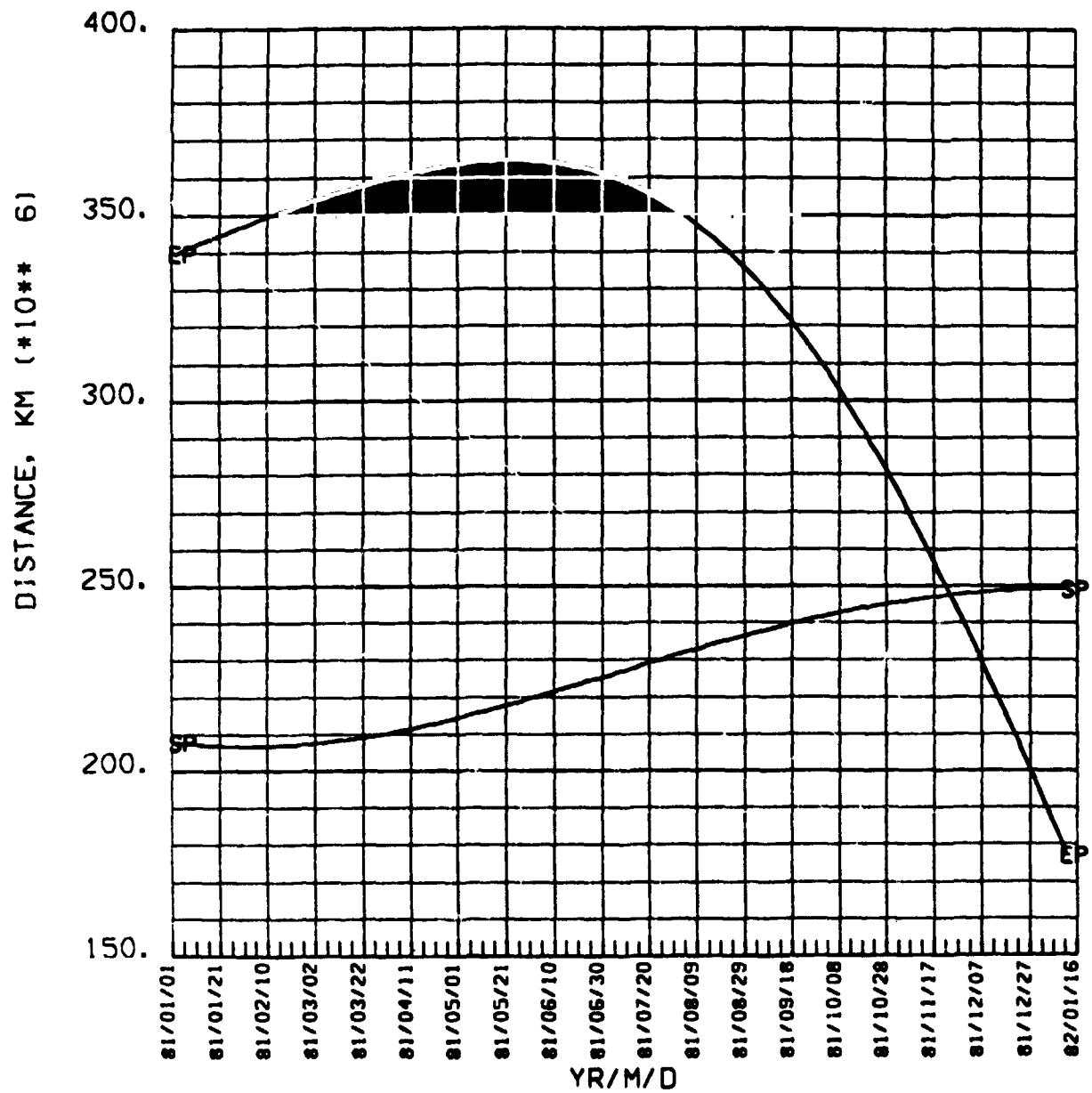
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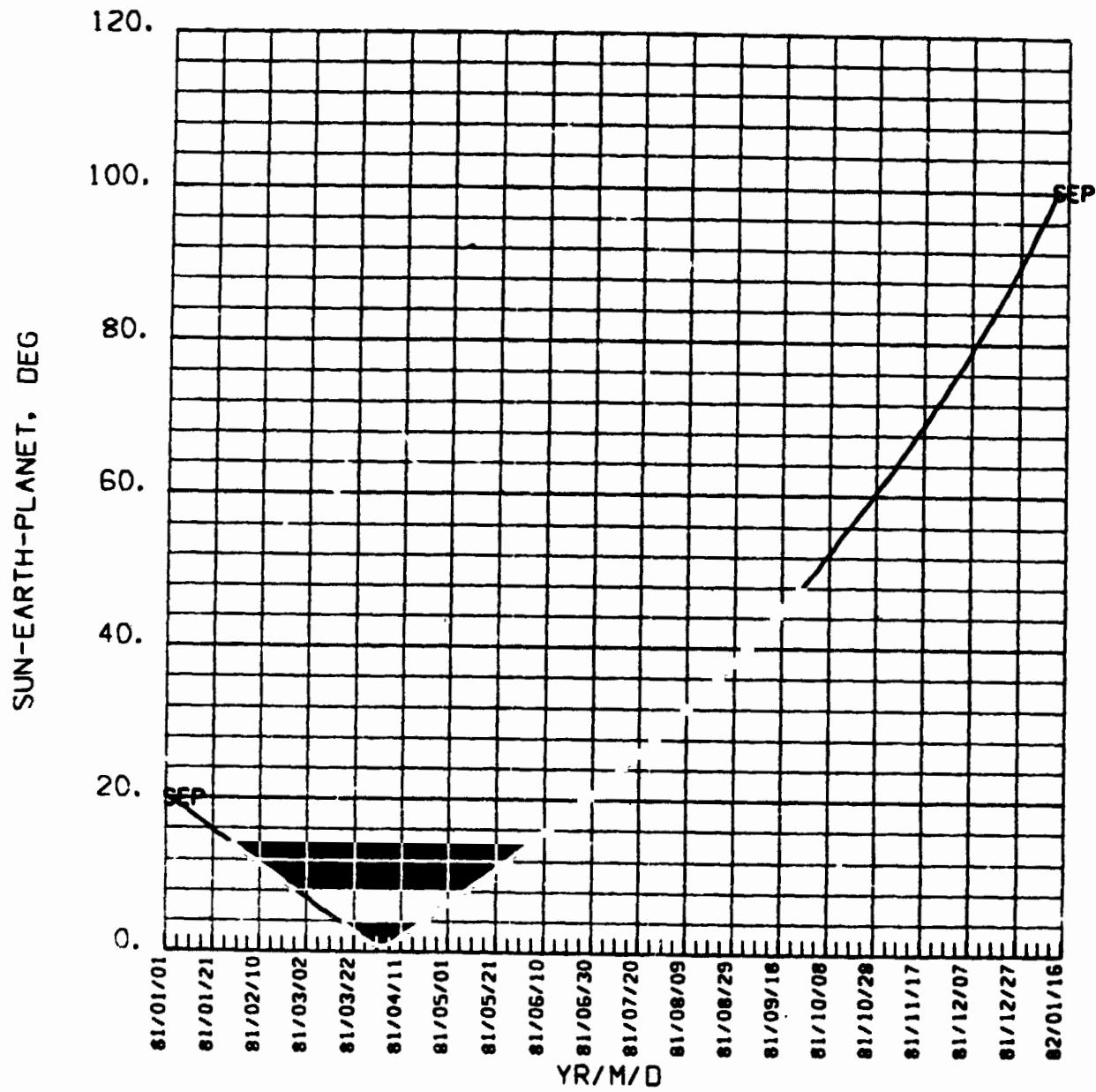
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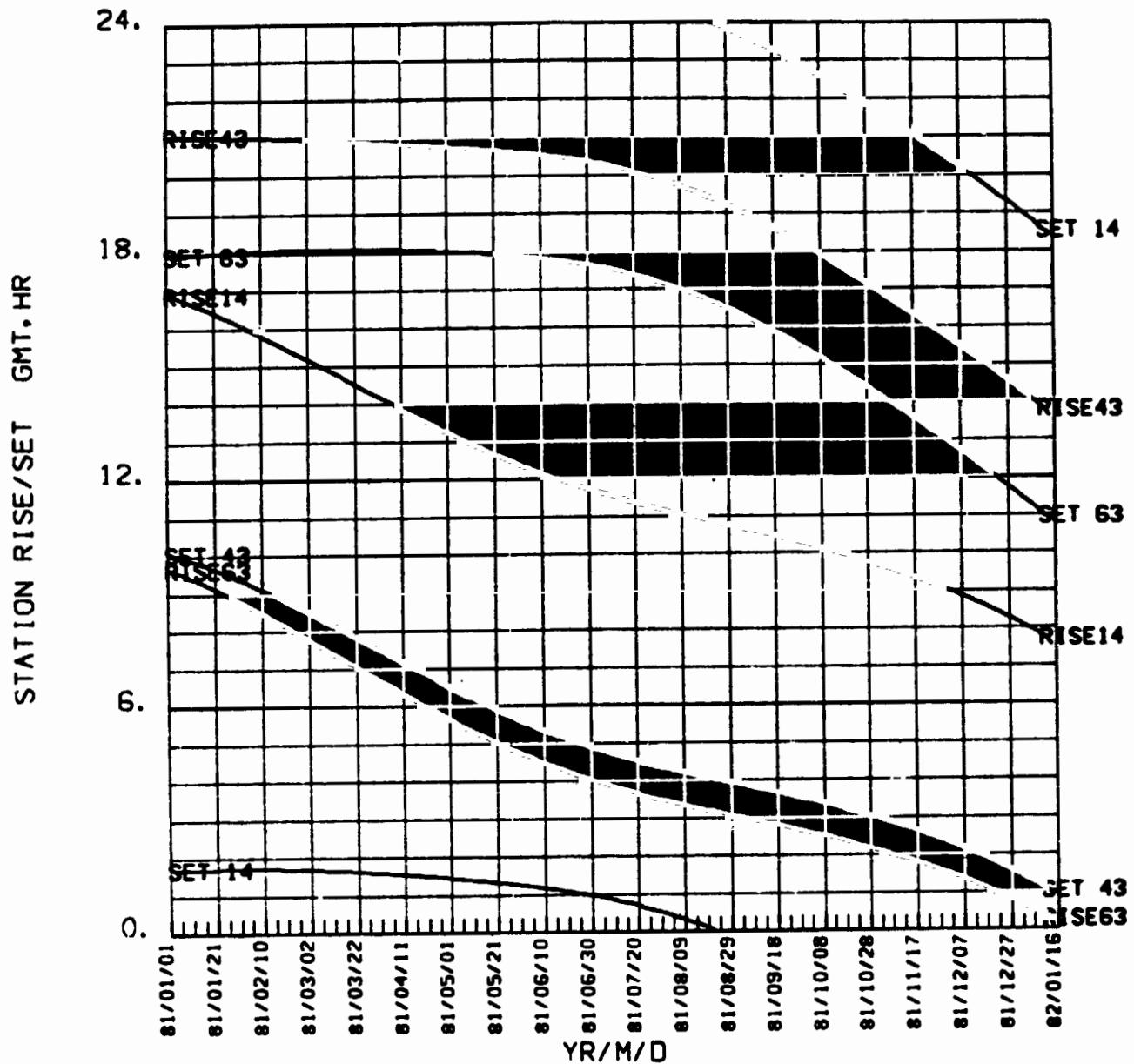
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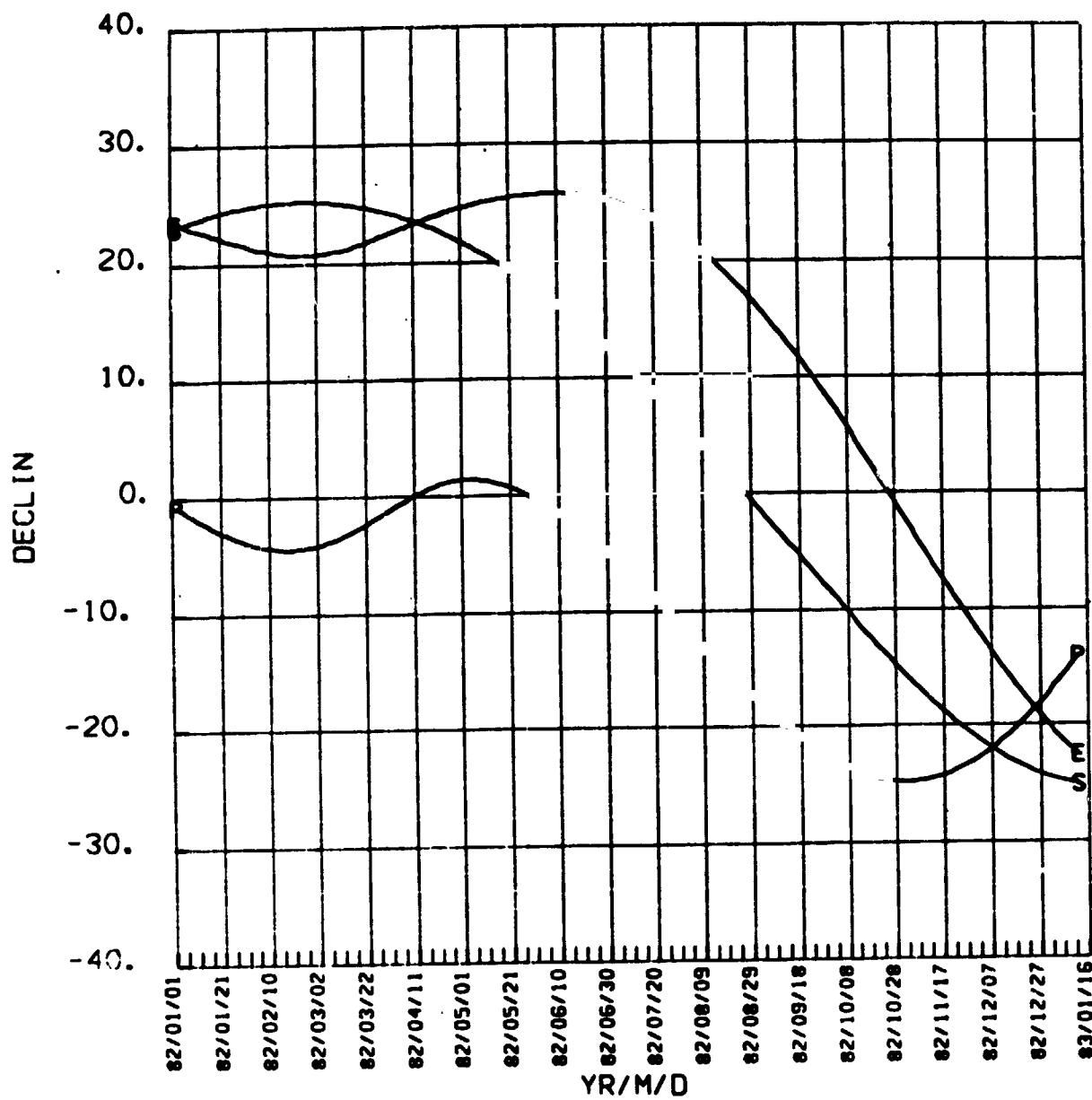
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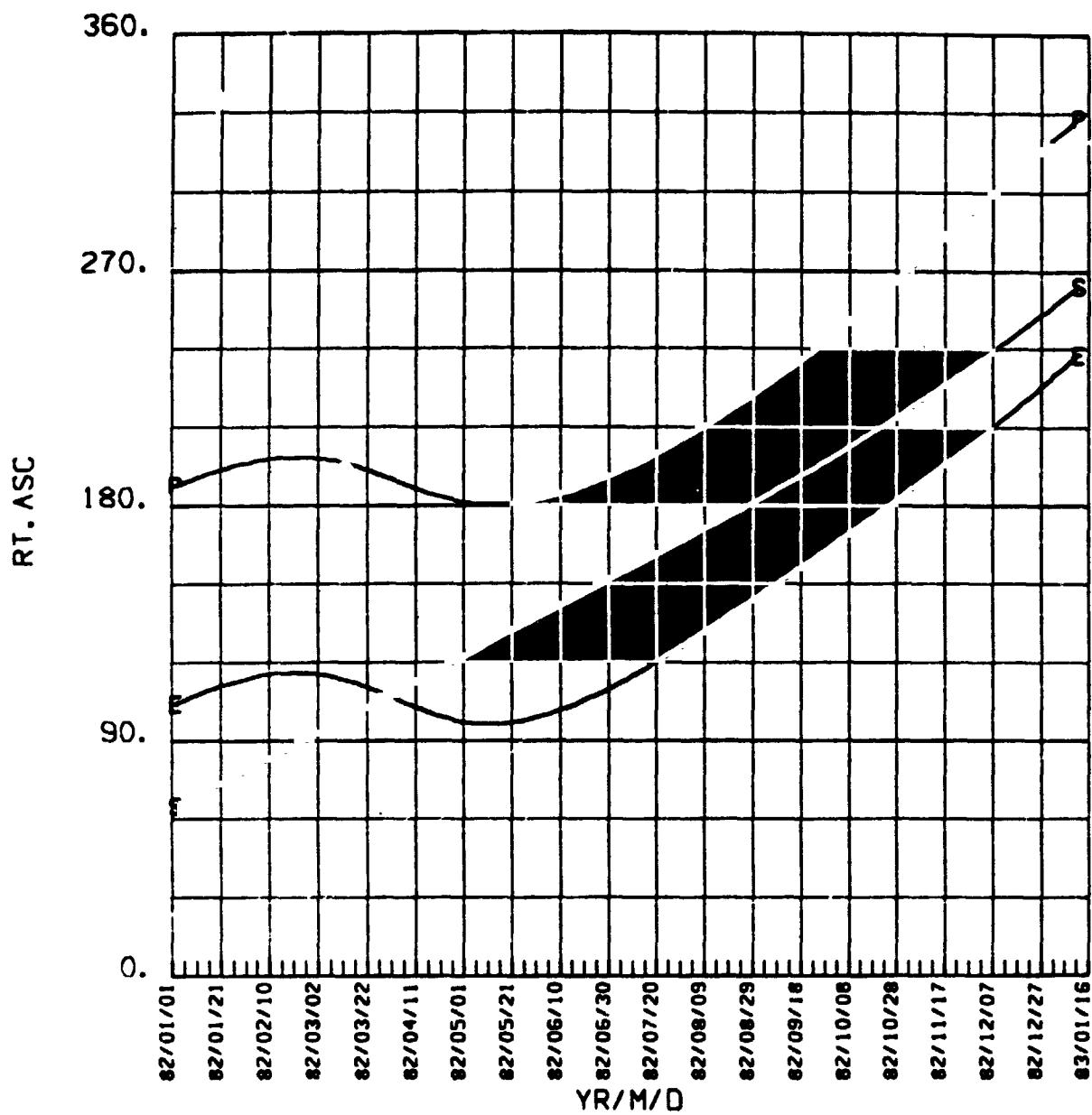
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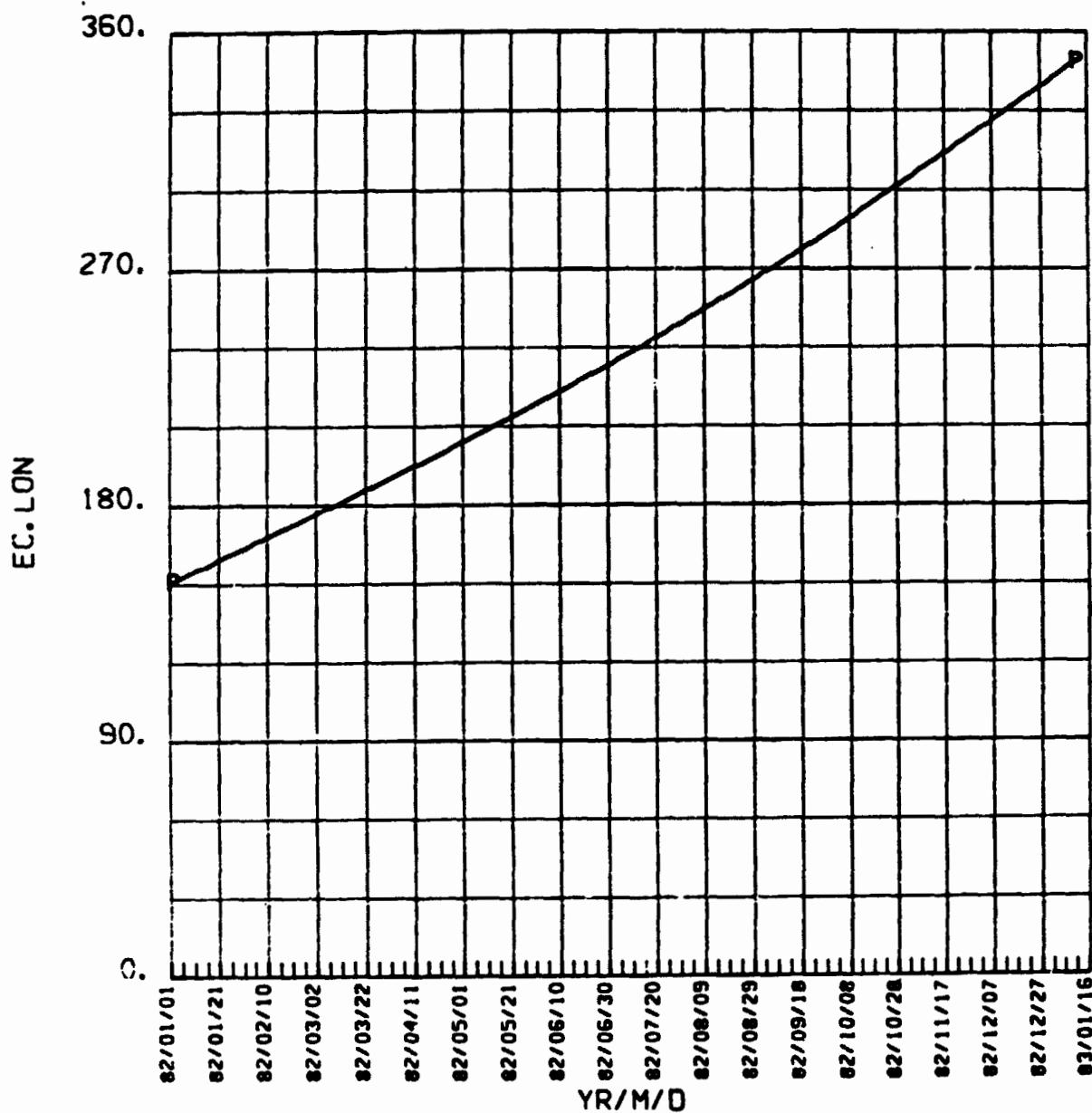
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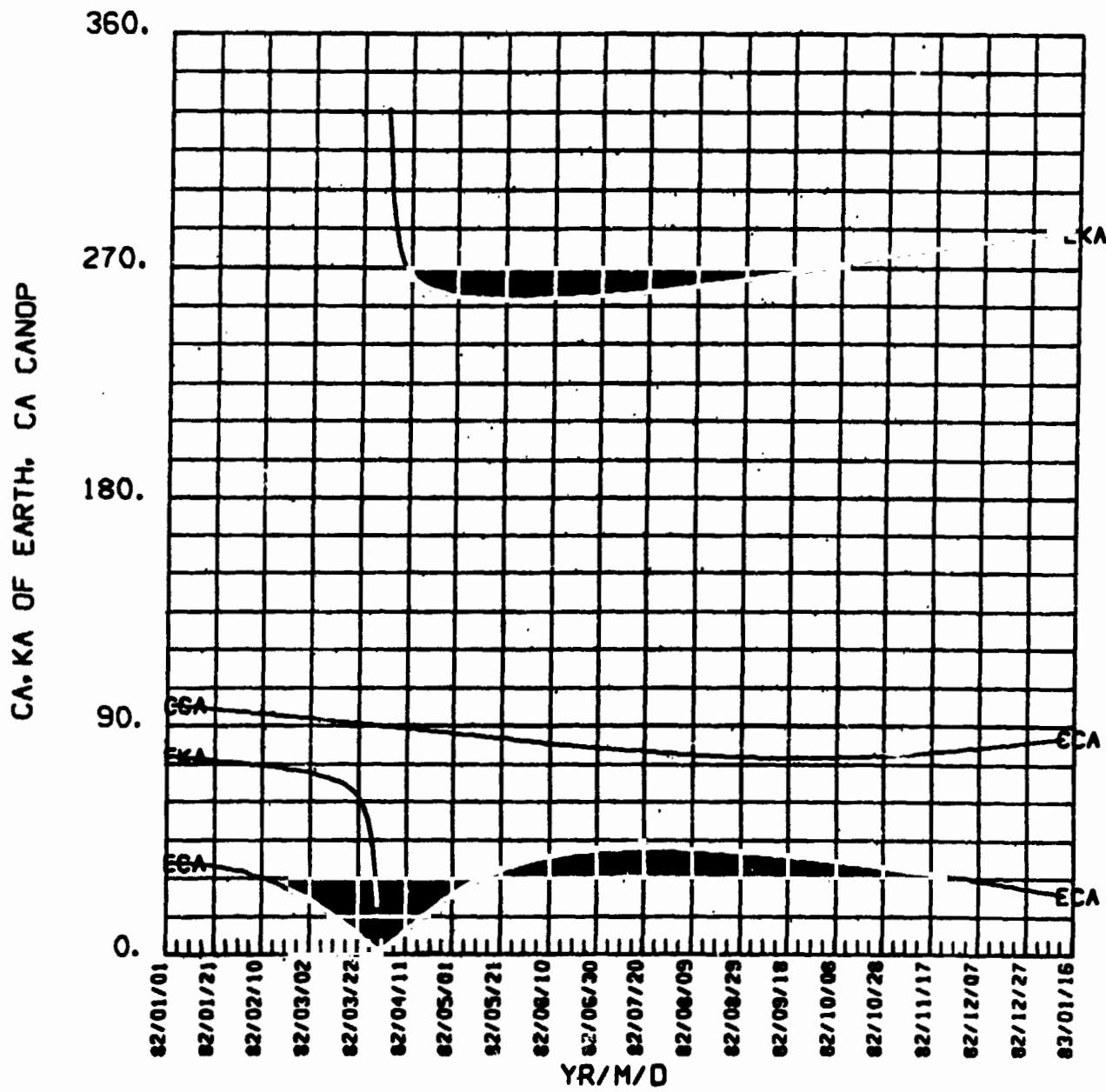


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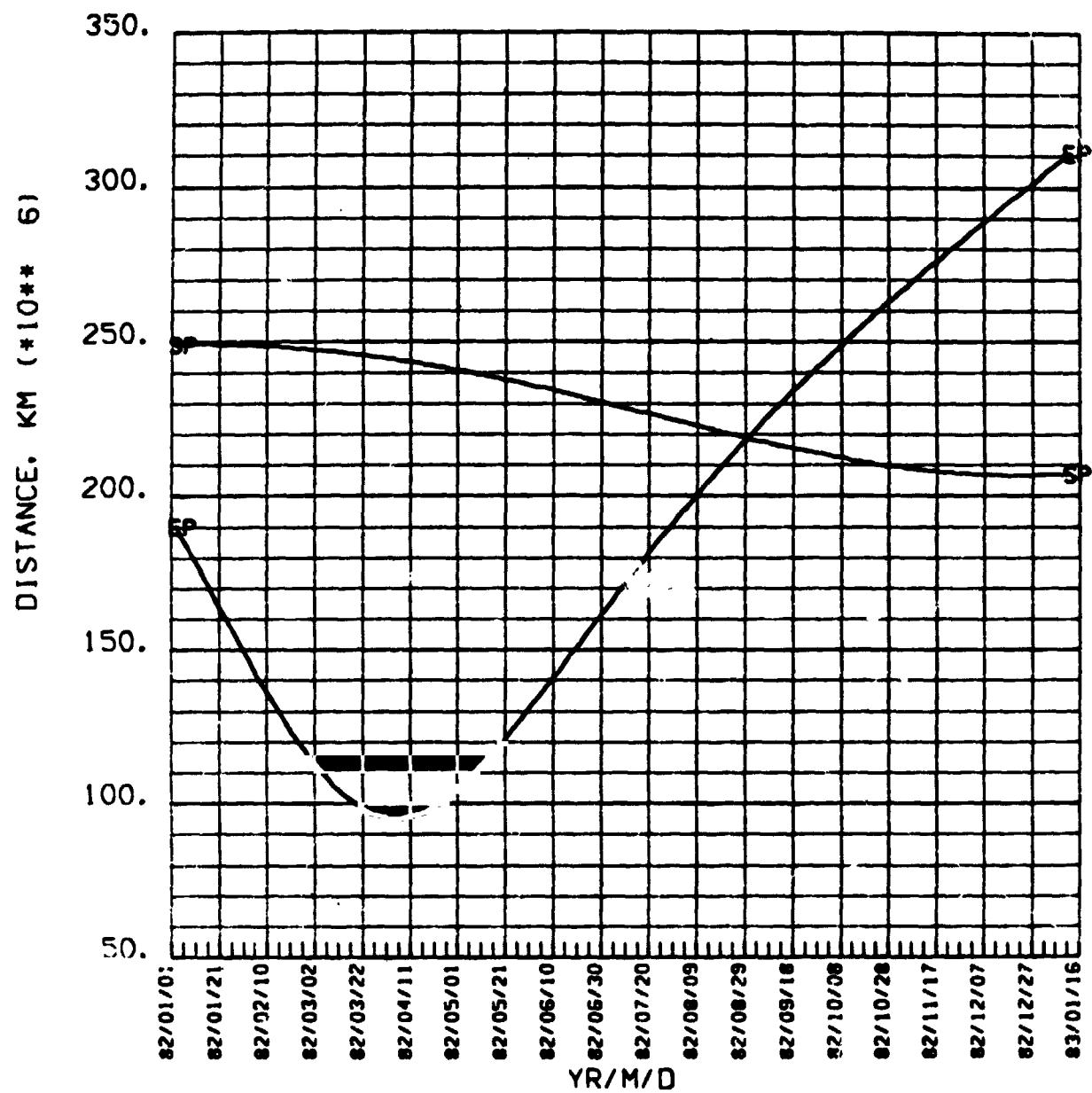


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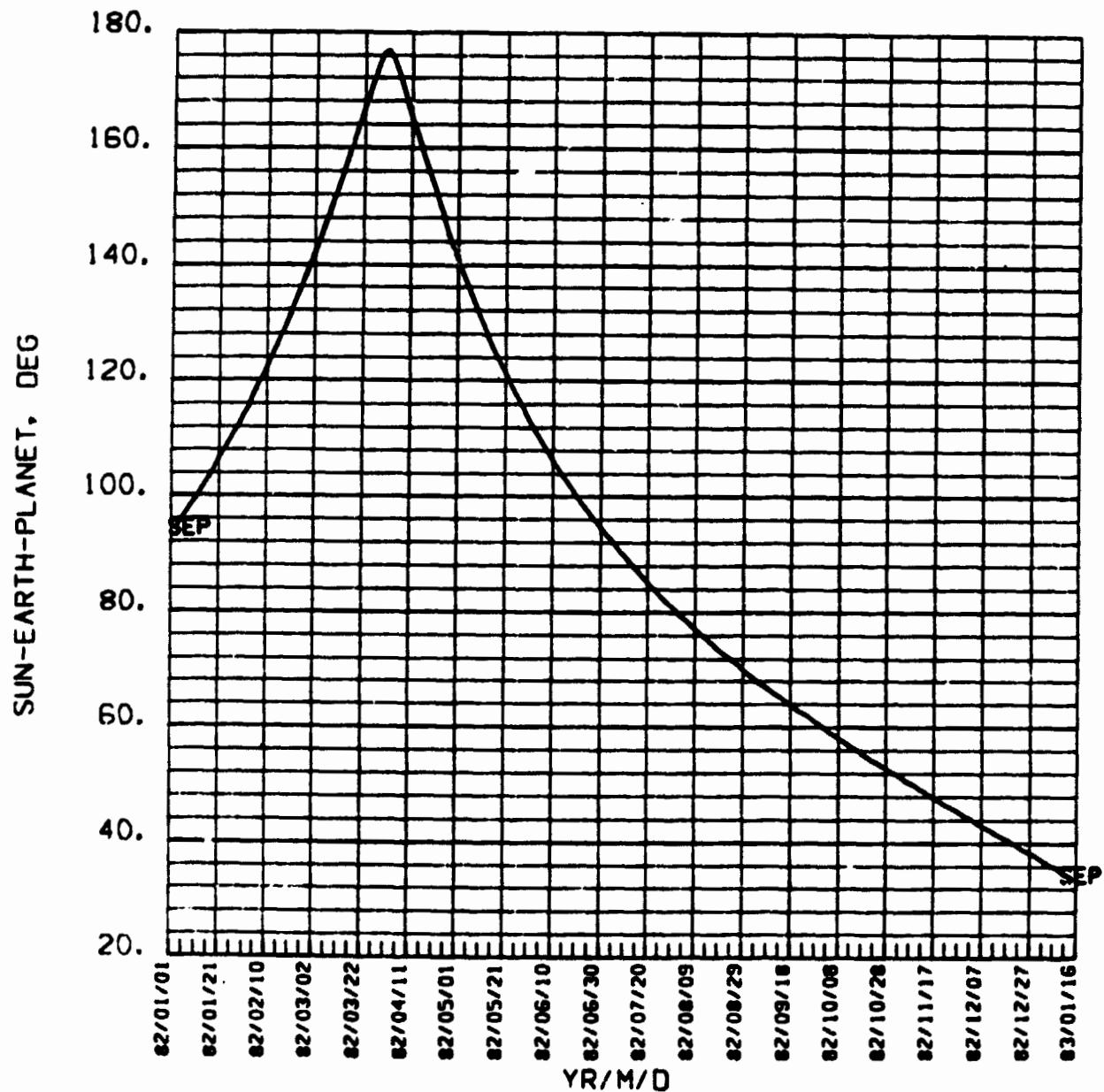
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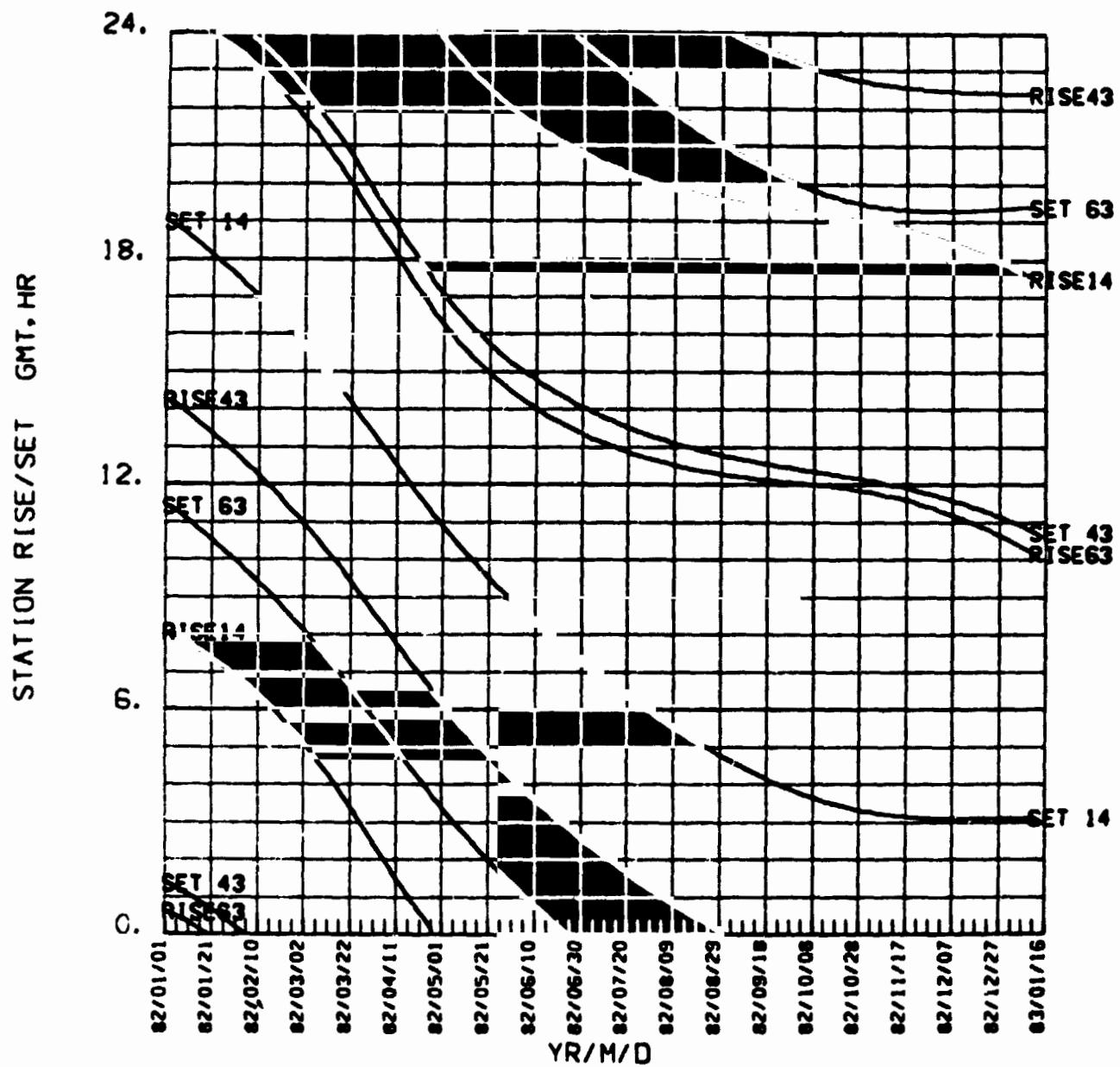
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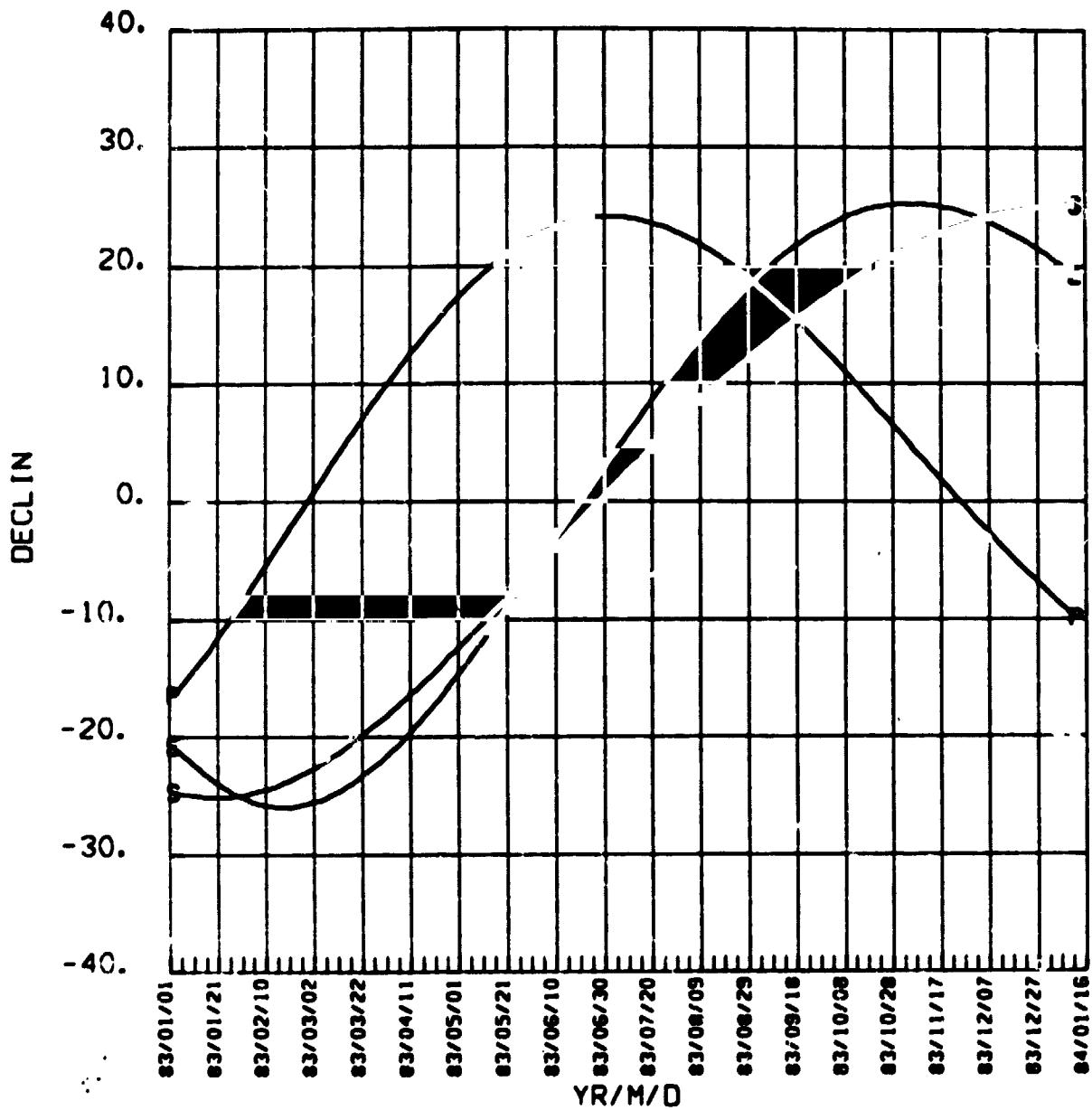
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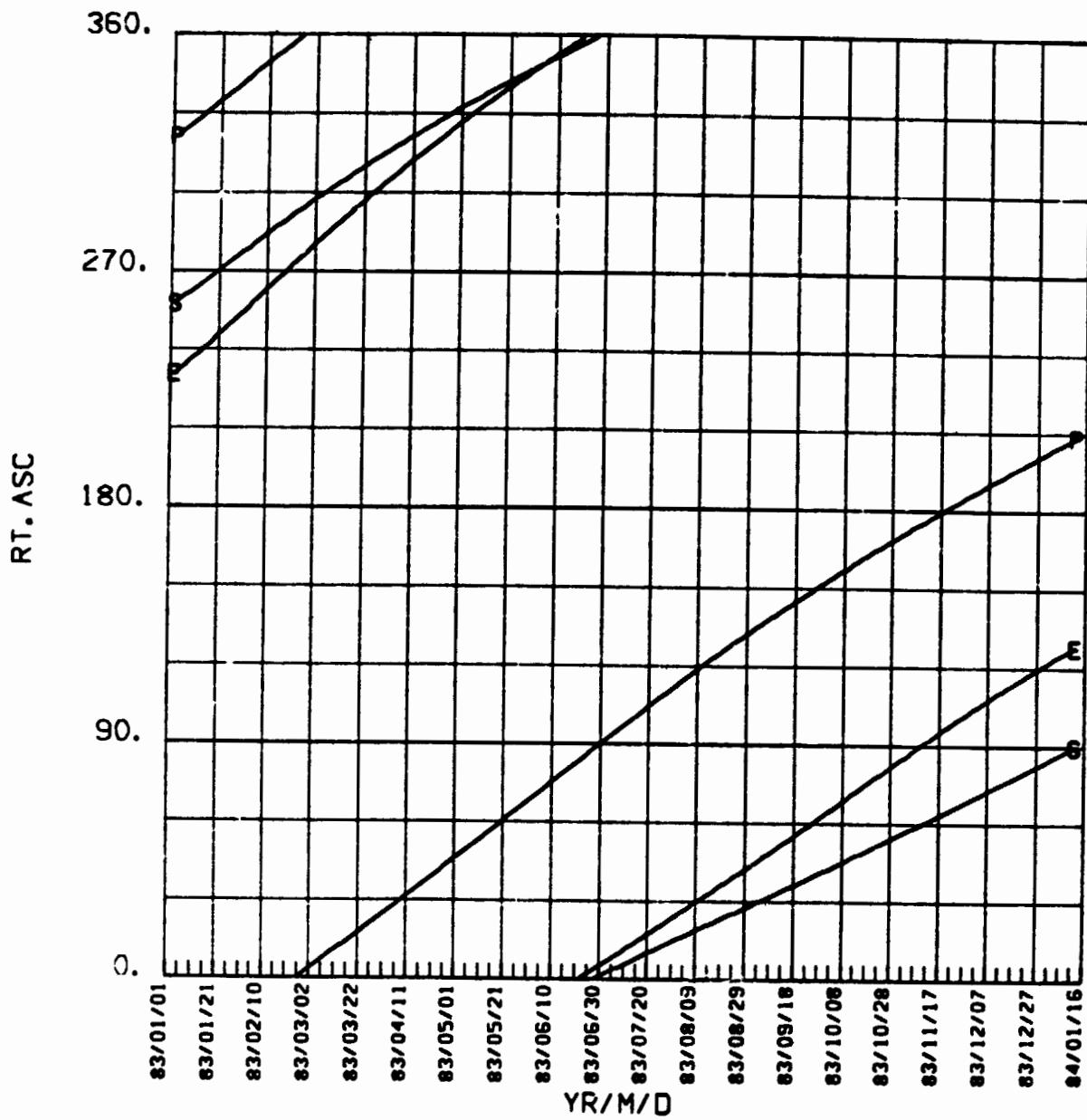
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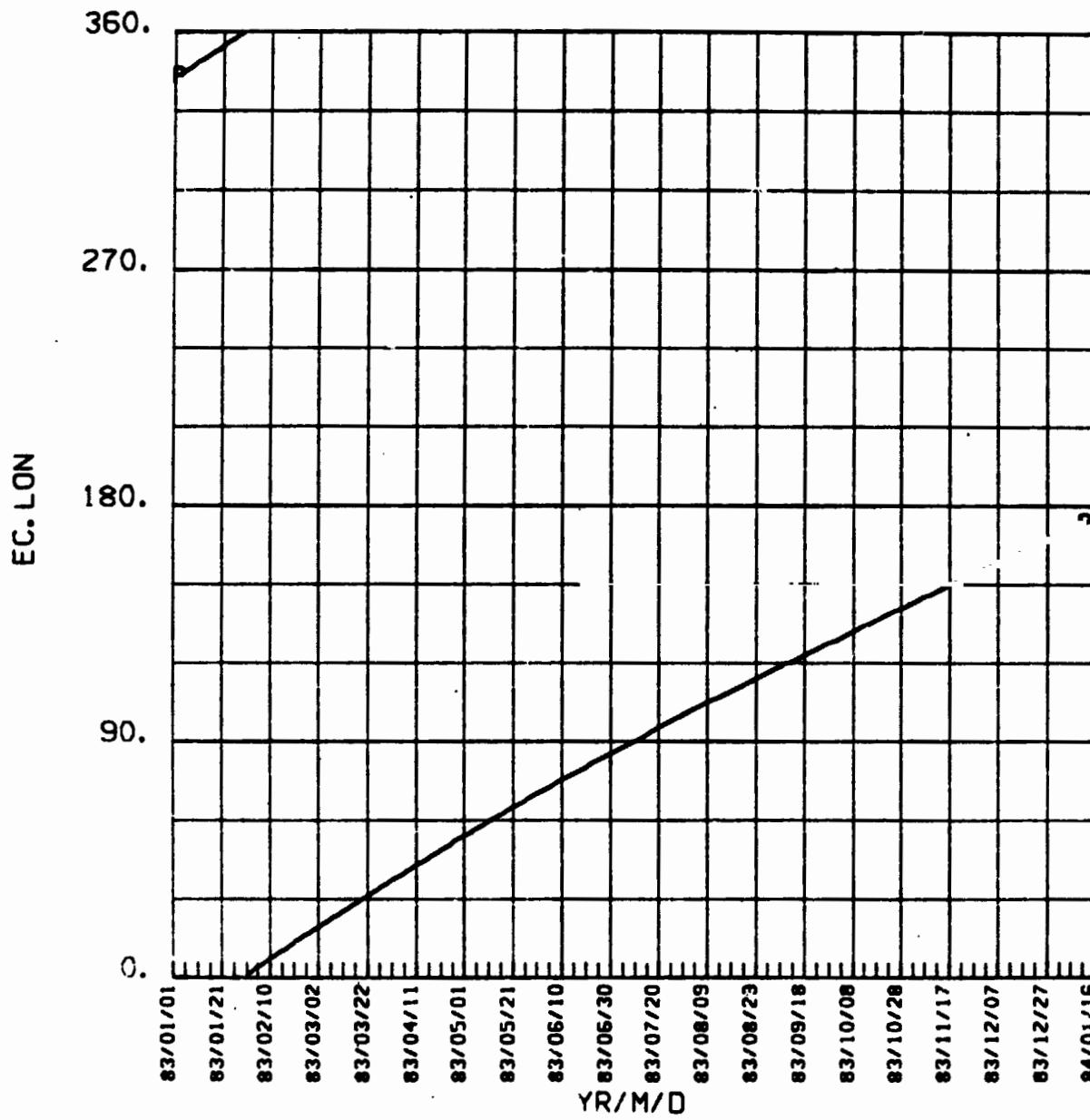
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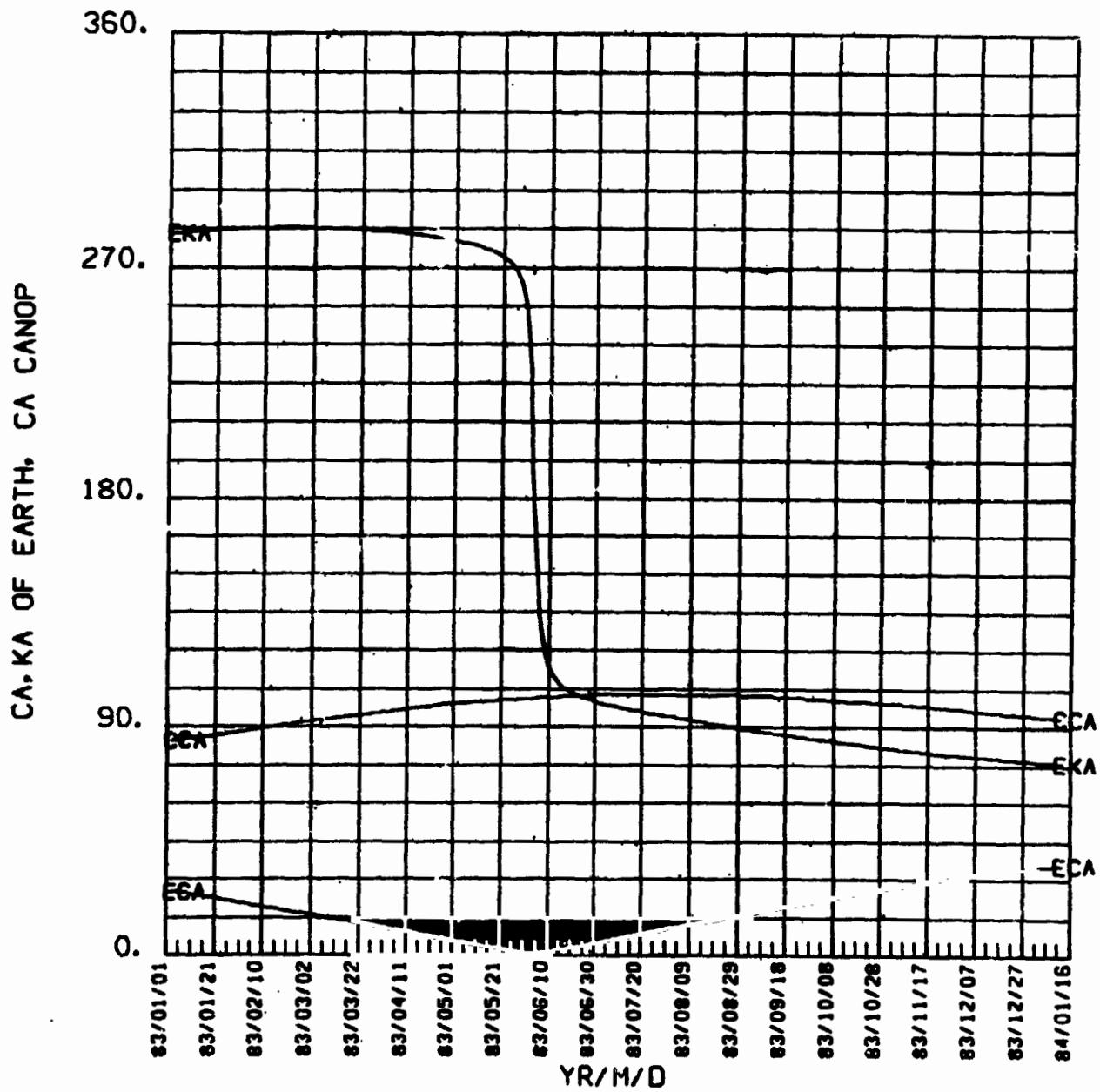
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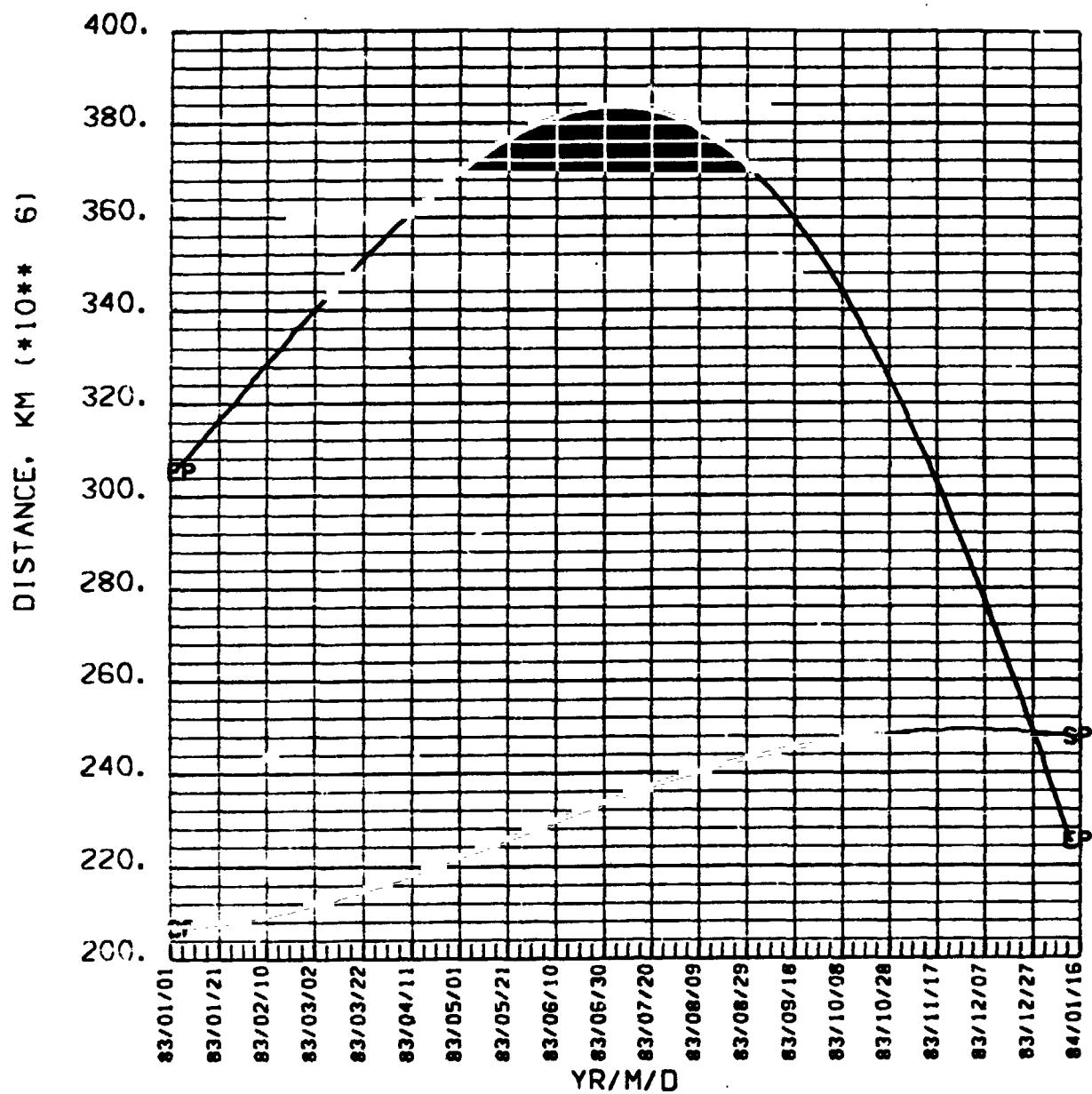


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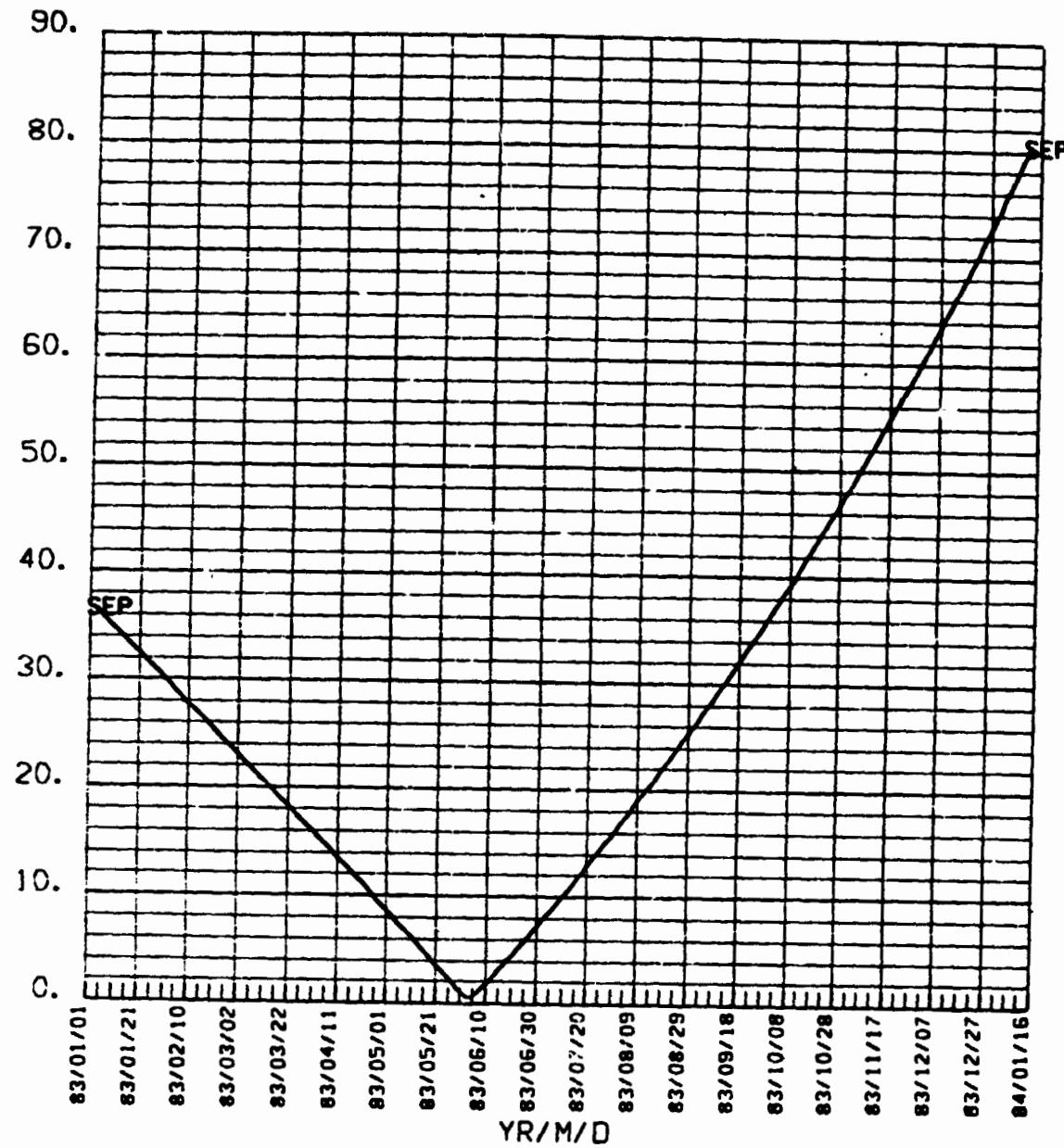
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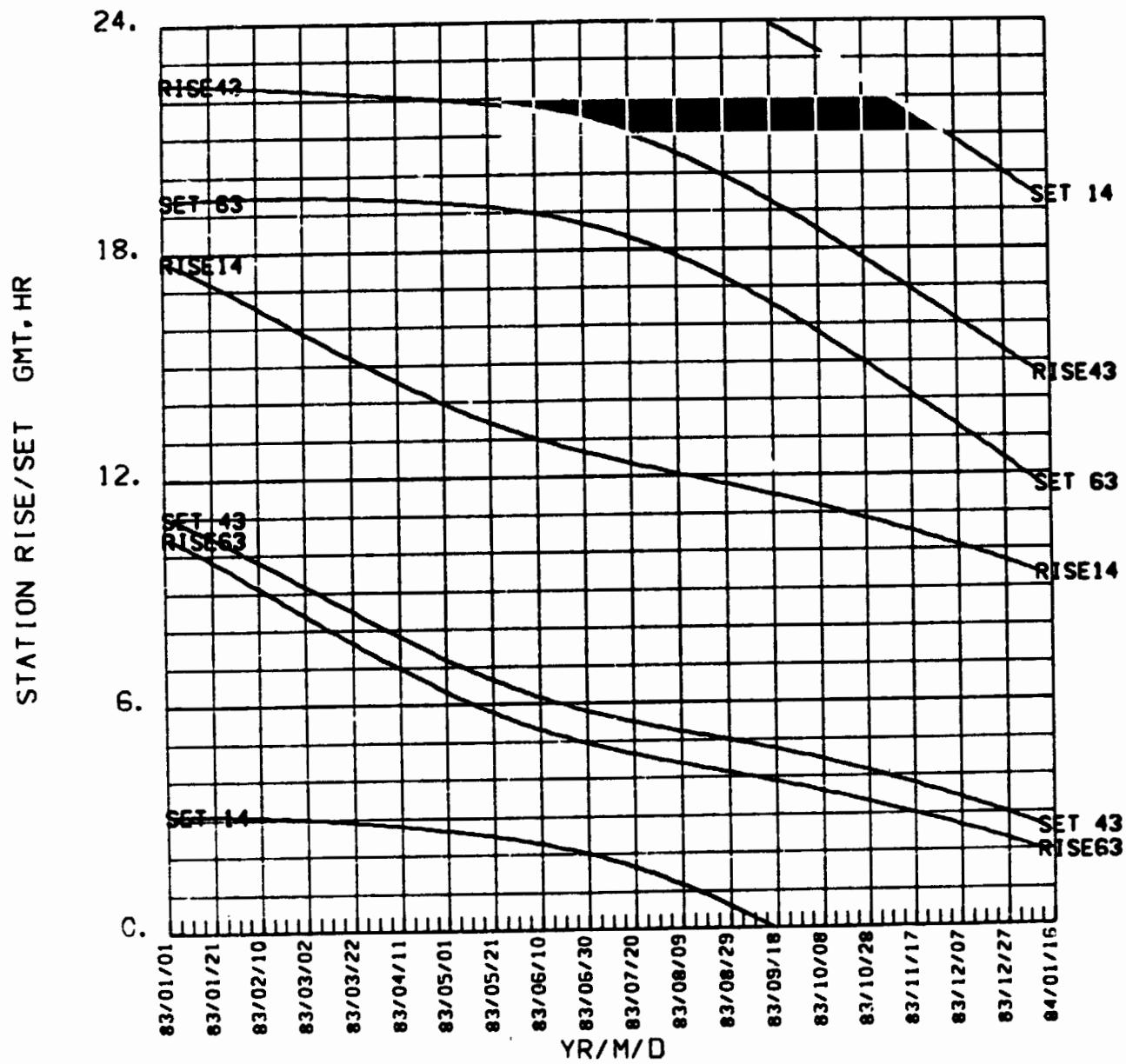


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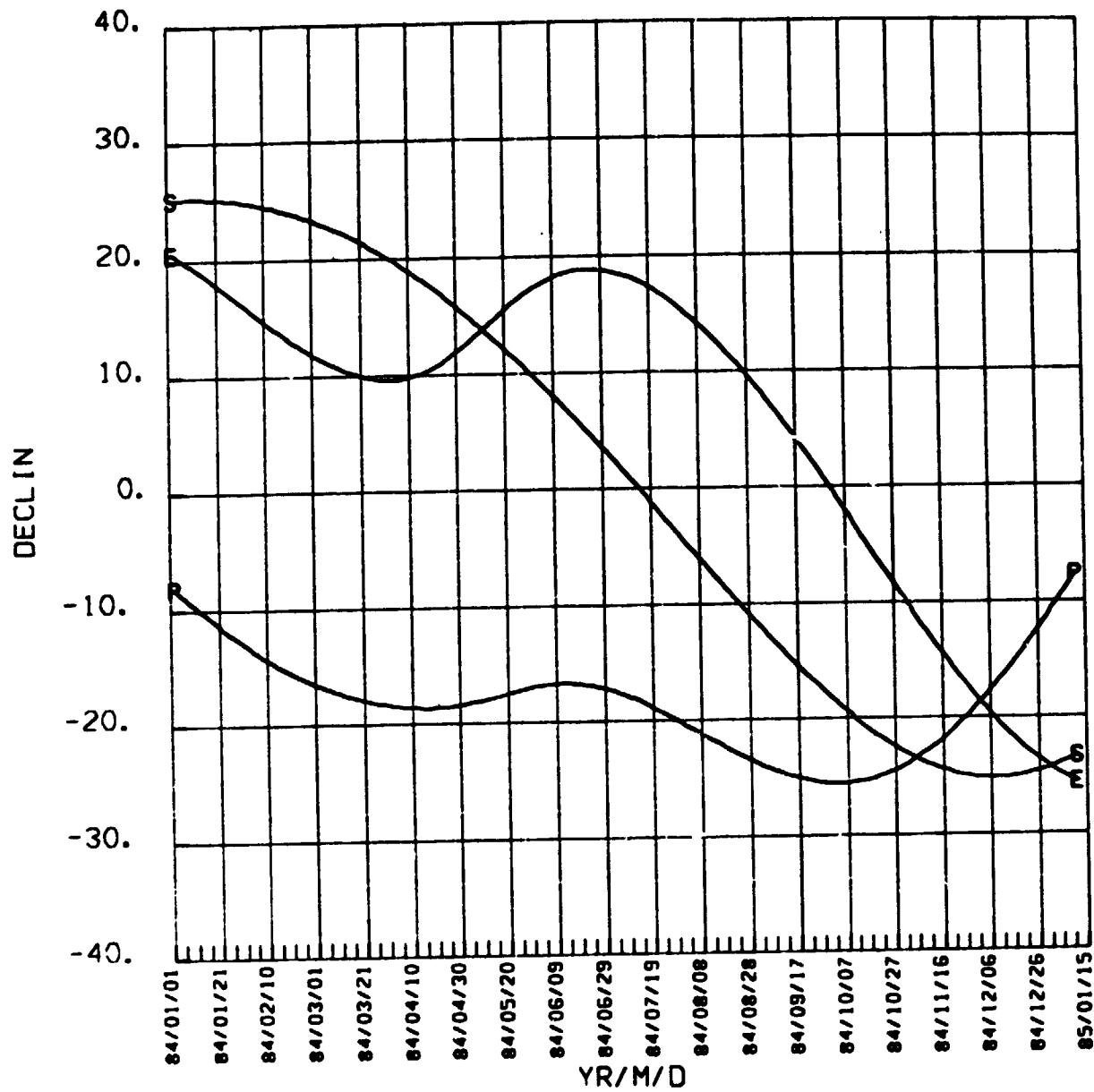
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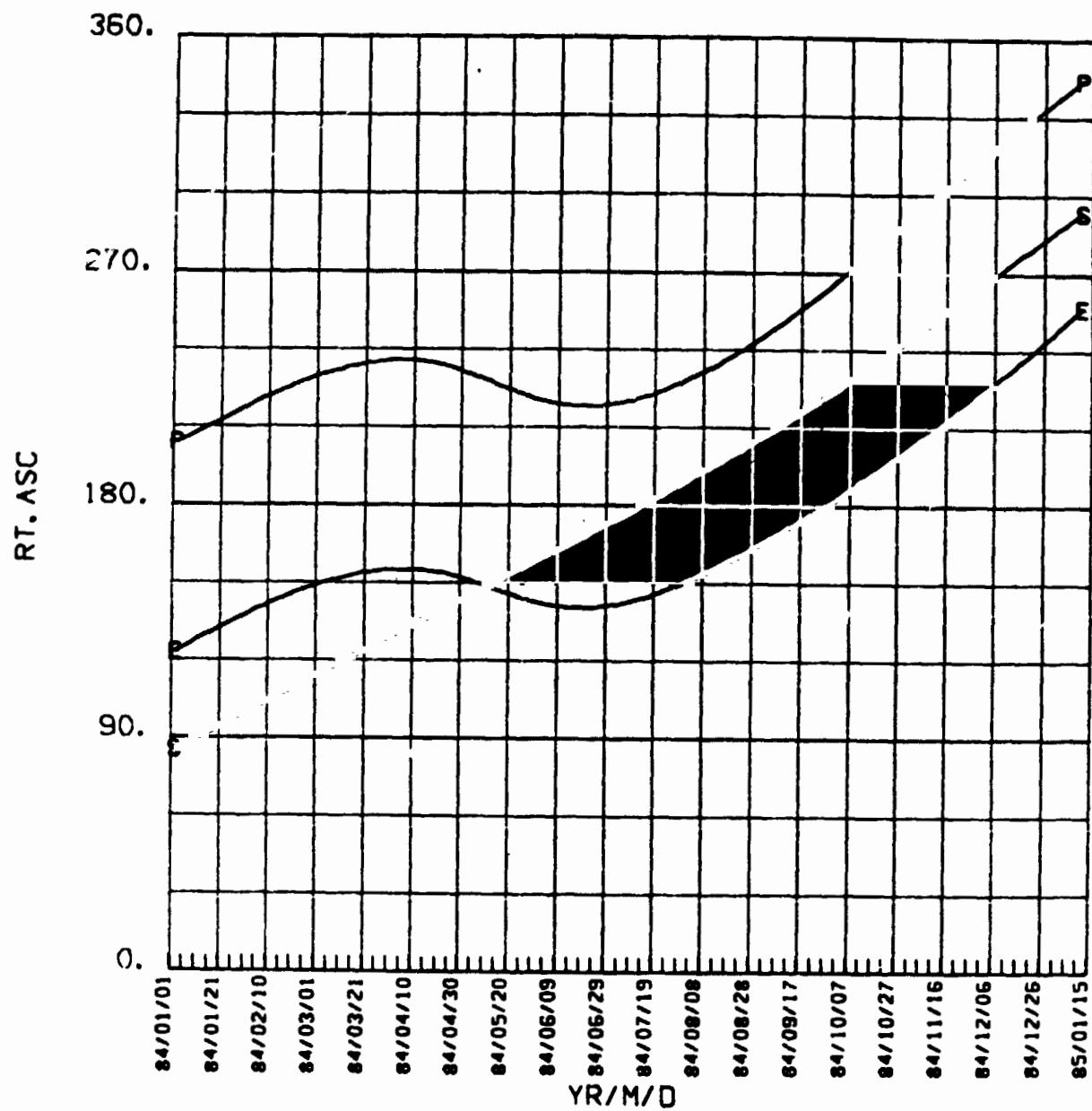
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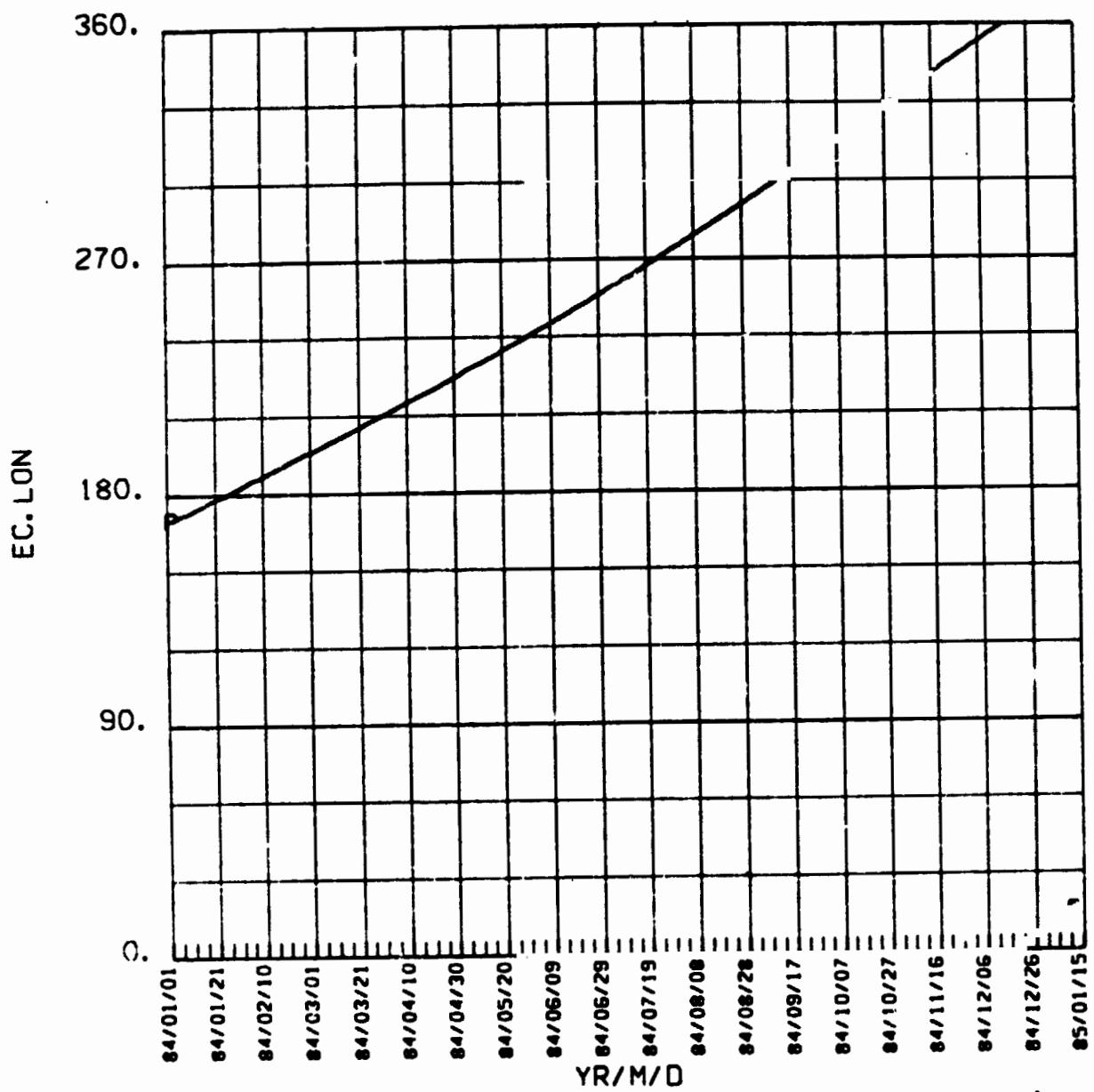


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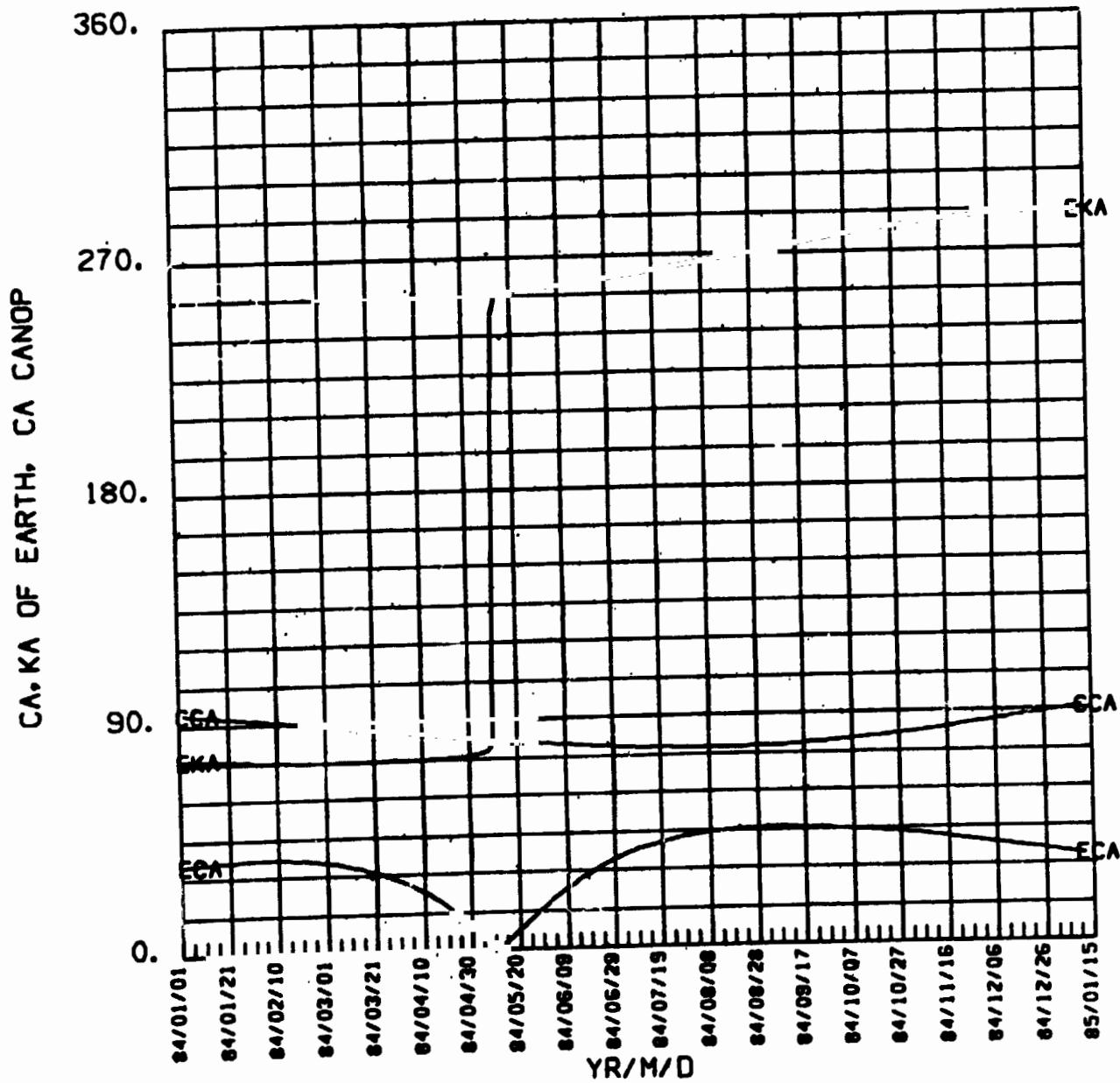
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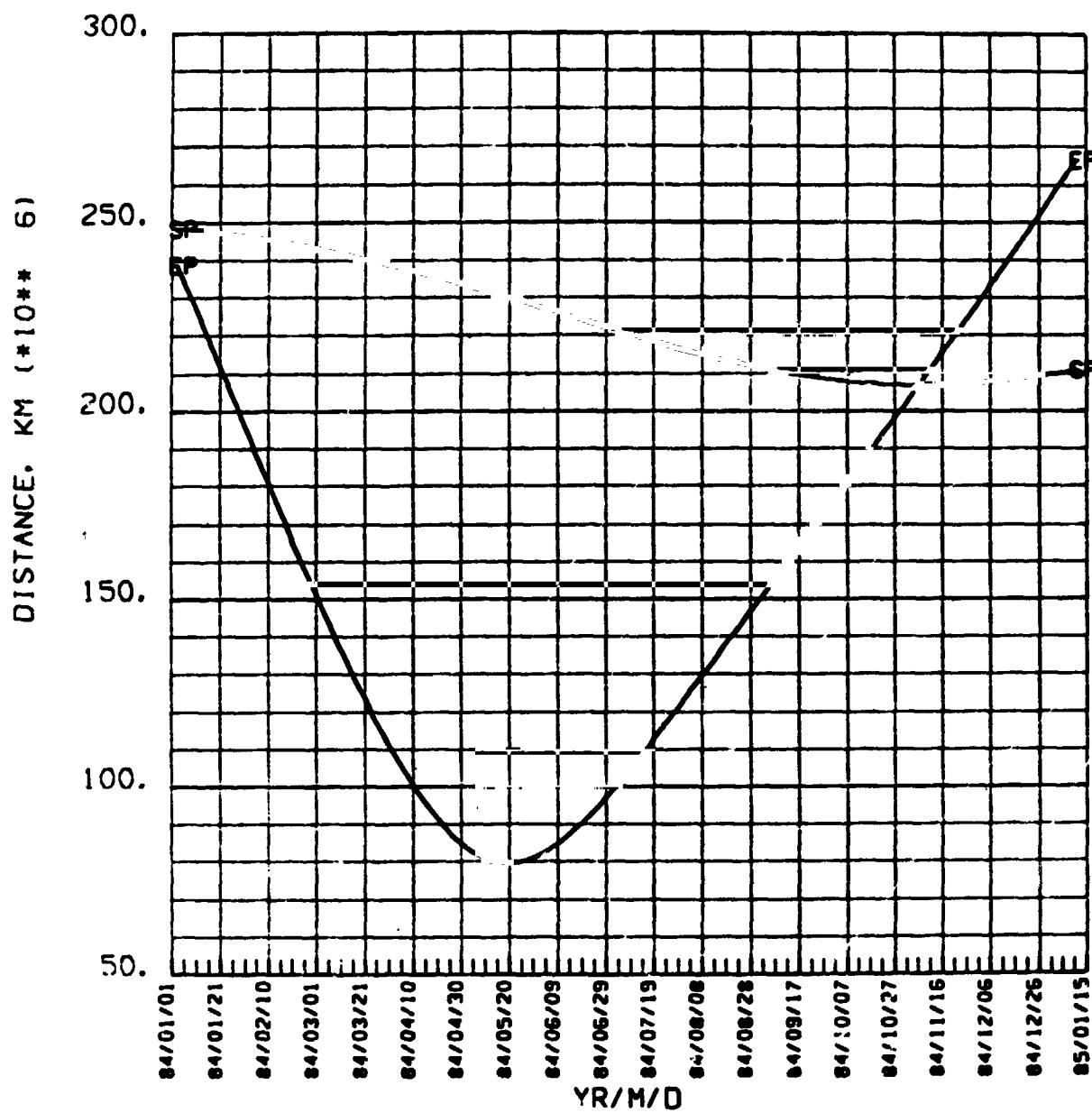


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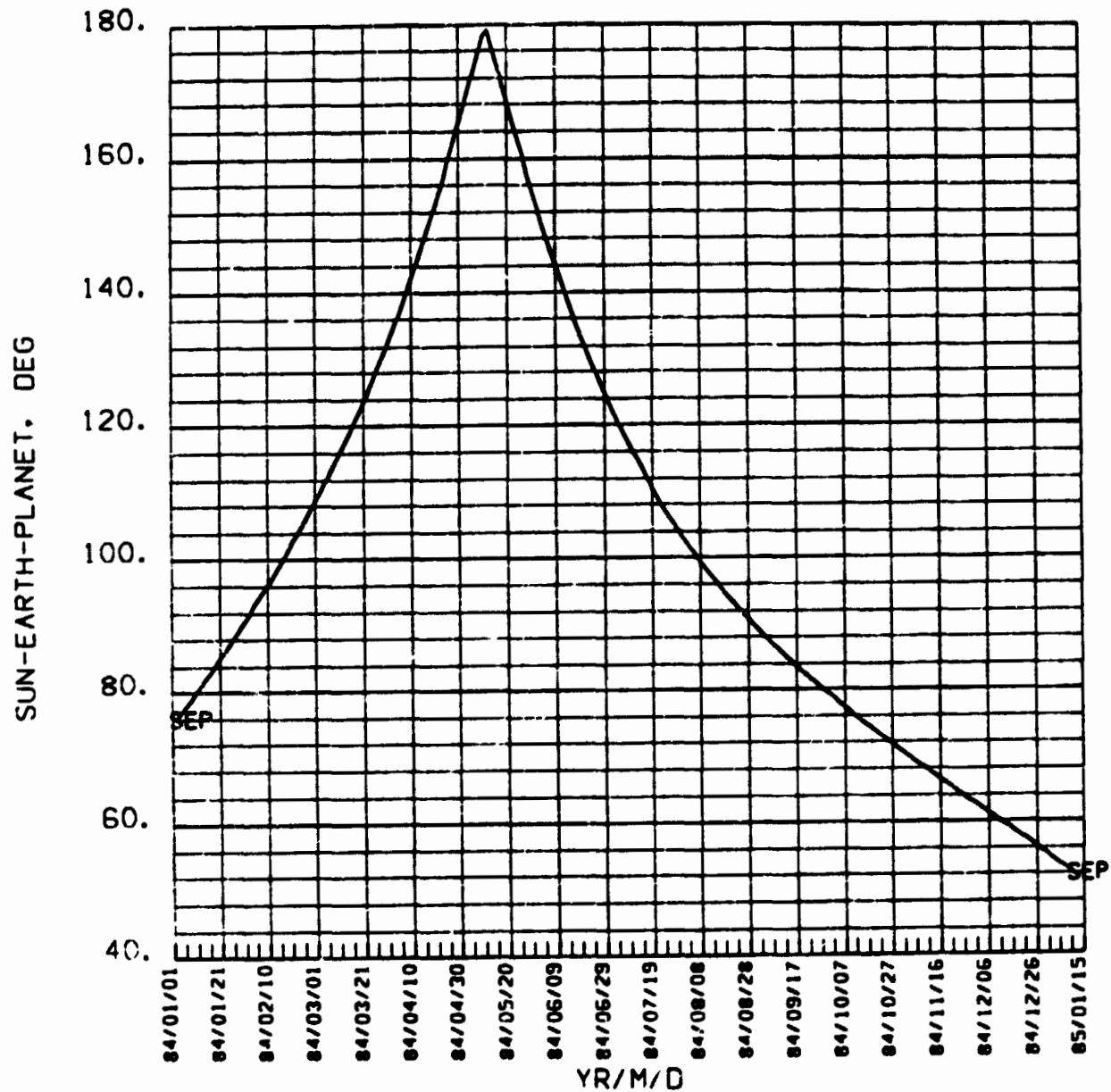
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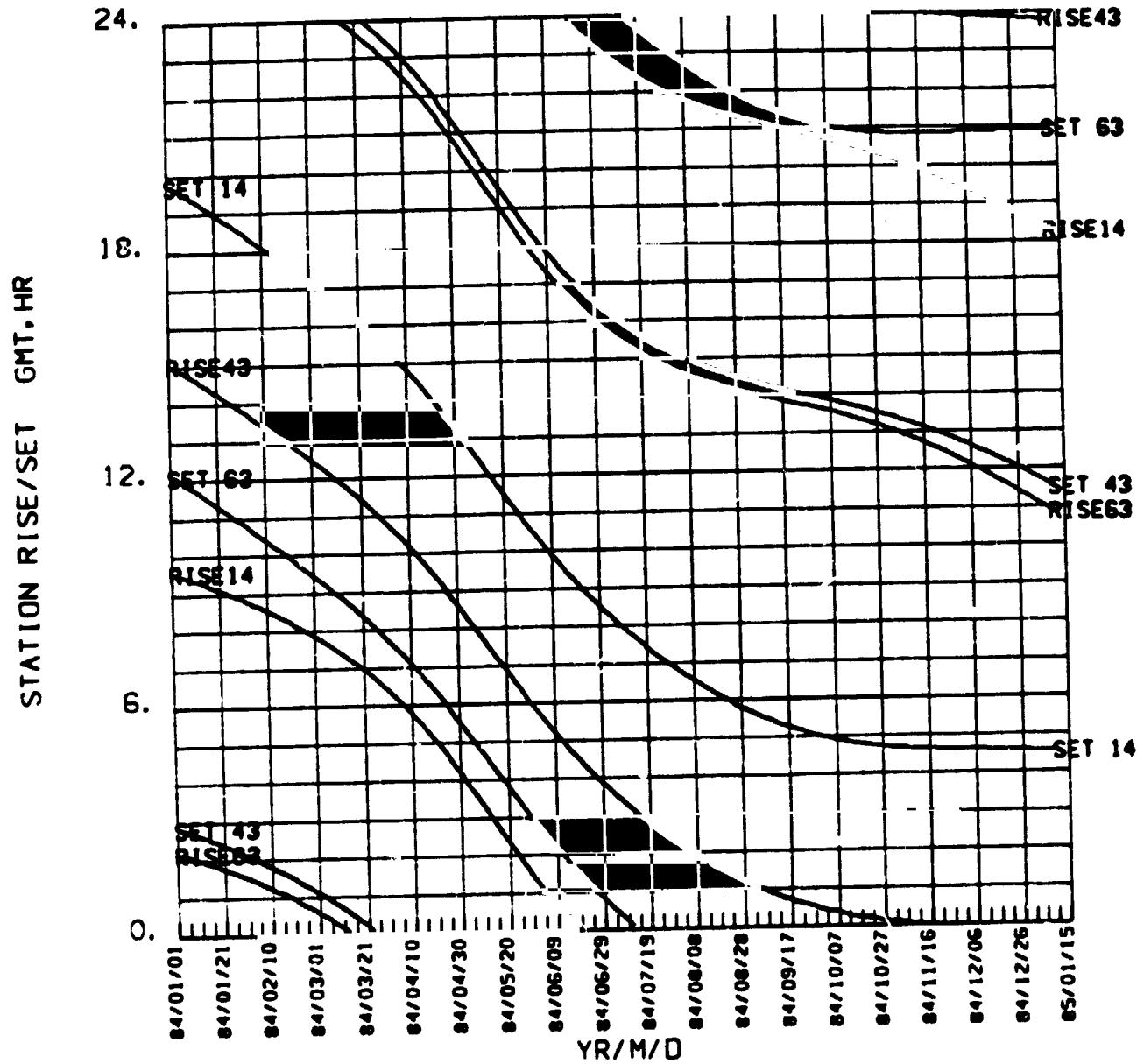
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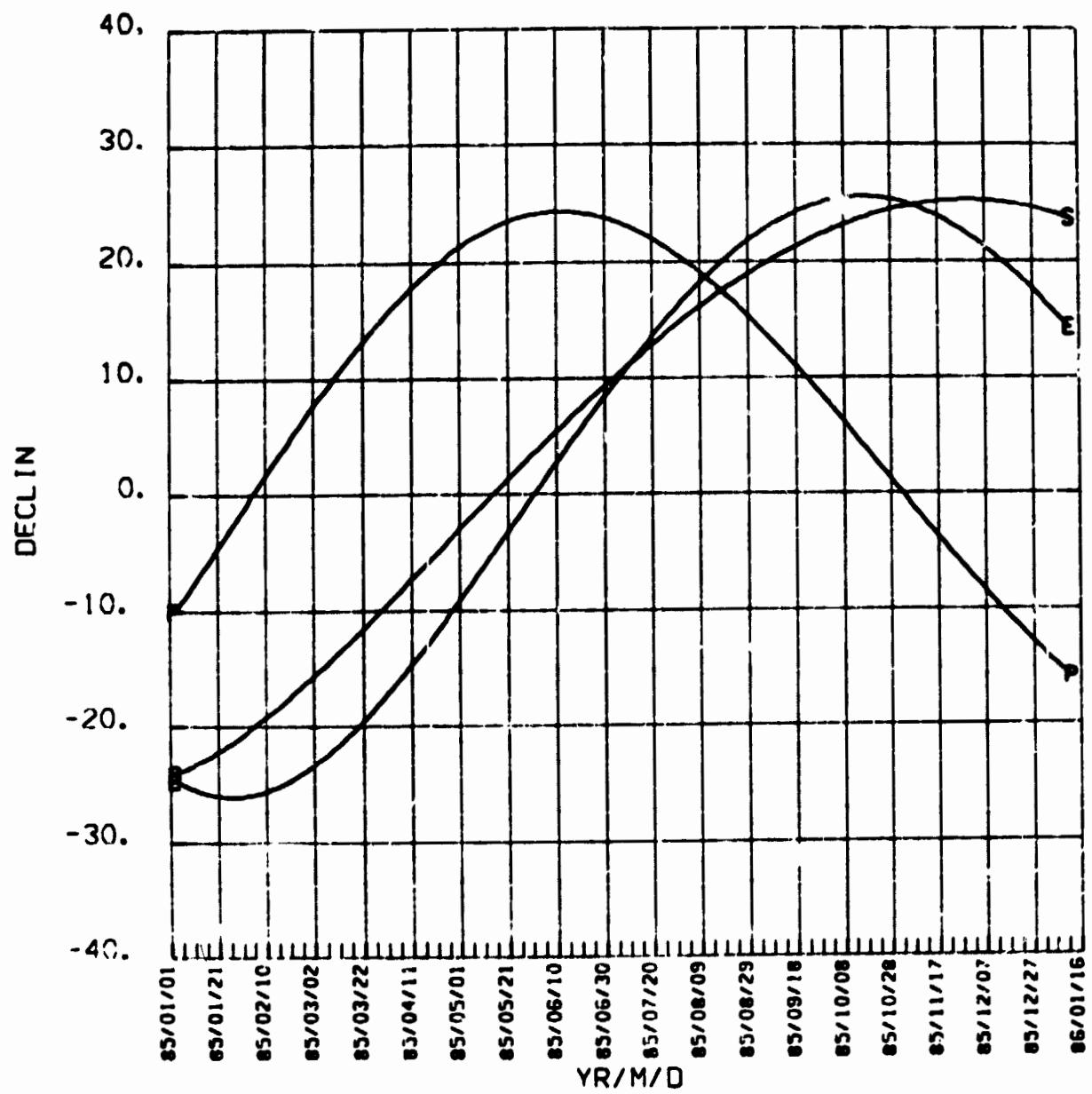
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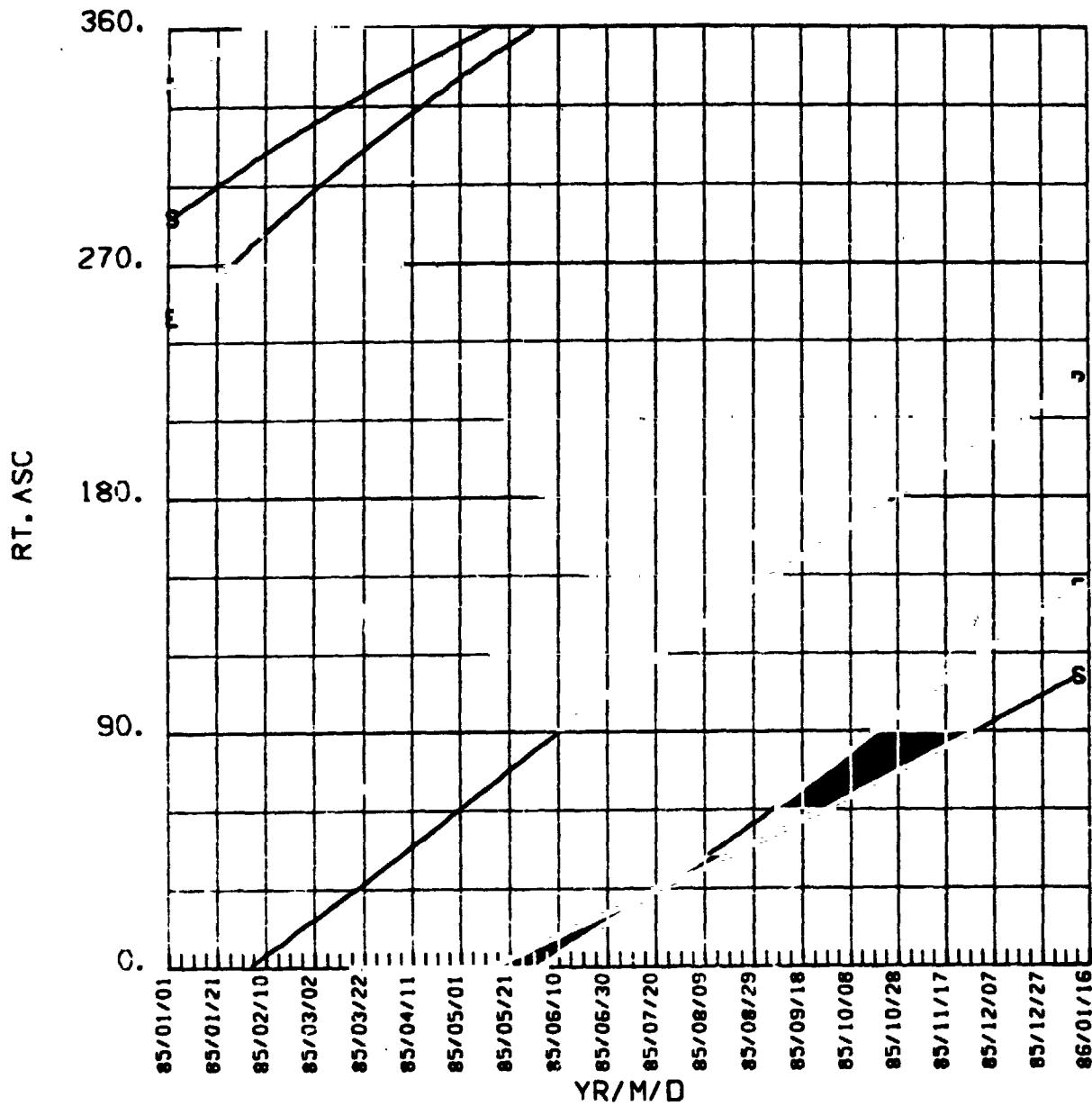
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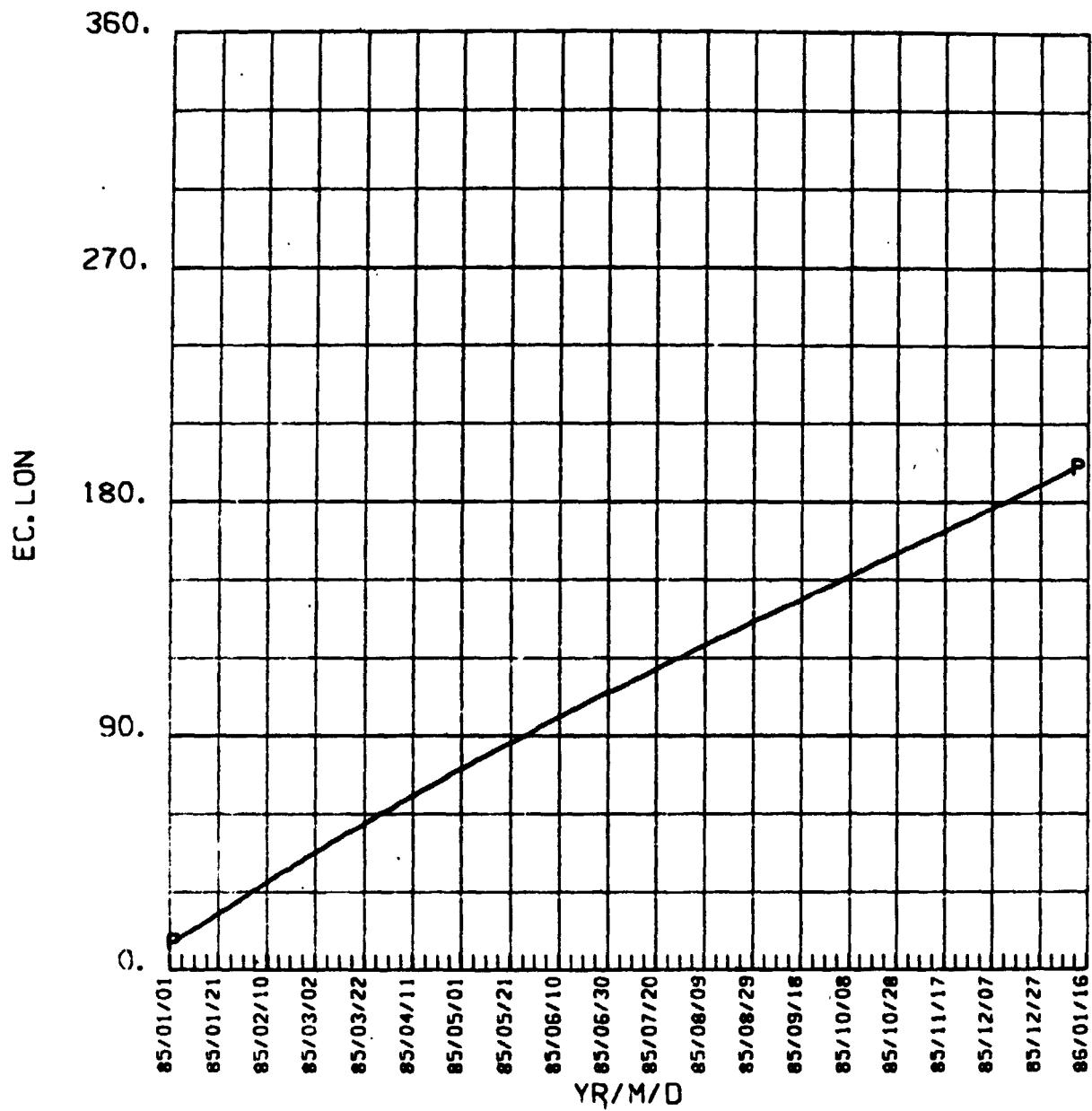
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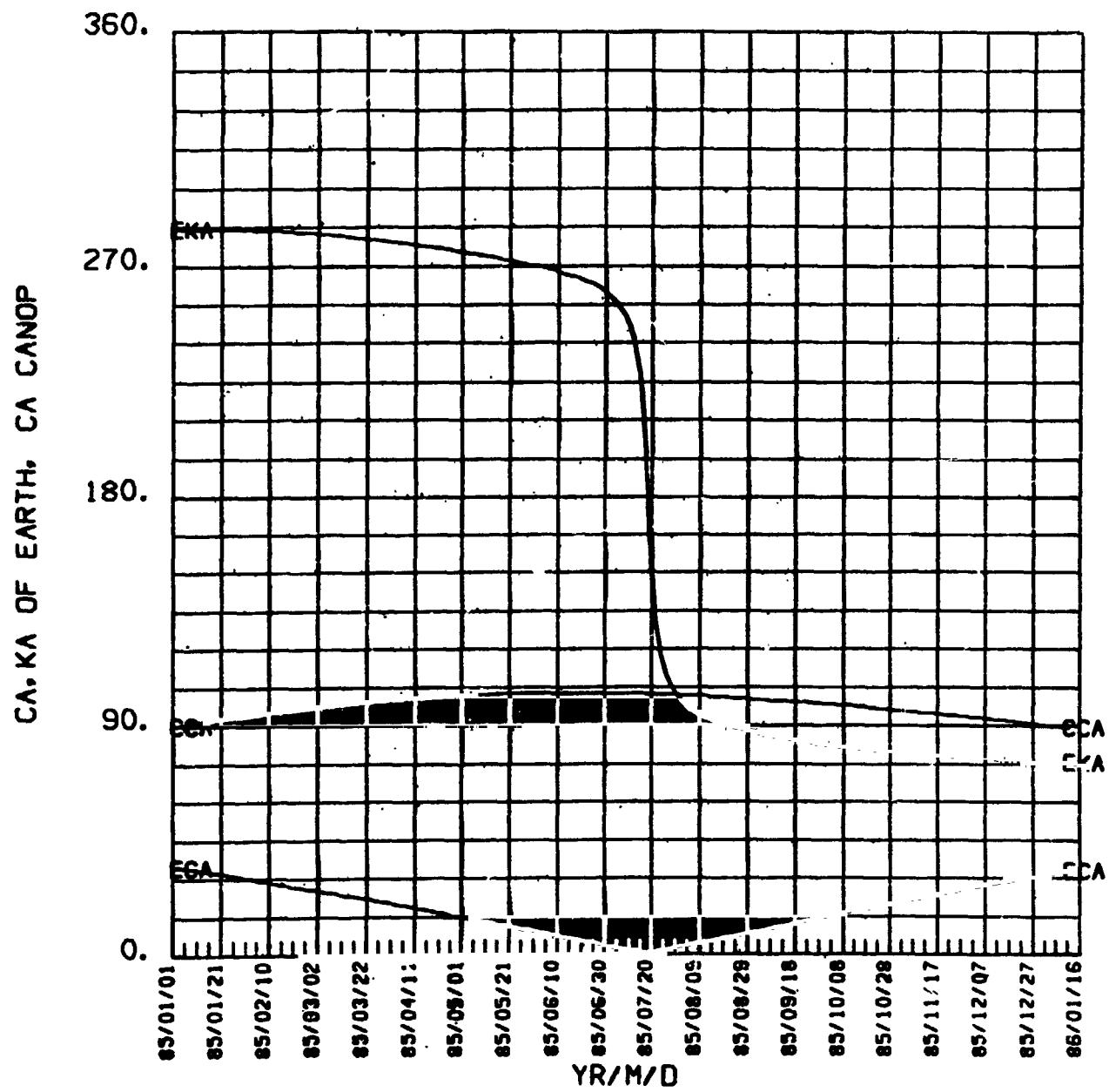
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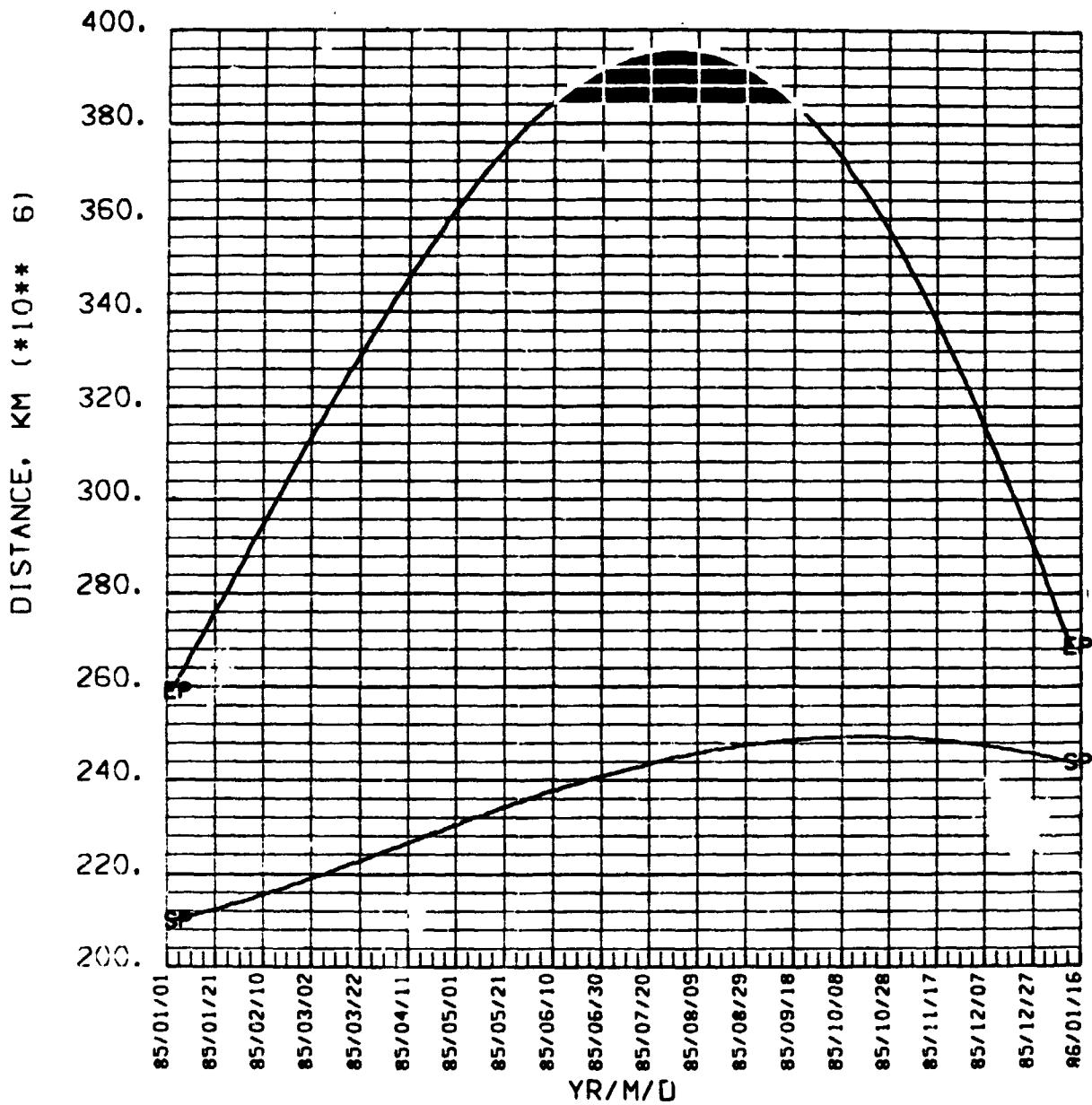
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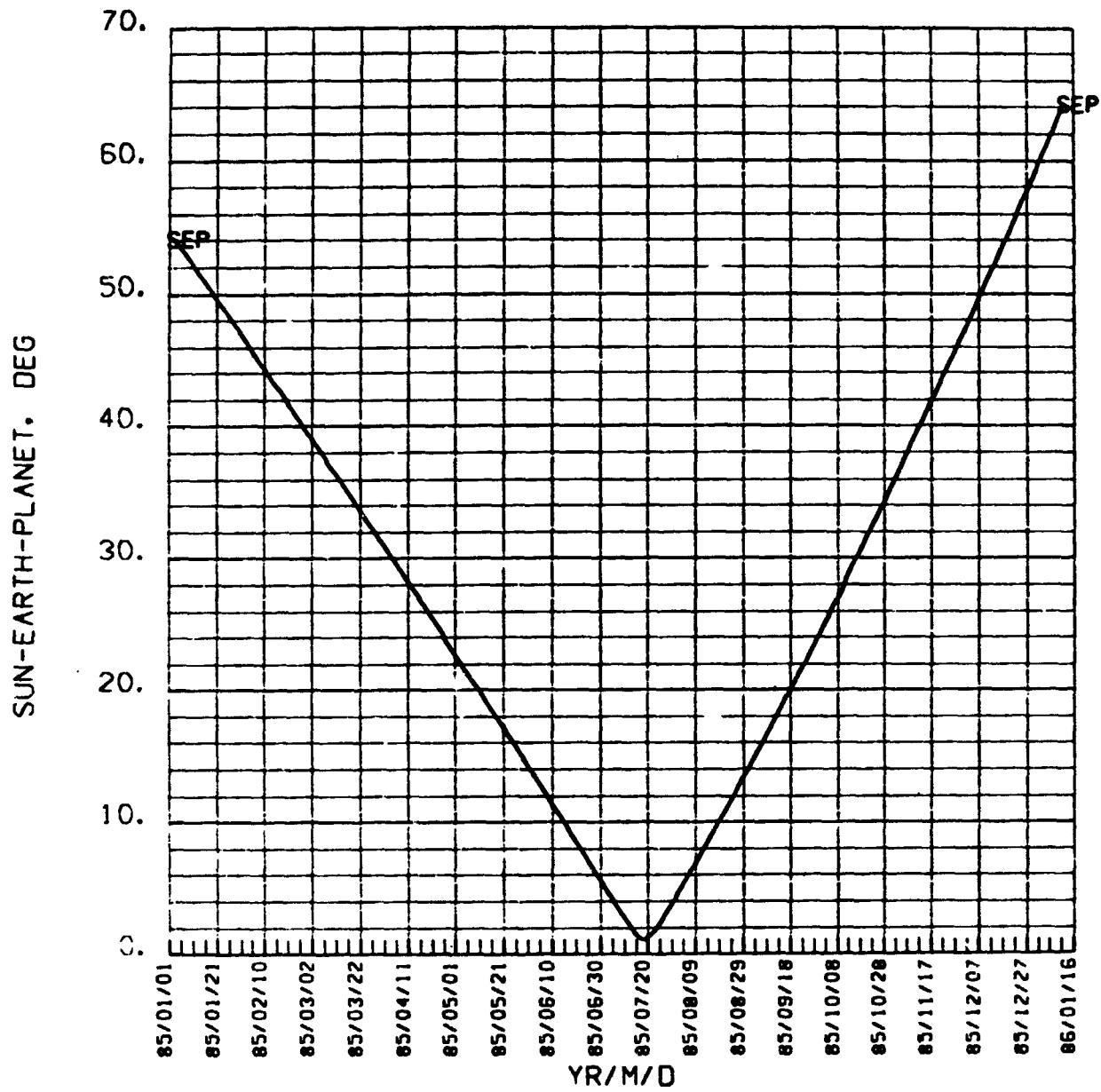
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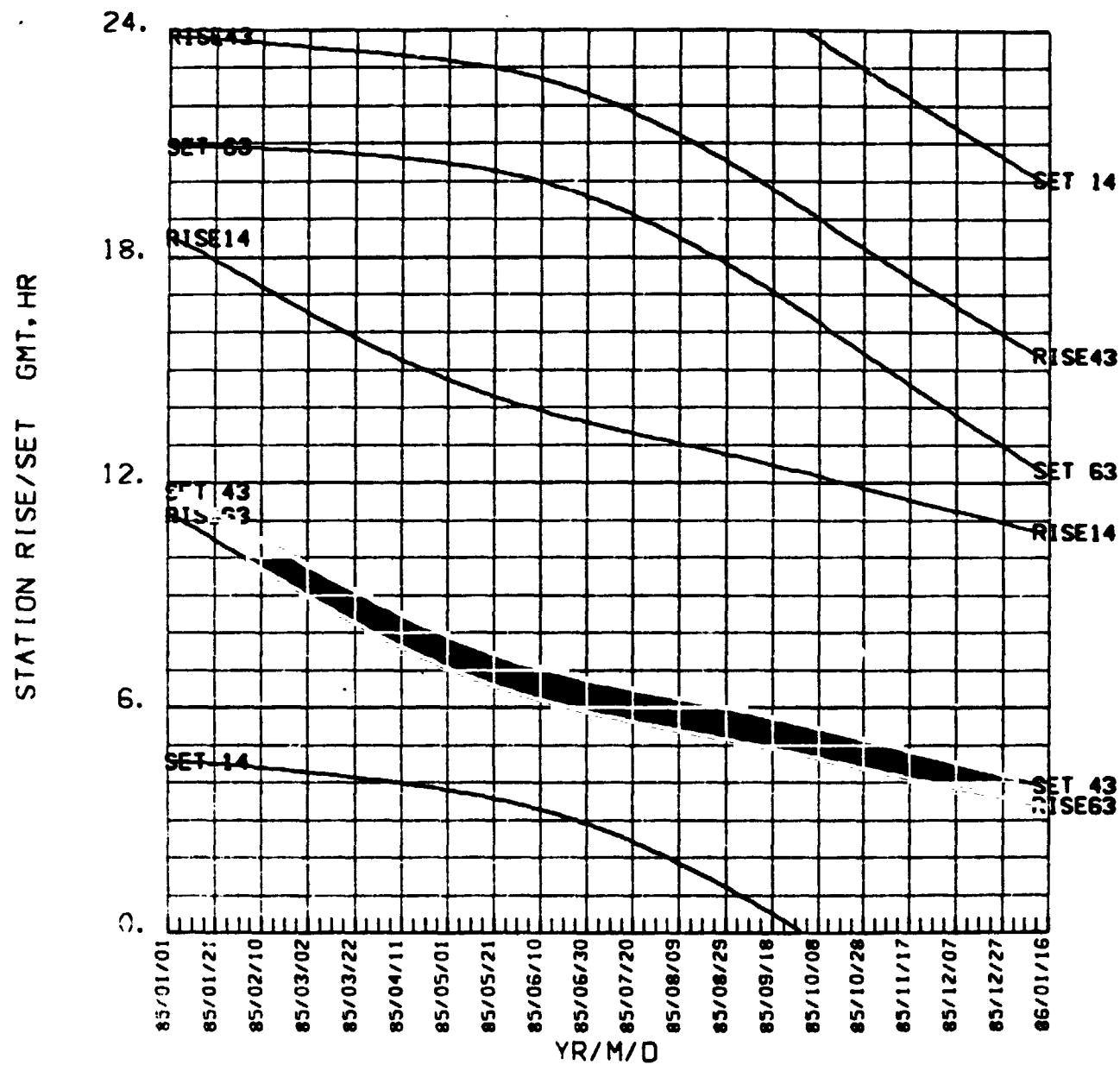
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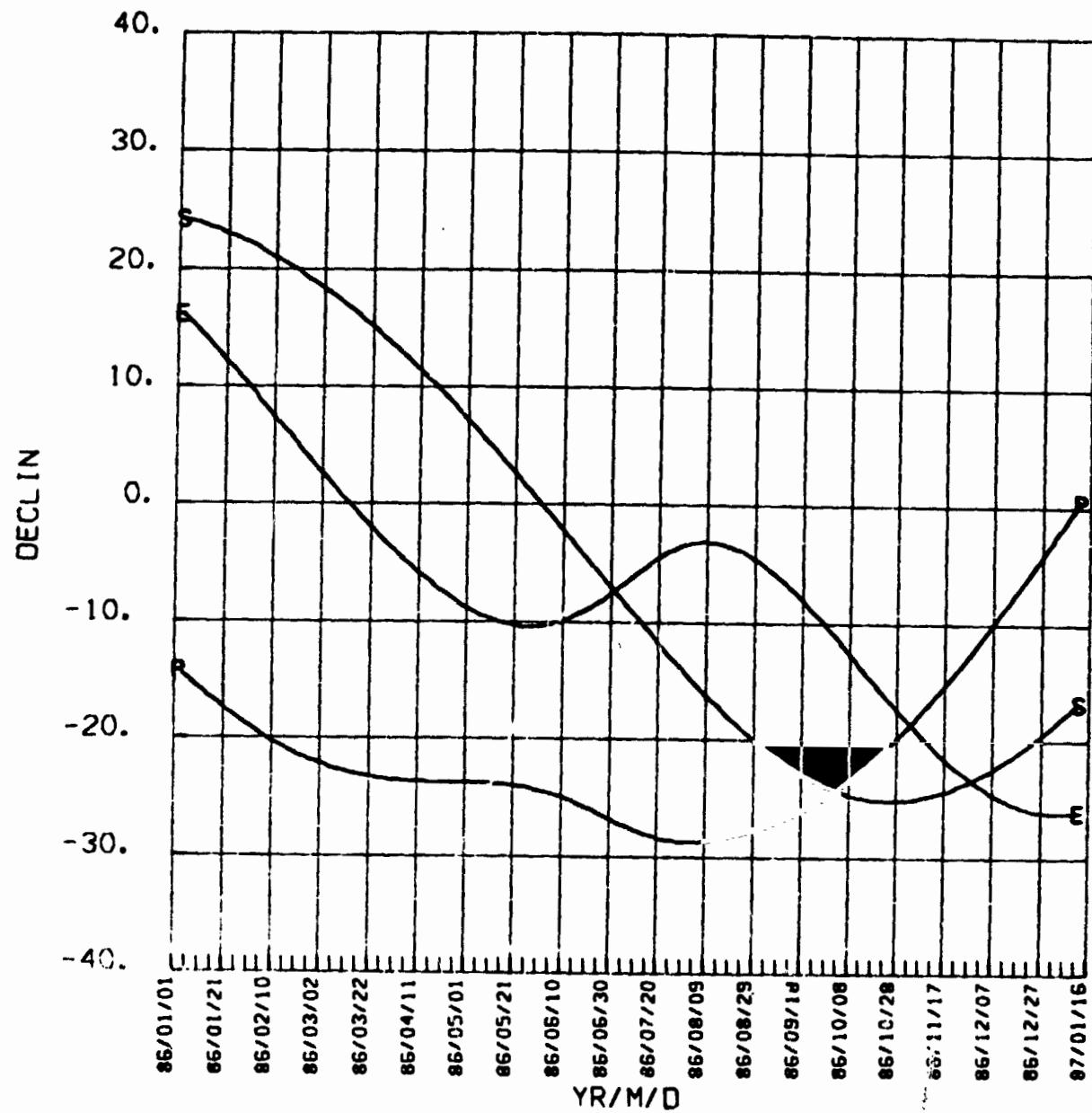
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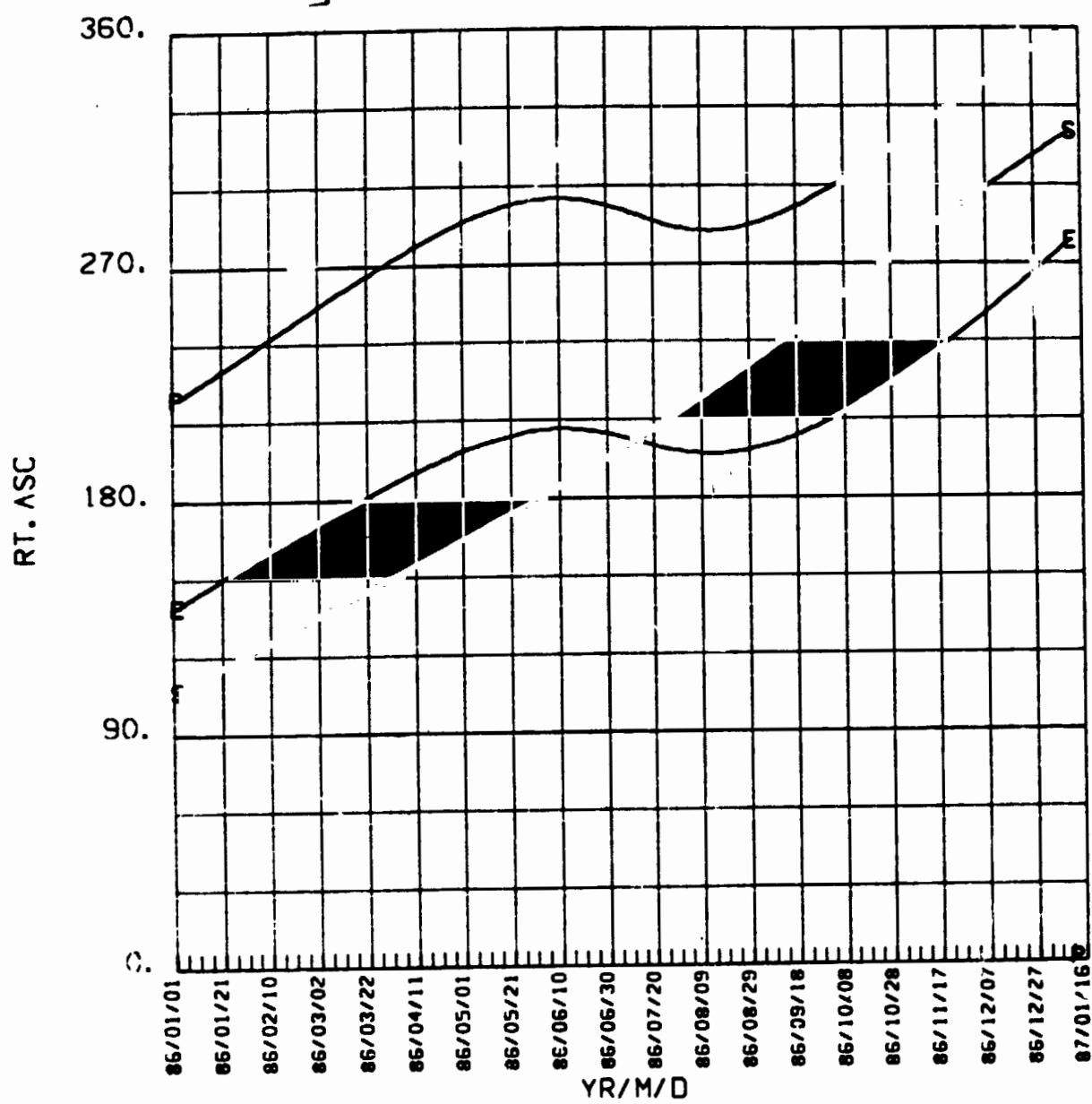
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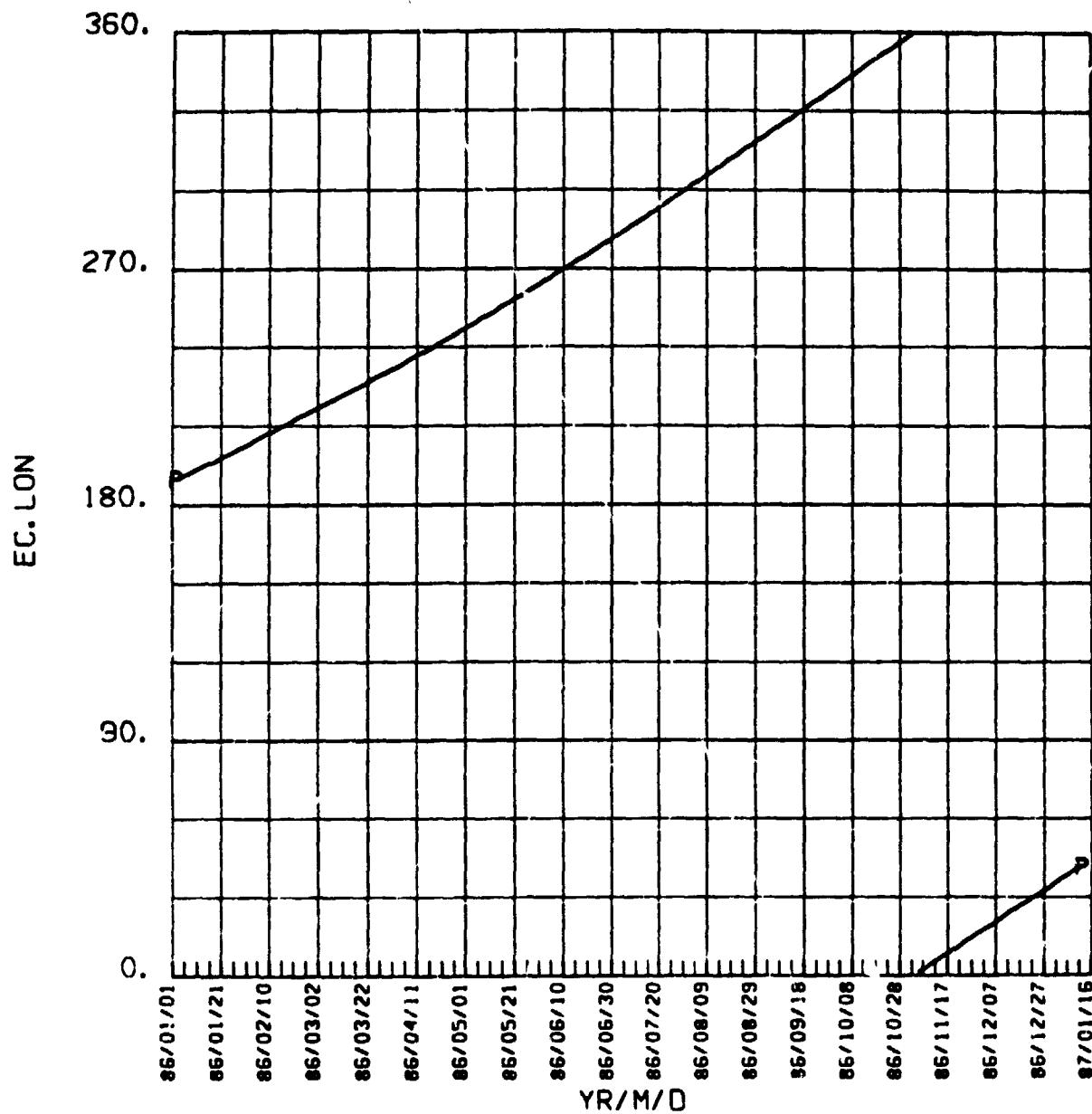
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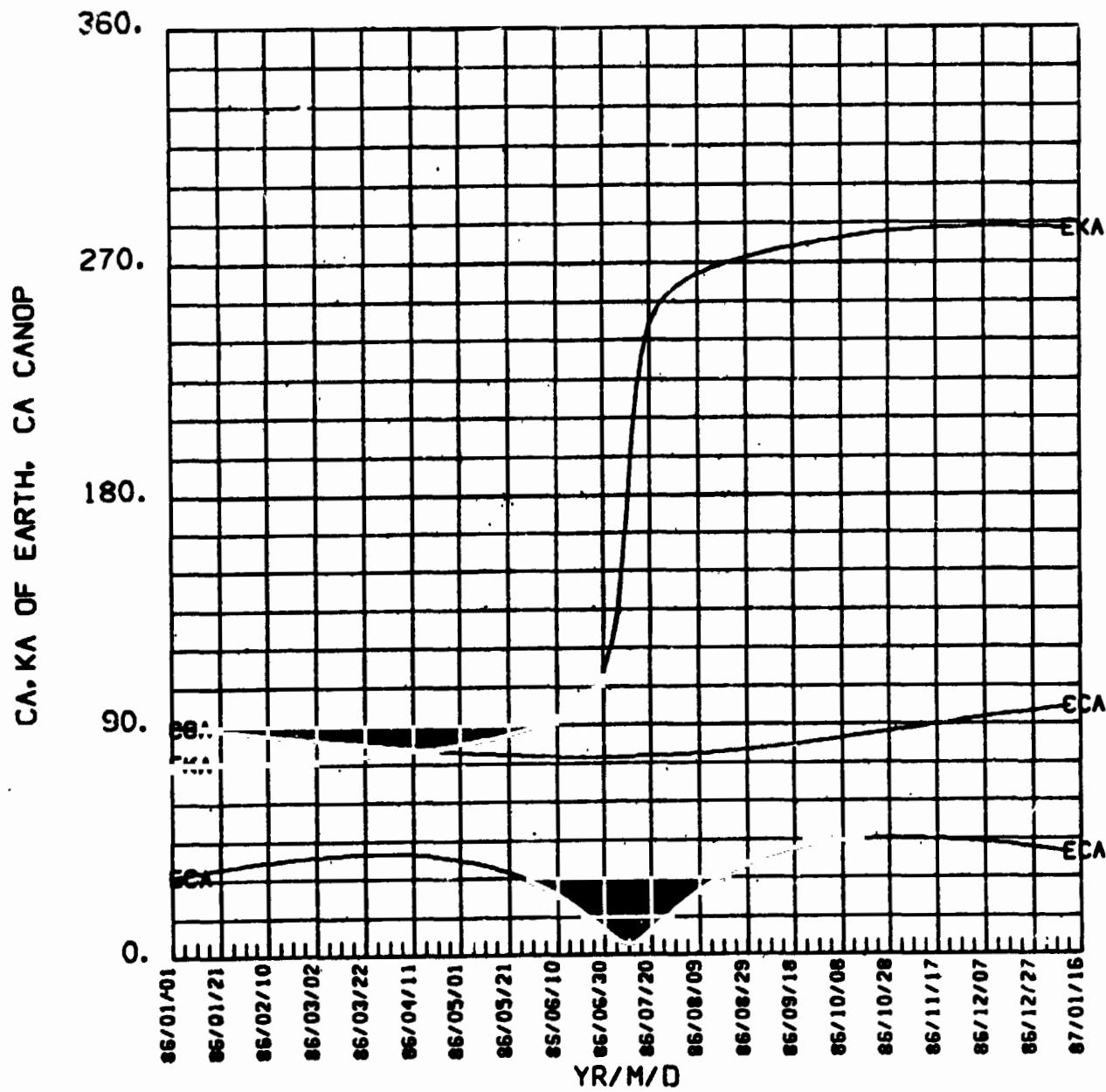
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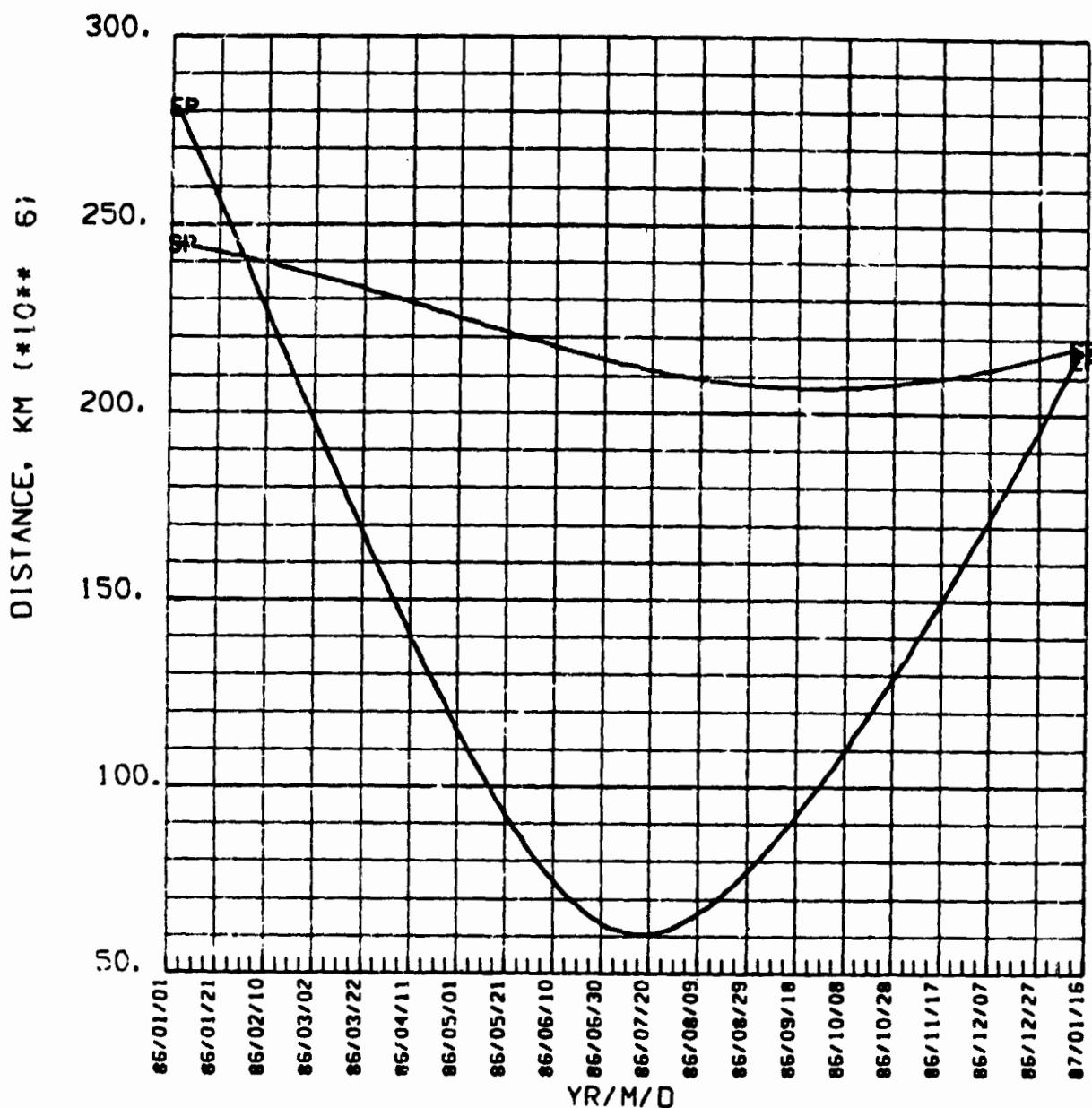
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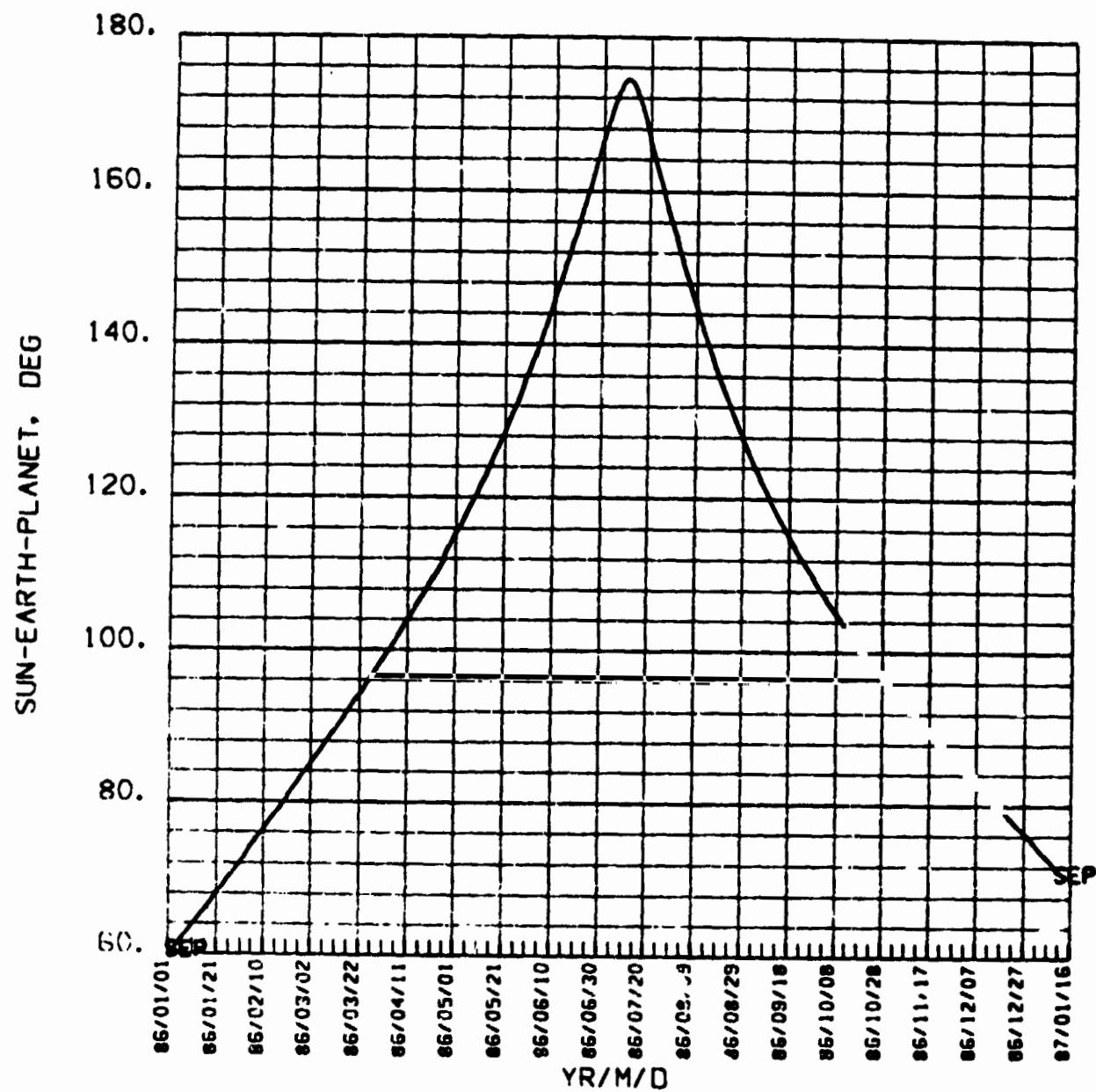
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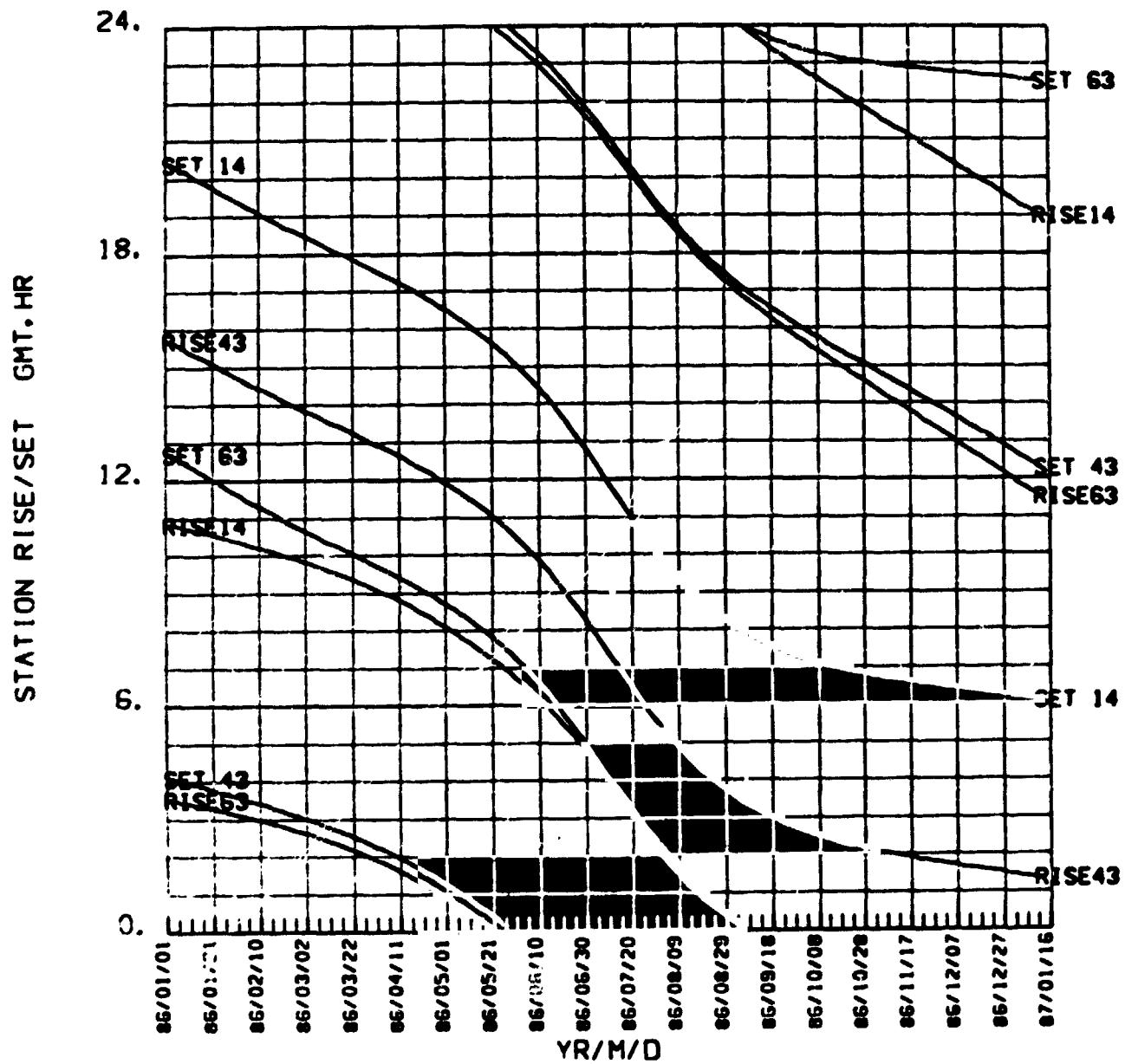
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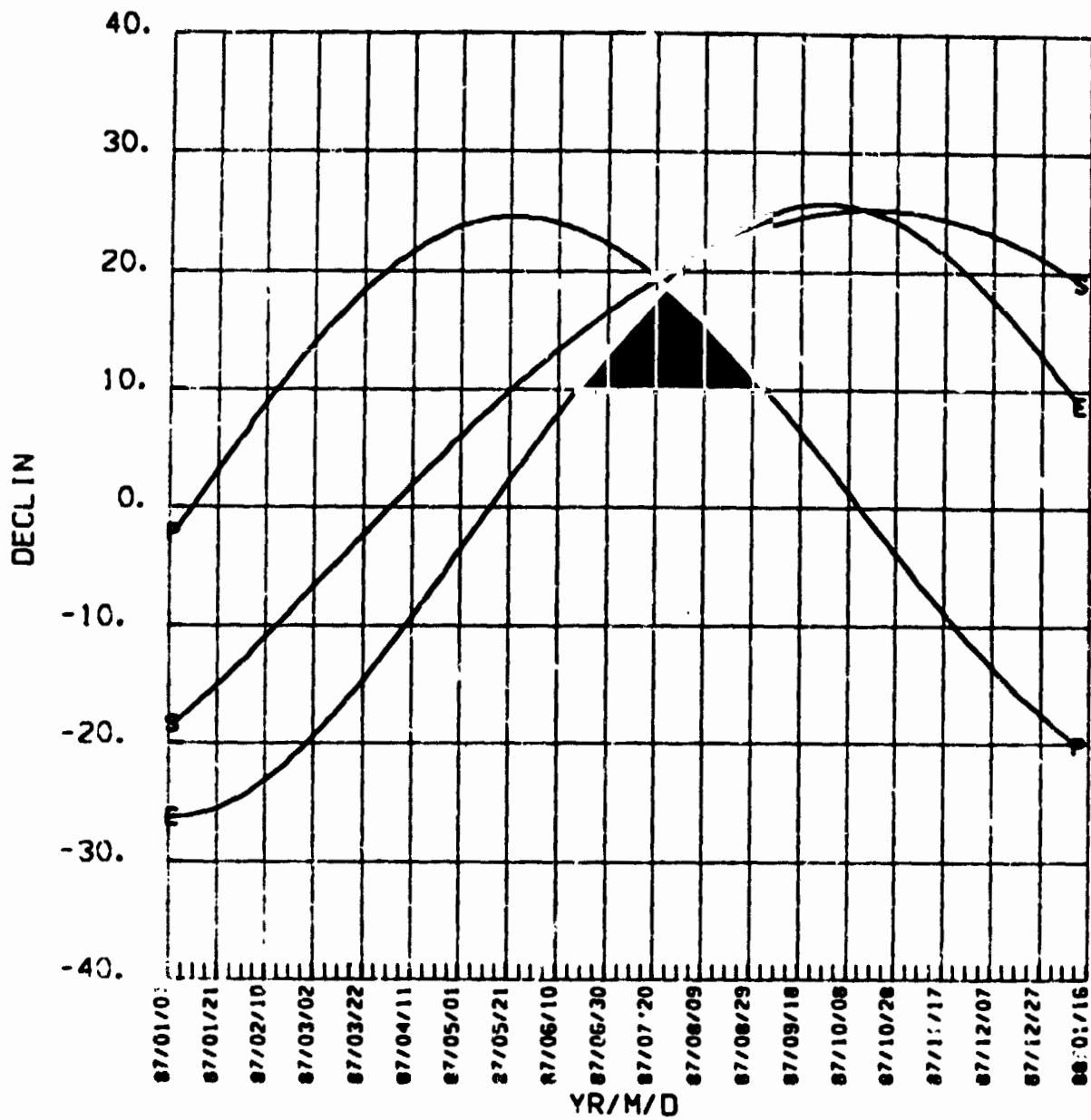
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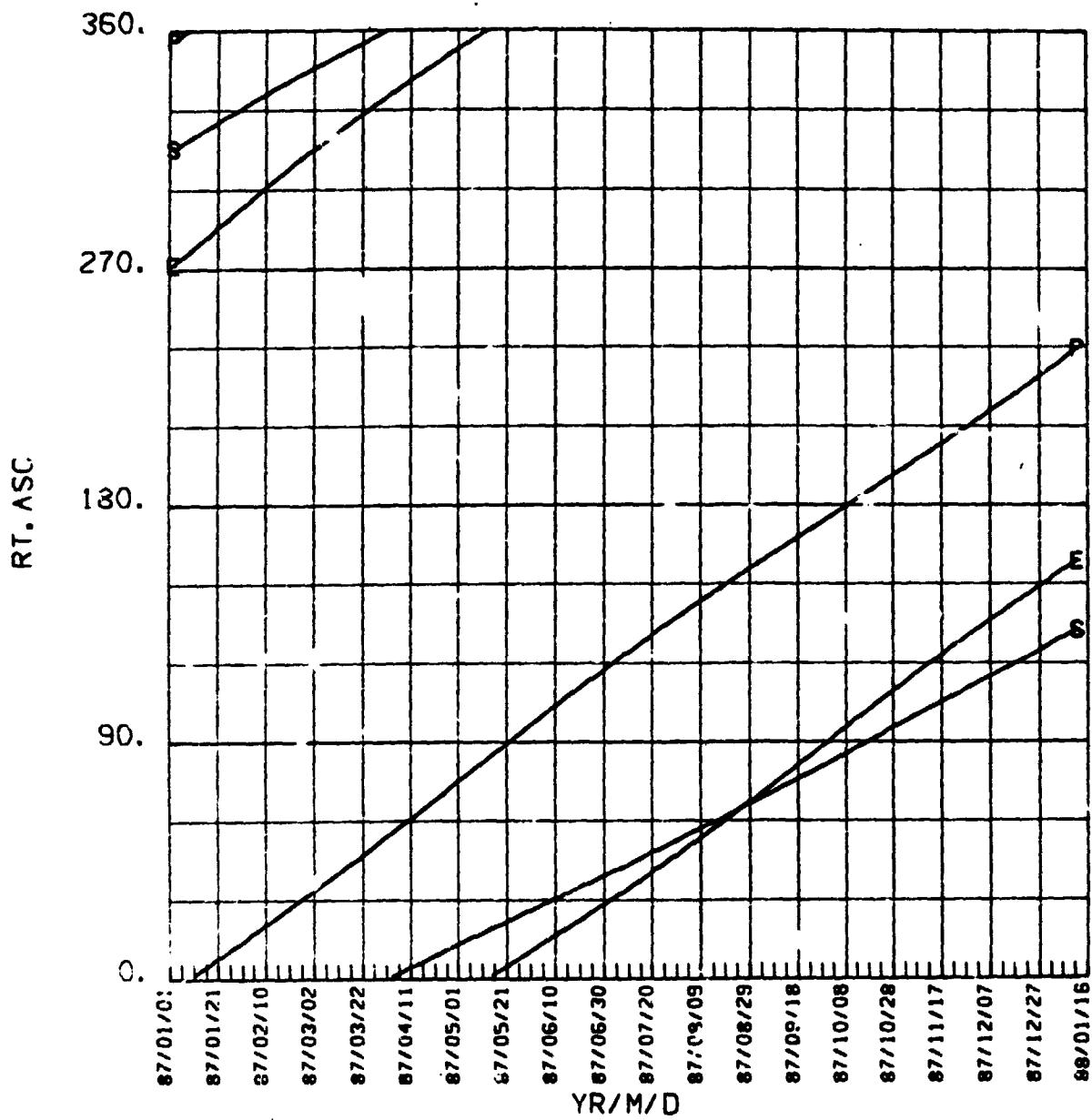
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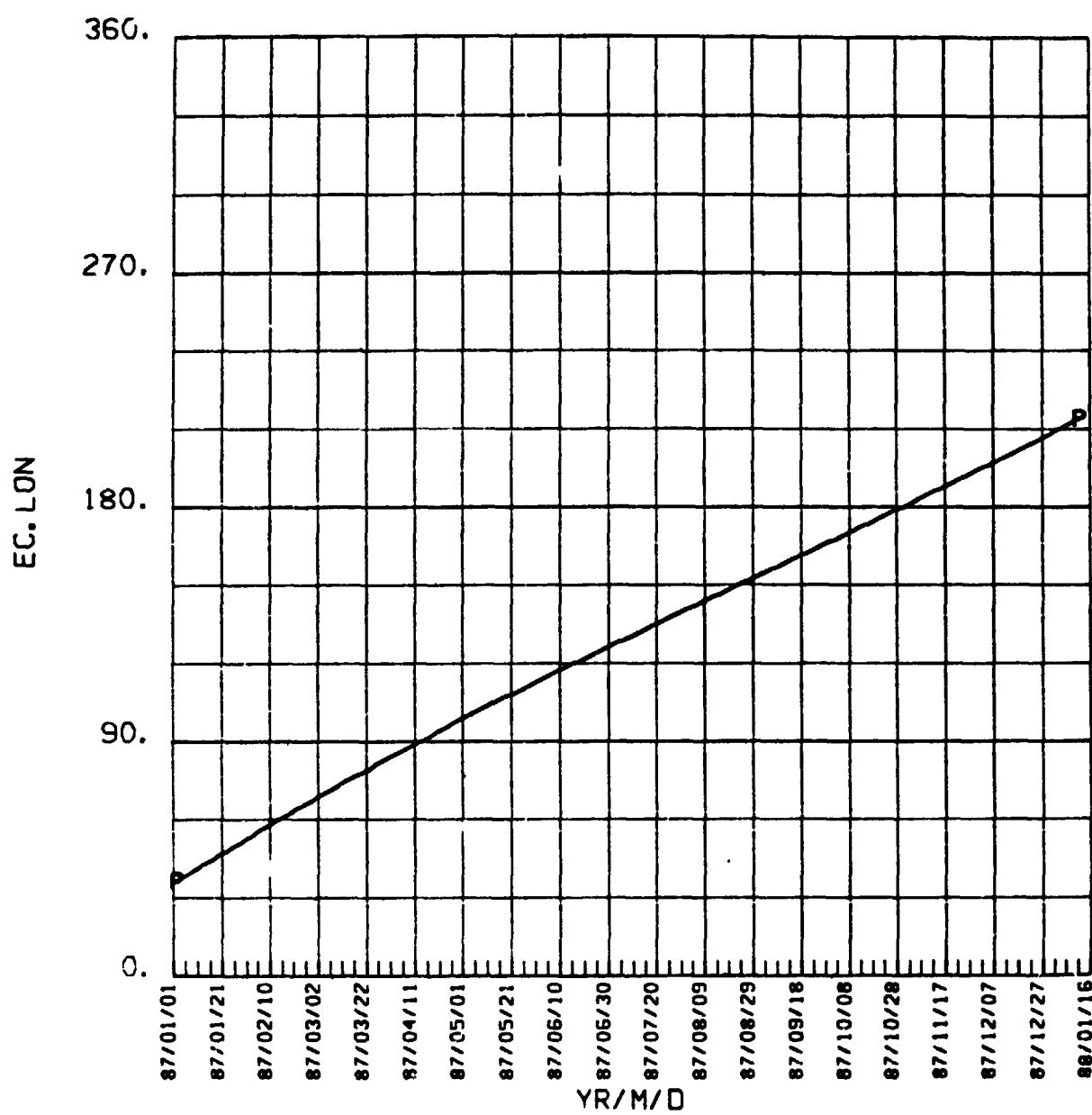
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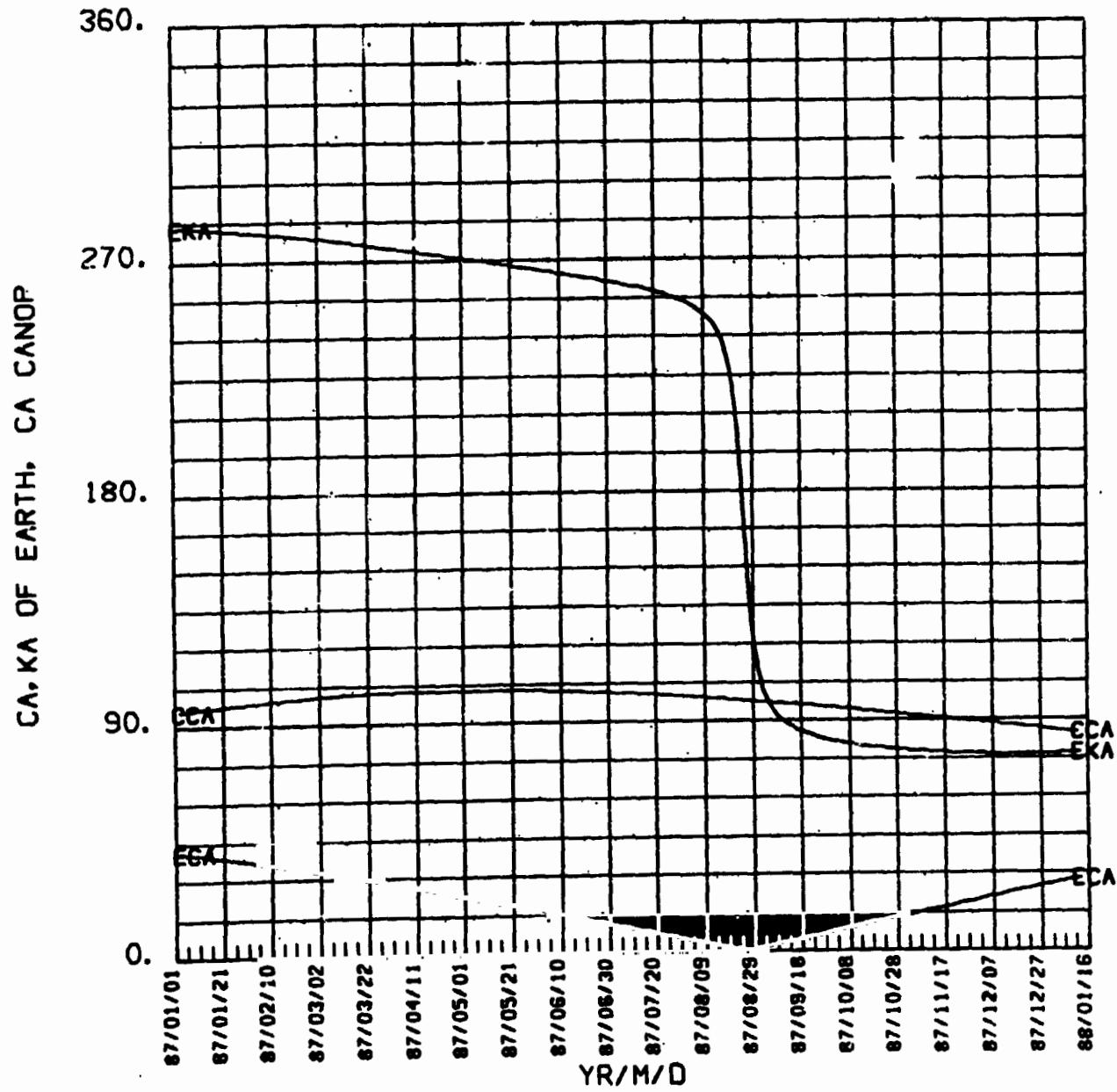
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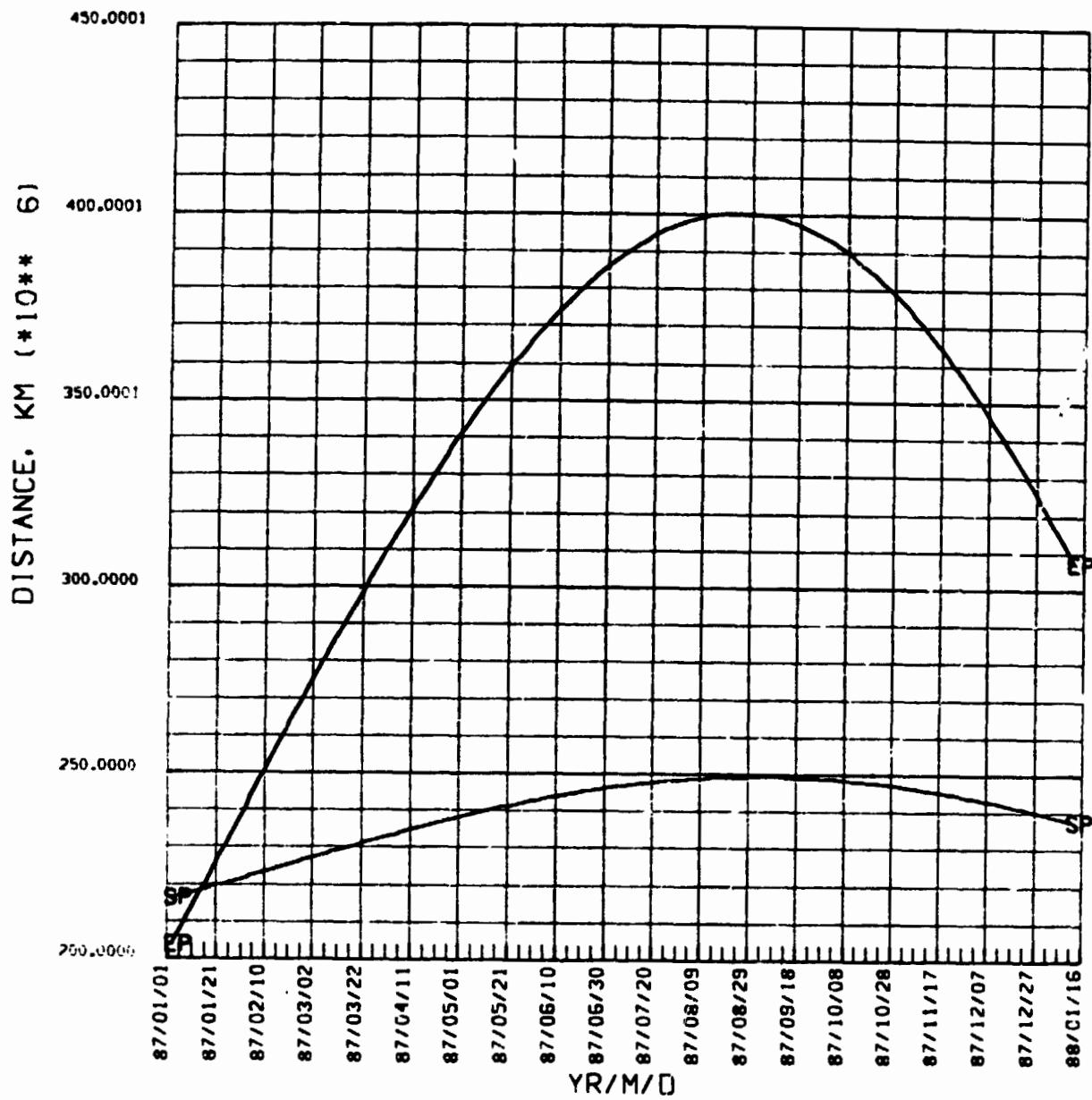
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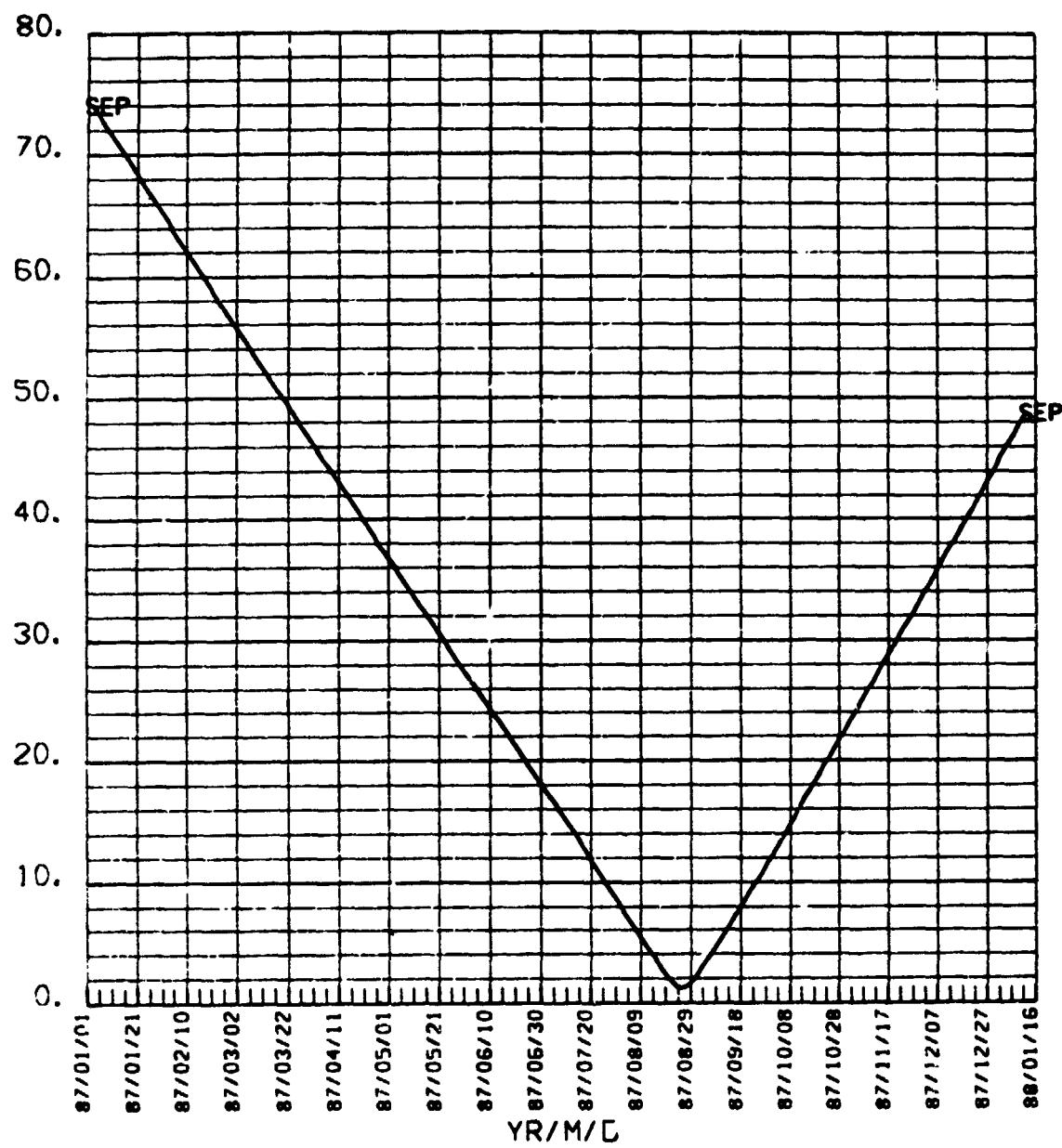


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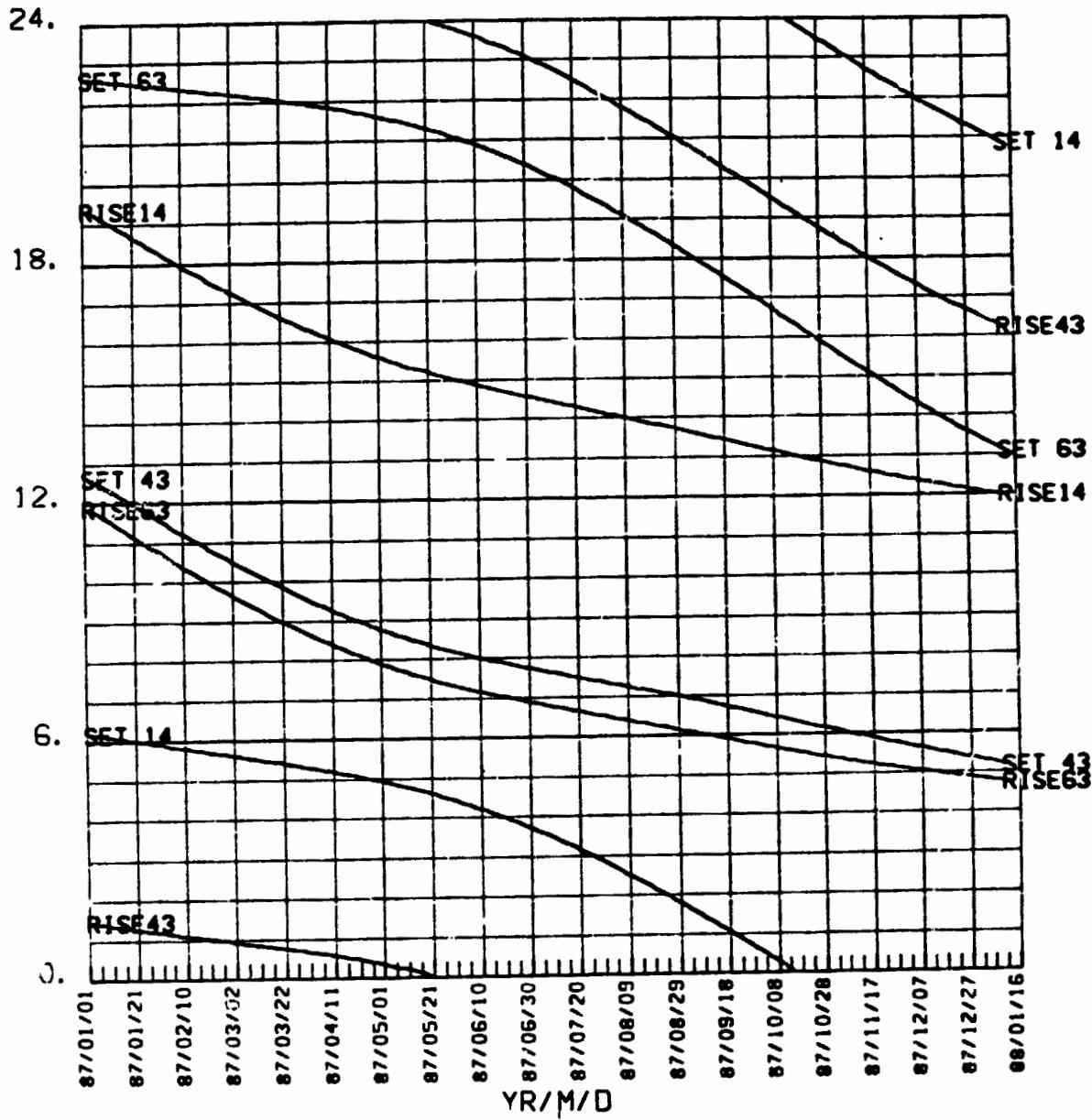
SUN-EARTH-PLANET. DEG

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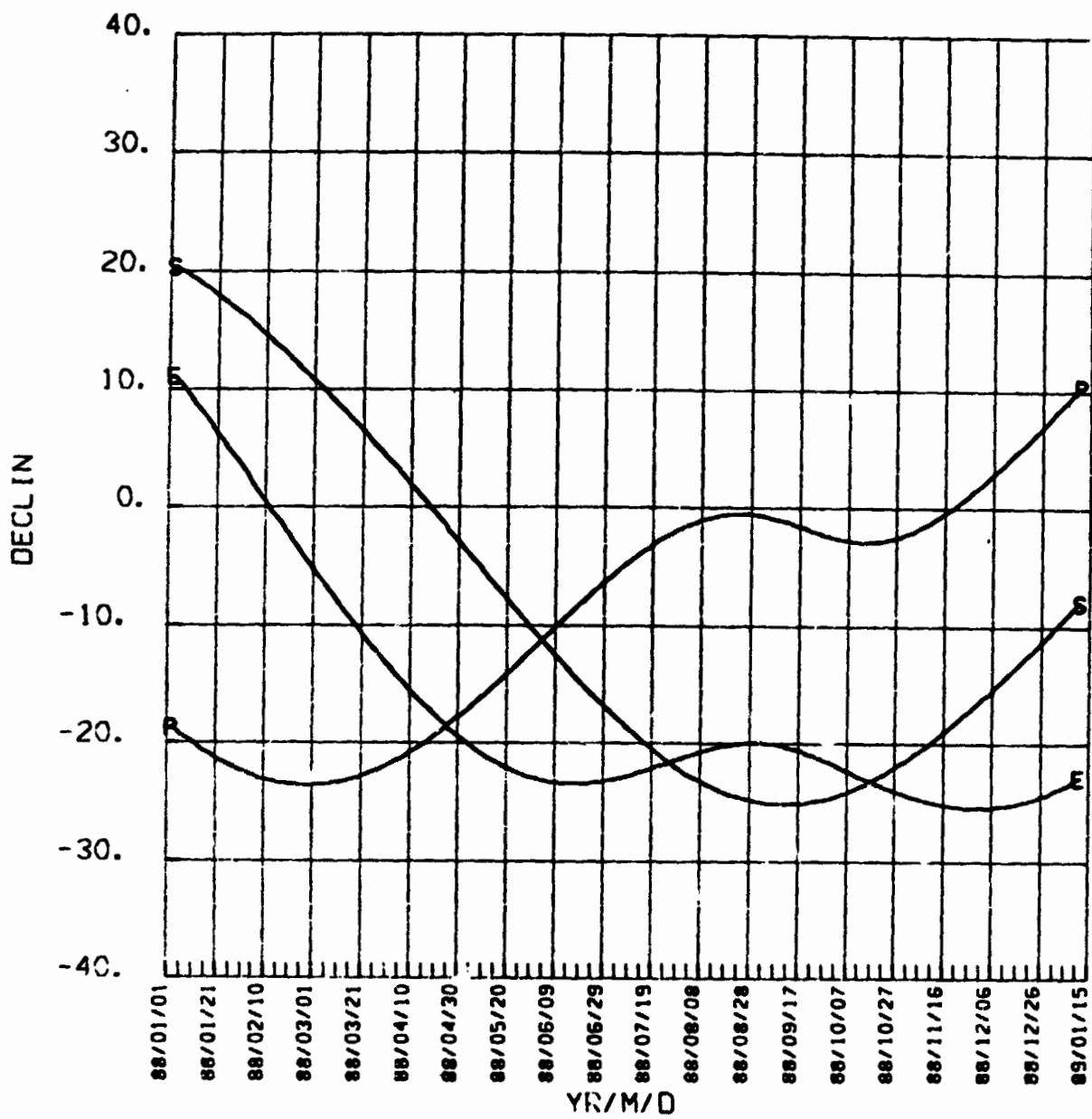


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STATION RISE/SET GMT. HR

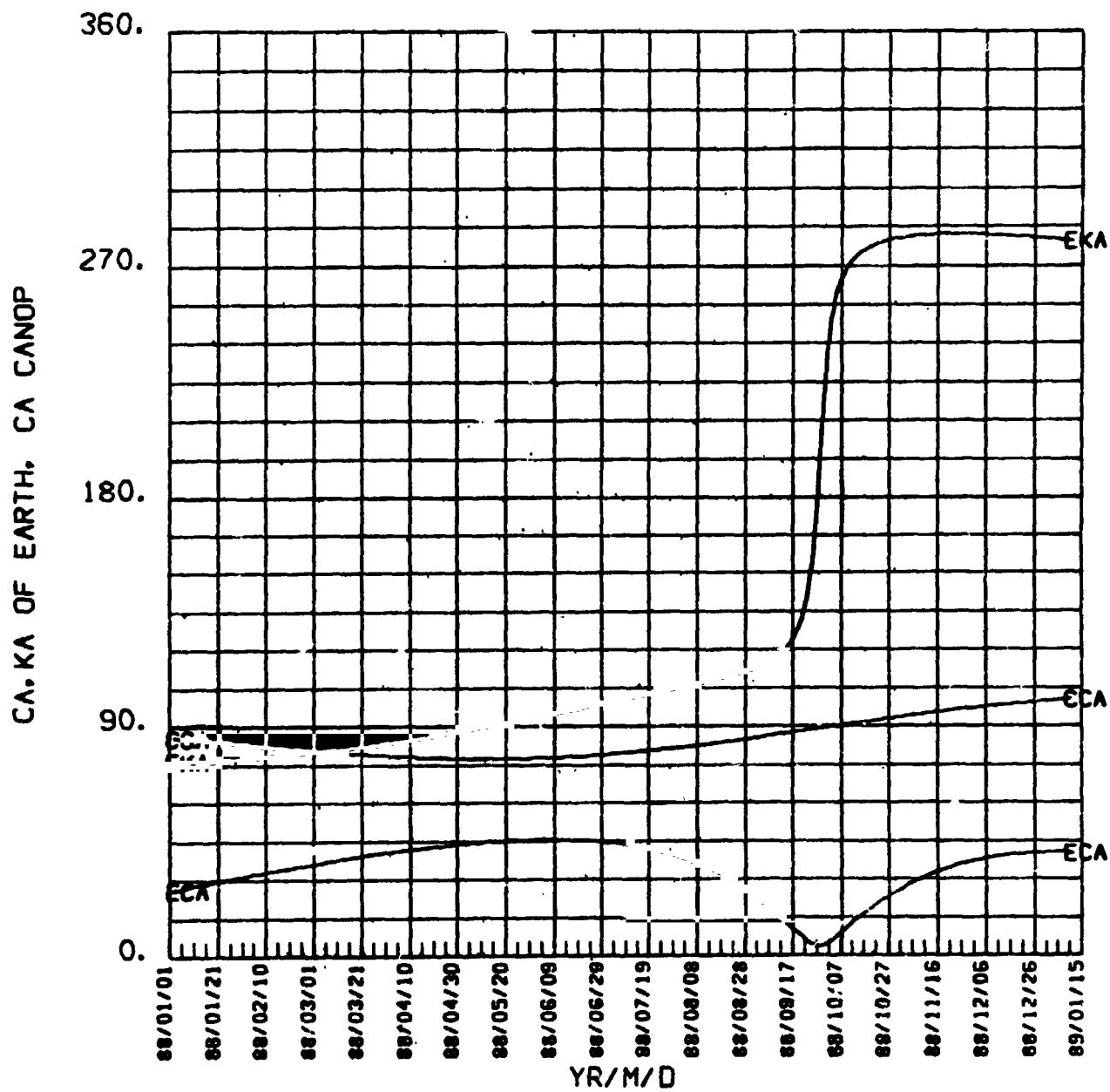


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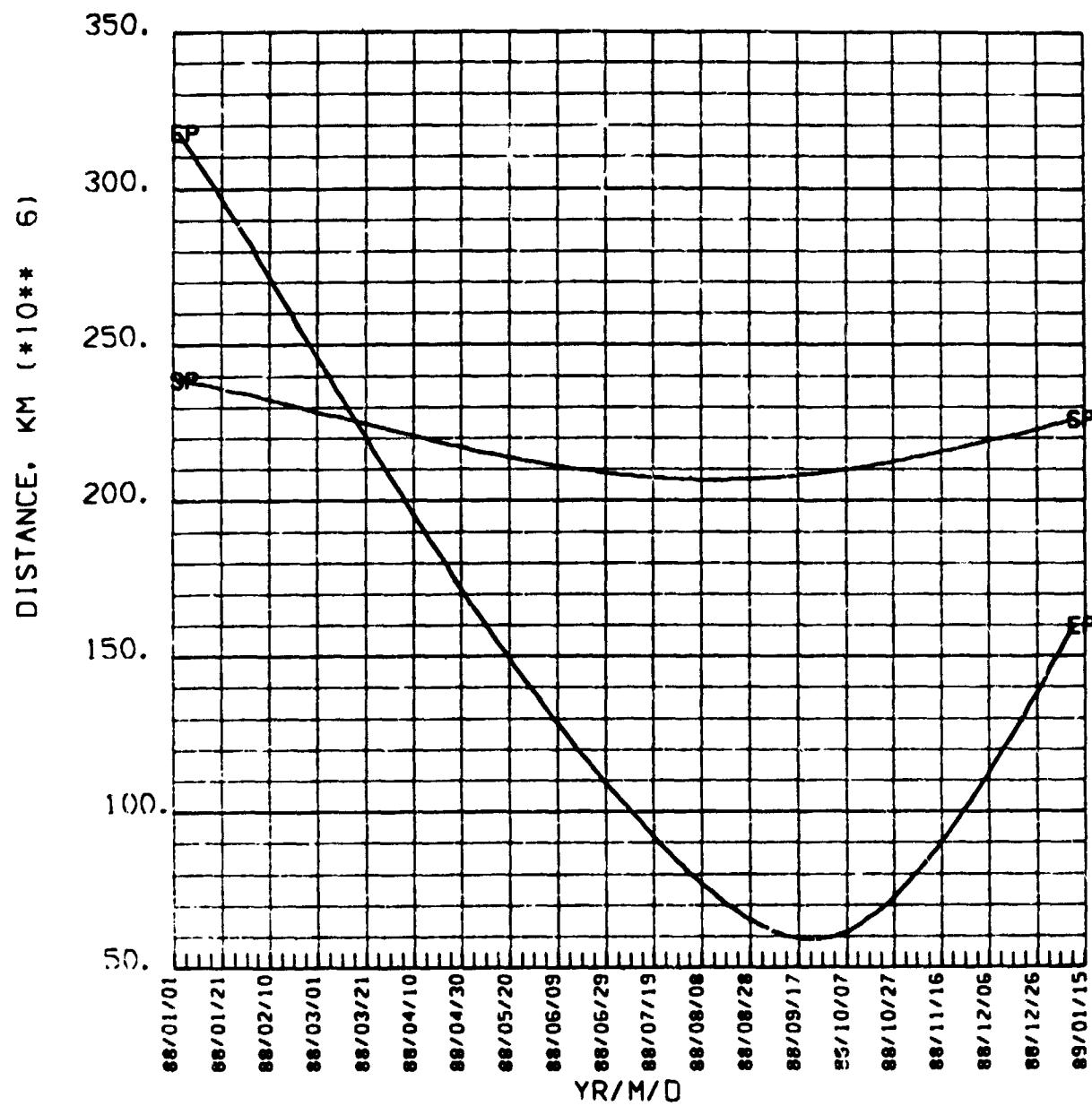
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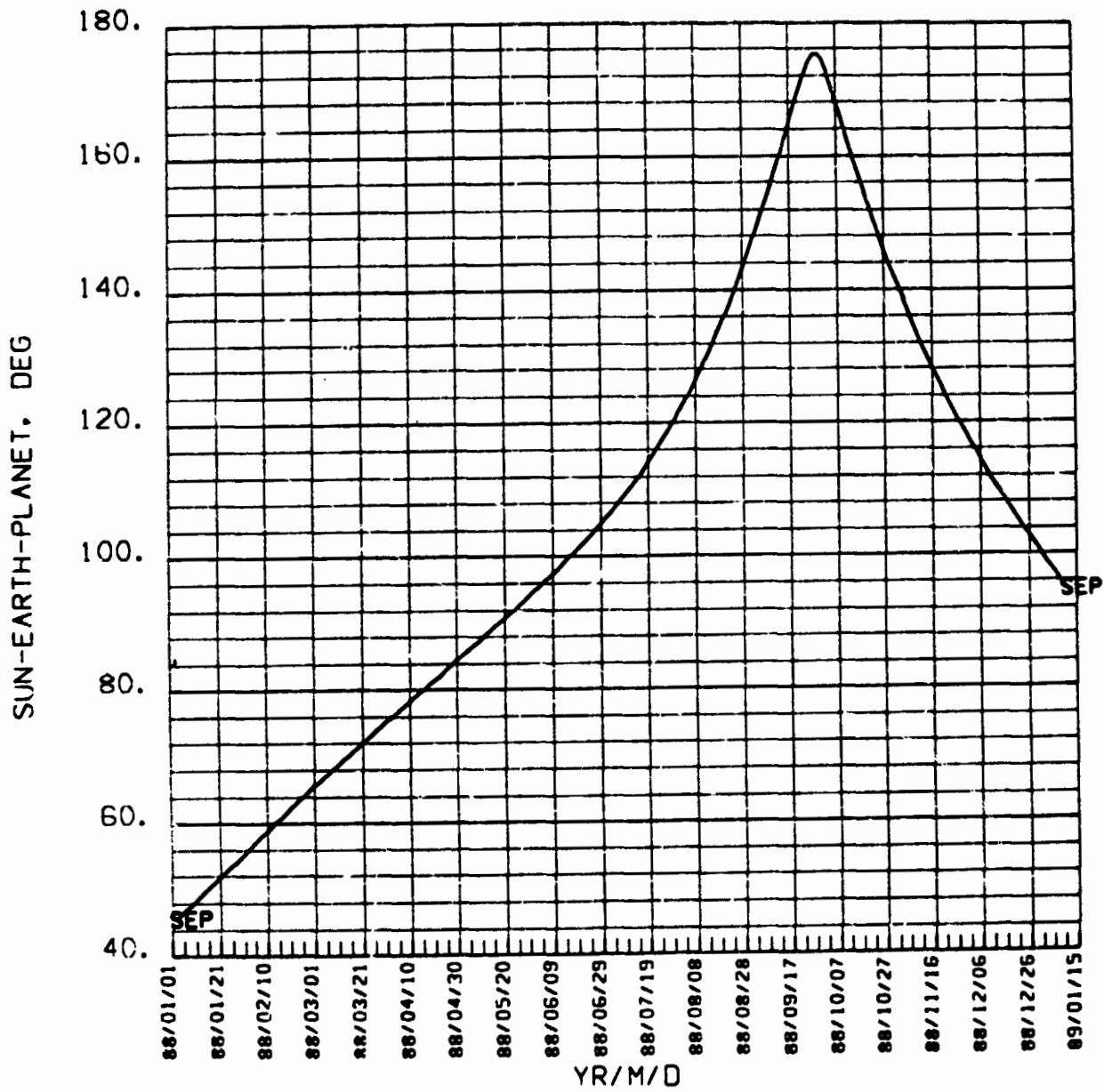
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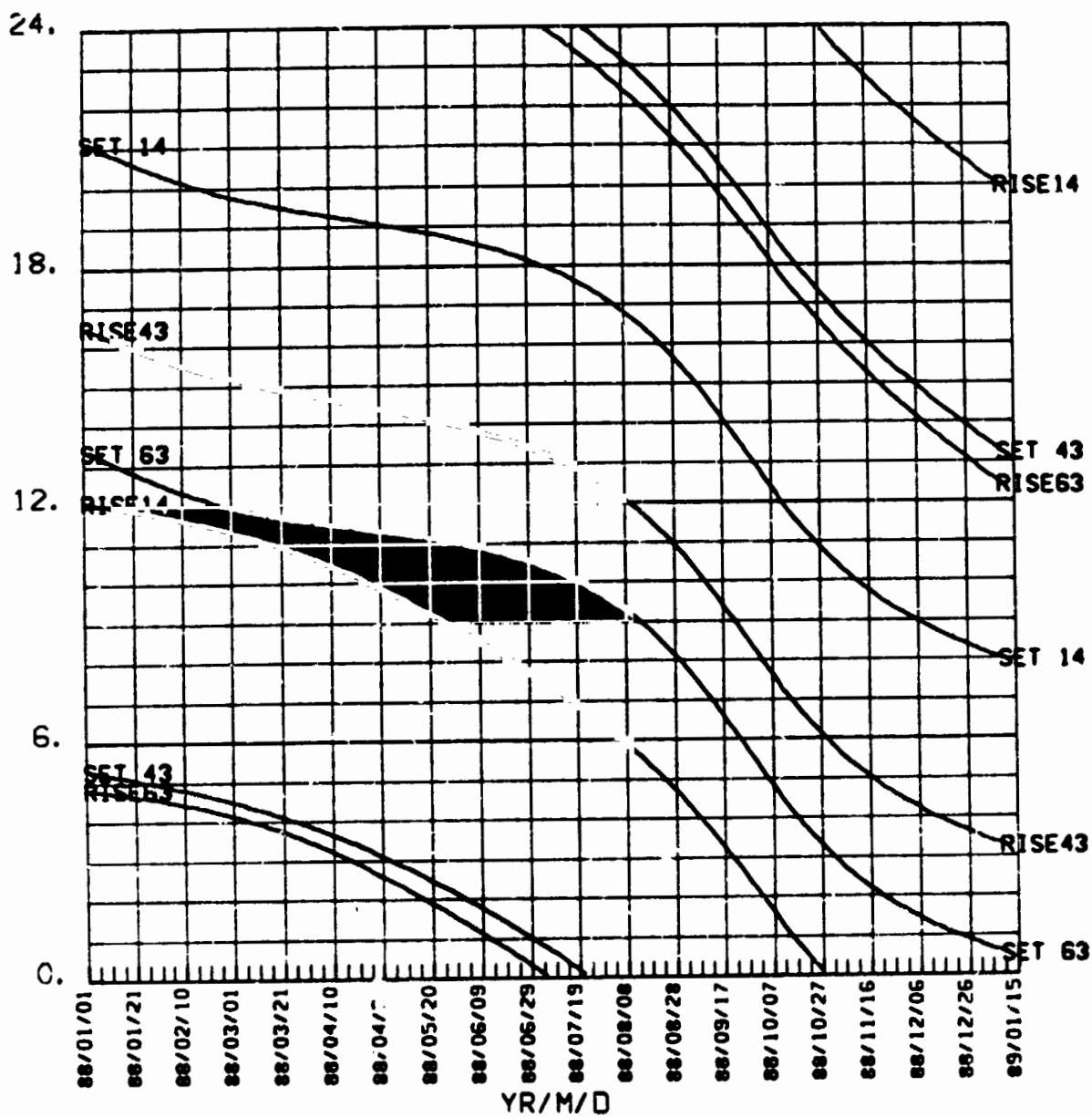


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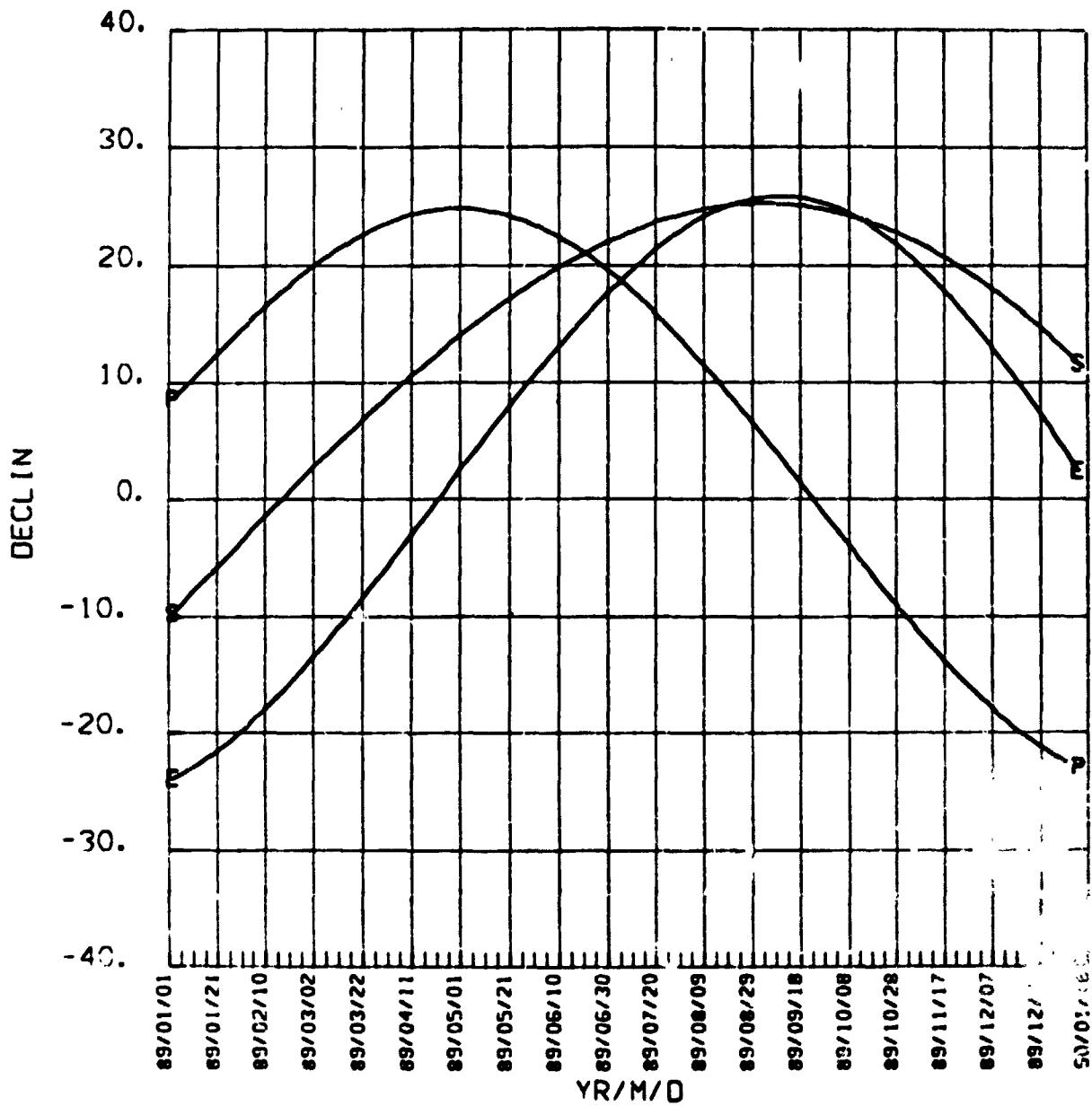


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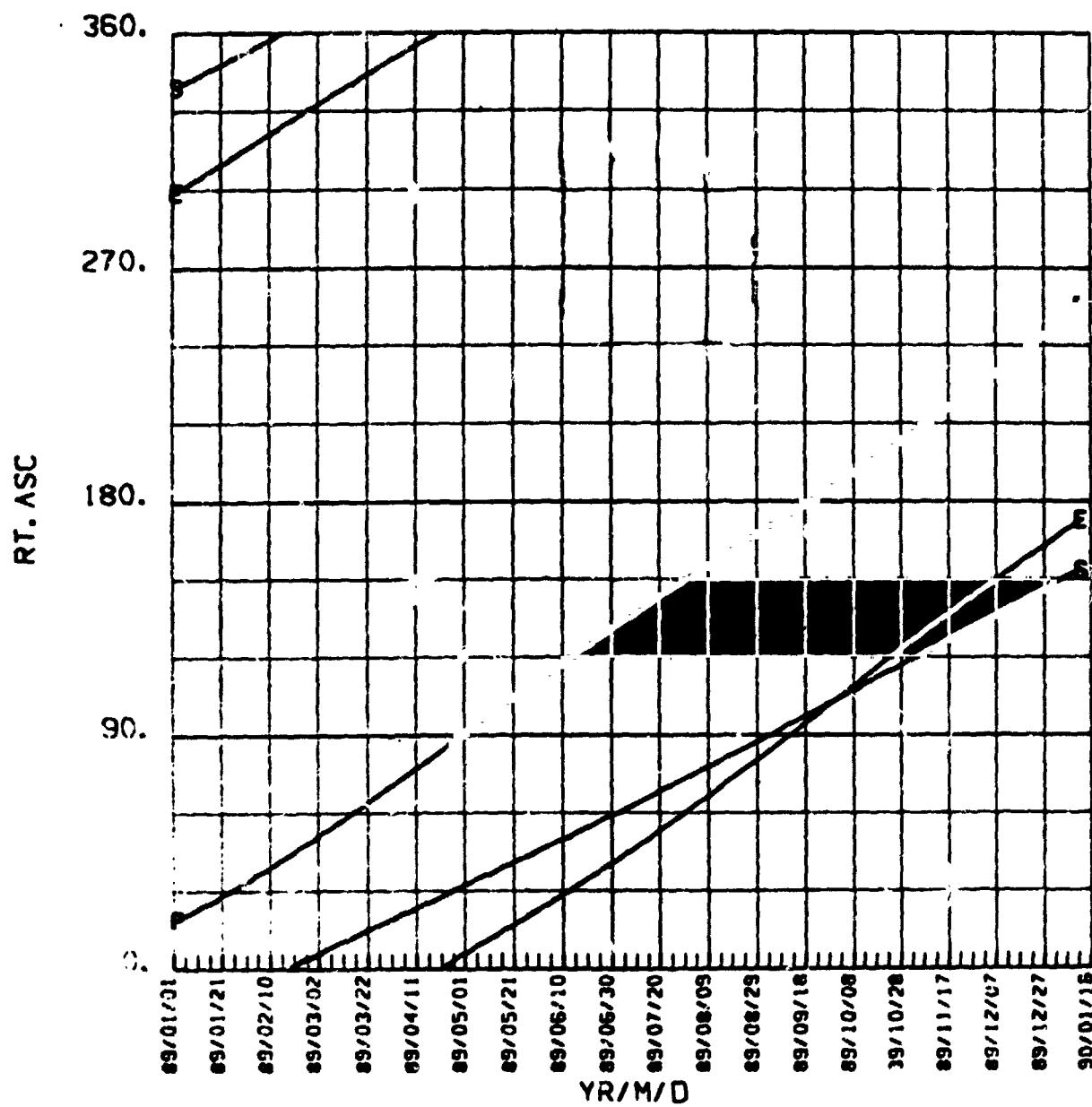
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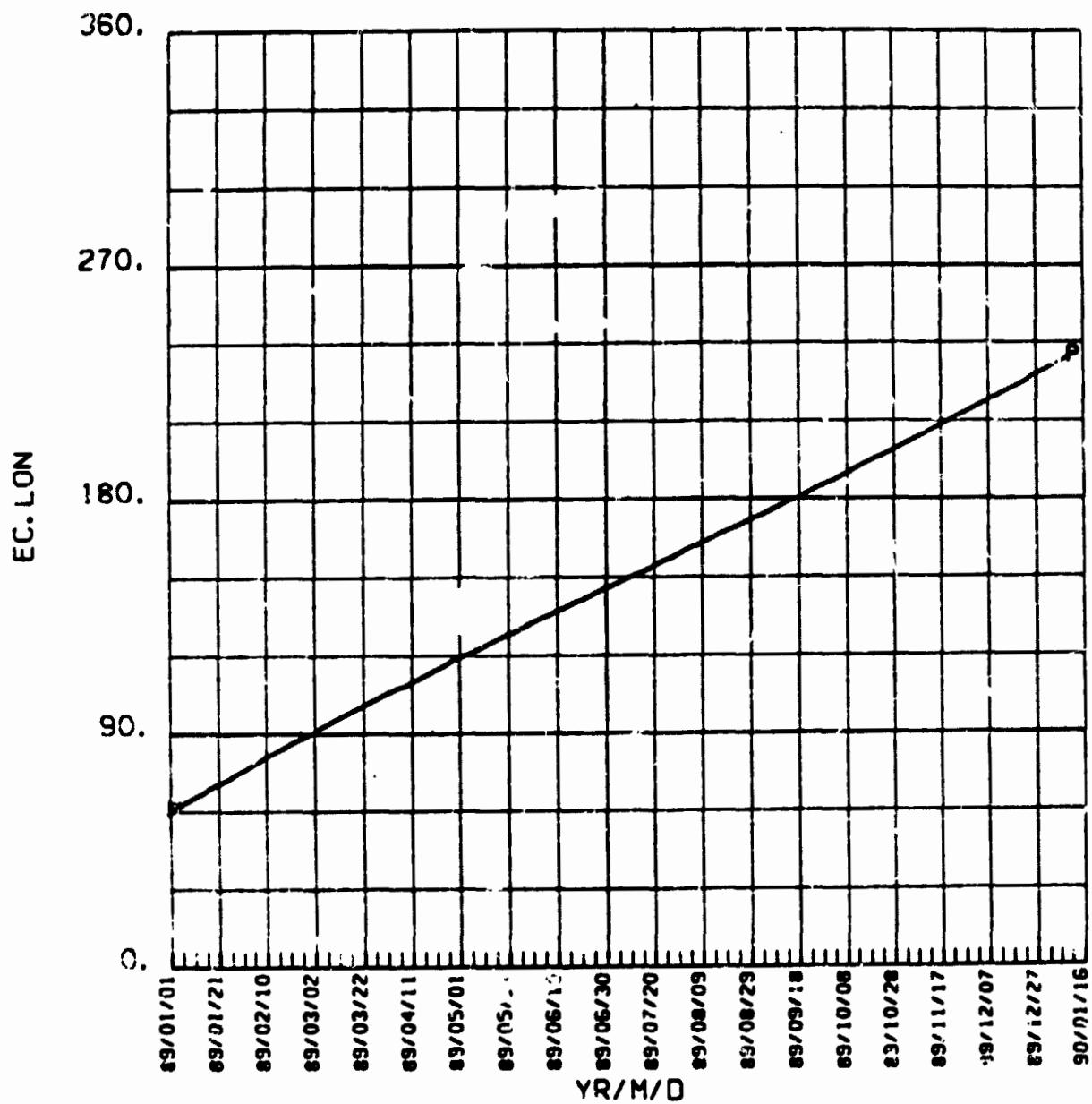
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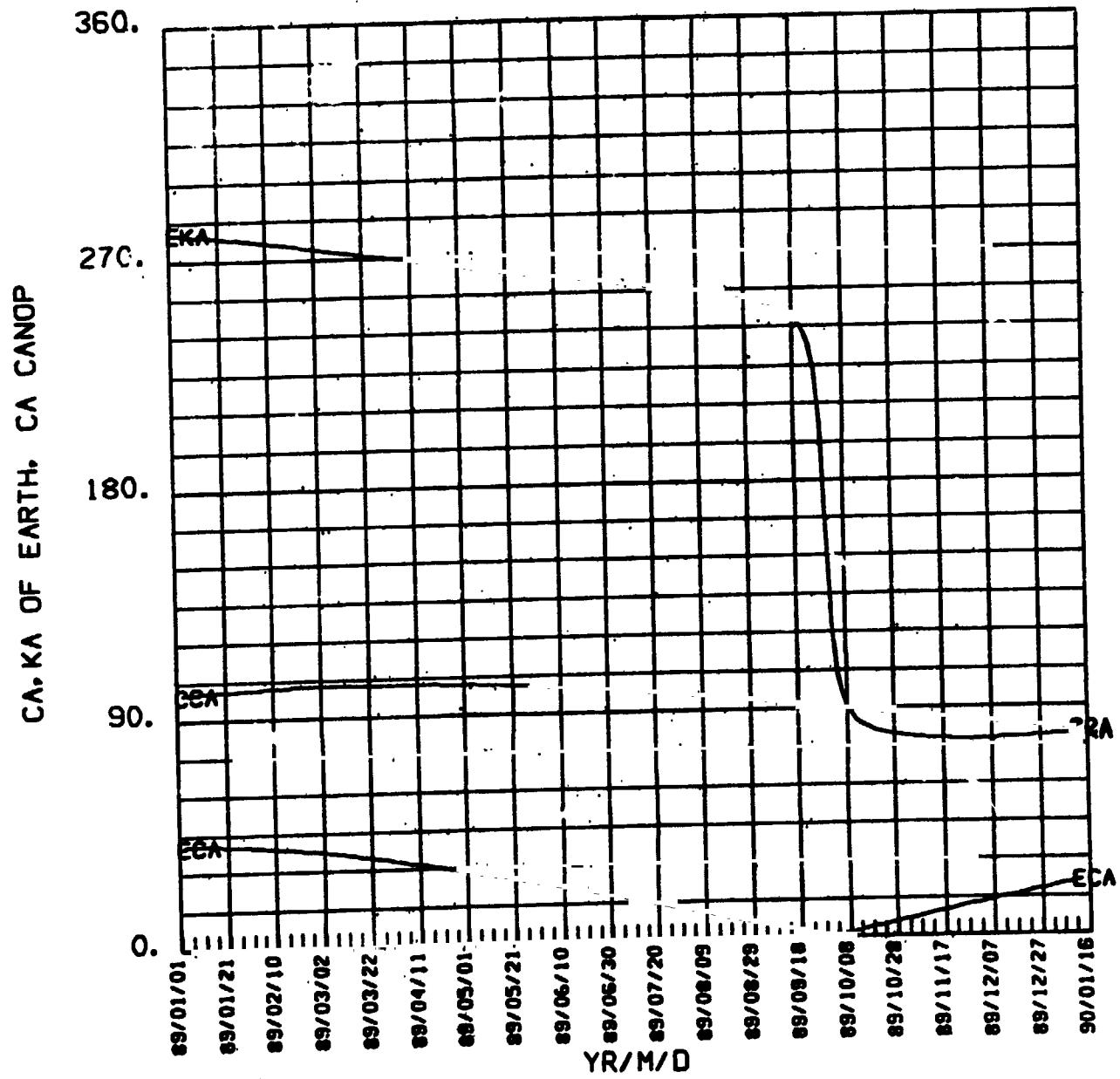


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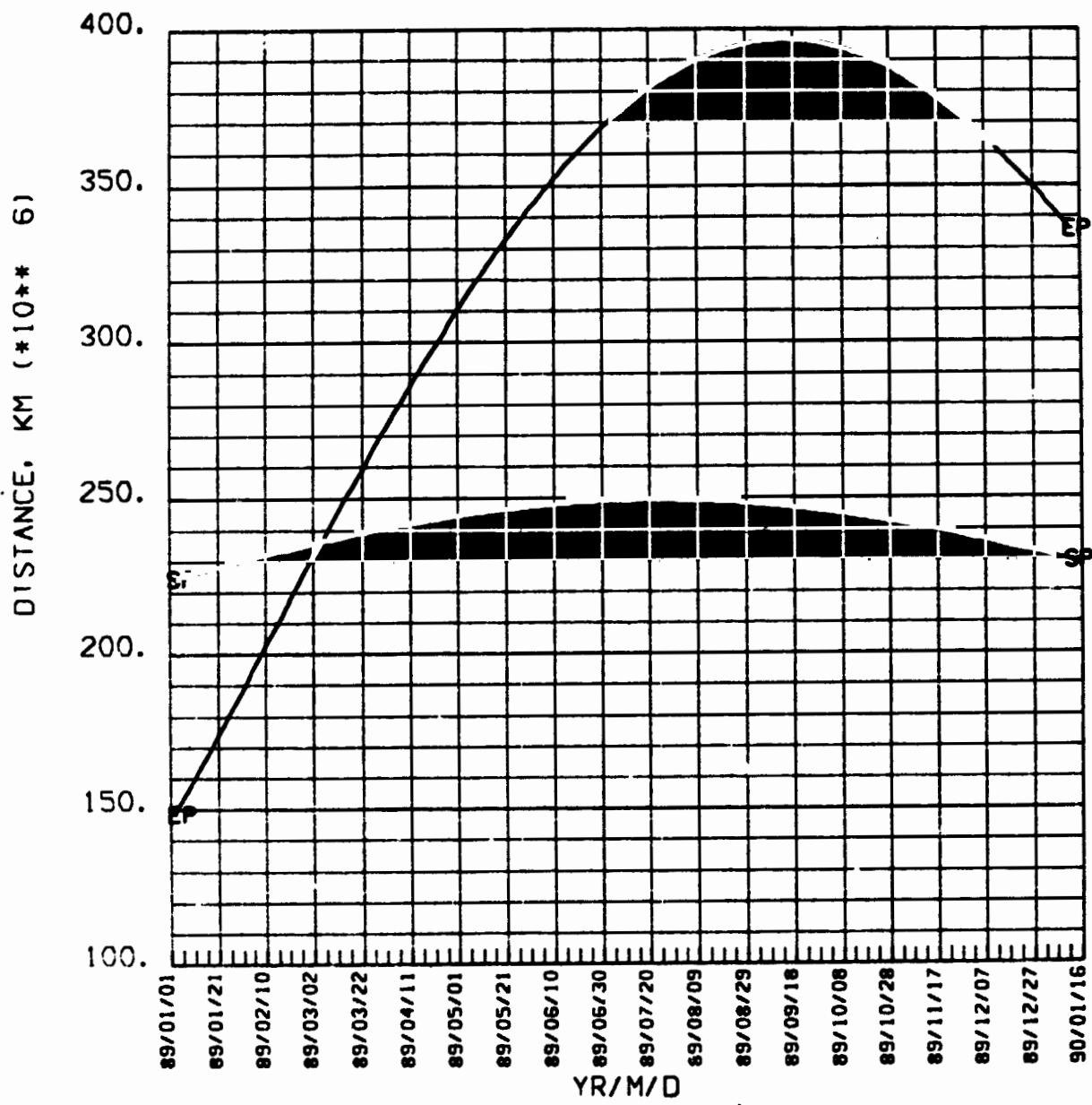


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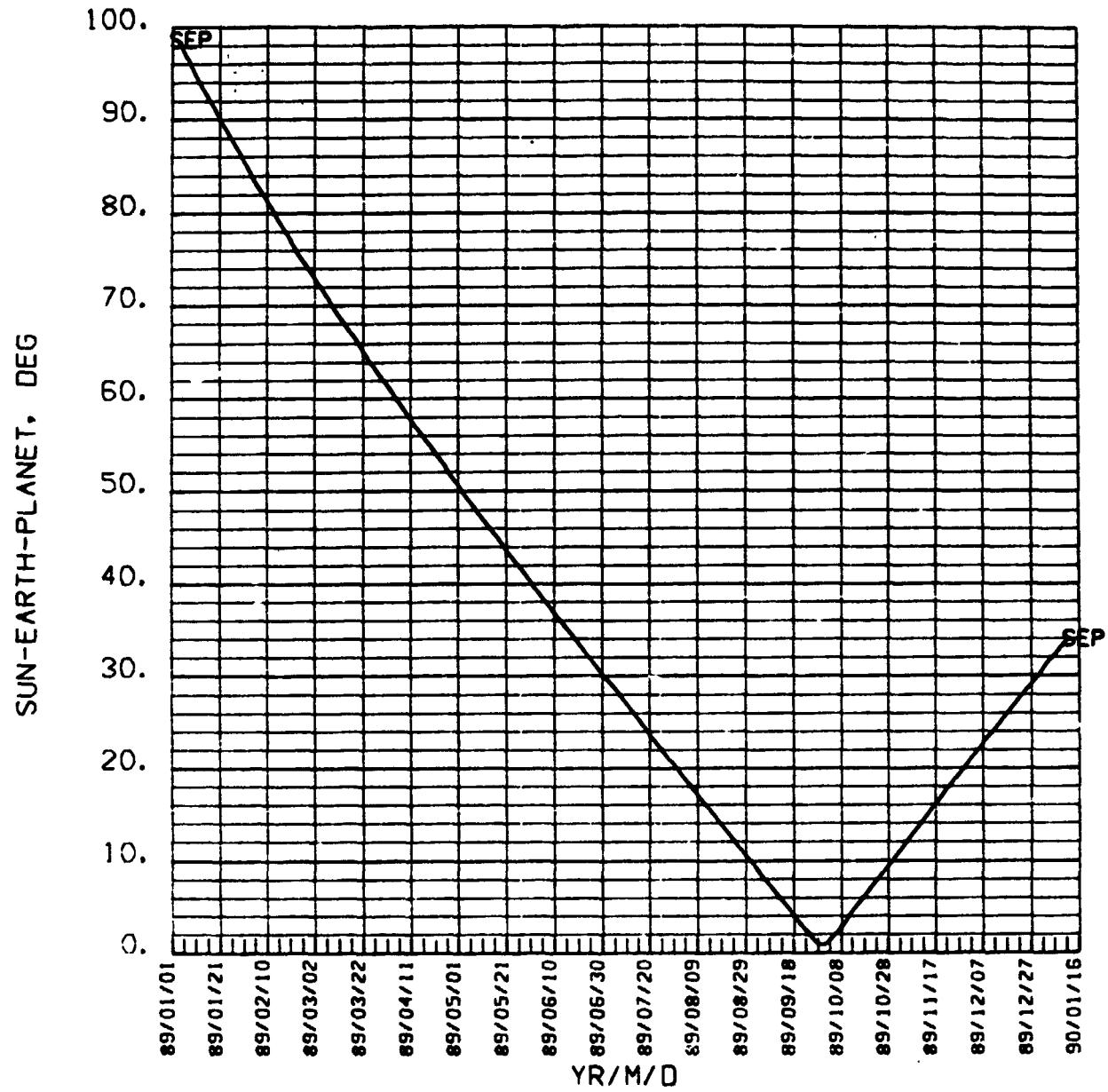
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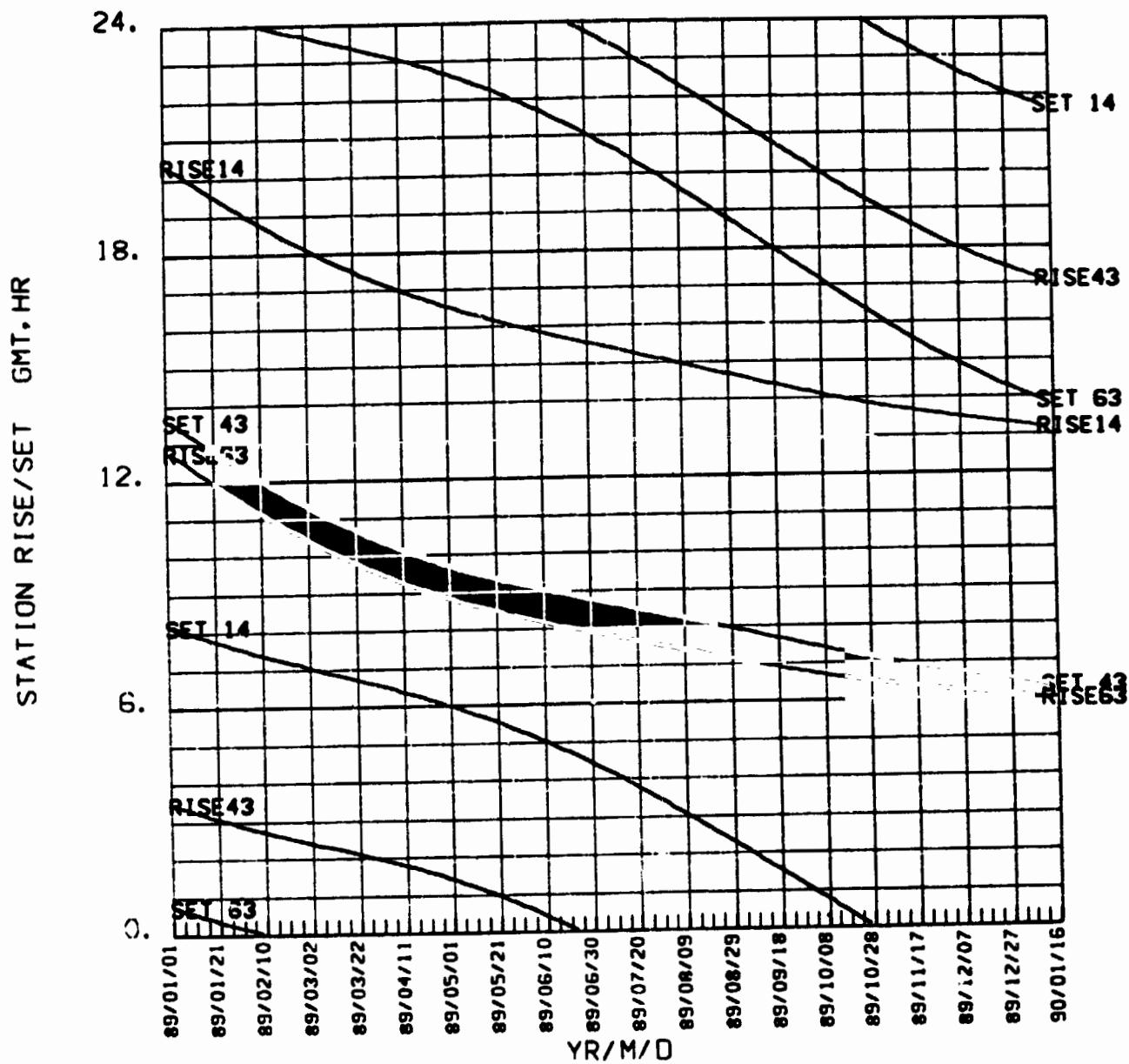
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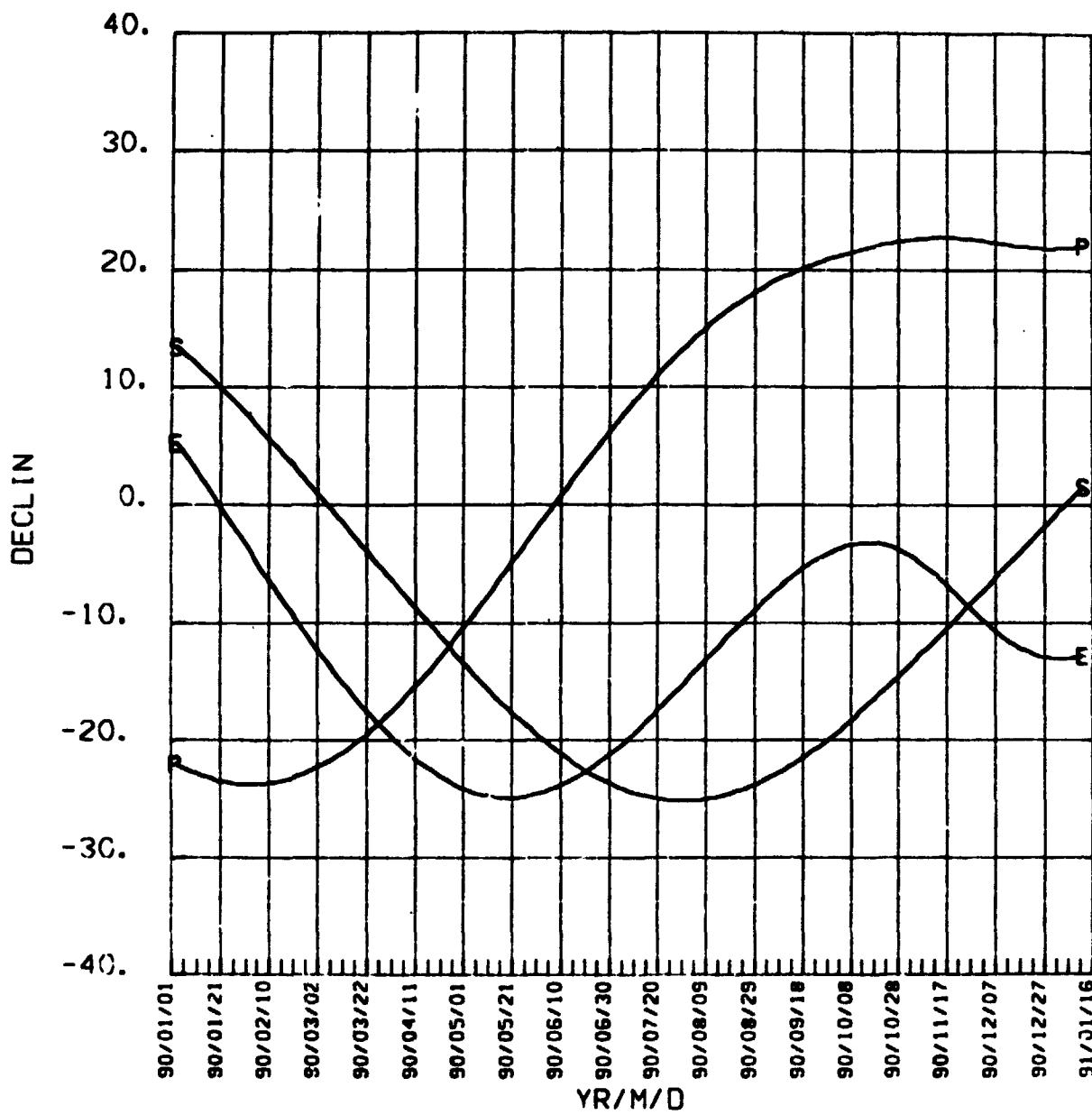
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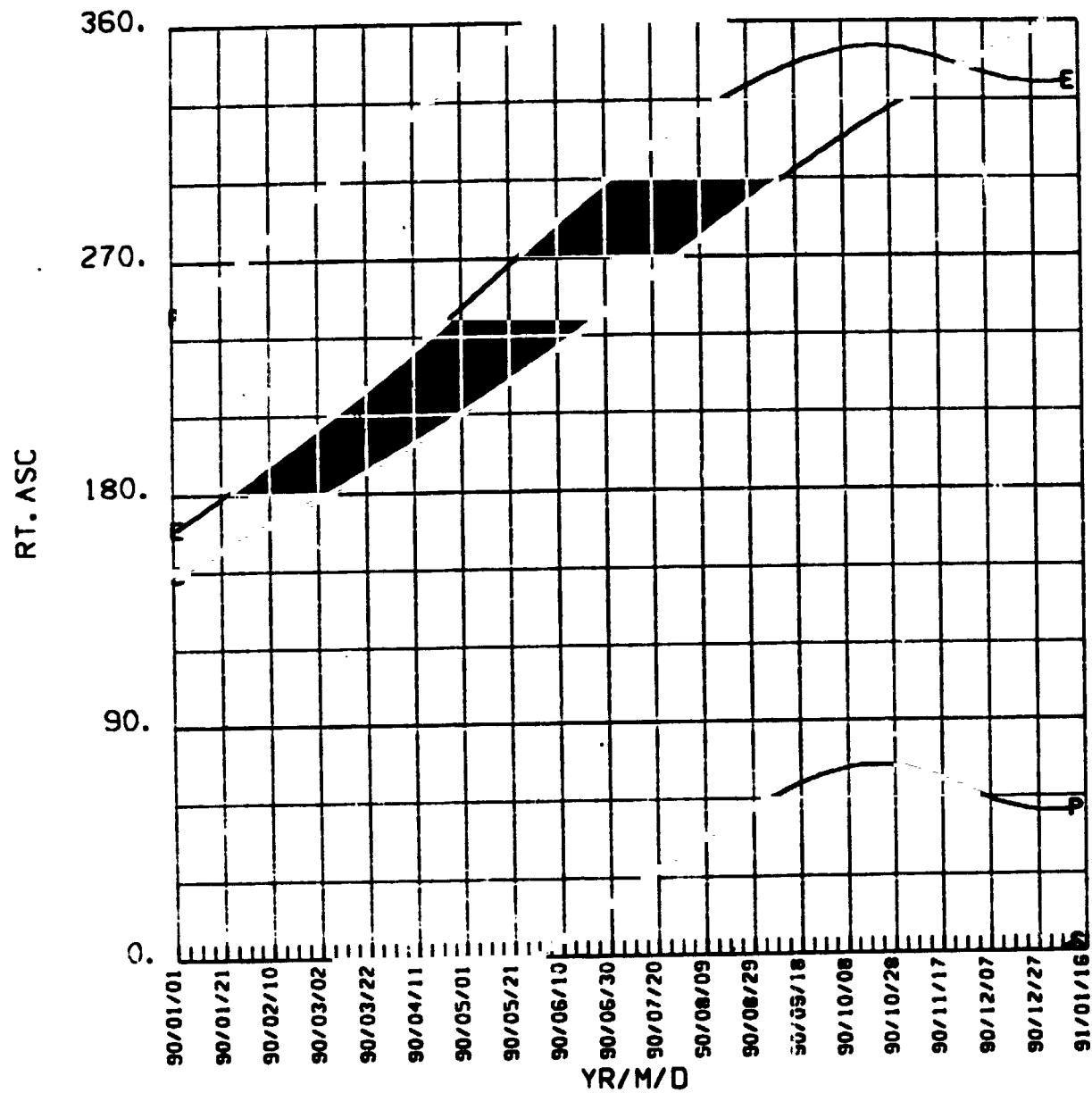
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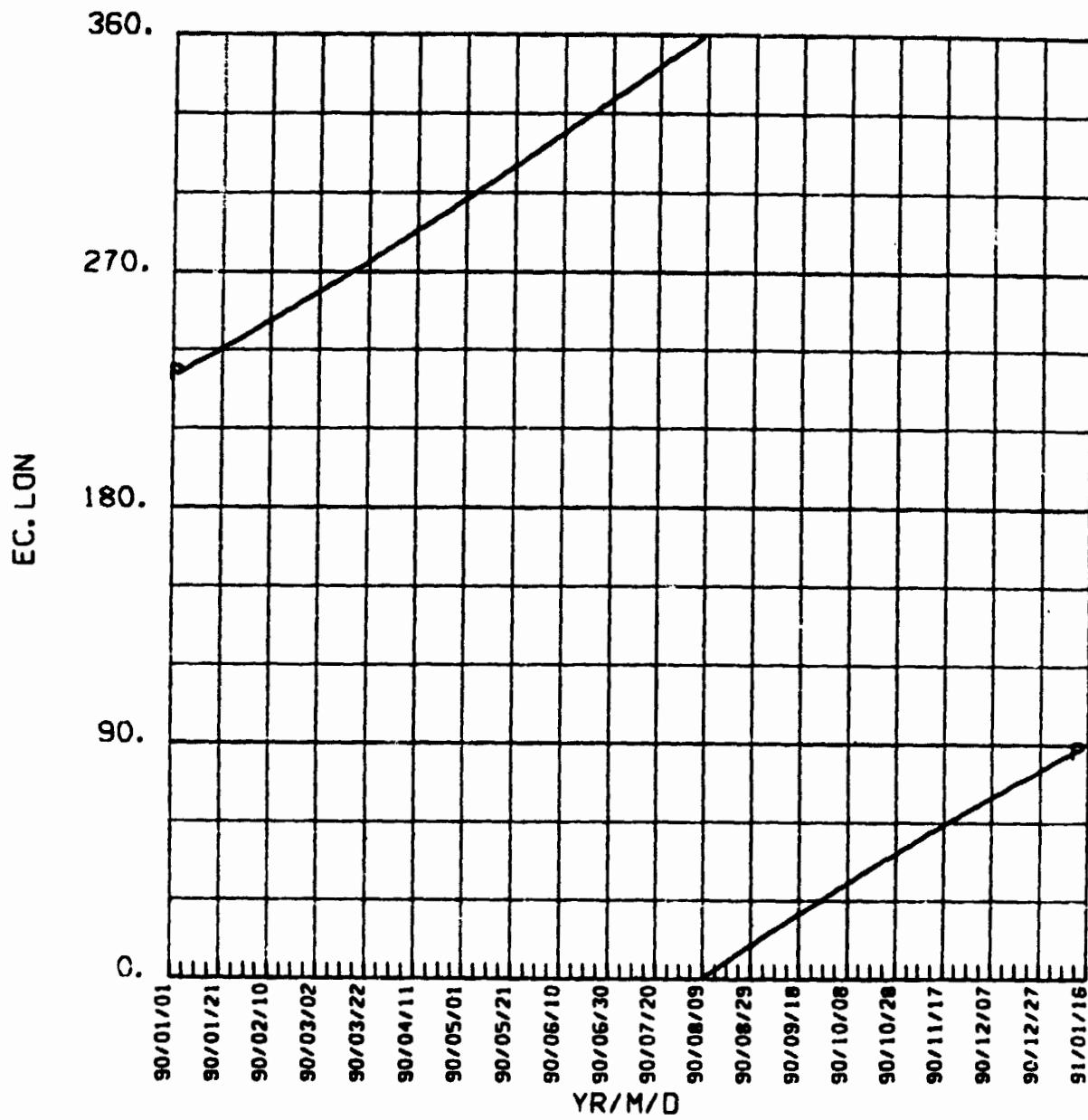
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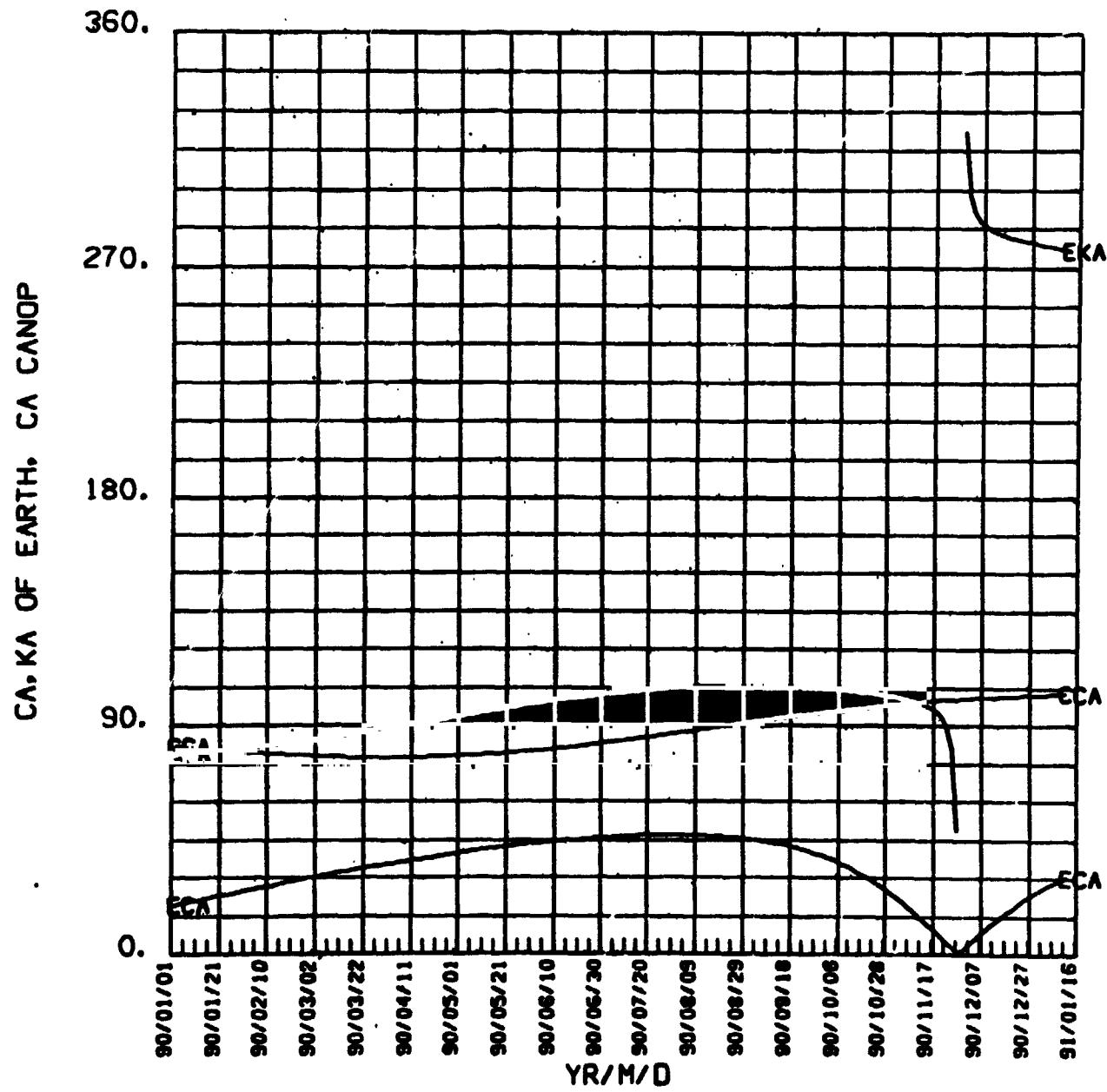
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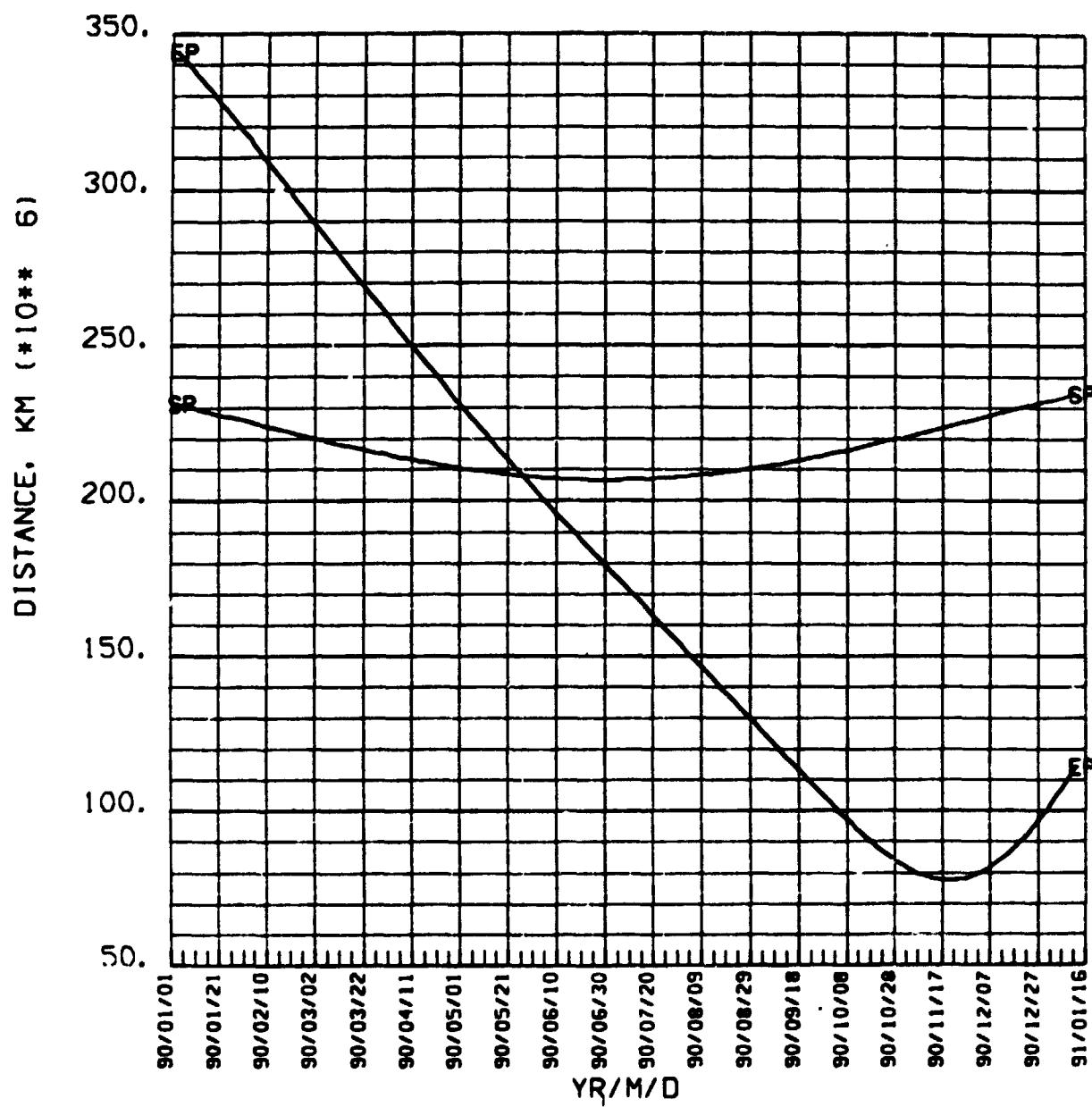


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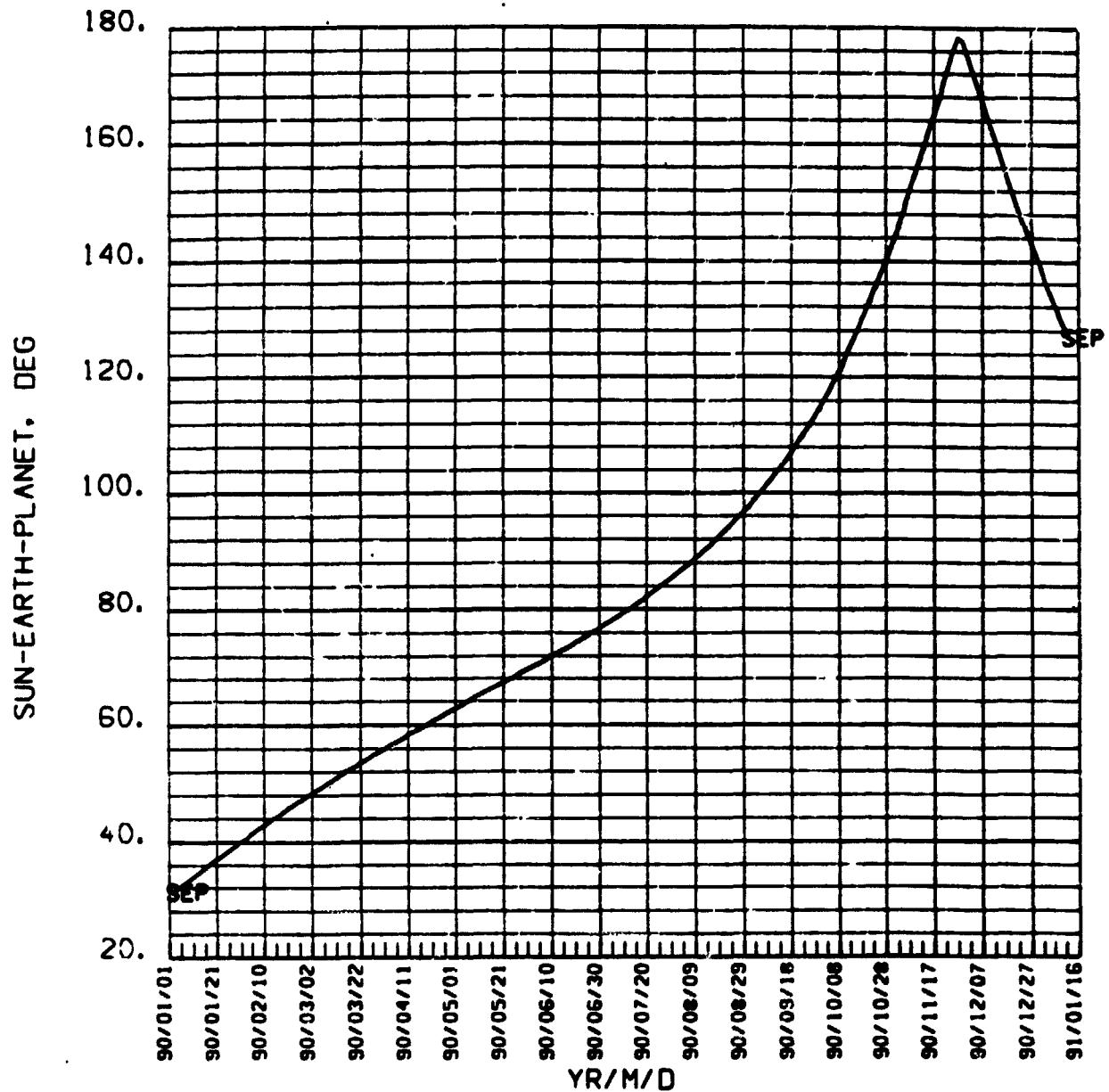


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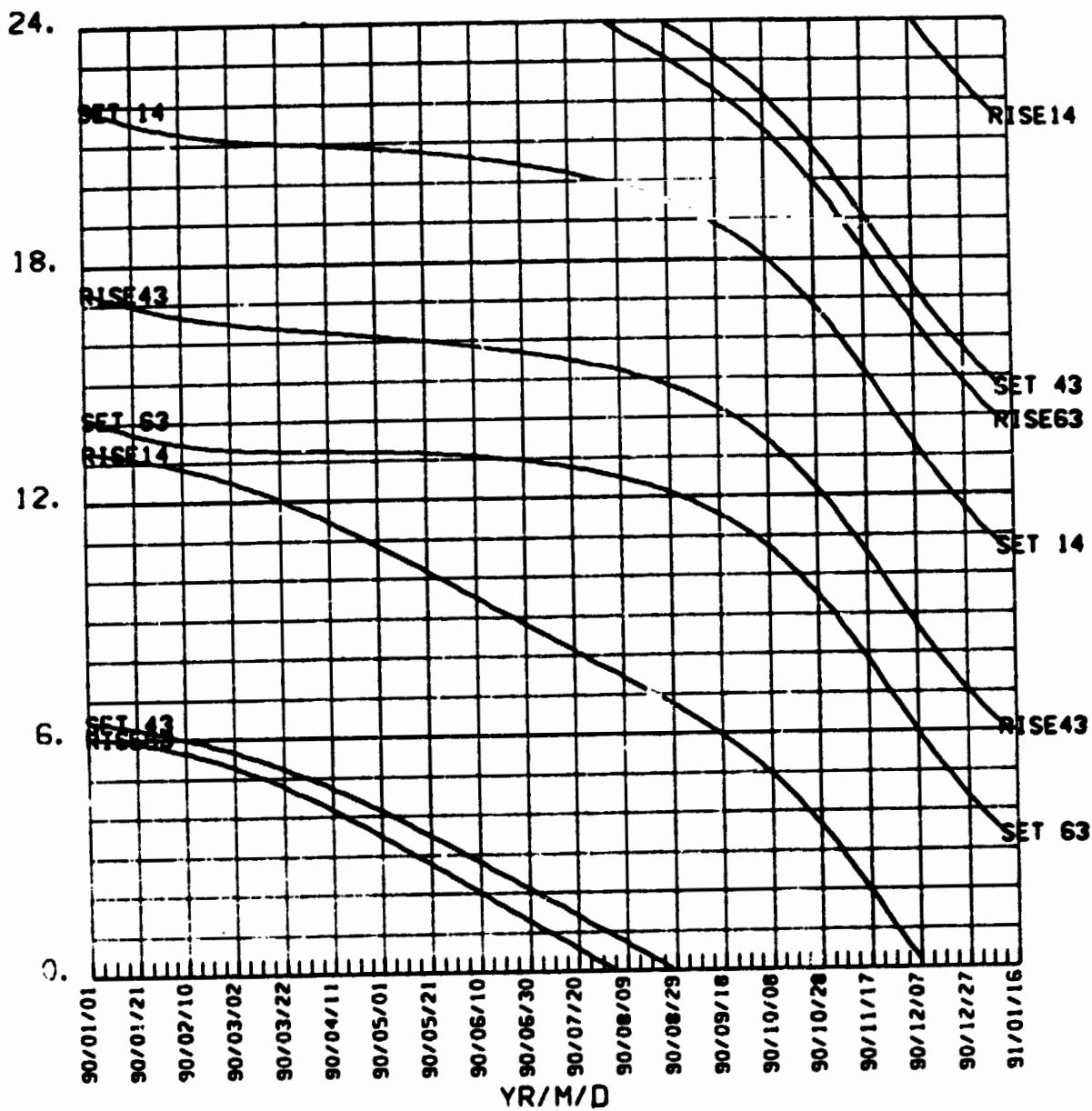


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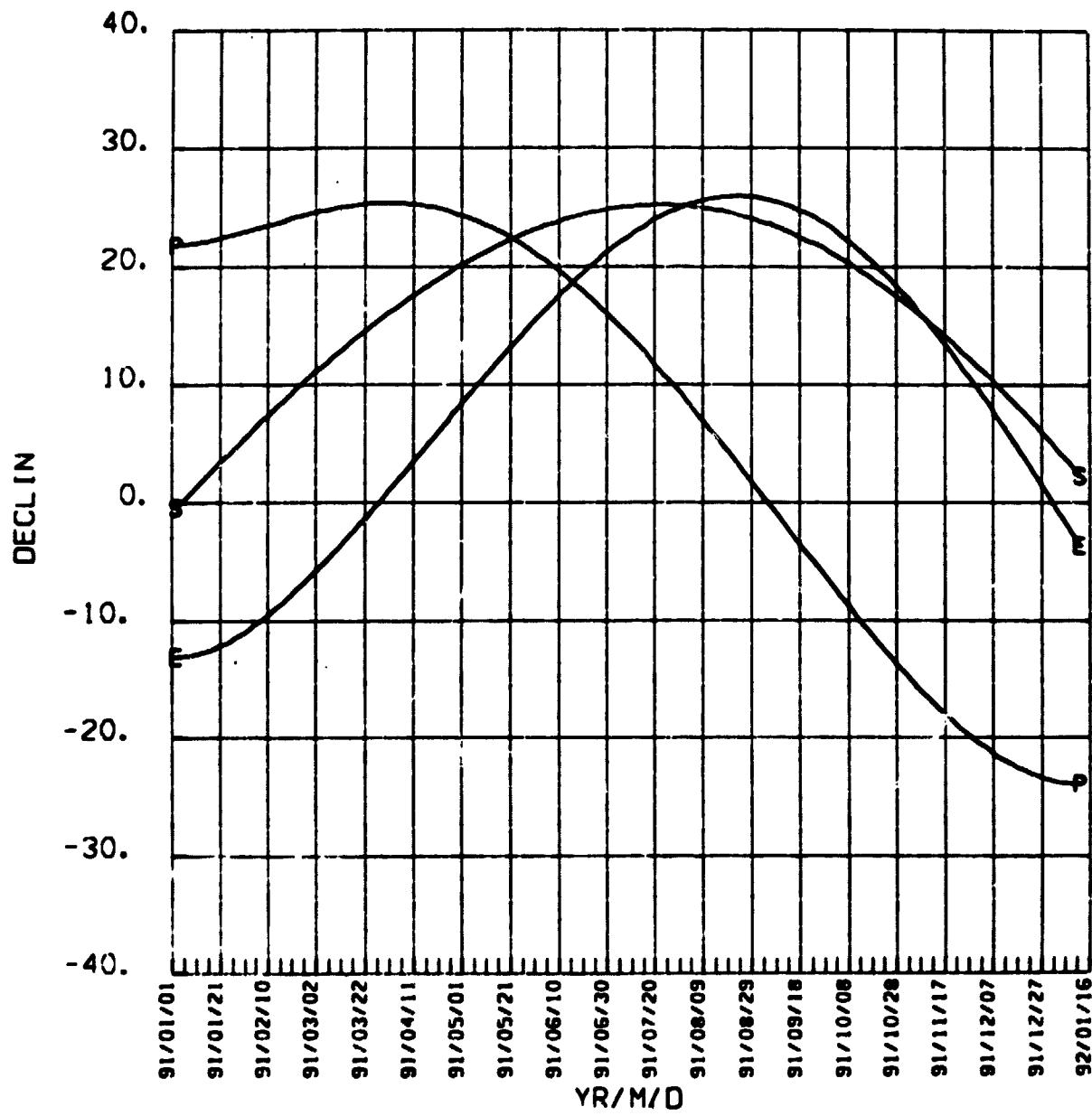


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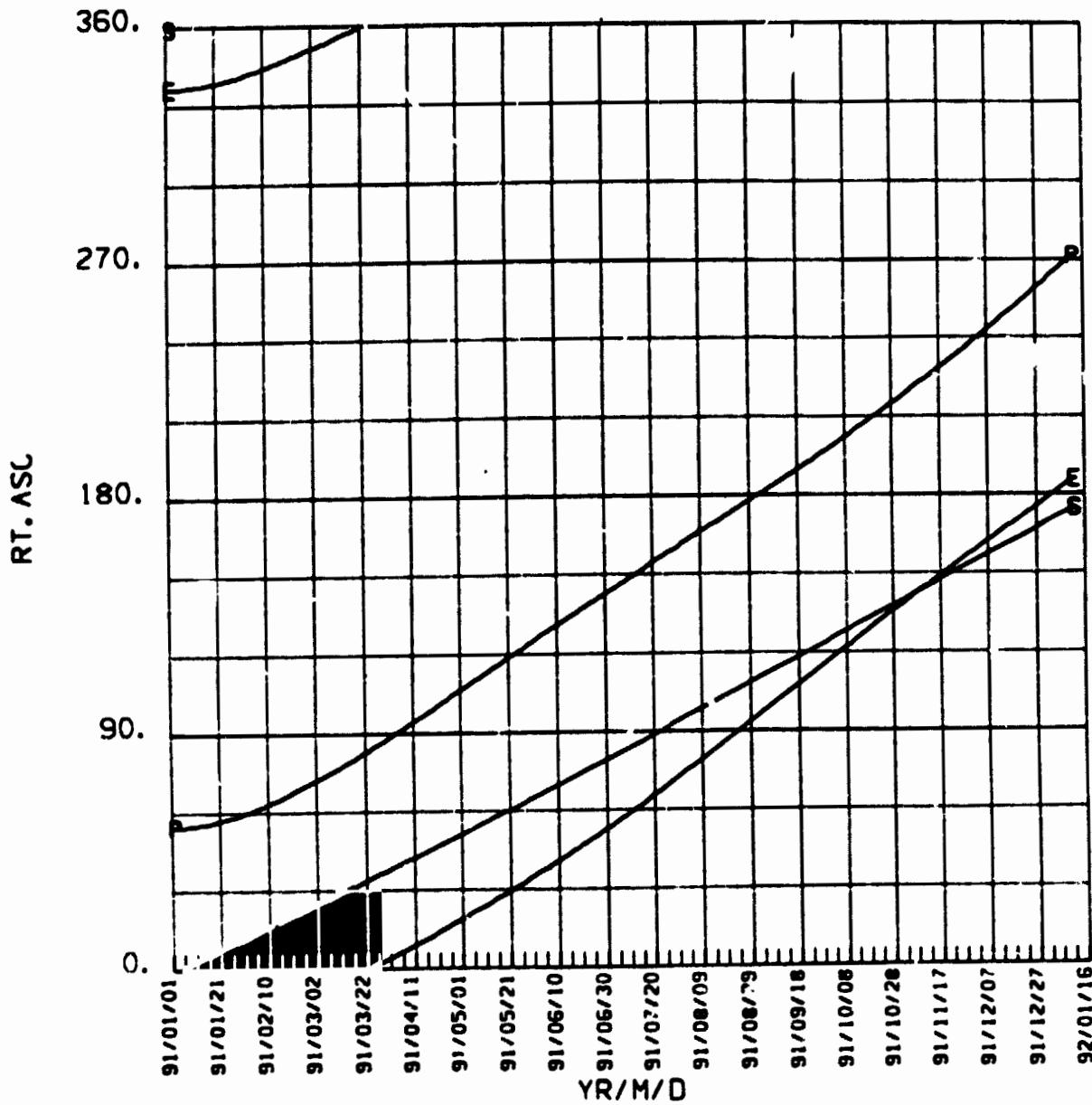
STATION RISE/SET GMT. HR



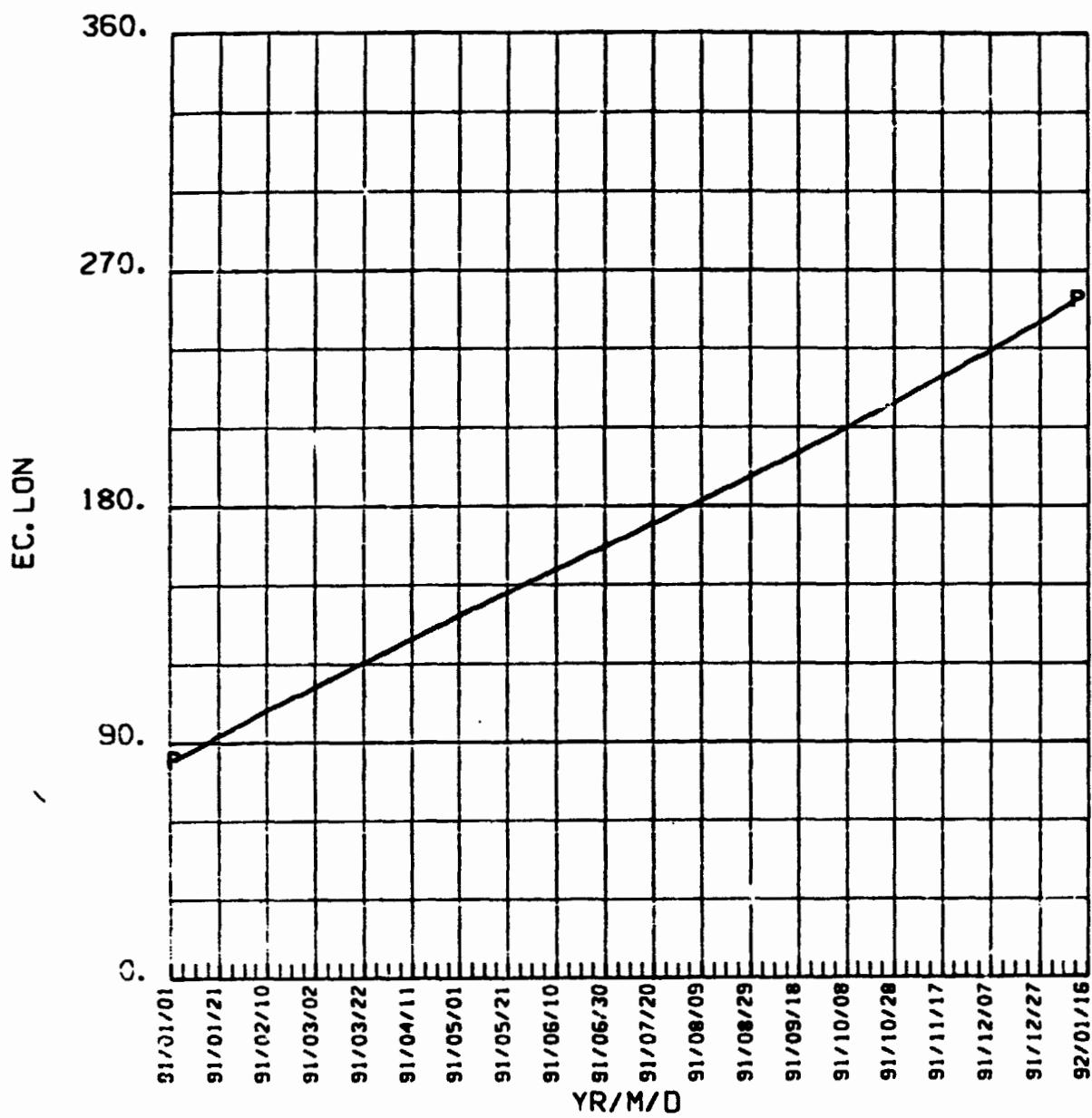
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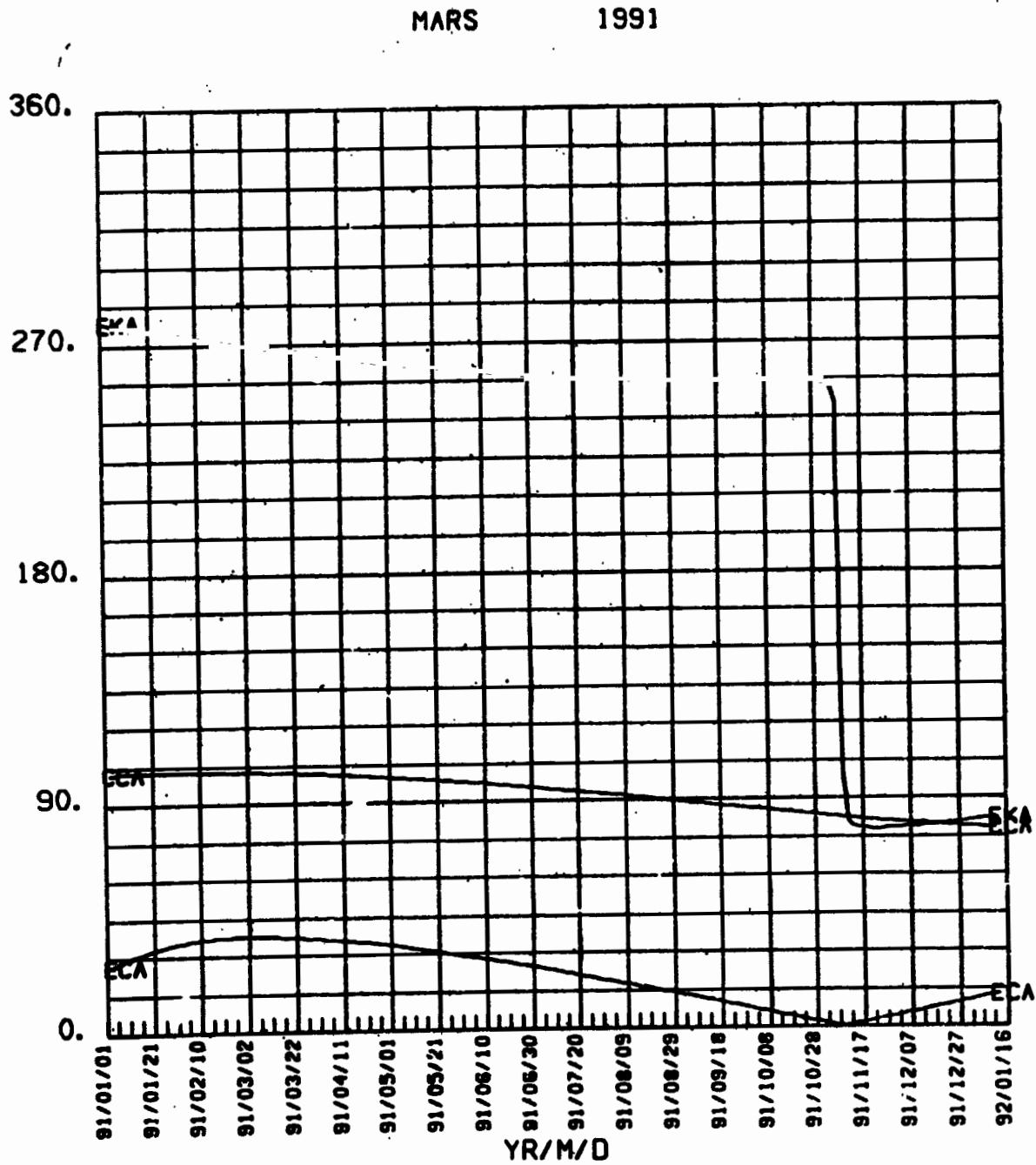
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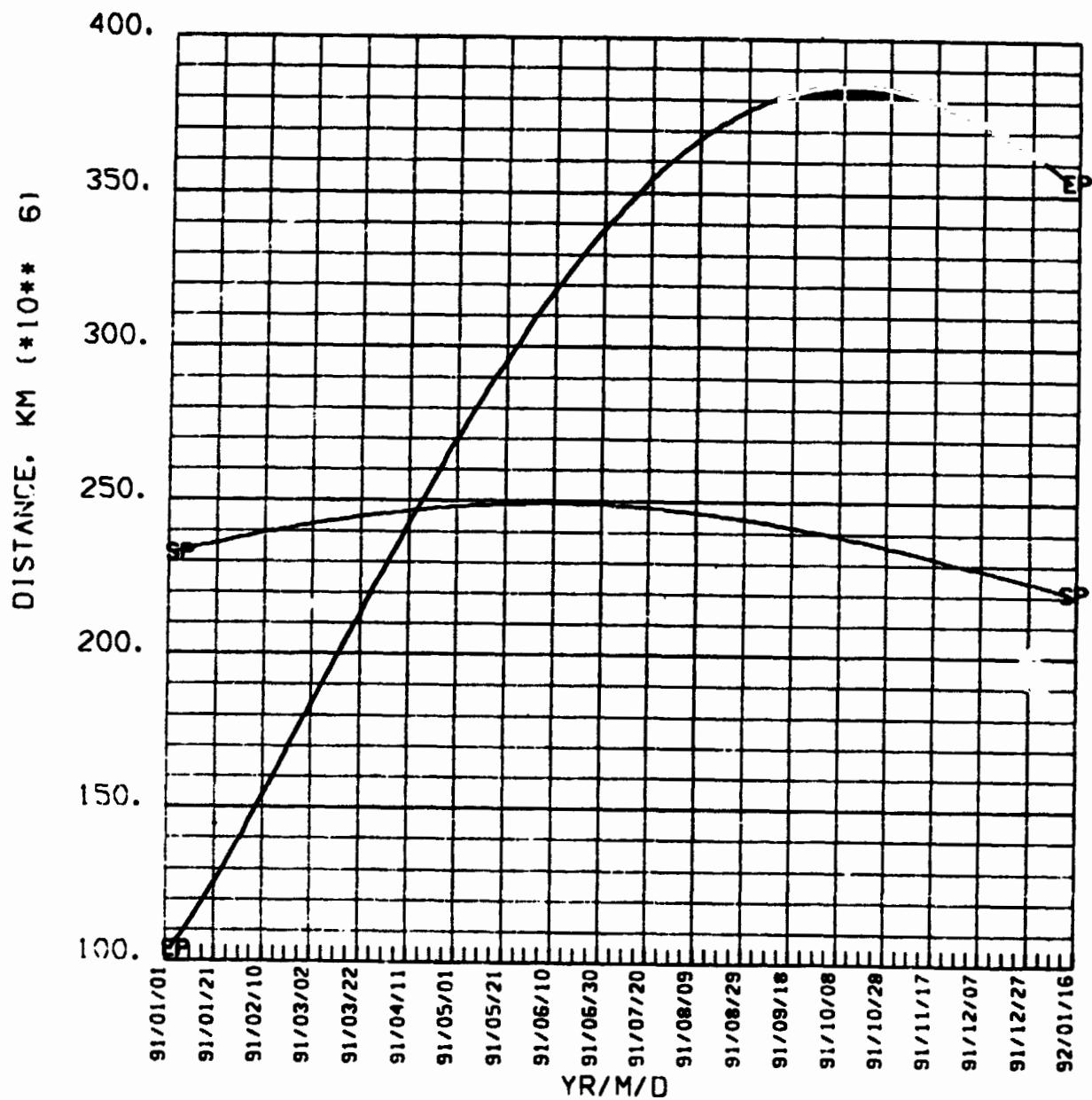


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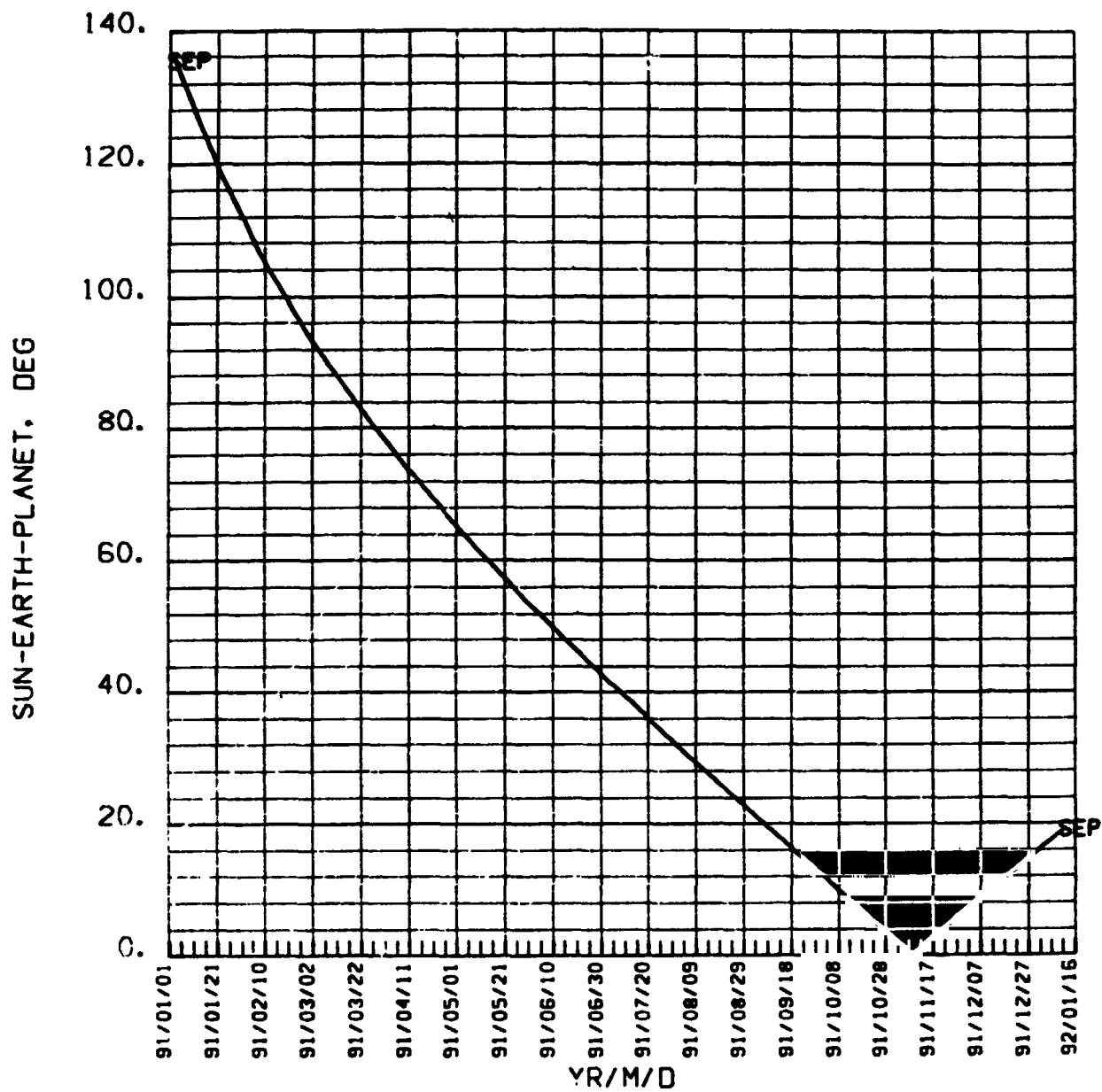
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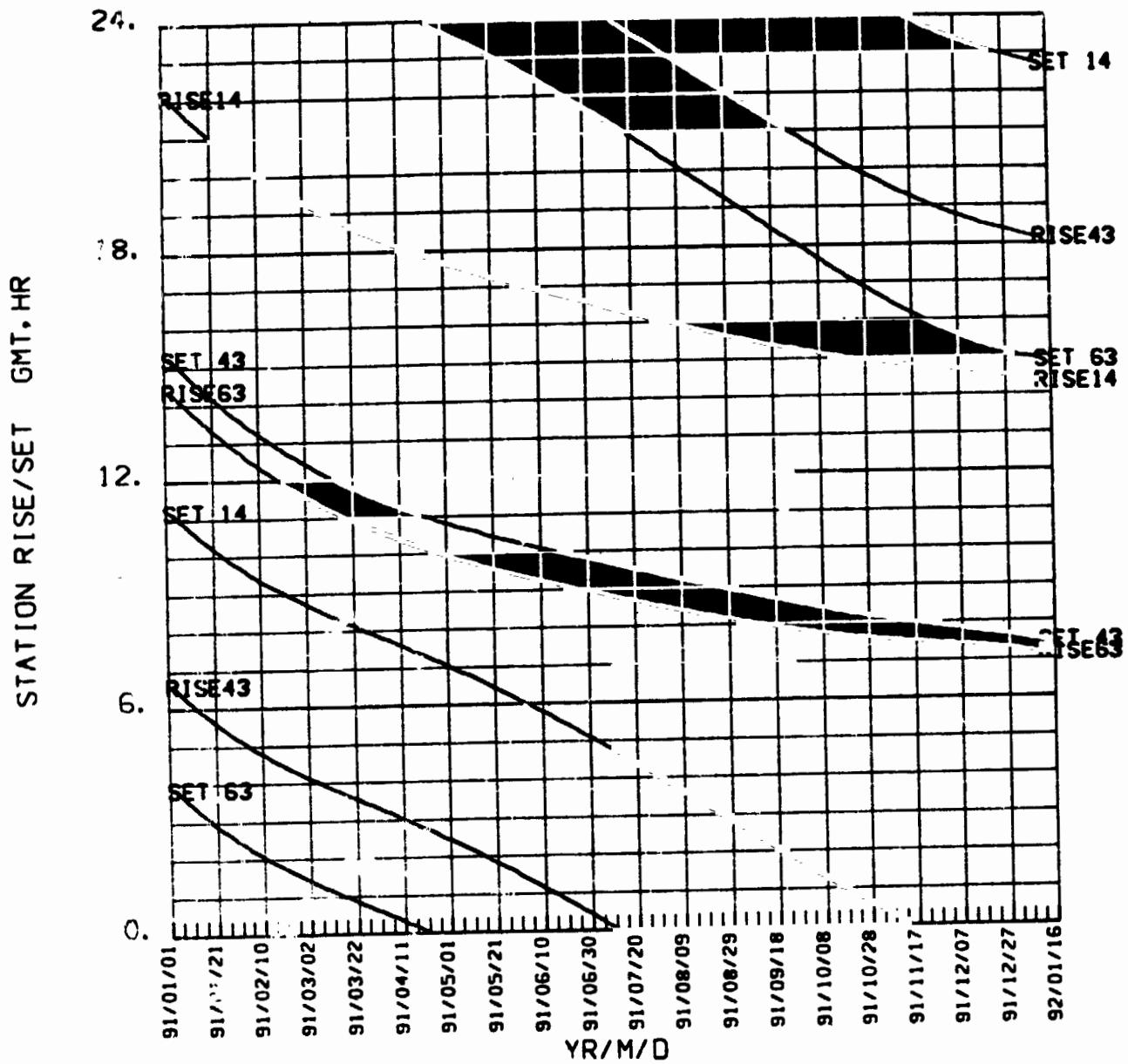


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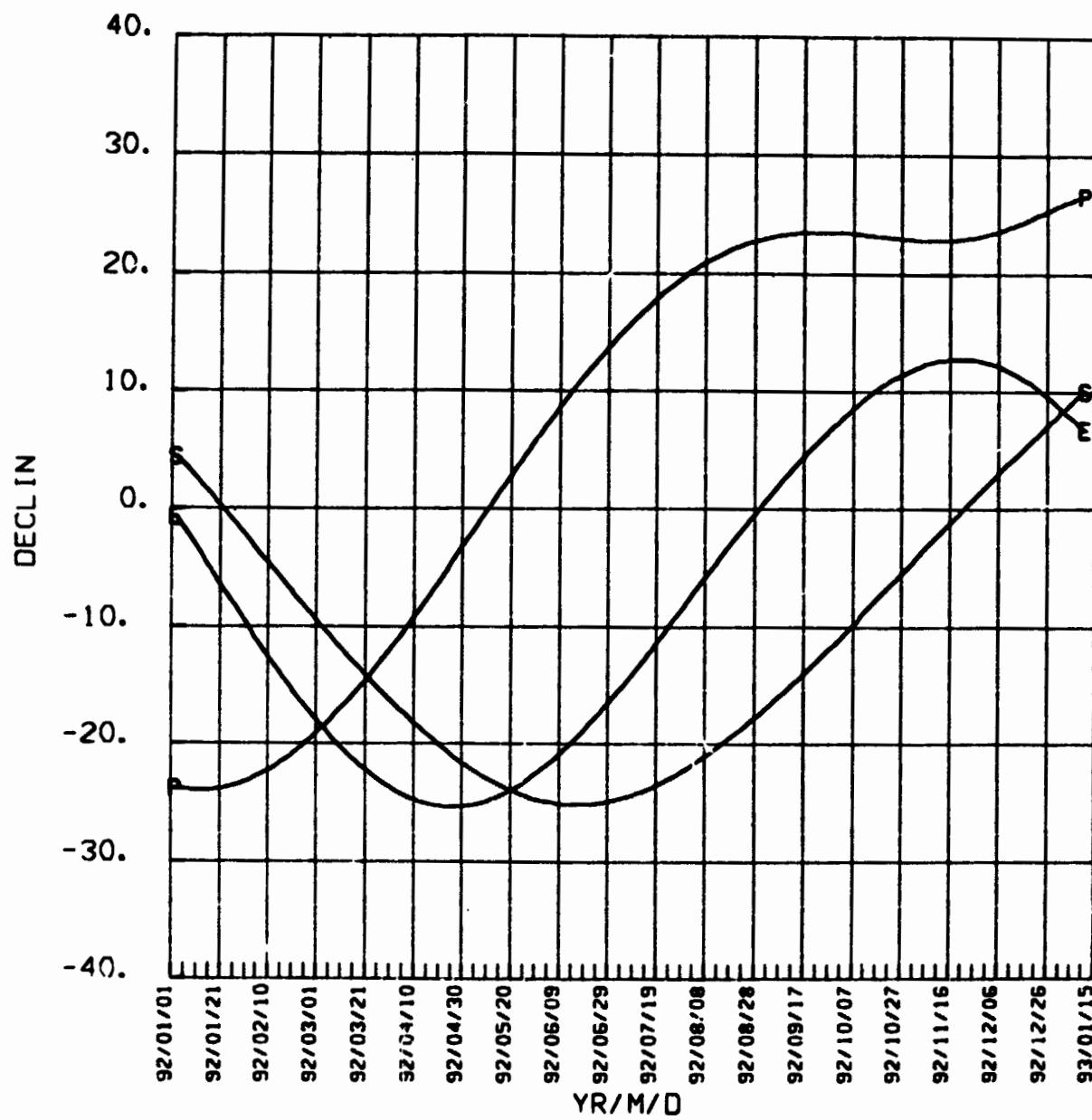
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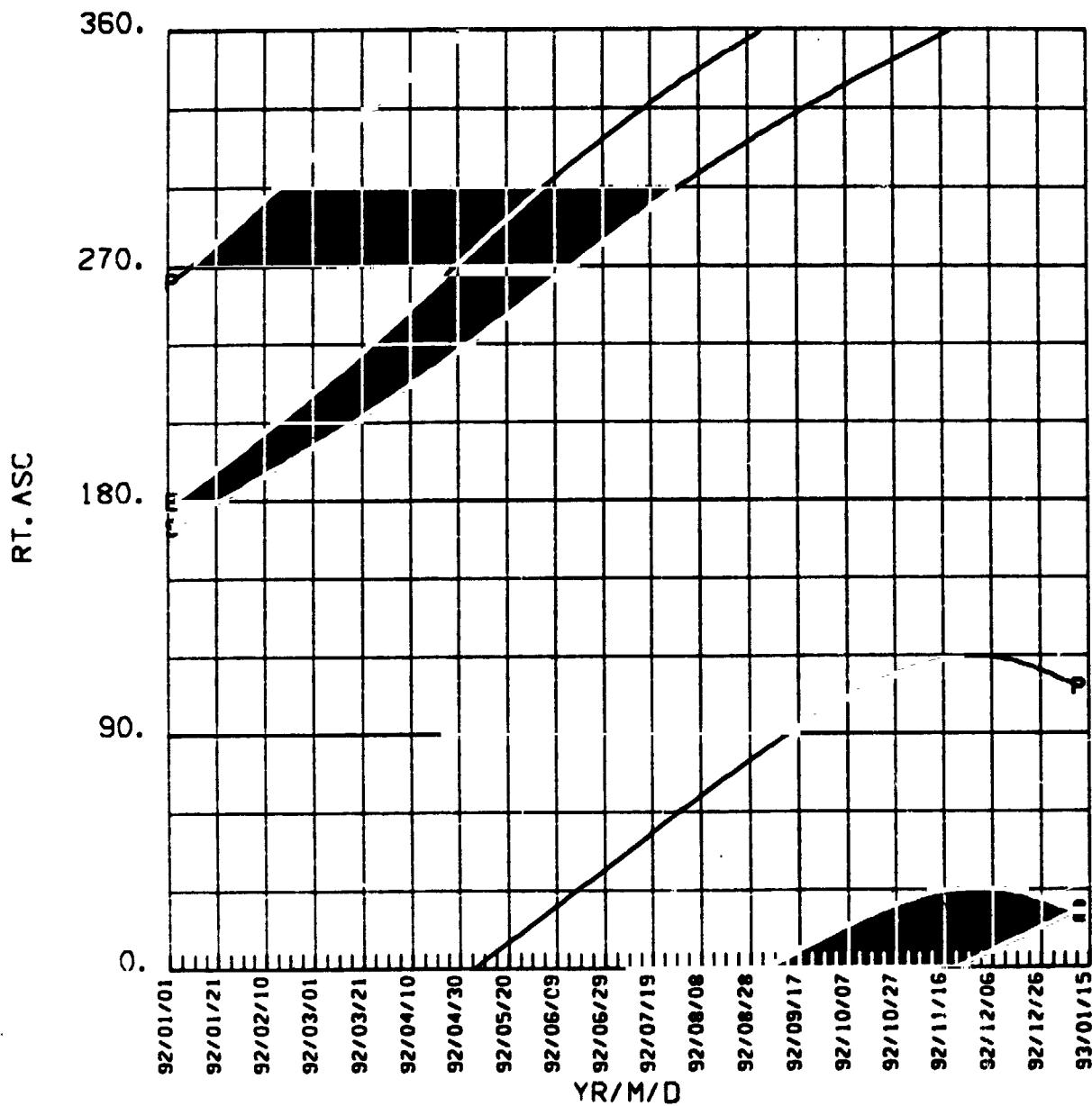
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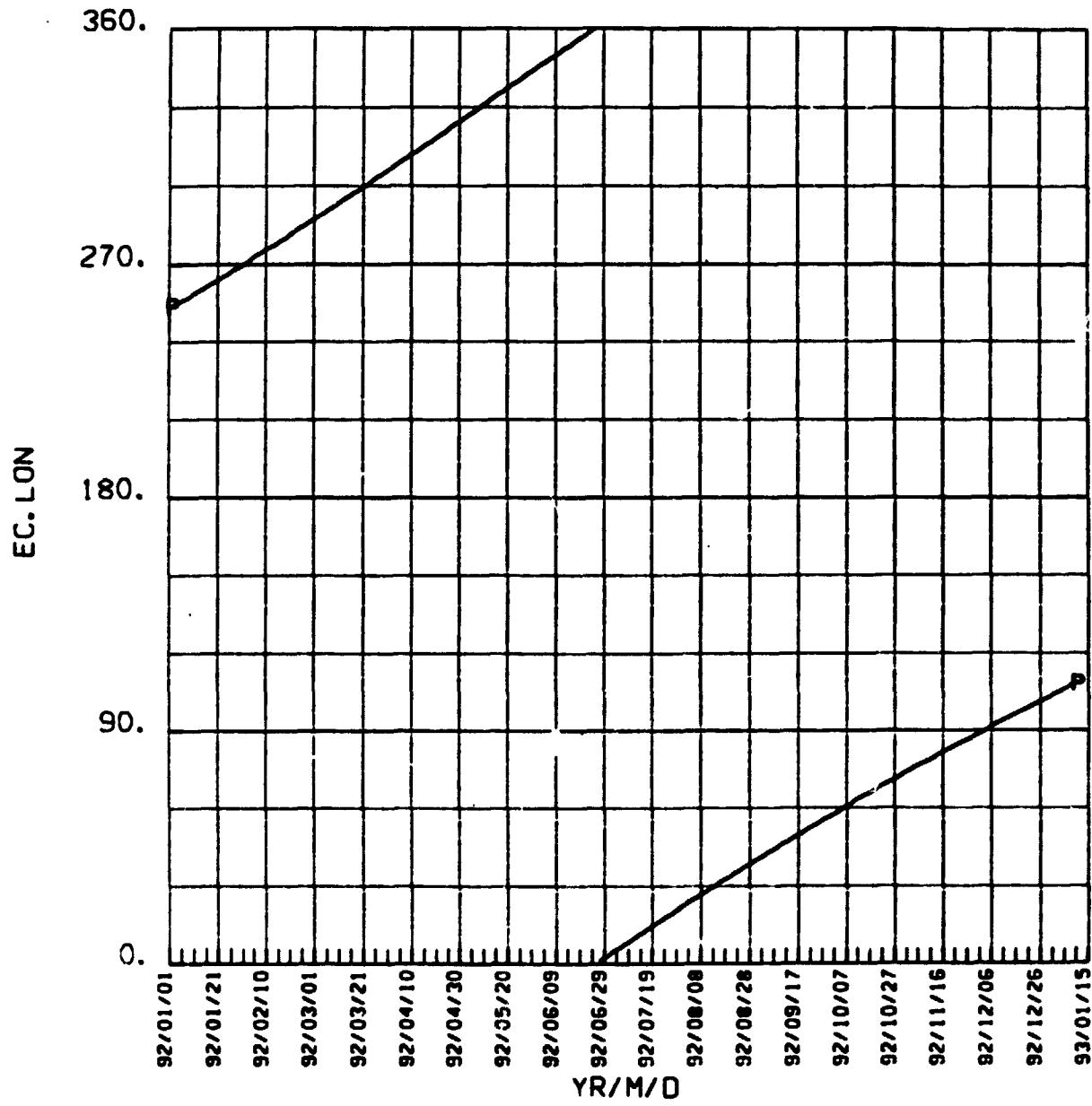
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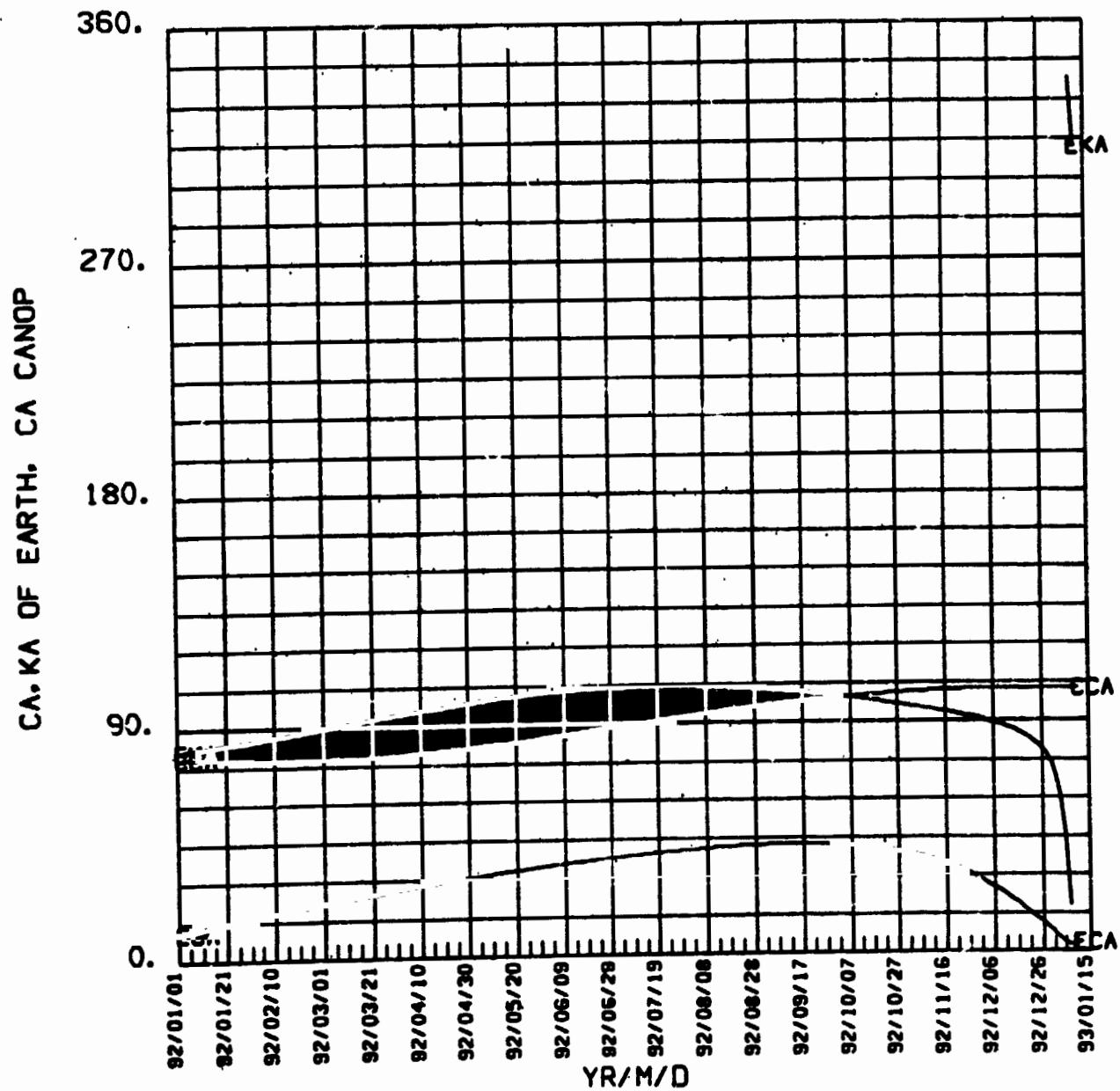
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MARS 1992

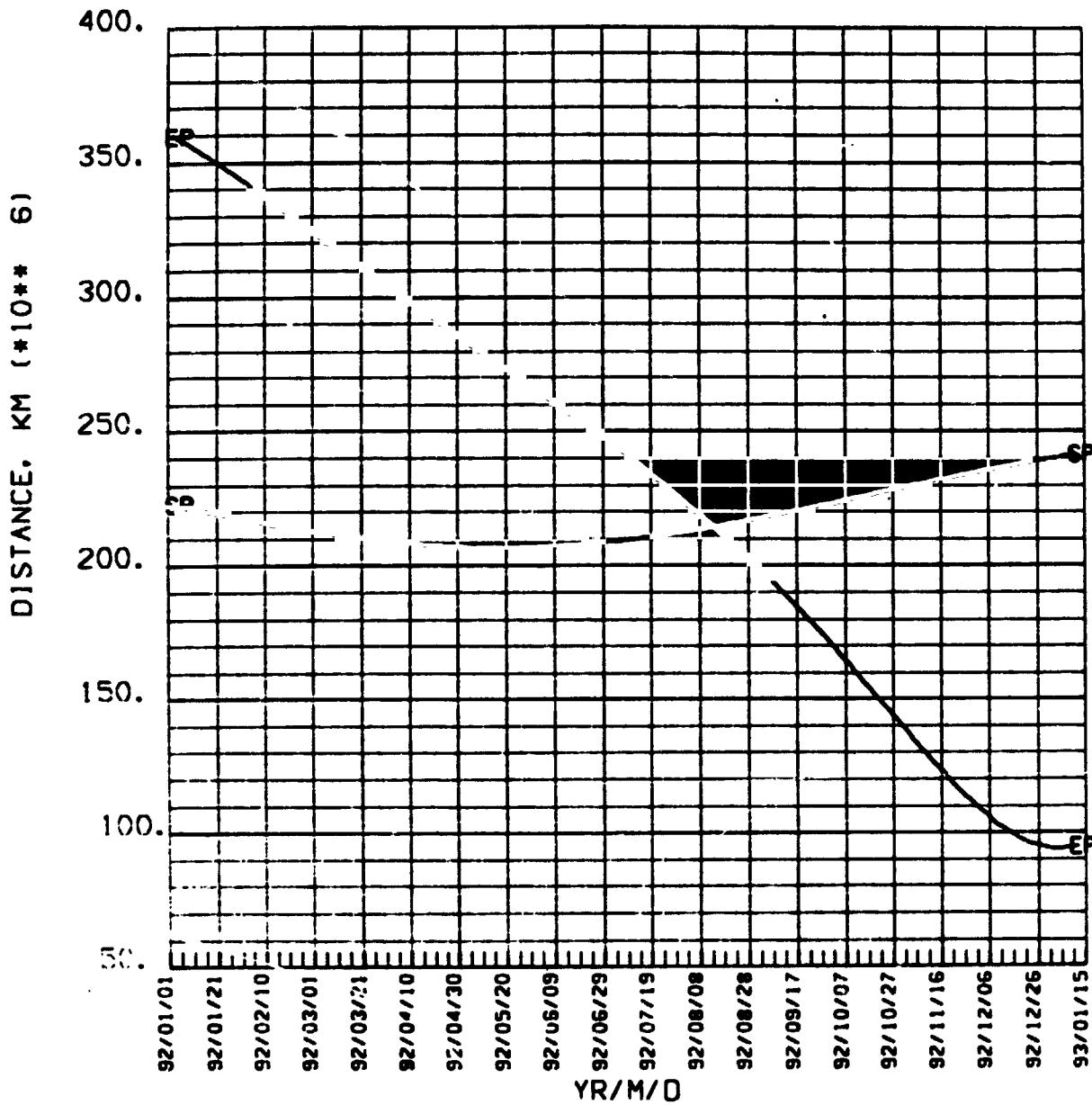


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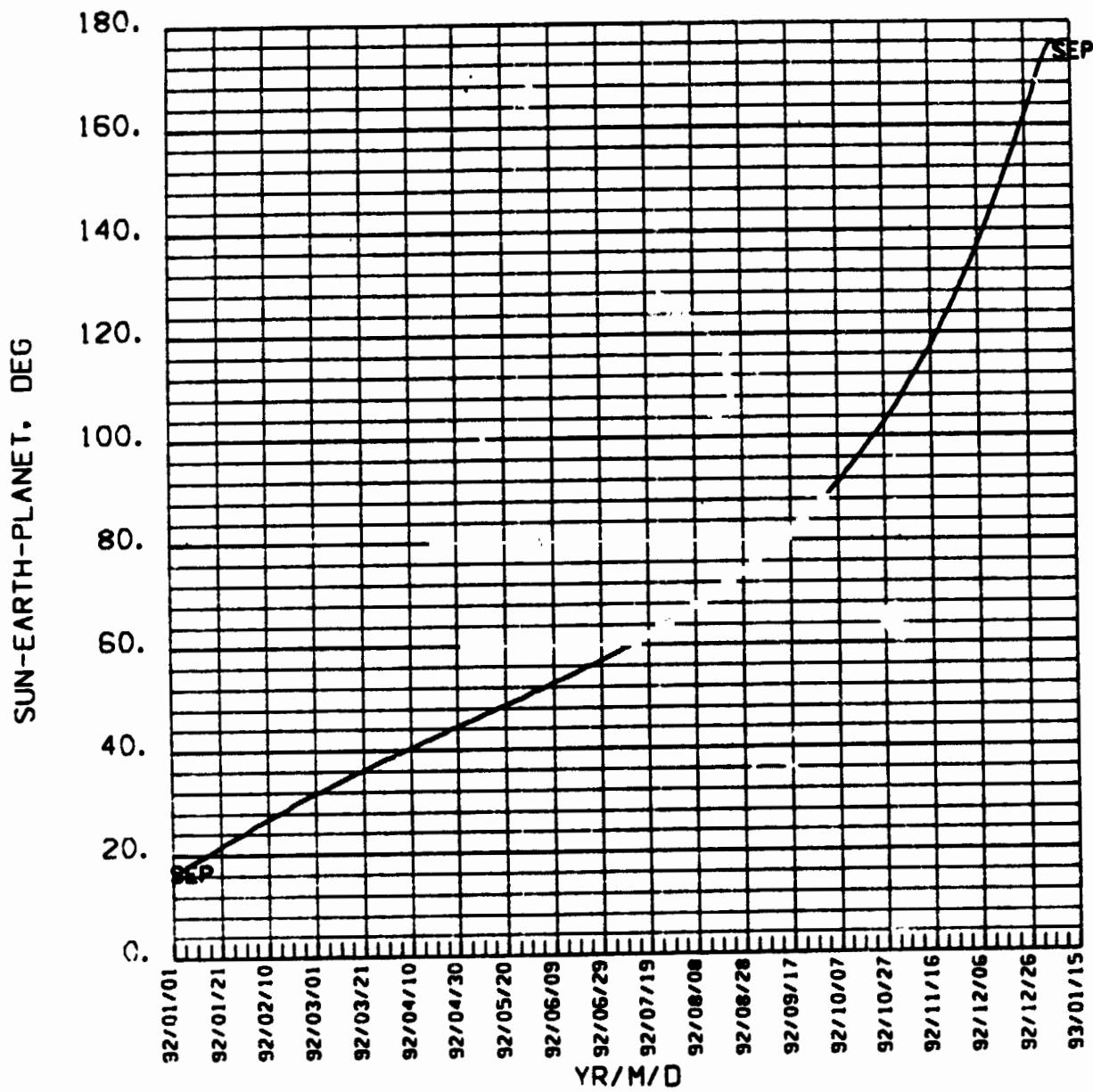


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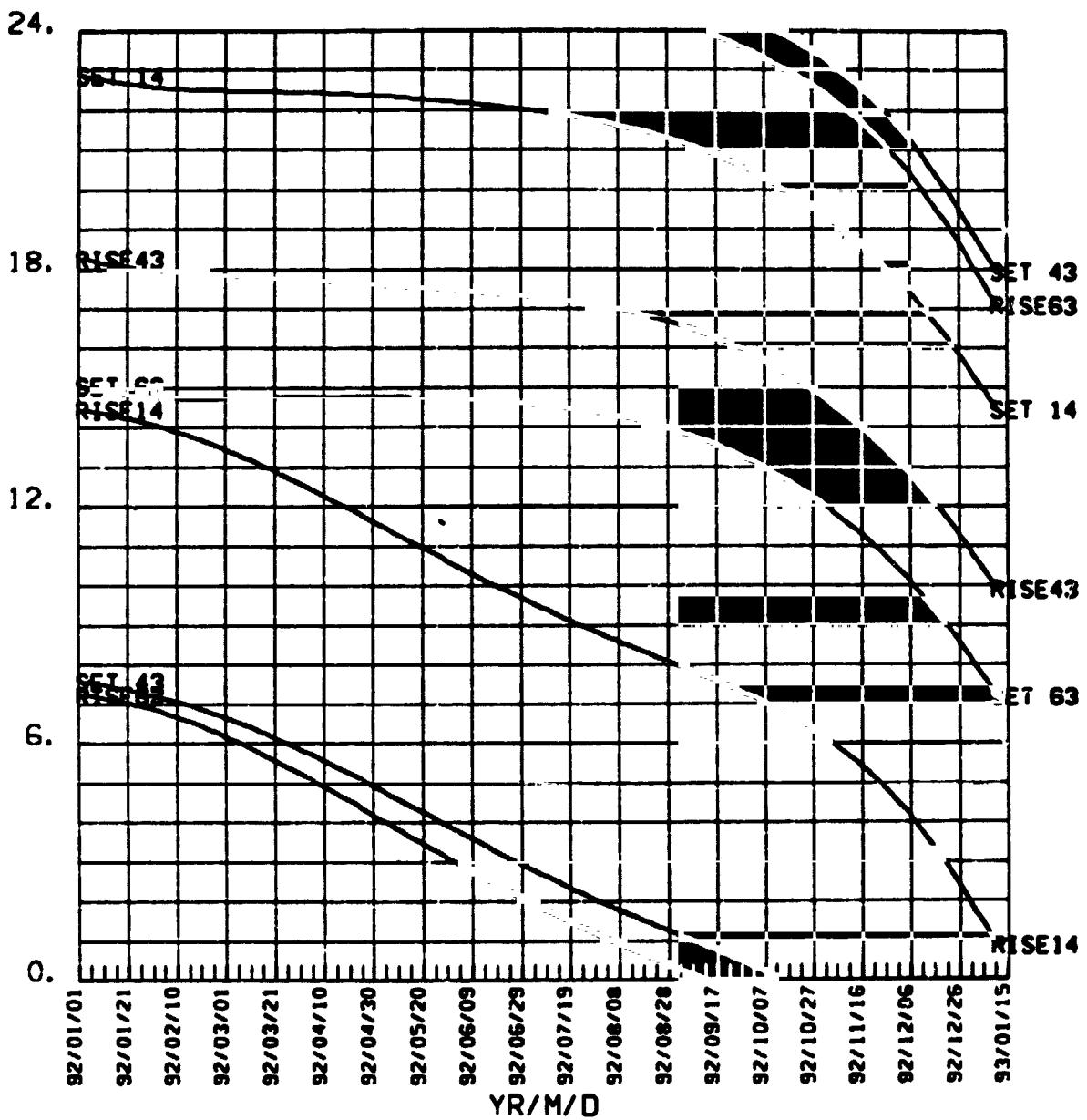


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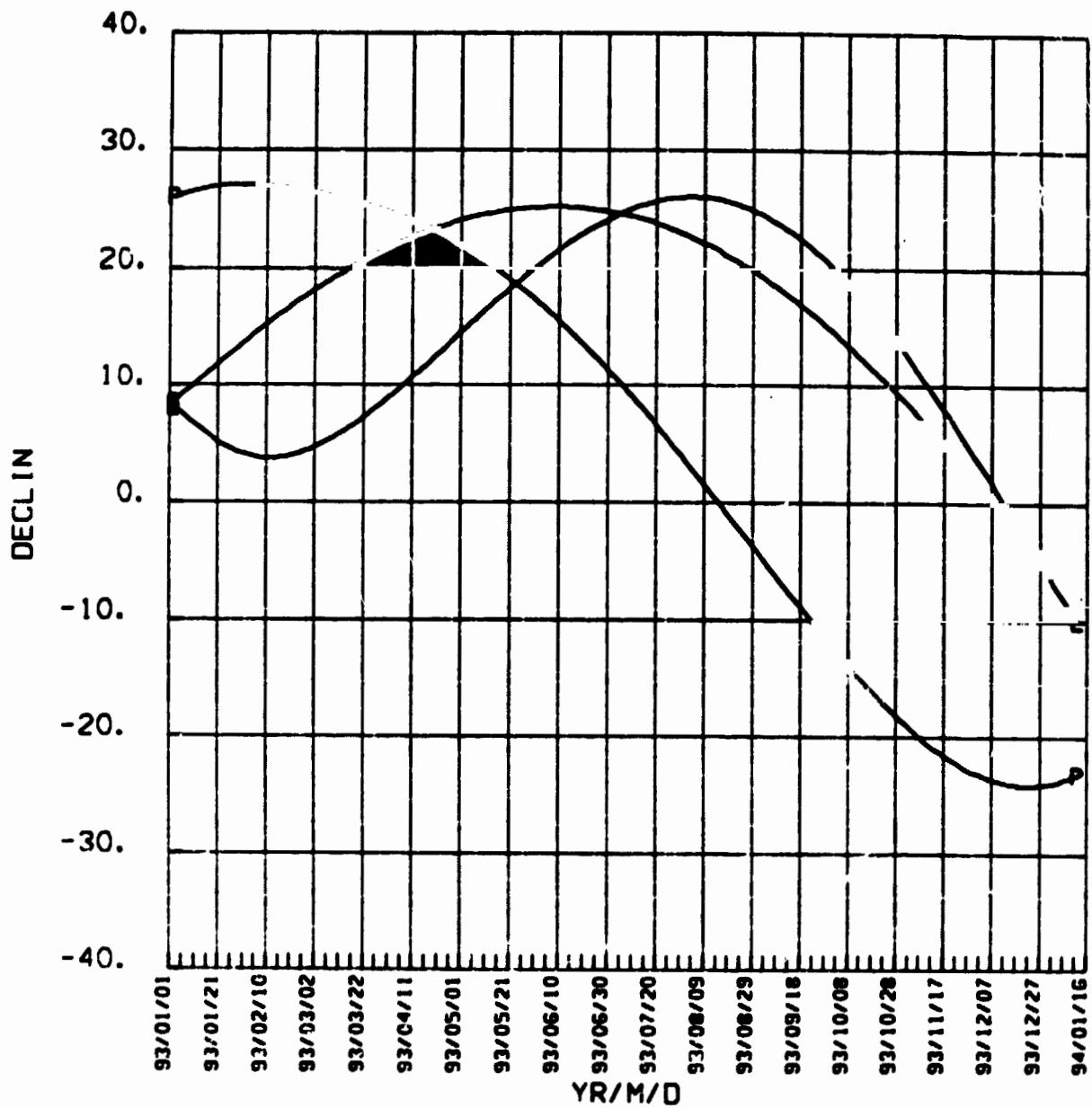
MARS 1992

STATION RISE/SET GMT. HR

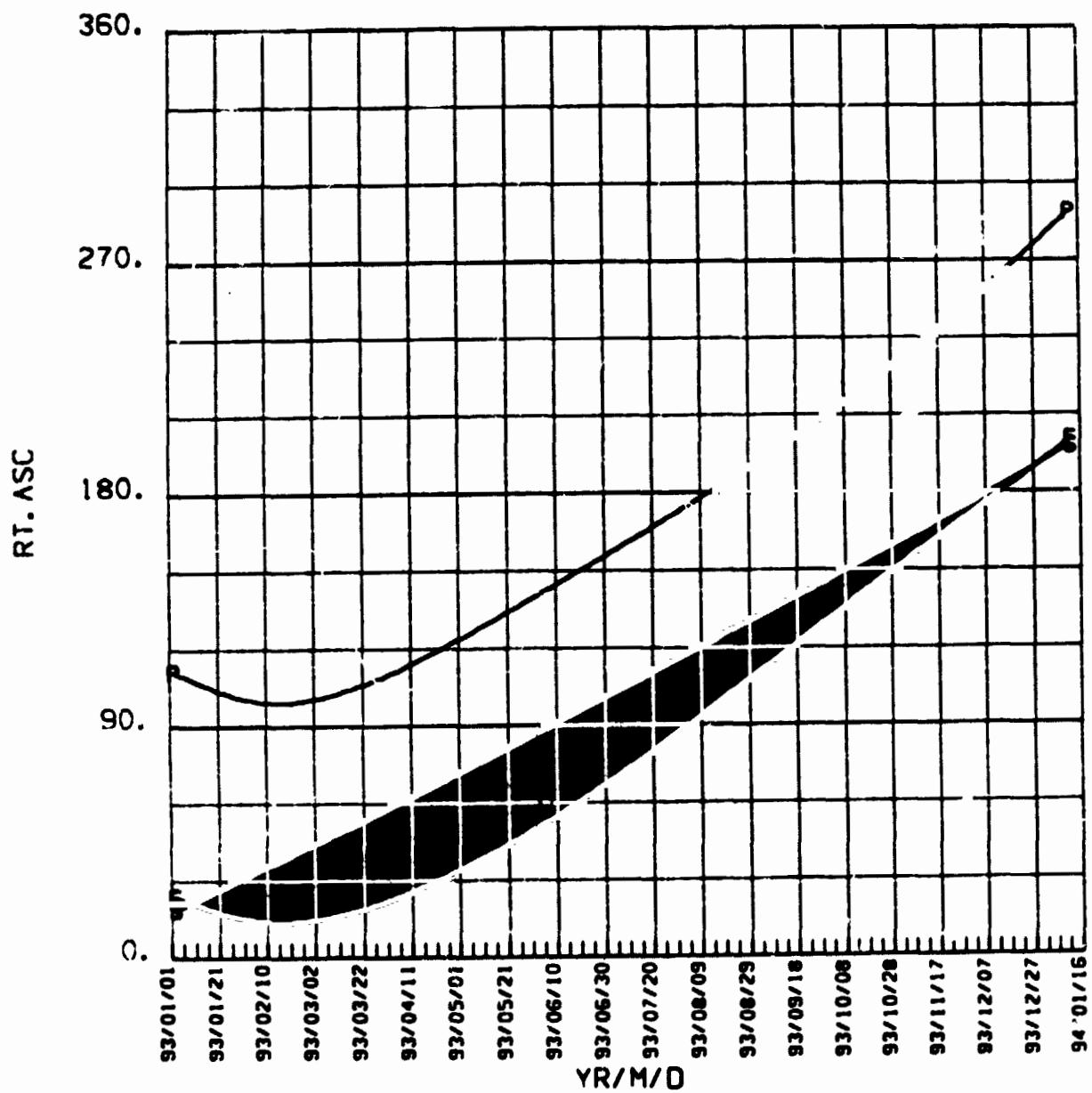


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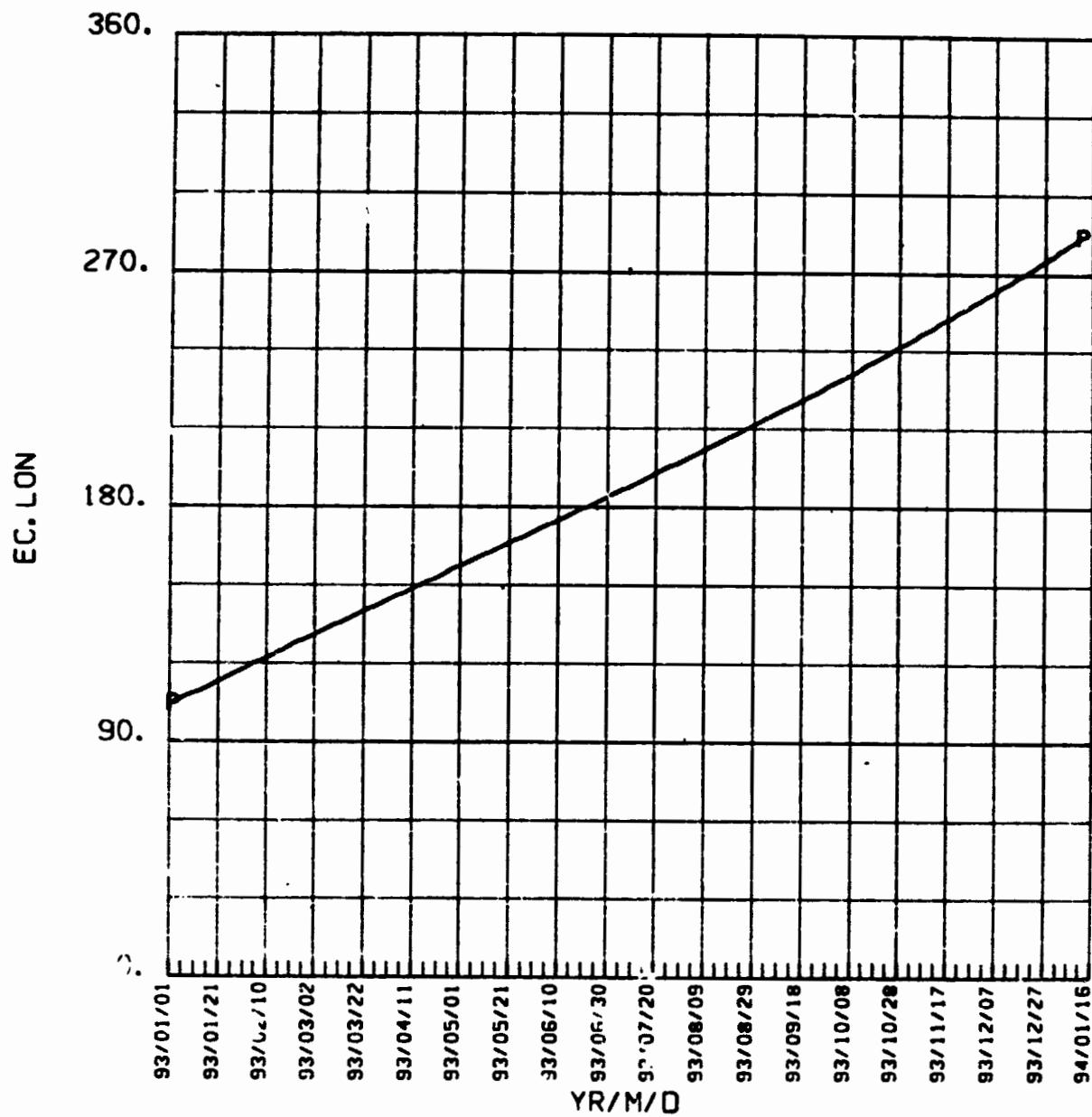
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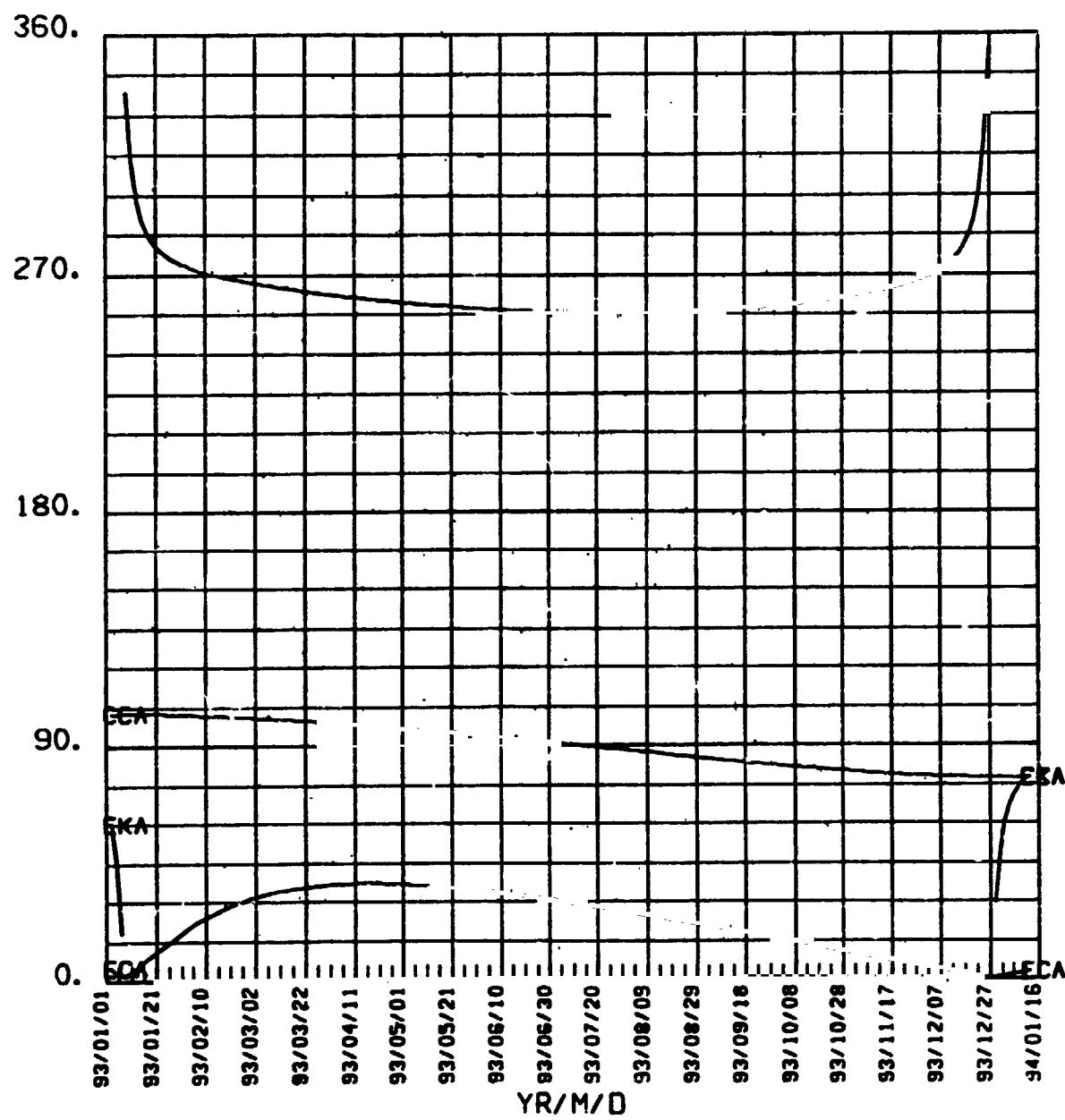


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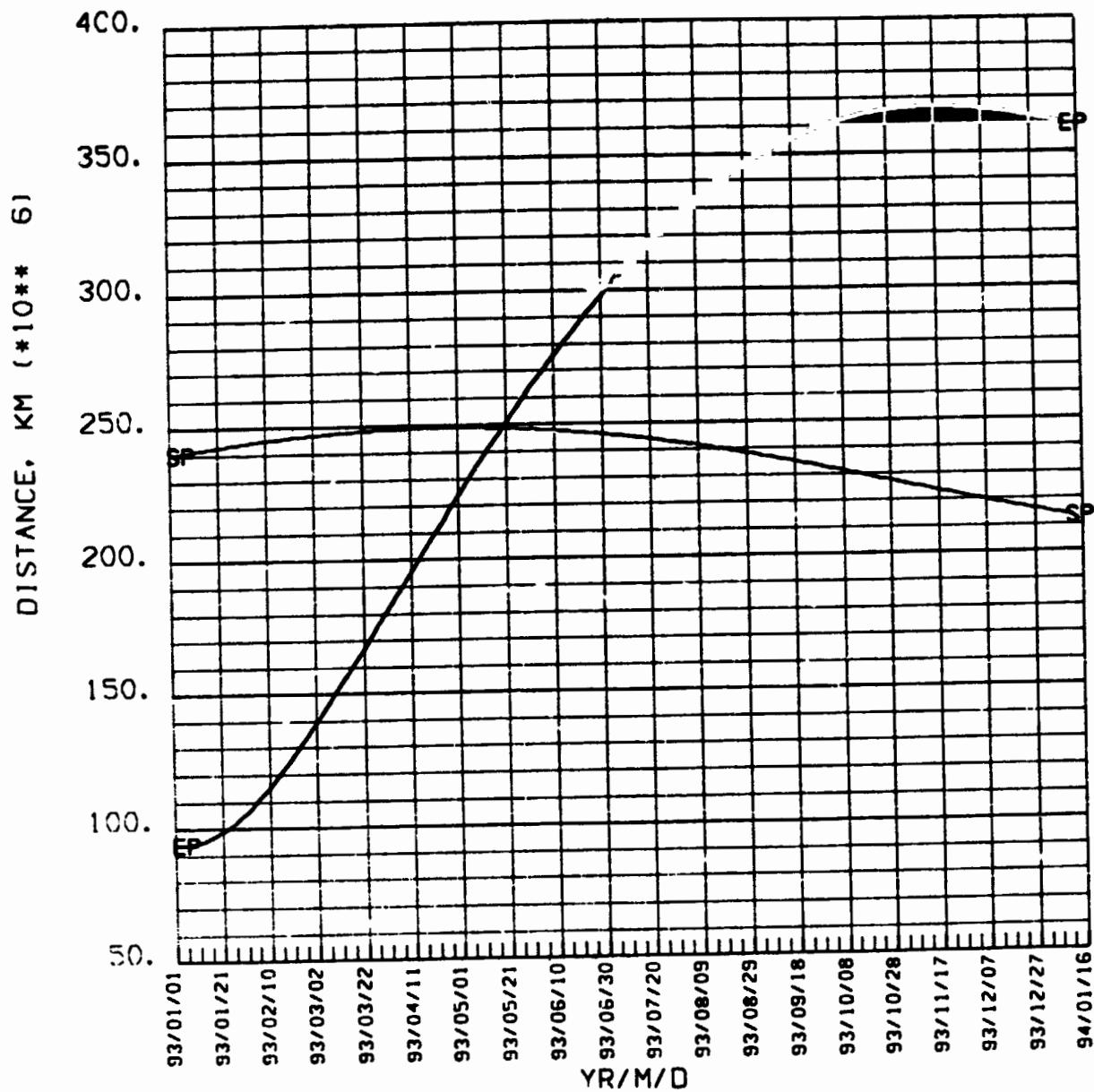


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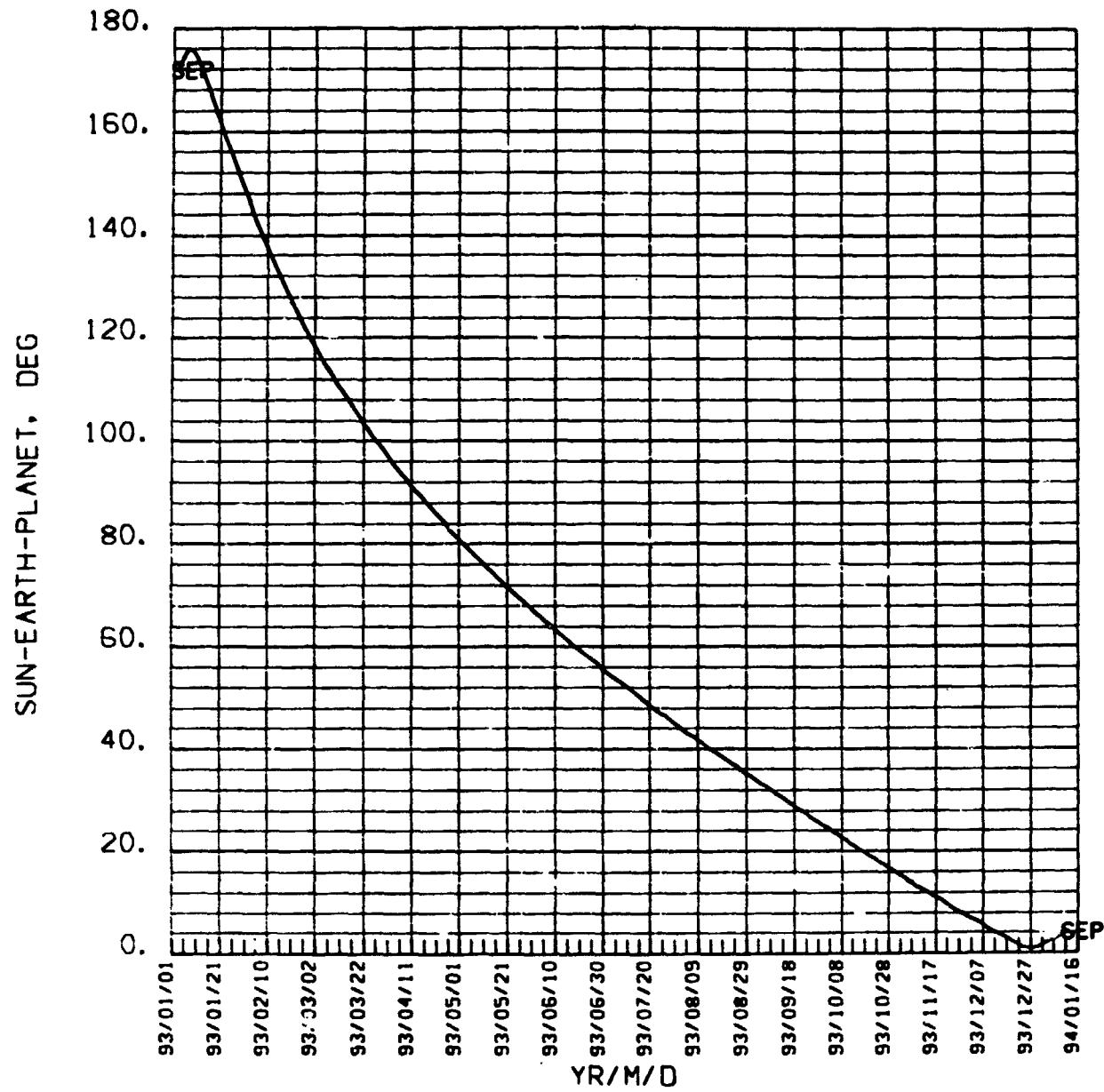
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MARS 1993

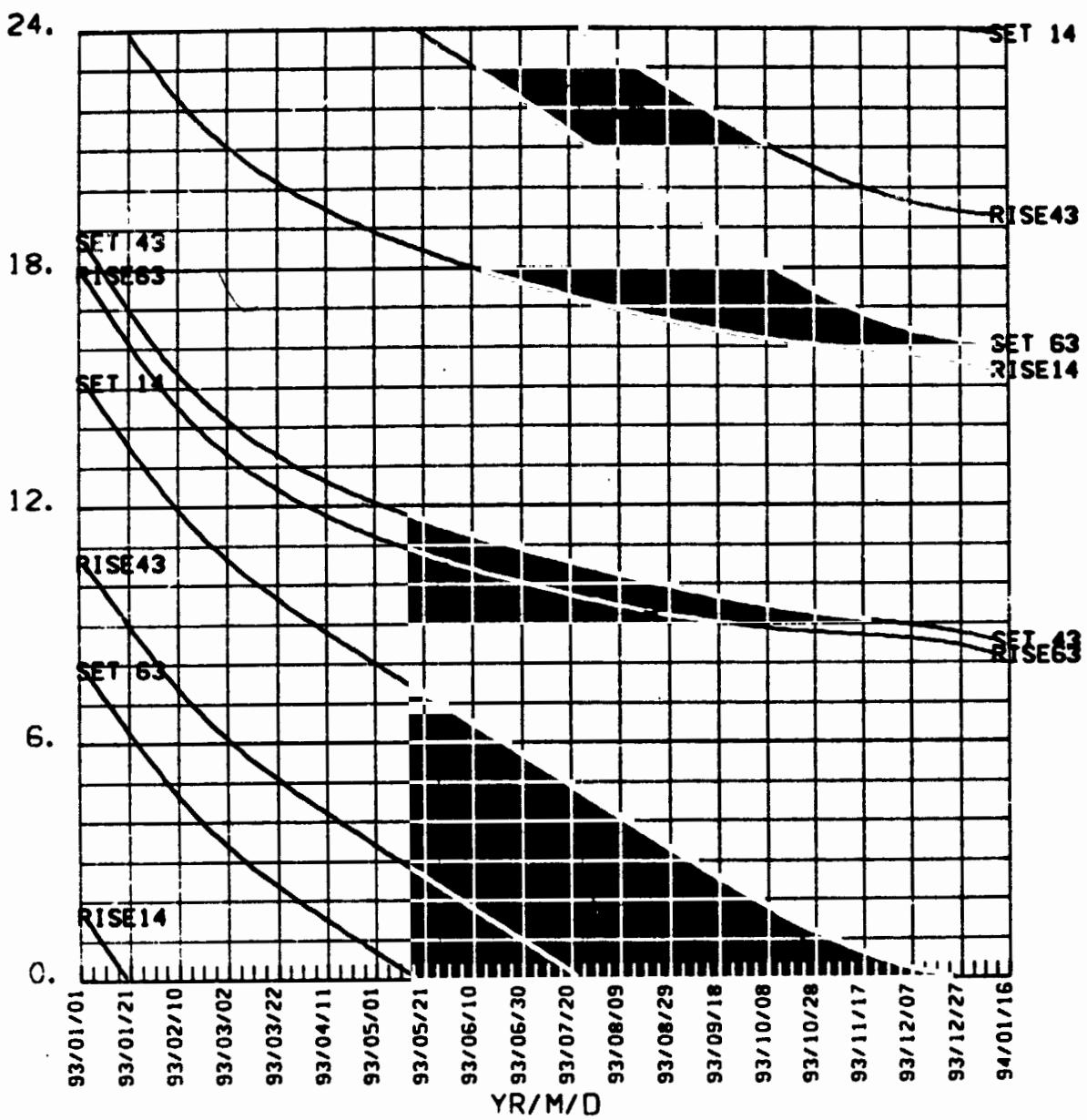


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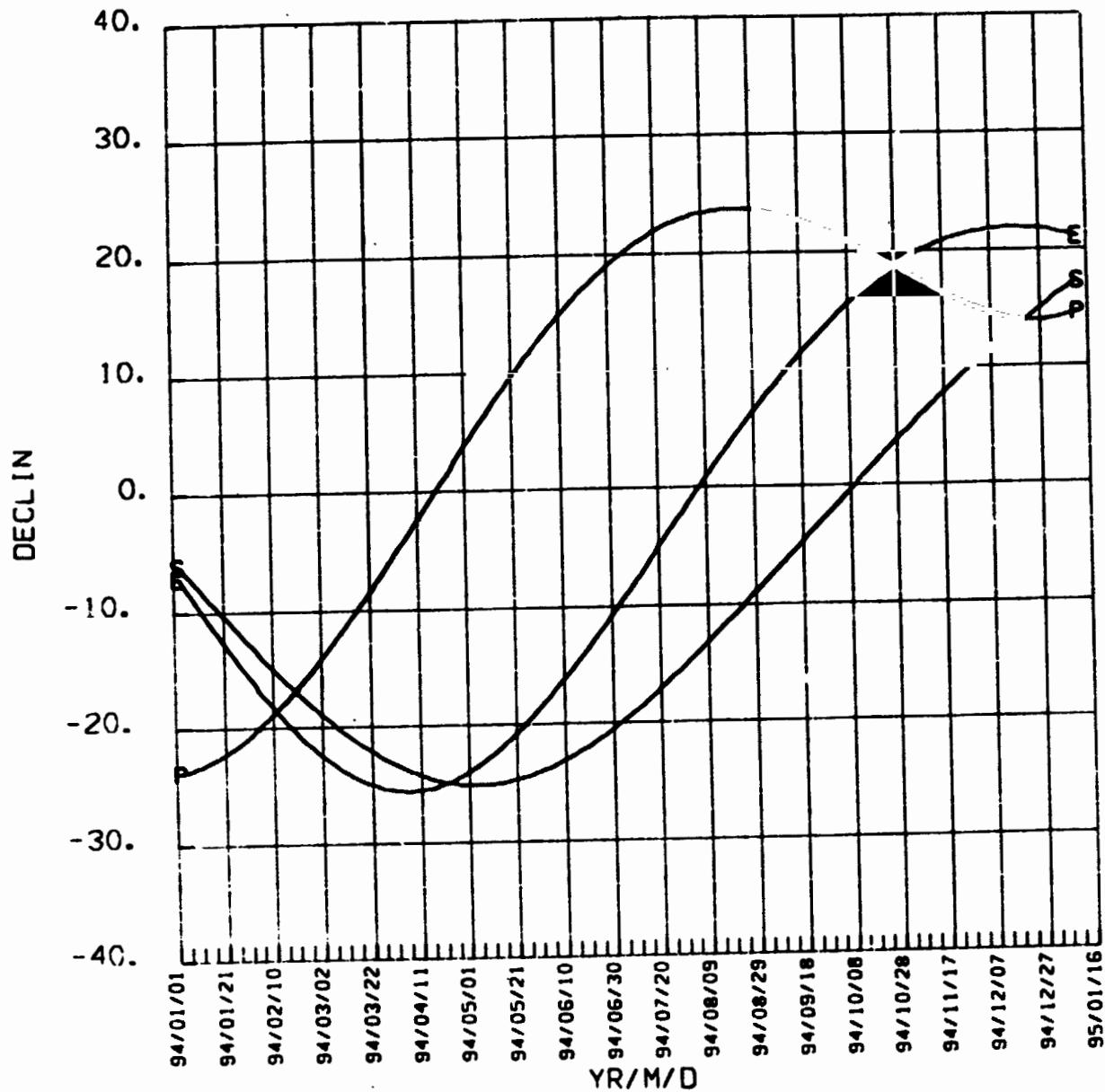
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STATION RISE/SET GMT, HR

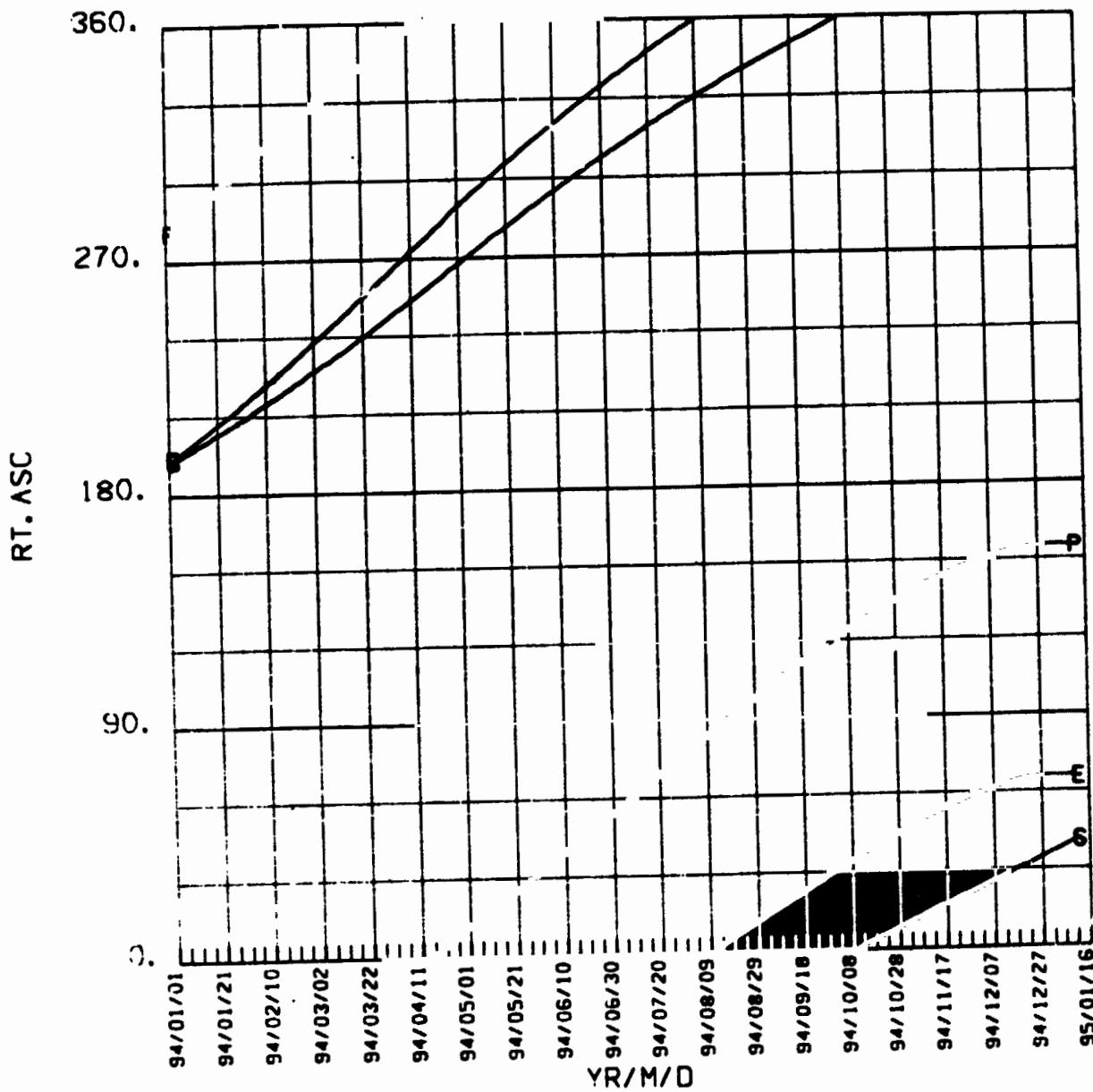


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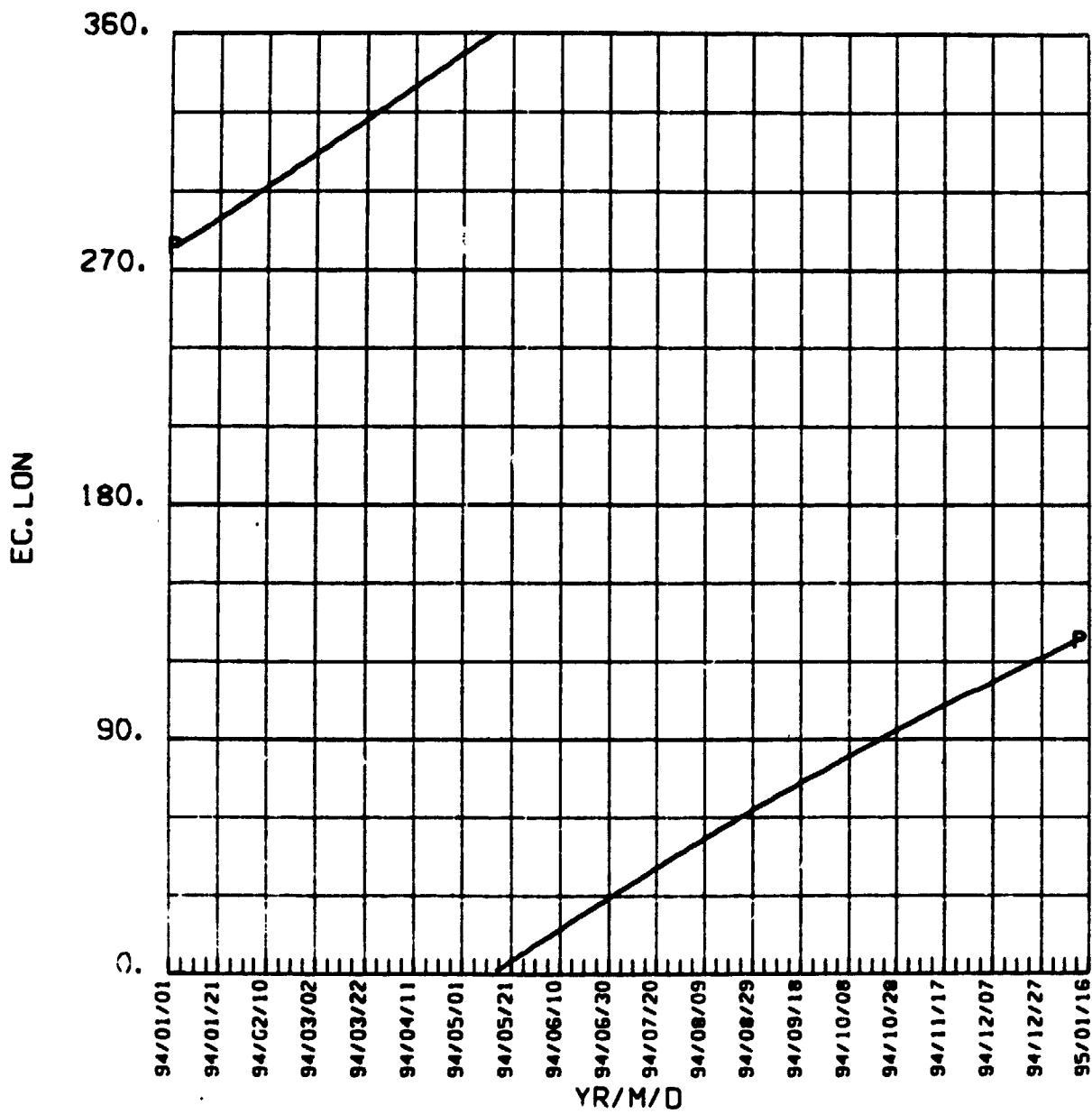
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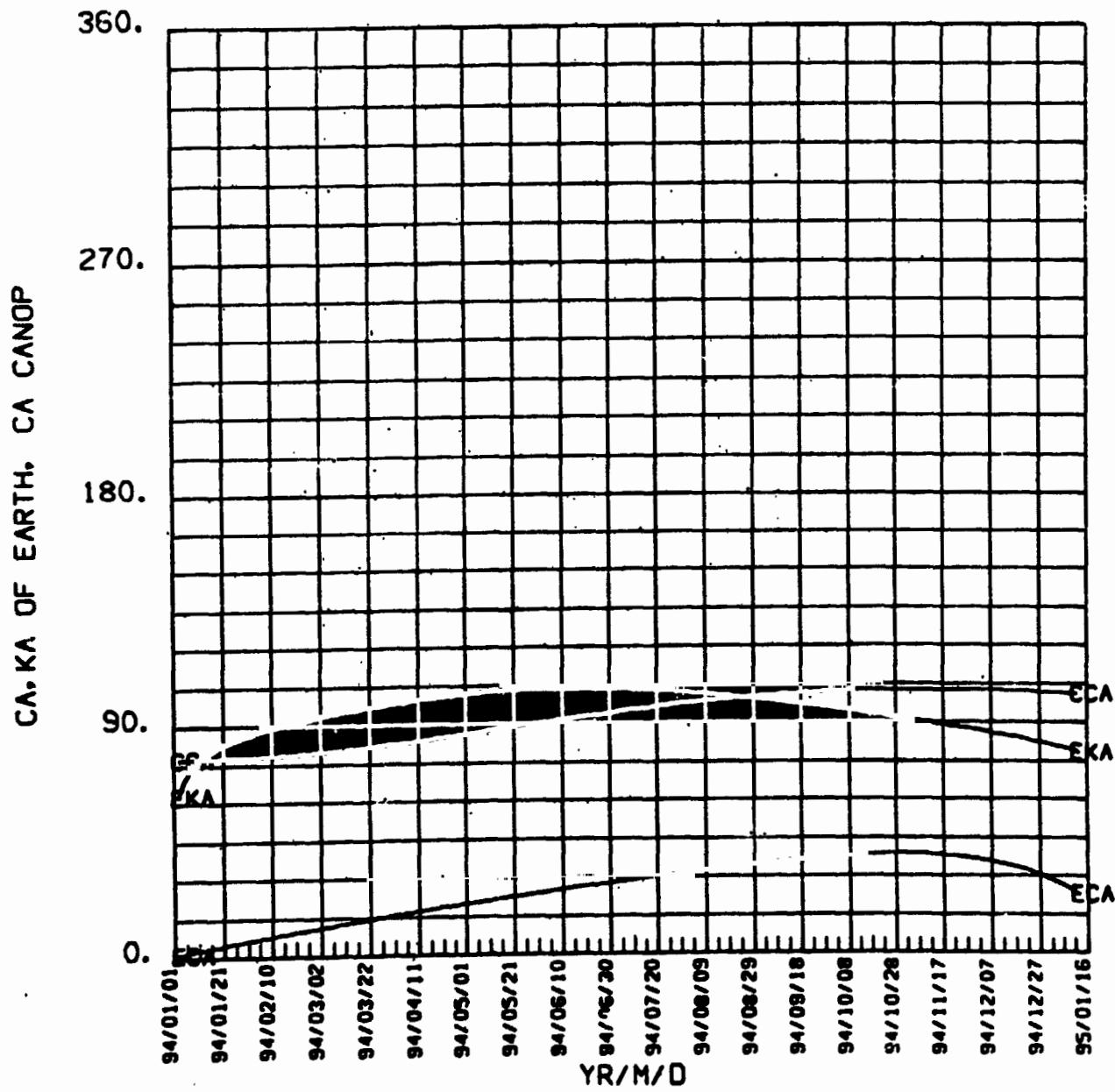


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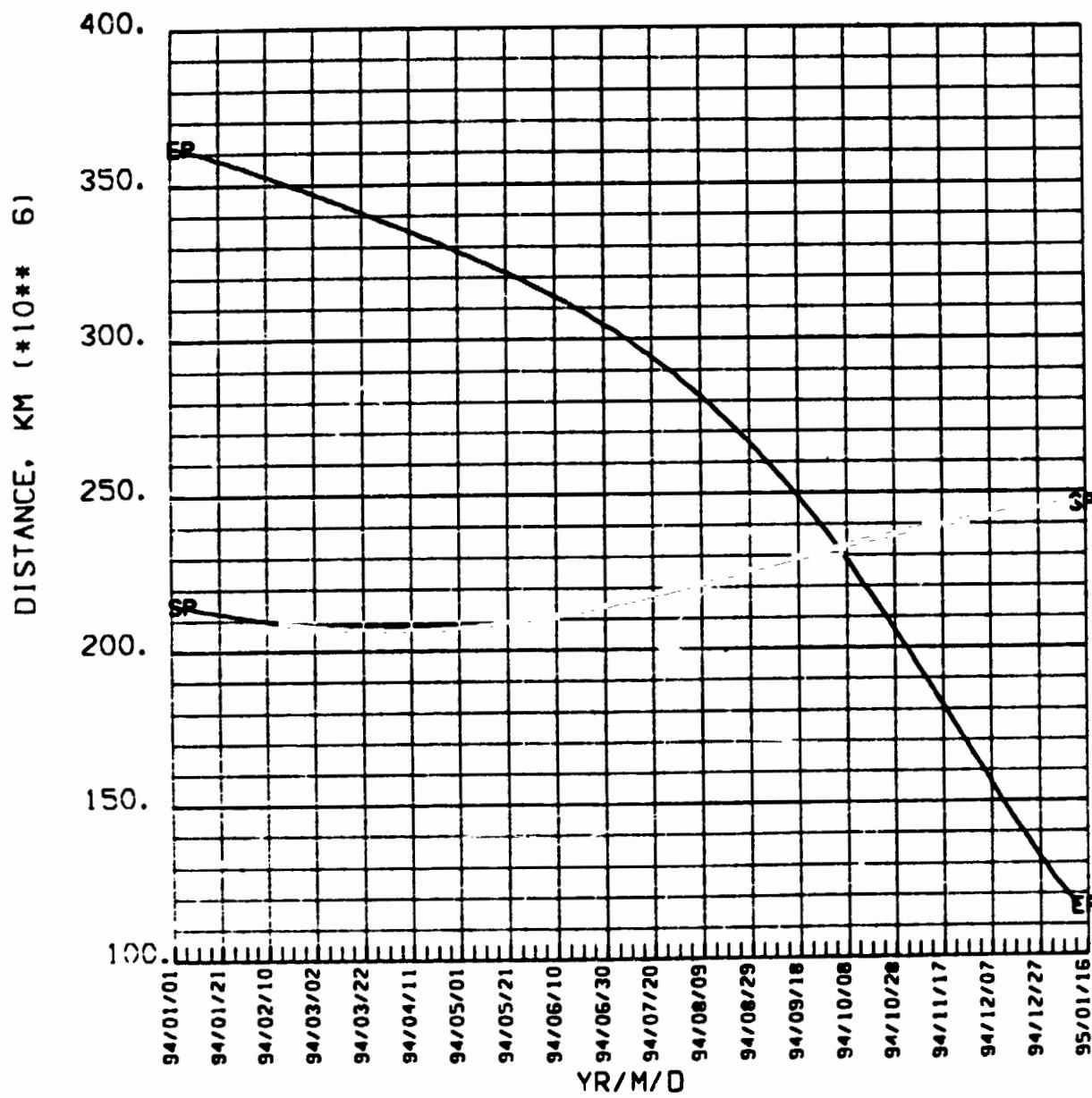


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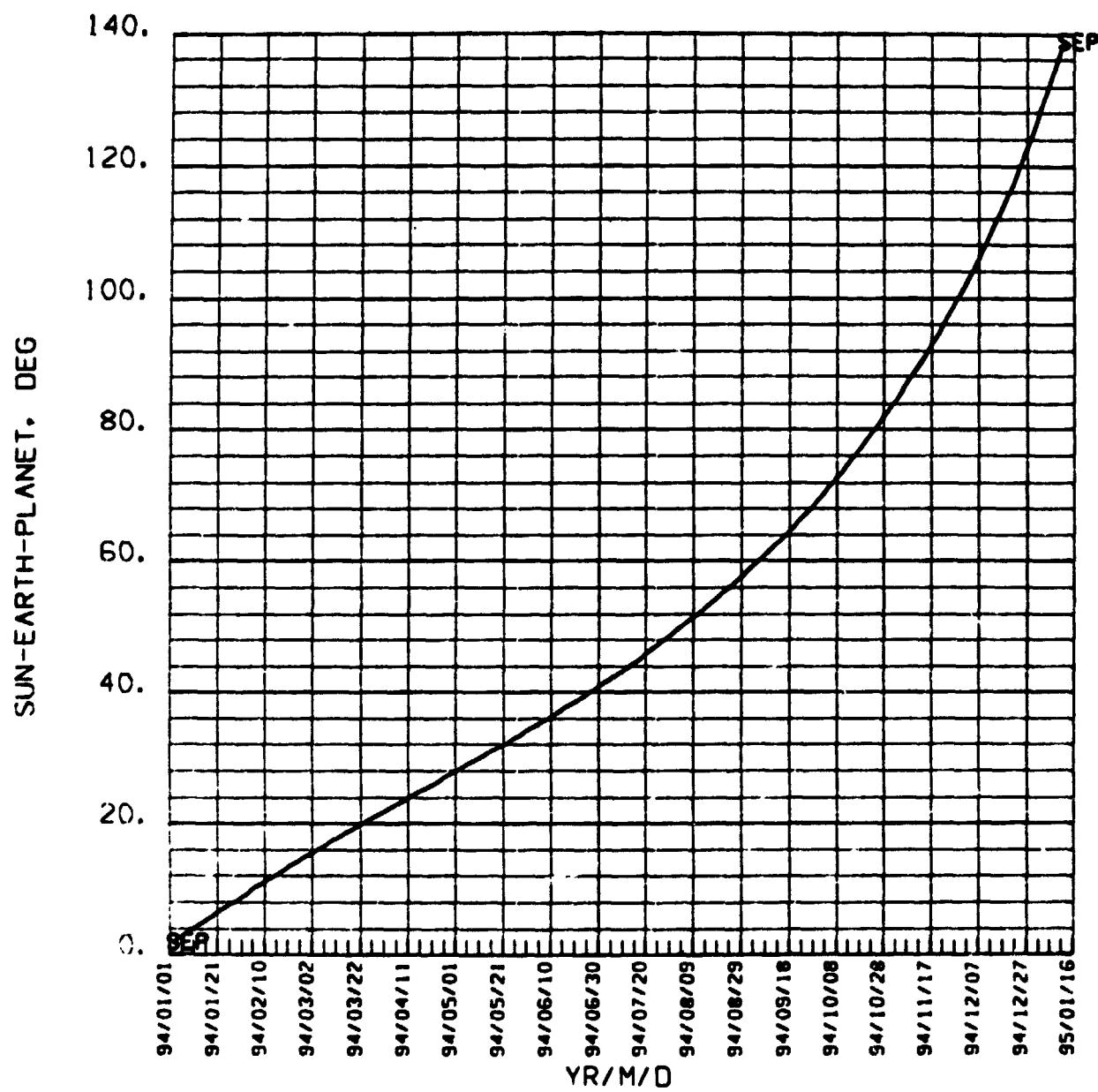
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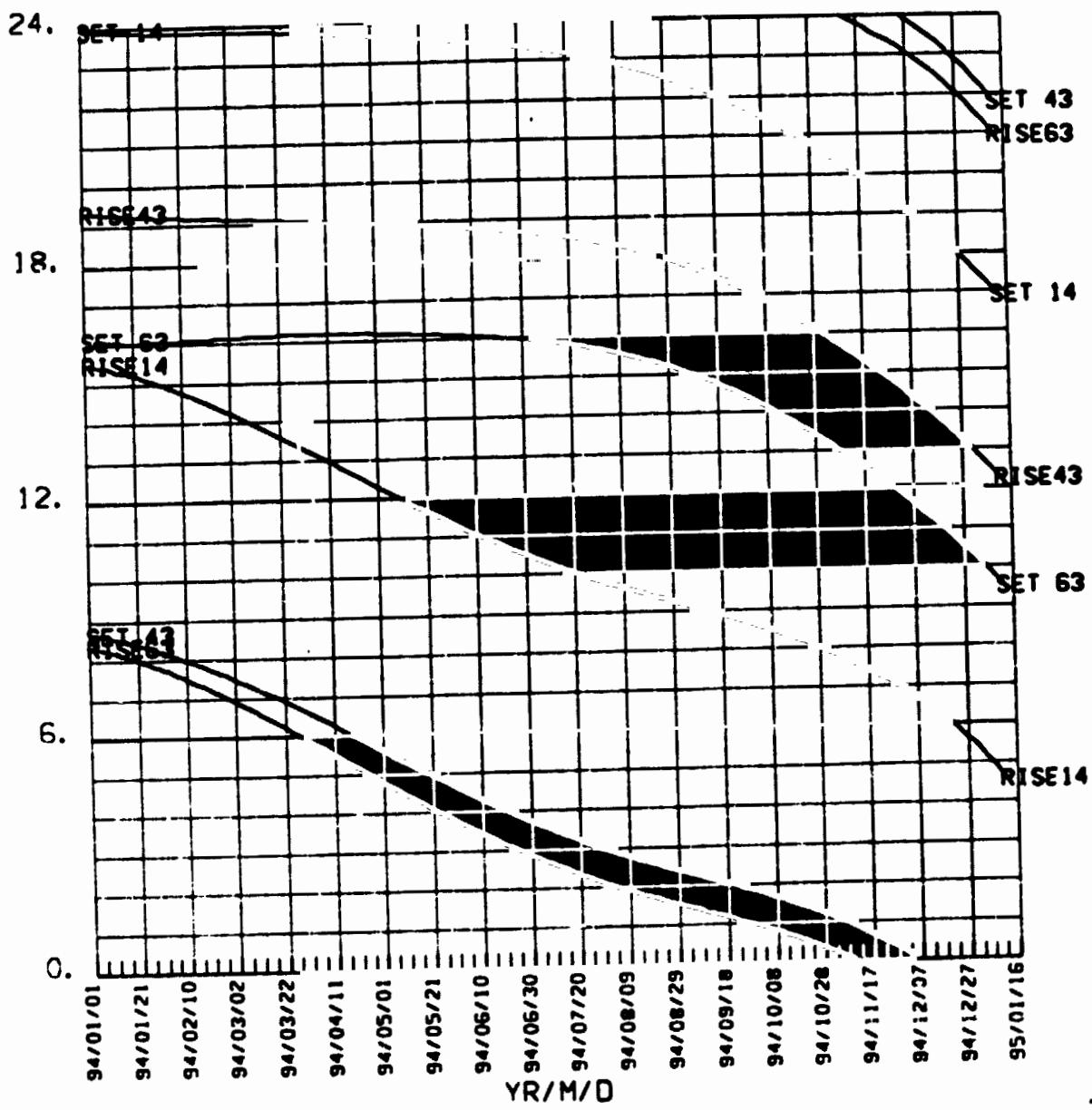


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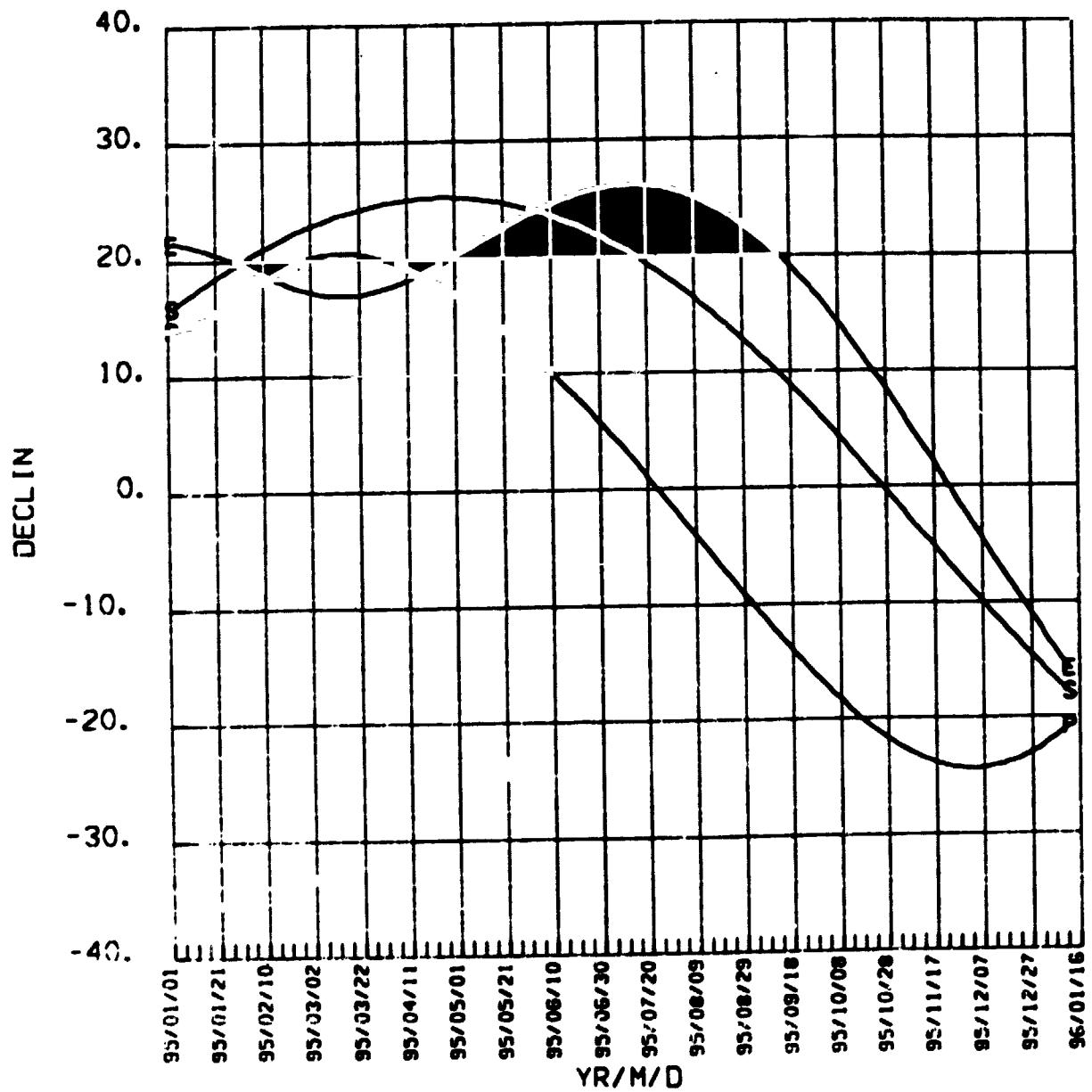


MARS 1994

STATION RISE/SET GMT. HR

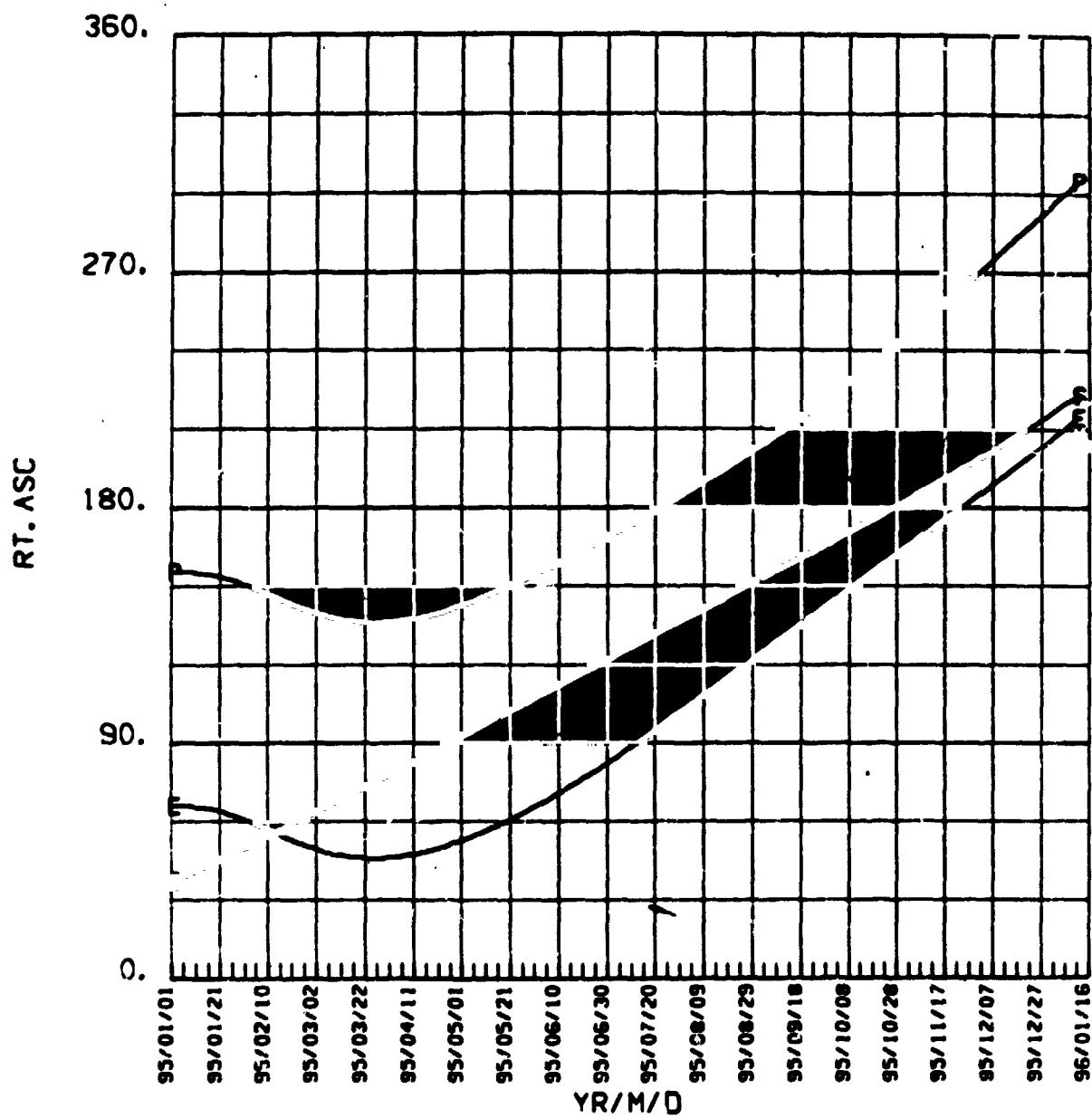


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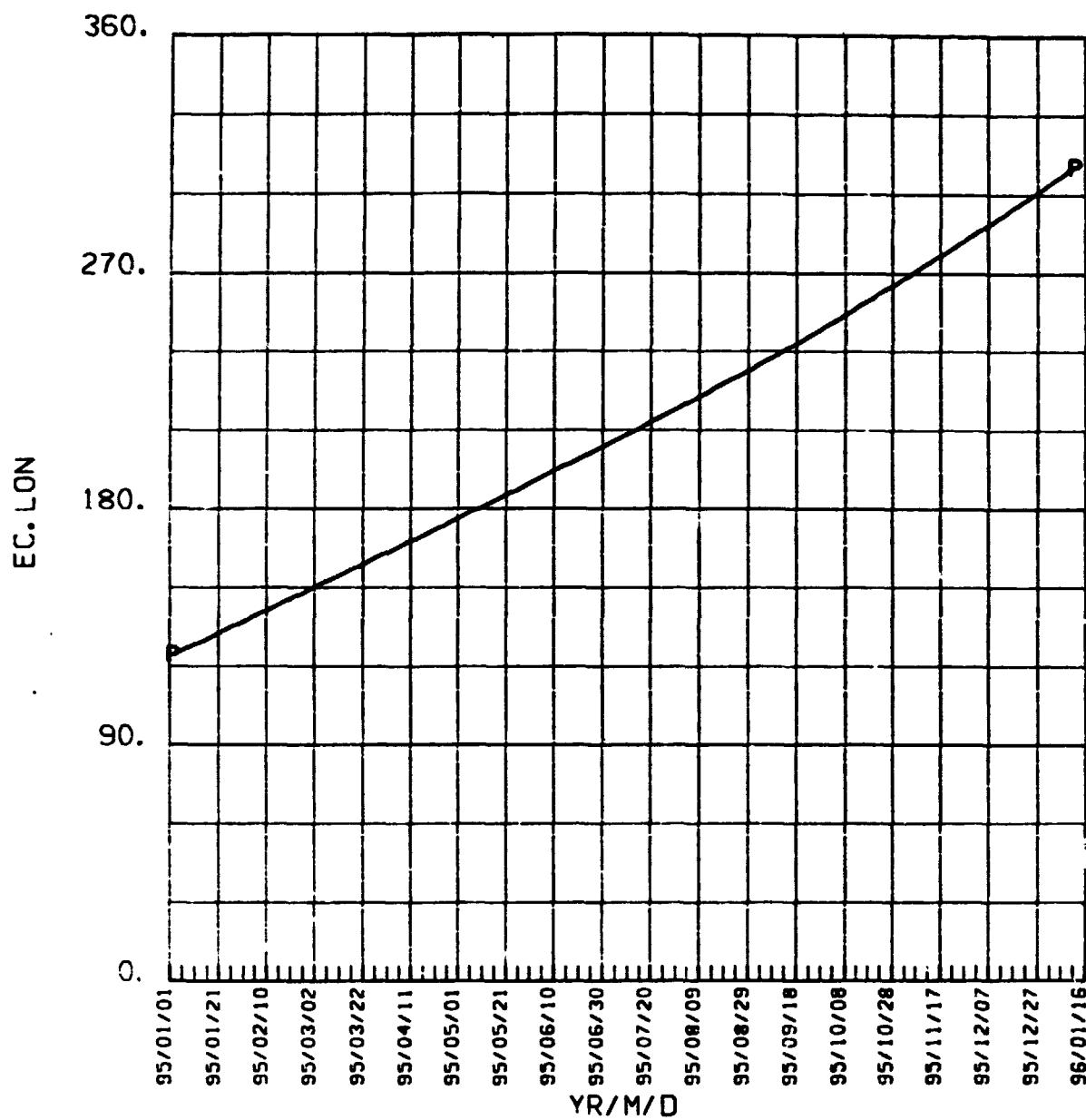


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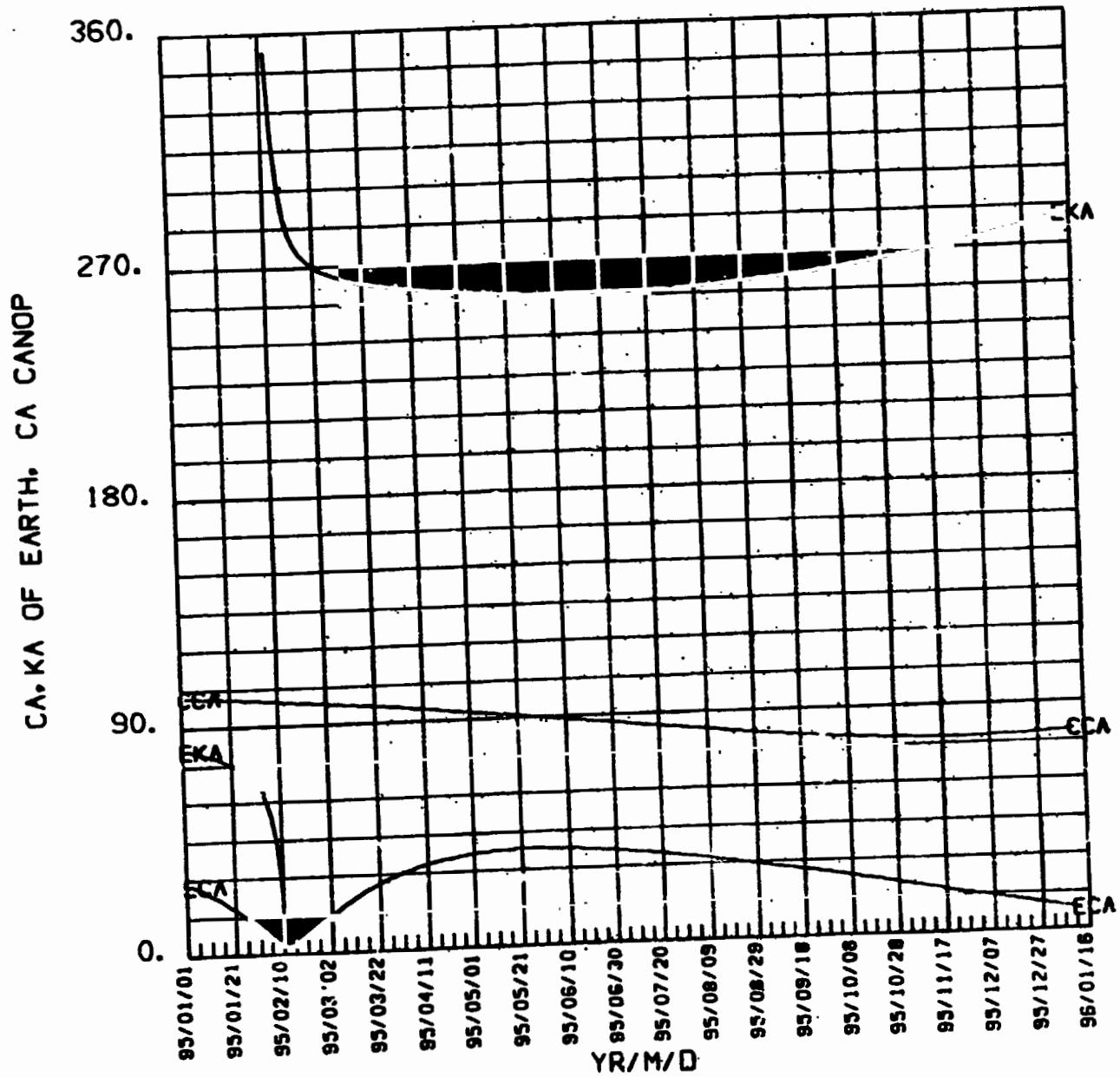
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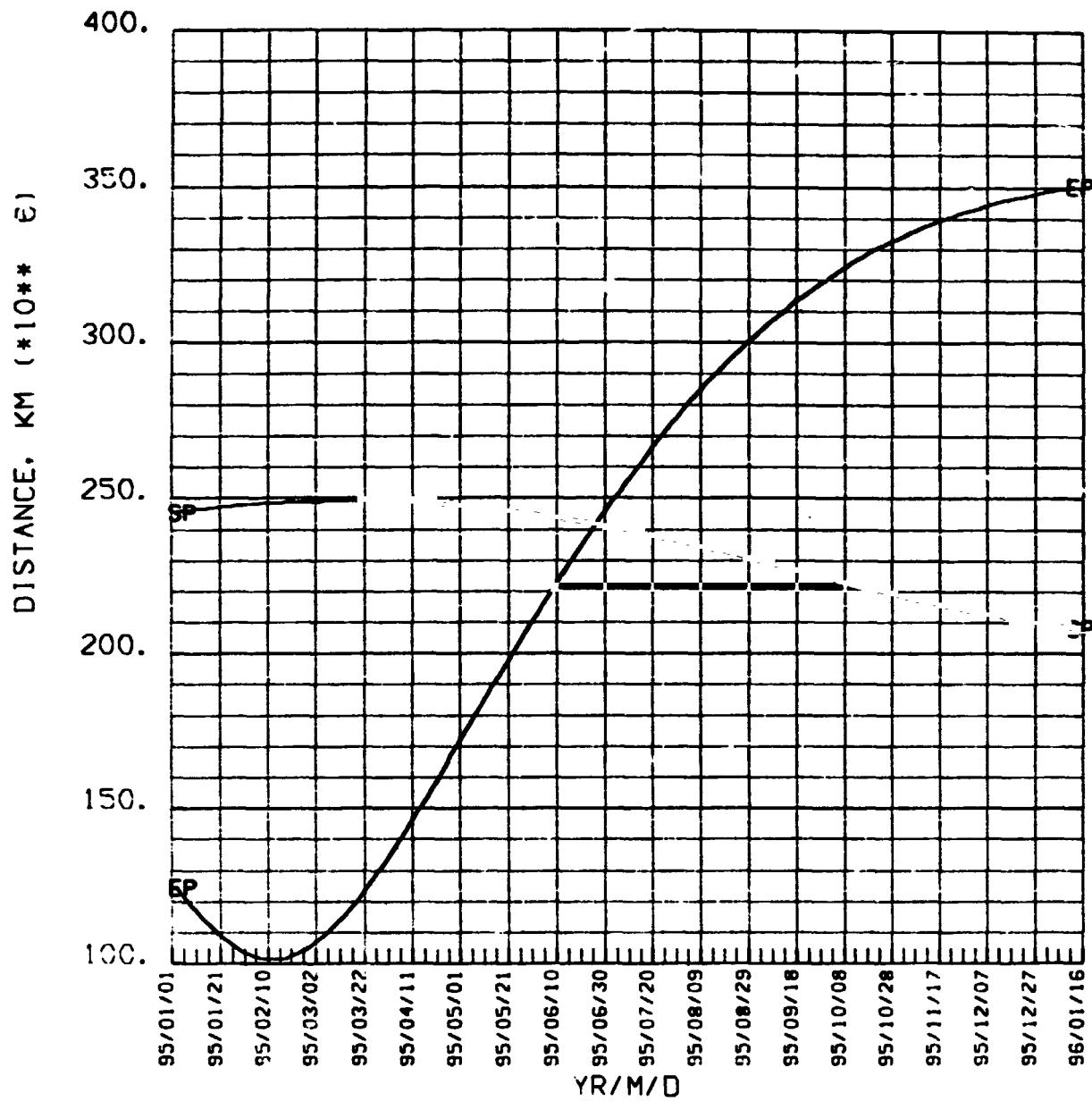


MARS 1995



MARS

1995



MARS 1995

